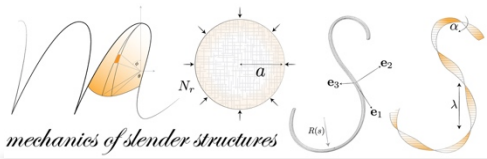


# Elastic Instabilities for Form and Function

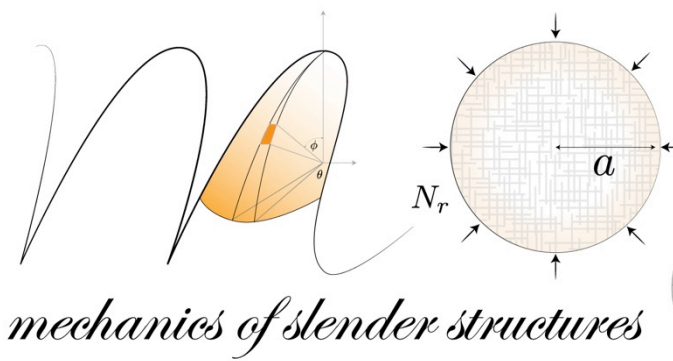
## Buckling, Wrinkling, Folding, and Snapping

Douglas P. Holmes

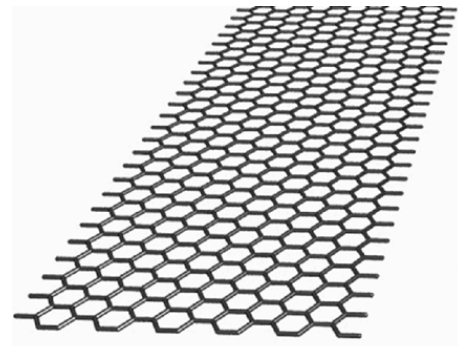
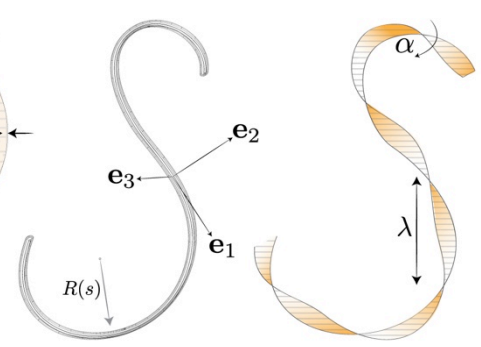
Mechanical Engineering  
Boston University



mechanics of slender structures



mechanics of slender structures

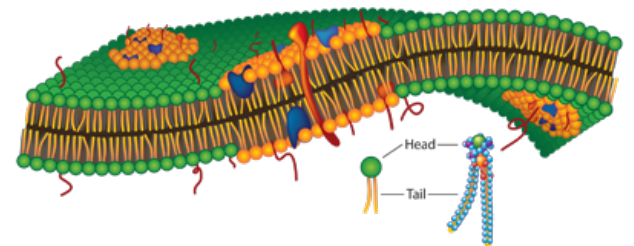


(a)



(b)

Source: Kreupl et al. (2004)



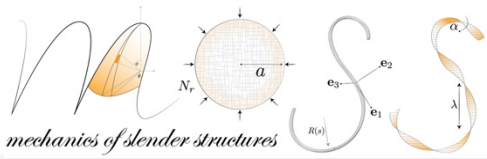
<http://www.lanl.gov/science/1663/august2011/images/Cell-Wave-Final.png>



<http://isabelleteo.deviantart.com/art/Just-hair-292904304>



<http://www.contactlensescomparison.com/wp-content/themes/smallbiz/images/lens.jpg>



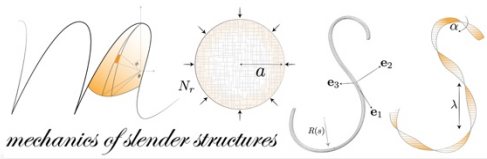
# Thin Structures



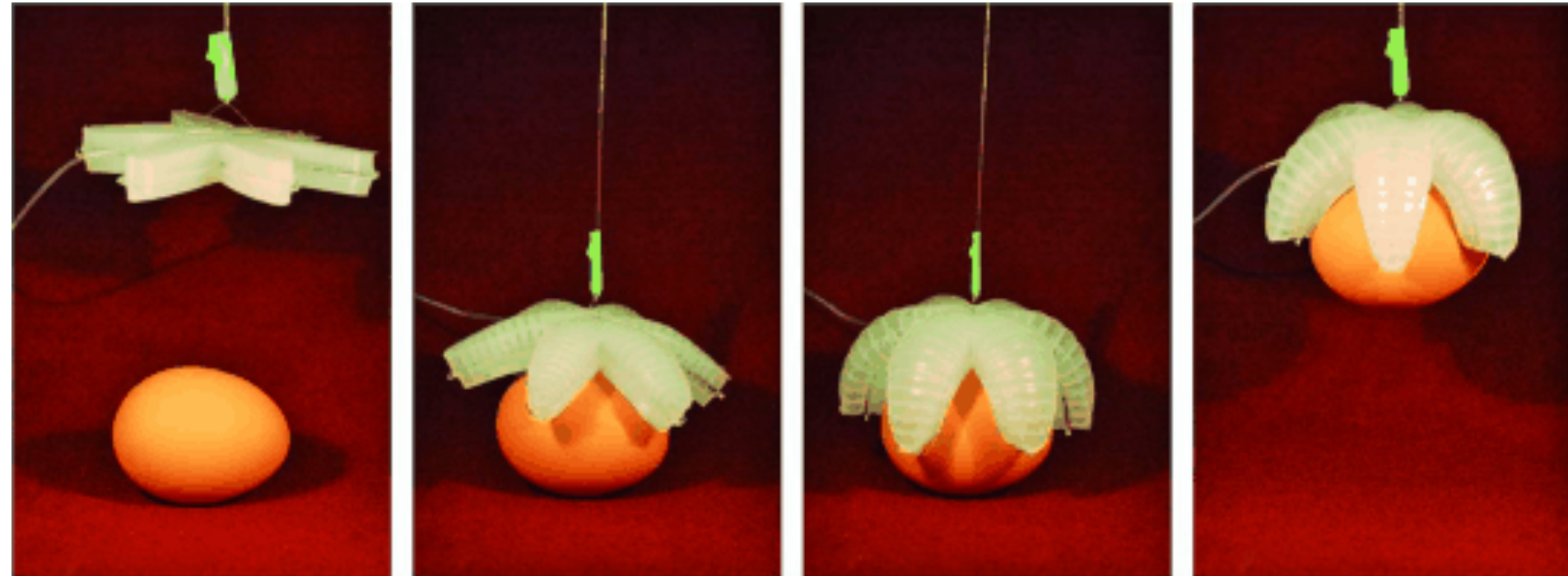
Leonardo da Vinci, "Drapery study for a Seated Figure", c1470

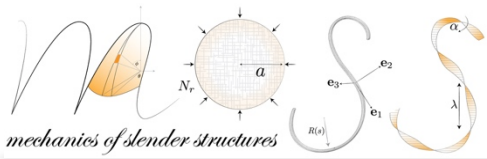


"British Lettuce", <http://www.britishleafysalads.co.uk/>



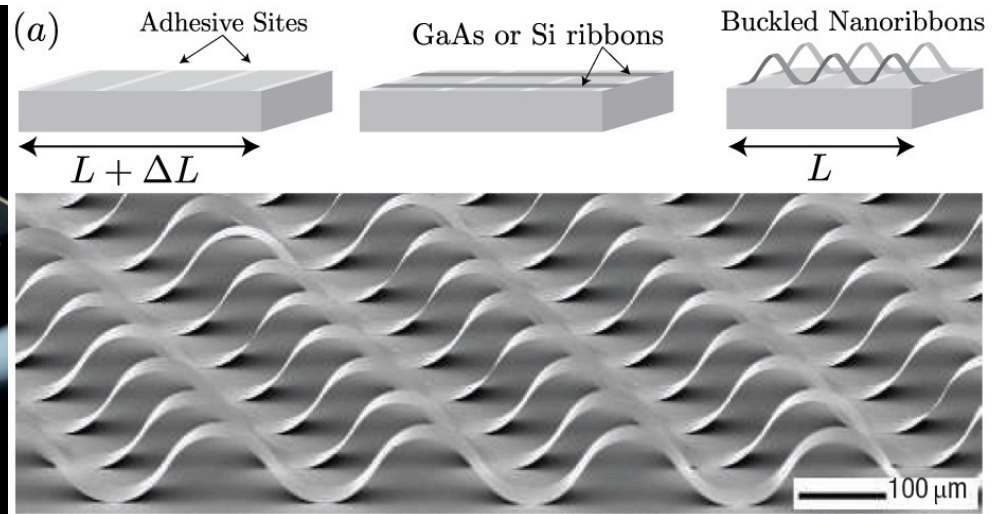
## Soft Robotics

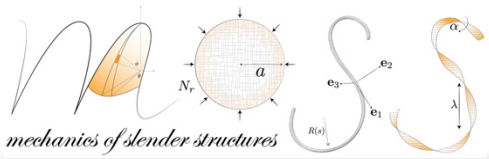




# Advanced Materials

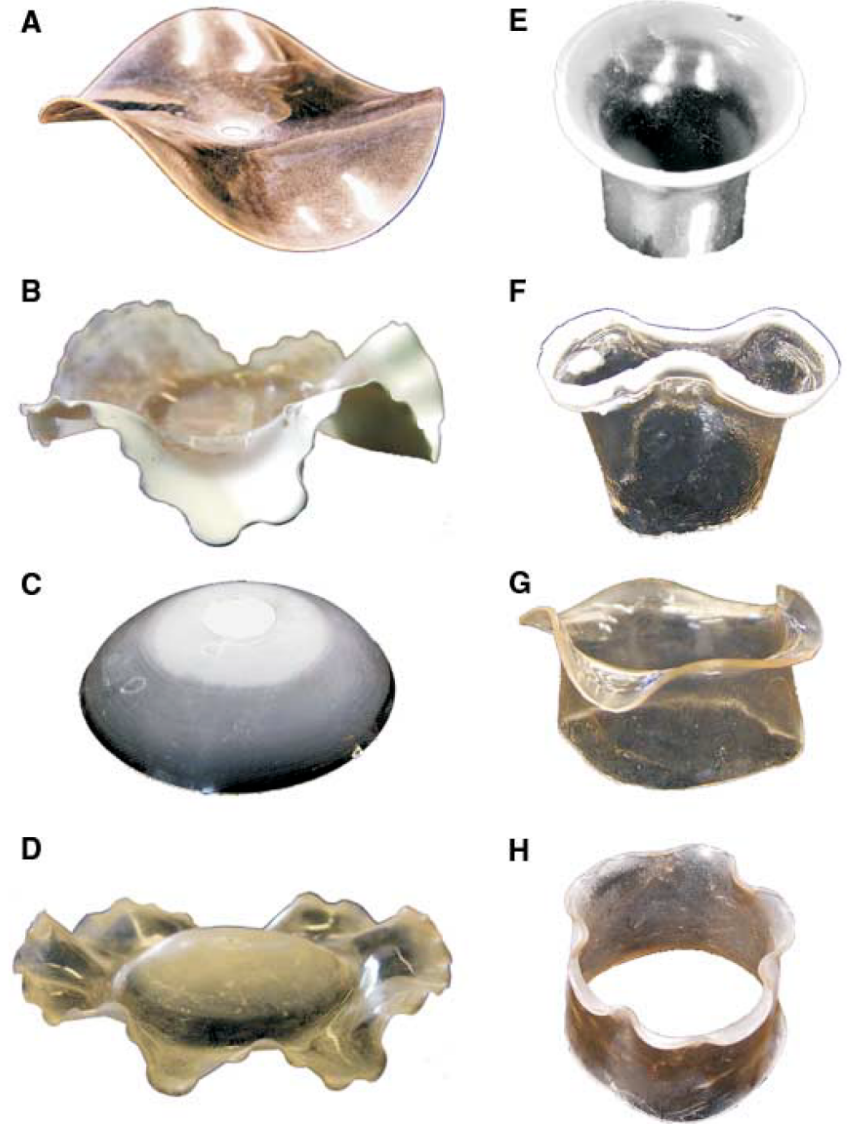
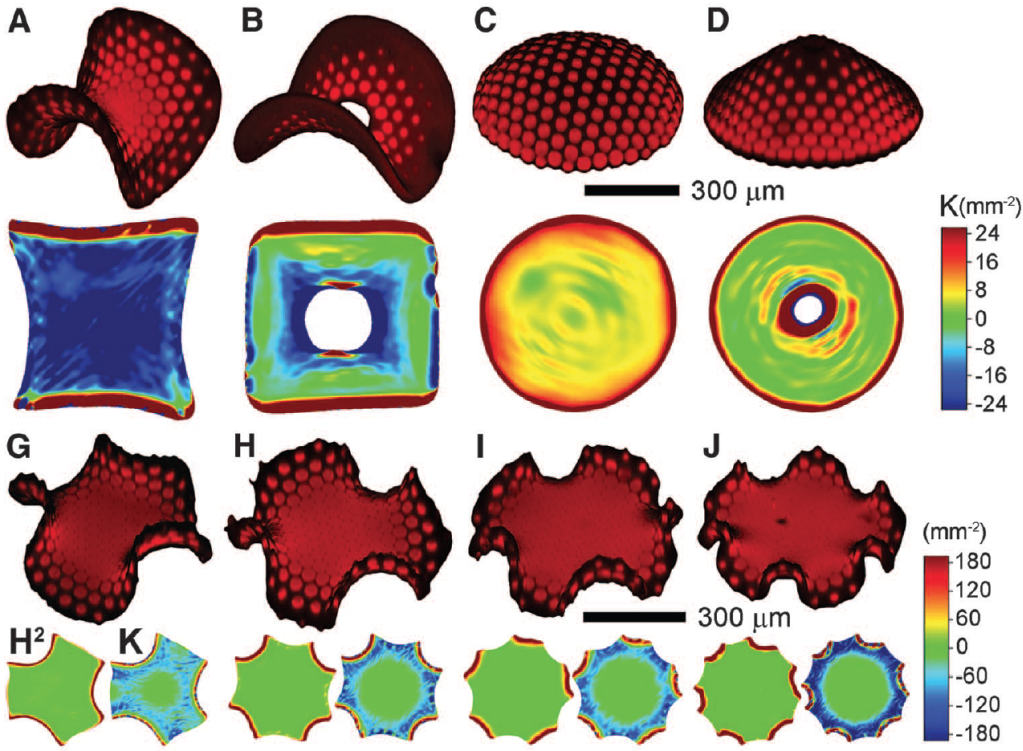
## Flexible Electronics

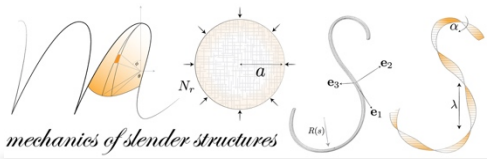




# Advanced Materials

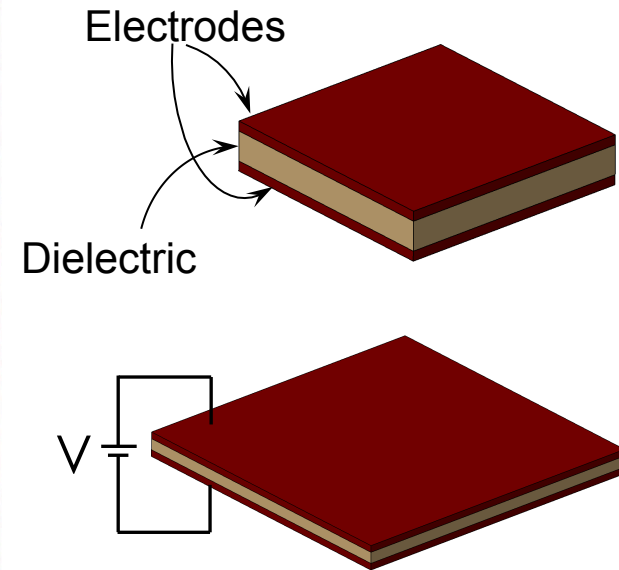
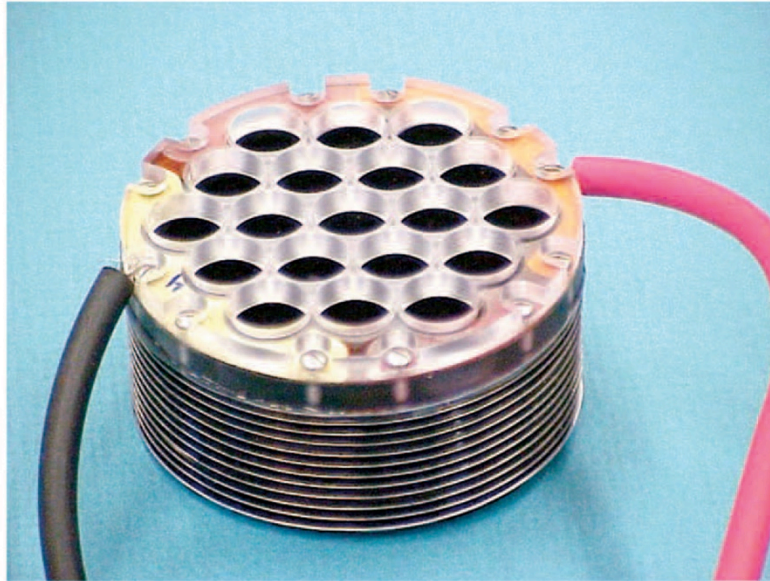
## Morphing Structures

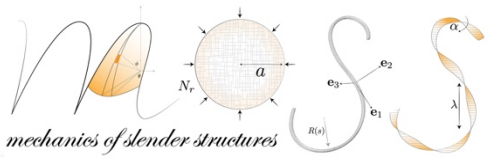




# Advanced Materials

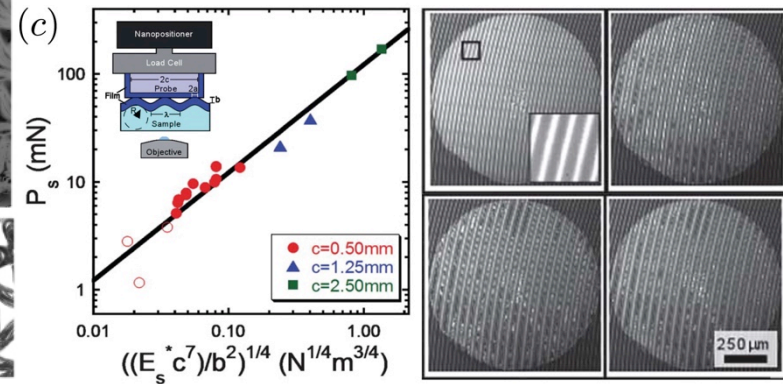
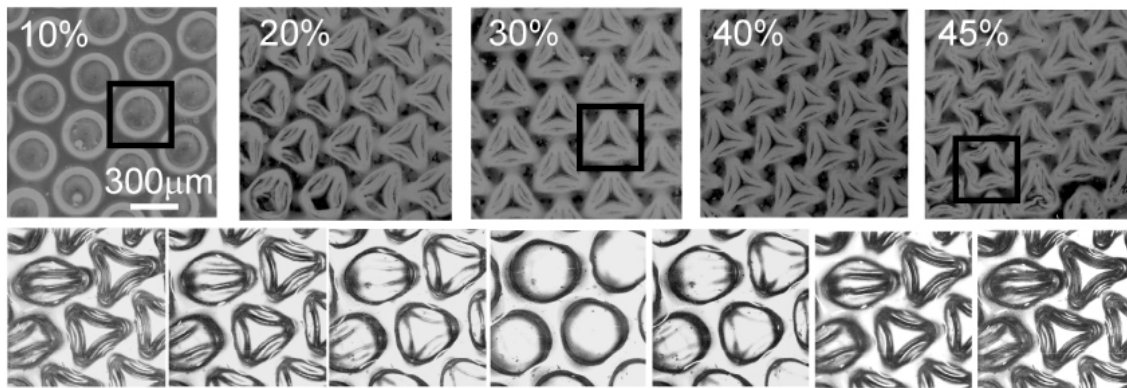
## Energy Harvesting





# Advanced Materials

## Adaptive Surface Patterning







## Elastic Instabilities for Form and Function Buckling, Wrinkling, Folding, and Snapping

### **Geometry and Mechanics:**

- Fundamental equations, geometric rigidity, morphing.

### **Buckling & Wrinkling:**

- Stability, wavelength, flexible electronics, mechanical metamaterials, adhesion.

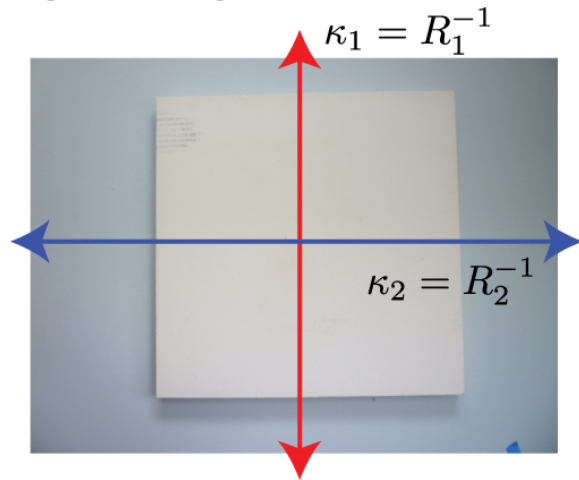
### **Stress Focusing – Folding & Creasing:**

- Wrinkle-to-fold, origami.

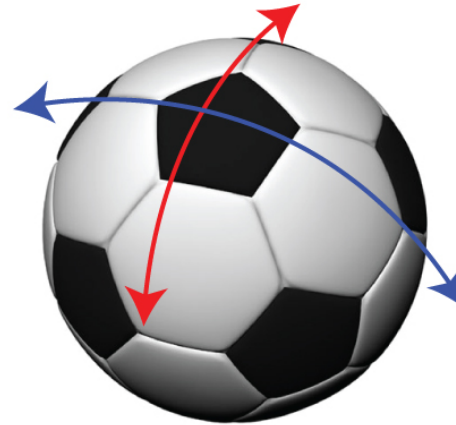
### **Snapping:**

- Snapping surfaces.

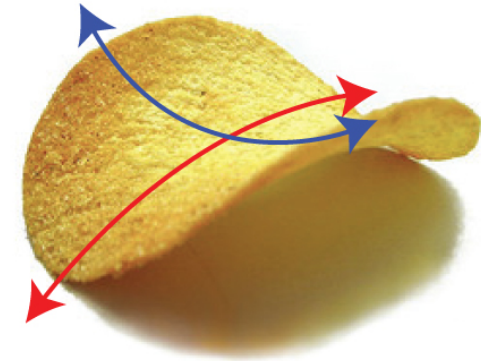
## Gauss Curvature



$$K = \kappa_1 \kappa_2 = \diamond^4[w, w, ] = 0$$

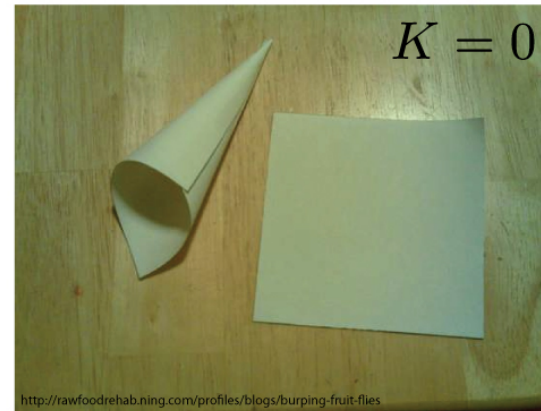
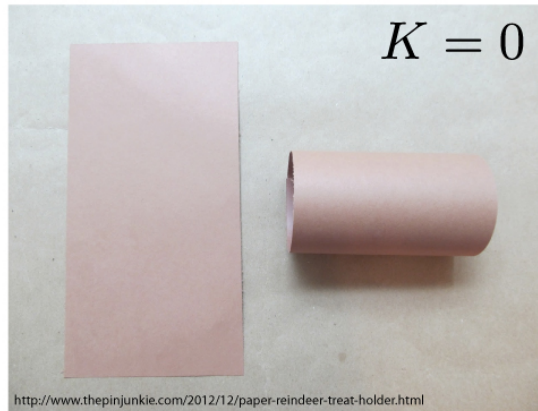


$$K = \kappa_1 \kappa_2 = \diamond^4[w, w, ] > 0$$



$$K = \kappa_1 \kappa_2 = \diamond^4[w, w, ] < 0$$

## Developable

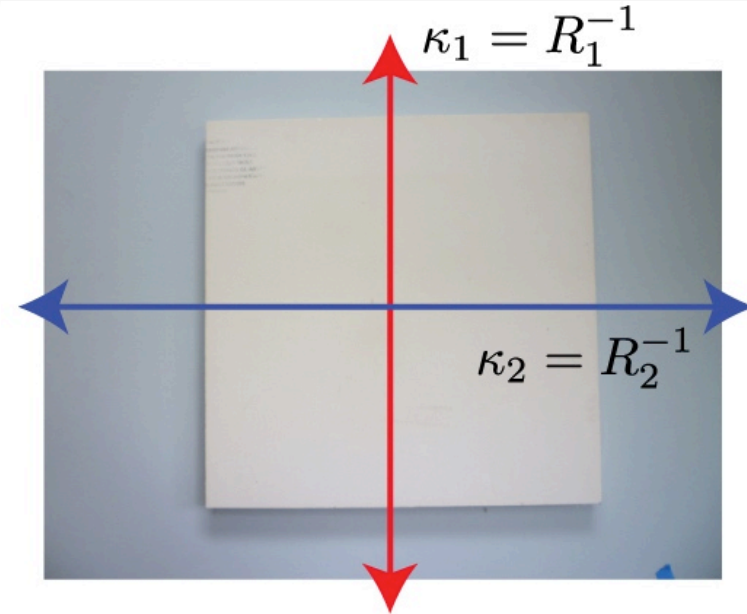




# Mechanics of Plates

**Elastic Energy** (variational form)

$$\delta \mathcal{U}_e = \iiint_V \sigma_{ij} \delta \varepsilon_{ij} \, dV$$

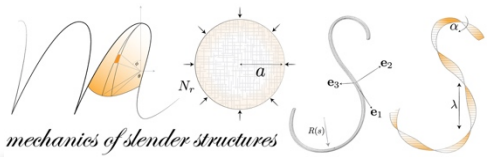


**Elastic Energy** (linear, isotropic material)

$$\mathcal{U}_e = \iiint_V \sigma_{ij}(\varepsilon_{kl}) \varepsilon_{ij} \, dV$$

Constitutive relationship:

$$\varepsilon_{ik} = \frac{1 + \nu}{E} \sigma_{ik} - \frac{\nu}{E} \sigma_{jj} \delta_{ik}$$

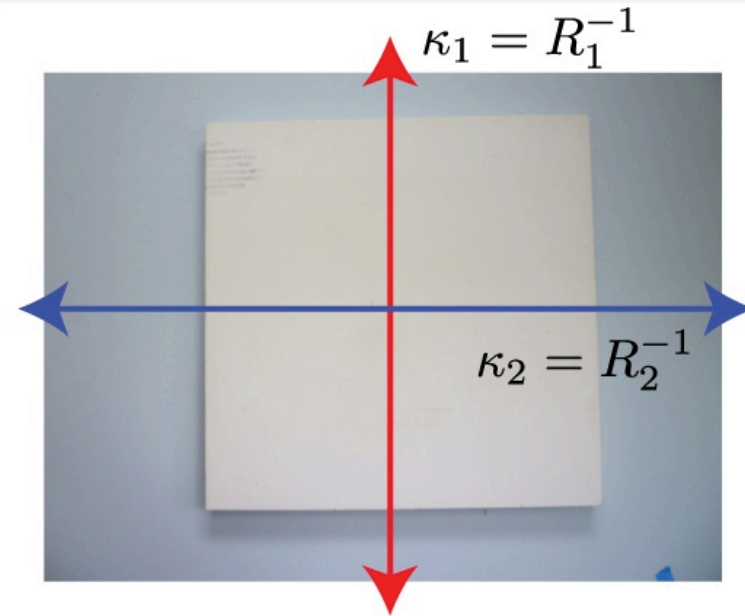


# Mechanics of Plates

**Elastic Energy** (linear, isotropic material)

$$\mathcal{U}_e = \iint_A dx dy \int_{-h/2}^{h/2} \sigma_{ij} \varepsilon_{ij} dz$$

Since  $h$  is small, it is often reasonable to **neglect** the **through thickness dependence** of the energy density.



**Membrane Approximation**

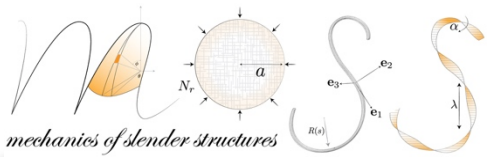
$$\mathcal{U}_m = \frac{h}{2} \iint_A dx dy (\sigma_{\alpha\beta} \varepsilon_{\alpha\beta})_{cs}$$

$$\mathcal{U}_m \sim Eh \iint_A dx dy (\varepsilon_{\alpha\beta})^2$$

Hooke's law:

$$\sigma_{\alpha\beta} \approx E \varepsilon_{\alpha\beta}$$

Scaling is valid to a numerical factor of order one, depending on Poisson's ratio.

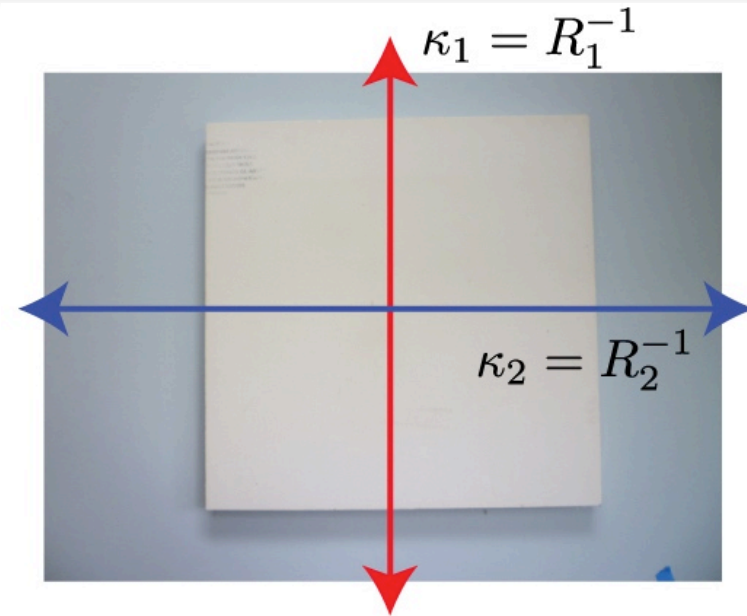


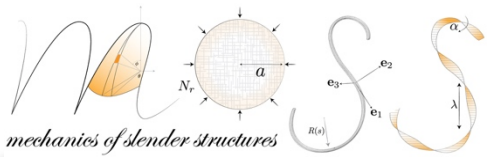
# Mechanics of Plates

## Membrane Approximation

$$U_m \sim Eh \iint_A dx dy (\varepsilon_{\alpha\beta})^2$$

- Any reference to the transverse direction ( $z$ ) has been eliminated by integration across the plate thickness.
- Membrane, or stretching energy, is linearly proportional to  $E$  and  $h$ .
- For an **isometric deformation**, the in-plane strain must vanish, therefore the membrane energy is zero.
- There are many ways to deform a plane isometrically – to break this degeneracy we need to consider the plate's **bending energy**.

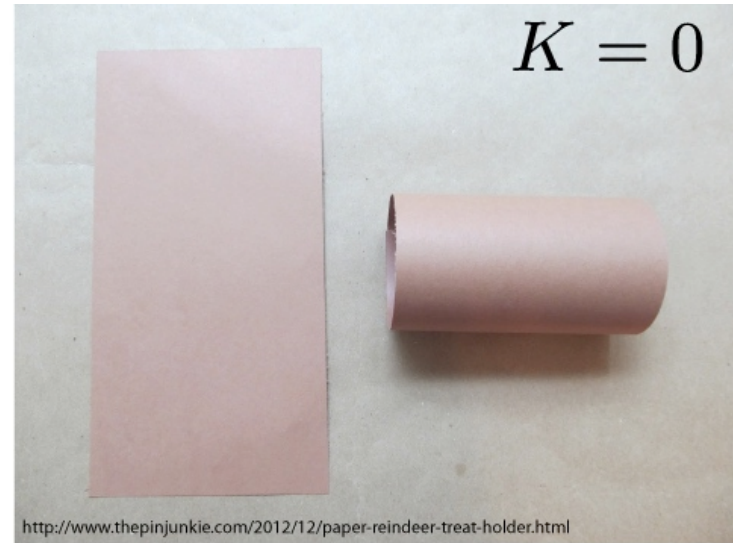
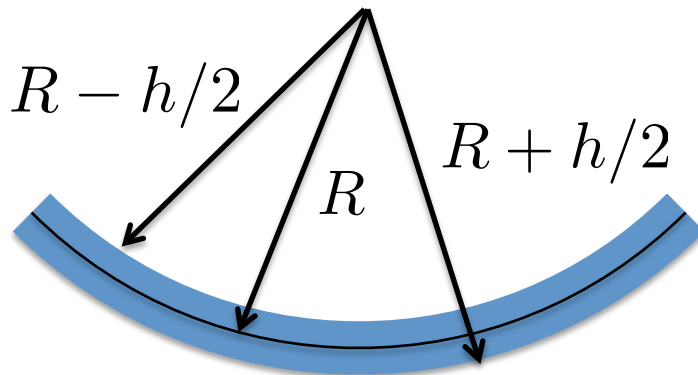




# Mechanics of Plates

## Bending energy

- Curved plate is under **compression** on one side and **extension** on the other.
- This contribution is **missed** by **membrane theory** since the stress is averaged over the thickness.



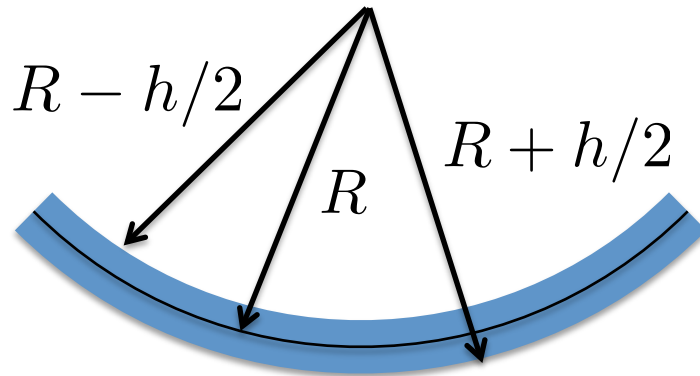
Radius of curvature of plate:

$$\frac{1}{R} \equiv \partial_x^2 w$$

Strain:

$$\varepsilon_{\alpha\beta}(x) = z \frac{\partial^2 w}{\partial x^2} = \frac{z}{R}$$

## Bending energy



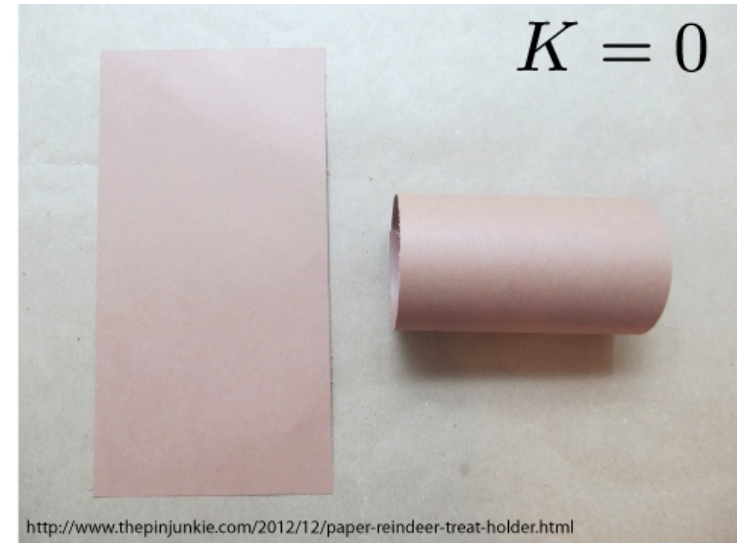
$$\mathcal{U}_e = \iint_A dx dy \int_{-h/2}^{h/2} \sigma_{ij} \varepsilon_{ij} dz$$

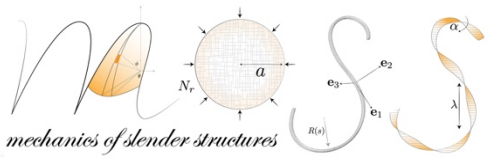
Stresses & strains in bending:

$$\sigma_{ij} = E z \partial_\alpha \partial_\beta w^2$$

$$\varepsilon_{ij} = z \partial_\alpha \partial_\beta w$$

$$\mathcal{U}_b \sim \frac{E h^3}{24} \iint_A dx dy |\partial_\alpha \partial_\beta w|^2$$





# Mechanics of Plates

## Elastic Energy of a Plate

$$\mathcal{U} \sim \underbrace{EhA|\varepsilon_{\alpha\beta}|^2}_{\text{stretching}} + \underbrace{Eh^3A|\partial_\alpha\partial_\beta w|^2}_{\text{bending}}$$

**Scaling** of the in-plane **strain** & out-of-plane **curvature**

$$|\varepsilon_{\alpha\beta}| \sim w^2/L^2 \quad |b_{\alpha\beta}| = |\partial_\alpha\partial_\beta w| \sim w/L^2$$

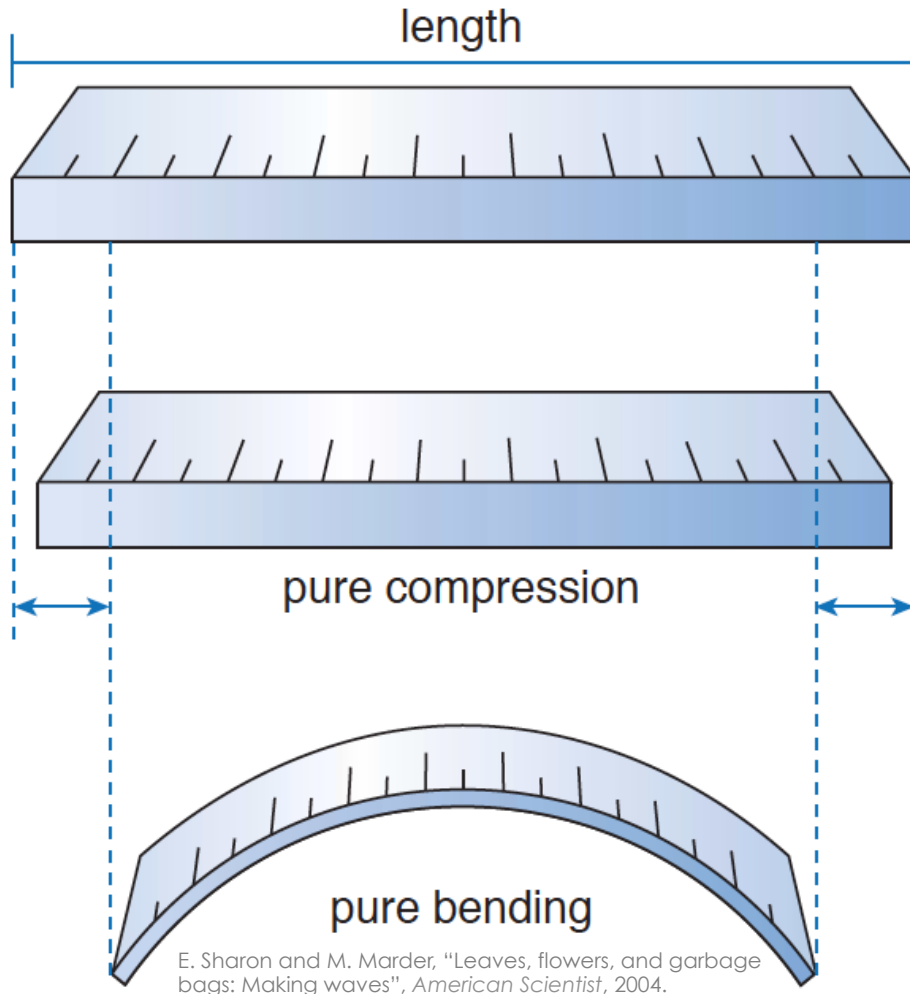
**Scaling** of the elastic energy of the plate

$$\mathcal{U} \sim A \left[ Eh \left( \frac{w}{L} \right)^4 + Eh^3 \left( \frac{w}{L^2} \right)^2 \right]$$





# Mechanics of Plates



## Bending vs. Stretching

$E$  – Elastic Modulus  
 $\varepsilon_{\alpha\beta}$  – in-plane strain

$h$  – thickness  
 $\kappa$  – curvature

$$U_m \sim E h \varepsilon_{\alpha\beta}^2$$

Energy in Compression  $\sim$  thickness

$$U_b \sim E h^3 \kappa^2$$

Energy in Bending  $\sim$  thickness<sup>3</sup>

Thin structures deform by **bending** & avoid **stretching**



# Mechanics of Plates

**Scaling** of the elastic energy of a plate

$$\mathcal{U} \sim A \left[ \underbrace{Eh \left( \frac{w}{L} \right)^4}_{\text{stretching}} + \underbrace{Eh^3 \left( \frac{w}{L^2} \right)^2}_{\text{bending}} \right]$$

Ratio of stretching to bending:

$$\mathcal{U}_m / \mathcal{U}_b \sim w^2 / h^2$$

If deflections of the plate are small, i.e.  $w \ll L$

Stretching can be neglected,

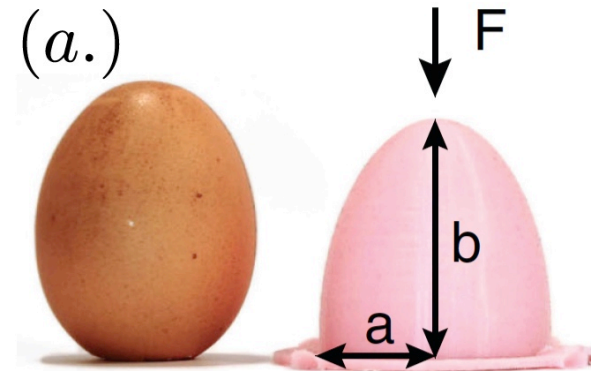
**Stretching can be decoupled from Bending.**



# Mechanics of Shells

**Plates:** Stretching and bending are decoupled. (a.)

**Shells: ?**



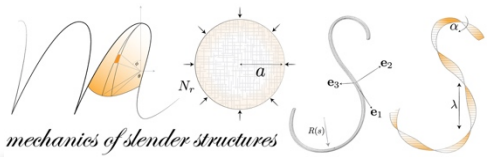
**Scaling** of the elastic energy of a thin structure

$$\mathcal{U} \sim EhA |\varepsilon_{\alpha\beta}|^2 + Eh^3 A |\partial_\alpha \partial_\beta w|^2$$

Assume deformations of an elliptical shell – material points are displaced radially by an amount  $w$

- The relative extension, or strain, is given by:

$$|\varepsilon_{\alpha\beta}| \sim w/R \quad \text{where } R \text{ is the typical radius of curvature}$$



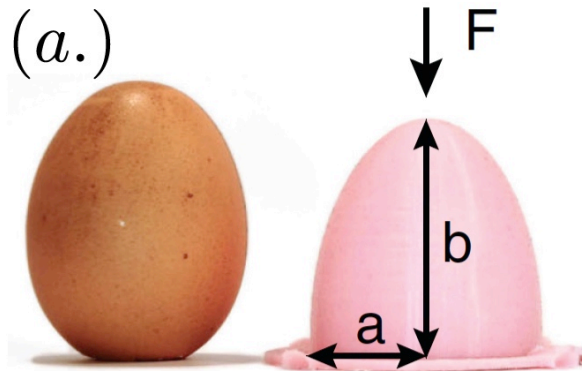
# Mechanics of Shells

**Scaling** of the elastic energy of a thin structure

(a.)

$$\mathcal{U} \sim EhA |\varepsilon_{\alpha\beta}|^2 + Eh^3 A |\partial_\alpha \partial_\beta w|^2$$

$$\text{Strain: } |\varepsilon_{\alpha\beta}| \sim w/R$$



**Stretching energy:**  $\mathcal{U}_s \sim EhA \left(\frac{w}{R}\right)^2$

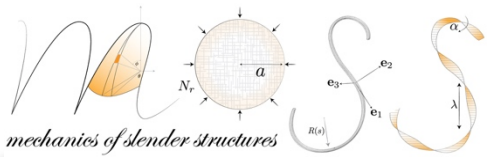
**Bending energy:**  $\mathcal{U}_b \sim Eh^3 A \left(\frac{w}{R^2}\right)^2$

For a uniform deformation...

Ratio of stretching to bending:

$$\mathcal{U}_s / \mathcal{U}_b \sim (R/h)^2 \gg 1$$

**Stretching with bending for a curved shell is always a first order effect.**



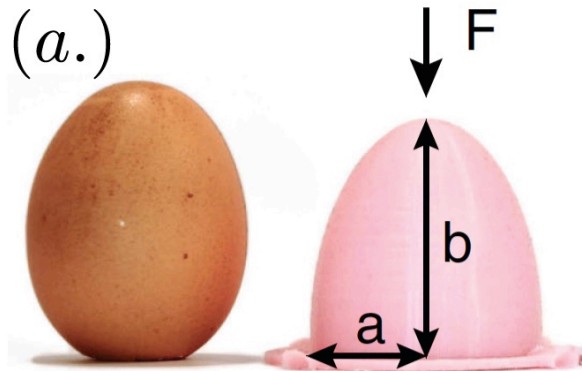
# Mechanics of Shells

**Scaling** of the elastic energy of a thin structure

(a.)

$$\mathcal{U} \sim EhA|\varepsilon_{\alpha\beta}|^2 + Eh^3A|\partial_\alpha\partial_\beta w|^2$$

$$\text{Strain: } |\varepsilon_{\alpha\beta}| \sim w/R$$



**Point force applied to the shell, causing a local deformation of  $\ell$**

**Bending energy:**

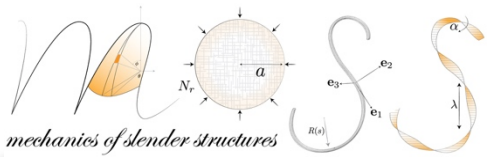
For a point force deformation...

$$\mathcal{U}_b \sim Eh^3A \left( \frac{w}{\ell^2} \right)^2$$

Scaling law for deformation depth:

$$\ell \sim \sqrt{Rh}$$

Balancing stretching and bending energies (equivalent to minimizing total elastic energy):

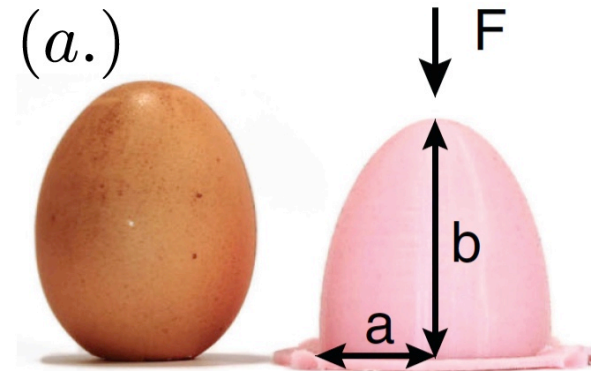


# Mechanics of Shells

## Fundamental equations of **thin shell theory**

$$\text{in-plane strain: } \varepsilon_{\alpha\beta} \equiv \frac{1}{2}(a_{\alpha\beta} - \bar{a}_{\alpha\beta})$$

$$\text{curvature strain: } \kappa_{\alpha\beta} \equiv b_{\alpha\beta} - \bar{b}_{\alpha\beta}$$

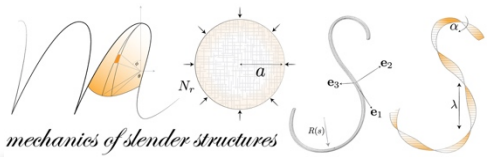


Stress and moment are derived from the total elastic energy

$$\sigma^{\alpha\beta} \equiv \frac{\delta\mathcal{U}}{\delta\varepsilon_{\alpha\beta}} \equiv \frac{Eh}{1-\nu^2} \left[ (1-\nu)\varepsilon^{\alpha\beta} + \nu\bar{a}^{\alpha\beta}\varepsilon^{\gamma}_{\gamma} \right] \quad \sigma^{\alpha\beta} \sim \mathcal{O}(Eh\varepsilon)$$

$$\mu^{\alpha\beta} \equiv \frac{\delta\mathcal{U}}{\delta\kappa_{\alpha\beta}} \equiv \frac{Eh^3}{12(1-\nu^2)} \left[ (1-\nu)\kappa^{\alpha\beta} + \nu\bar{a}^{\alpha\beta}\kappa^{\gamma}_{\gamma} \right] \quad \mu^{\alpha\beta} \sim \mathcal{O}(Eh^3\kappa)$$

Shell equations are derived by expressions the variations in the in-plane and curvature strains.



# Mechanics of Shells

Fundamental equations of **thin shell theory**

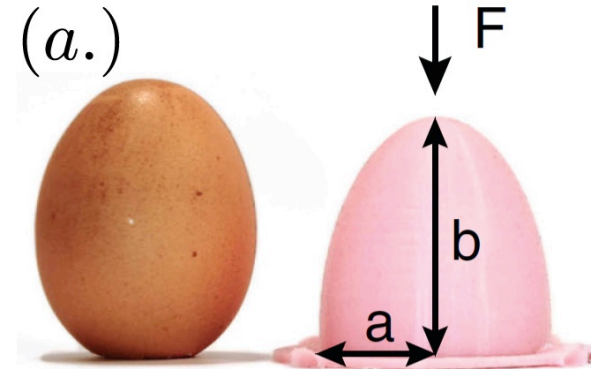
$$\underbrace{\bar{\nabla}_\alpha \sigma^{\alpha\beta}}_{\sim \mathcal{O}\left(\frac{Eh\varepsilon}{l}\right)} + \underbrace{2\bar{b}^\beta_\gamma \bar{\nabla}_\alpha \mu^{\gamma\alpha}}_{\sim \mathcal{O}\left(\frac{Eh^3\kappa}{Rl}\right)} + \underbrace{\mu^{\gamma\alpha} \bar{\nabla}_\alpha \bar{b}^\beta_\gamma}_{\sim \mathcal{O}\left(\frac{Eh^3\kappa}{Rl}\right)} + f^\beta = 0$$

Ratio of first term to next two

$$\text{ratio} \sim \mathcal{O}\left(\frac{\varepsilon}{h\kappa}\right) \mathcal{O}\left(\frac{R}{h}\right)$$

If  $h\kappa \lesssim \varepsilon$

$$\bar{\nabla}_\alpha \sigma^{\alpha\beta} + f^\beta = 0$$



$$\sigma^{\alpha\beta} \sim \mathcal{O}(Eh\varepsilon)$$

$$\mu^{\alpha\beta} \sim \mathcal{O}(Eh^3\kappa)$$

$$l \sim (Rh)^{1/2}$$

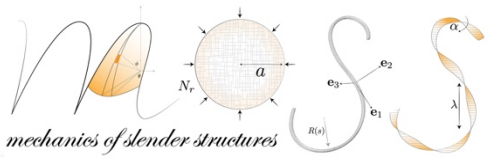
$$\bar{\nabla}_\alpha \bar{\nabla}_\beta \mu^{\alpha\beta} - \underbrace{\bar{b}_{\alpha\gamma} \bar{b}^\beta_\gamma \mu^{\alpha\beta}}_{\sim \mathcal{O}\left(\frac{Eh^3\kappa}{R^2}\right)} - \underbrace{\bar{b}_{\alpha\beta} \sigma^{\alpha\beta}}_{\sim \mathcal{O}\left(\frac{Eh\varepsilon}{R}\right)} - p = 0$$

Ratio of second term to third term

$$\text{ratio} \sim \mathcal{O}\left(\frac{\varepsilon}{h\kappa}\right) \mathcal{O}\left(\frac{R}{h}\right)$$

If  $h\kappa \lesssim \varepsilon$

$$\bar{\nabla}_\alpha \bar{\nabla}_\beta \mu^{\alpha\beta} - \bar{b}_{\alpha\beta} \sigma^{\alpha\beta} - p = 0$$



# Mechanics of Shells

Approximate equations of **thin shell theory**

$$\bar{\nabla}_{\alpha} \sigma^{\alpha\beta} + f^{\beta} = 0$$

$$\bar{\nabla}_{\alpha} \bar{\nabla}_{\beta} \mu^{\alpha\beta} - \bar{b}_{\alpha\beta} \sigma^{\alpha\beta} - p = 0$$

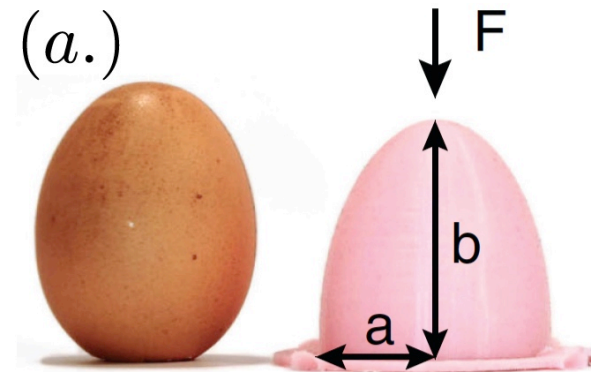
As long as the Gaussian curvature is much less the curvature set by the deformation length:

$$\mathcal{K} \ll 1/\ell^2$$

This problem can be solved by the introduction of particular scalar field (i.e. Airy potential), and ensuring compatibility between the strains.

$$\frac{1}{Eh} \bar{\nabla}^4 \phi + \bar{b}_{\alpha\beta} \epsilon^{\alpha\gamma} \epsilon^{\beta\gamma} \bar{\nabla}_{\gamma} \bar{\nabla}_{\lambda} w = 0$$

$$B \bar{\nabla}^4 w - \bar{b}_{\alpha\beta} \epsilon^{\alpha\gamma} \epsilon^{\beta\gamma} \bar{\nabla}_{\gamma} \bar{\nabla}_{\lambda} \phi = 0$$



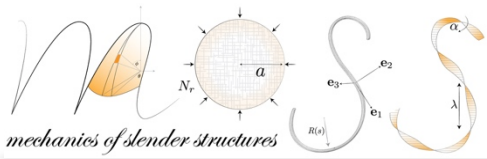
$$\sigma^{\alpha\beta} \sim \mathcal{O}(Eh\varepsilon)$$

$$\mu^{\alpha\beta} \sim \mathcal{O}(Eh^3\kappa)$$

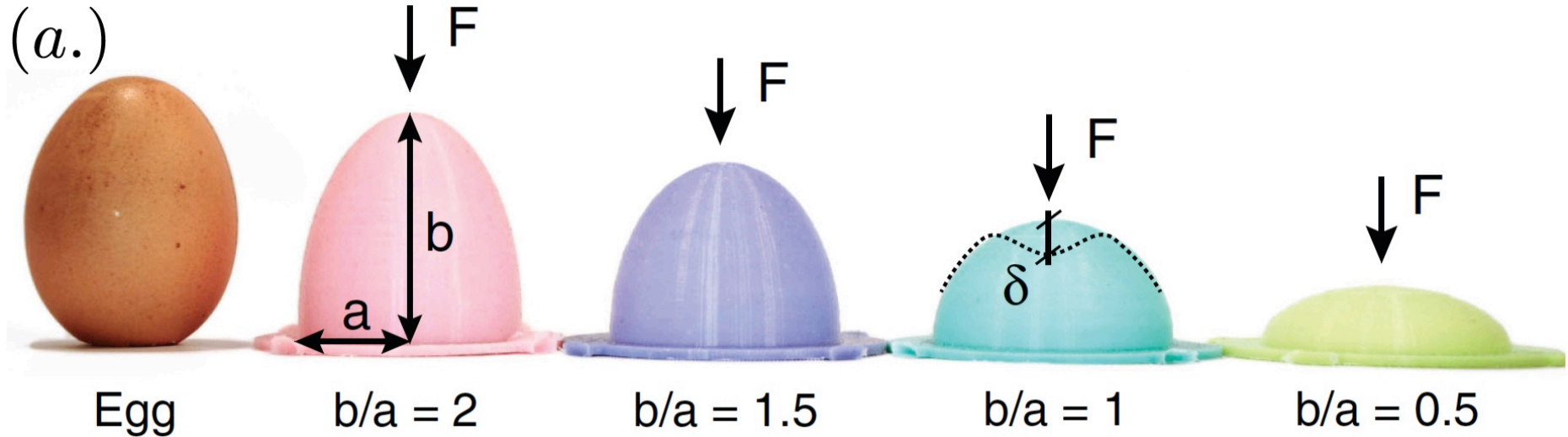
$$\ell \sim (Rh)^{1/2}$$

**Donnell-Mushtaru-Vlasov (DMV) Equations**





# Geometric Rigidity



DMV Equations (point load at the apex)

Introducing the operators:

$$\nabla_k^2(f) \equiv R_\beta^{-1} f_{,\alpha\alpha} + R_\alpha^{-1} f_{,\beta\beta}$$

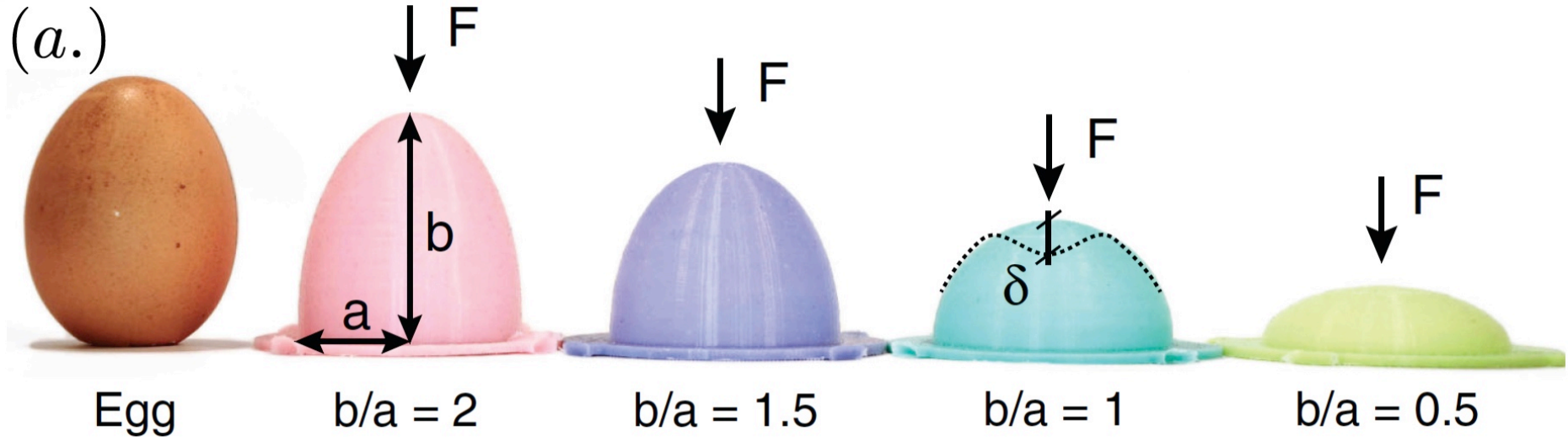
$$\diamond^4[f, g] \equiv f_{,\alpha\alpha} g_{,\beta\beta} - 2f_{,\alpha\beta} g_{,\alpha\beta} + f_{,\beta\beta} g_{,\alpha\alpha}$$

$$B\nabla^4 w - \nabla_k^2 \phi - \diamond^4[\phi, w] = p - \frac{F}{2\pi} \frac{\delta(r)}{r}$$

Can be reduced to an ODE by assuming the pressurized shell is experiencing a uniform state of stress:  $\sigma = pR/2$

$$B\nabla^4 w - \sigma \nabla_k^2 w - \frac{Eh}{R^2} w = -\frac{F}{2\pi} \frac{\delta(r)}{r}$$

# Geometric Rigidity



Integration of the ODE leads to a linear relationship between force and displacement:

$$F = k_1 w_0$$

The shell's rigidity is given by:  $k_1 \sim \frac{4\pi B}{\ell_b^2} \frac{\tau}{\log 2\tau}$

Dimensionless pressure:

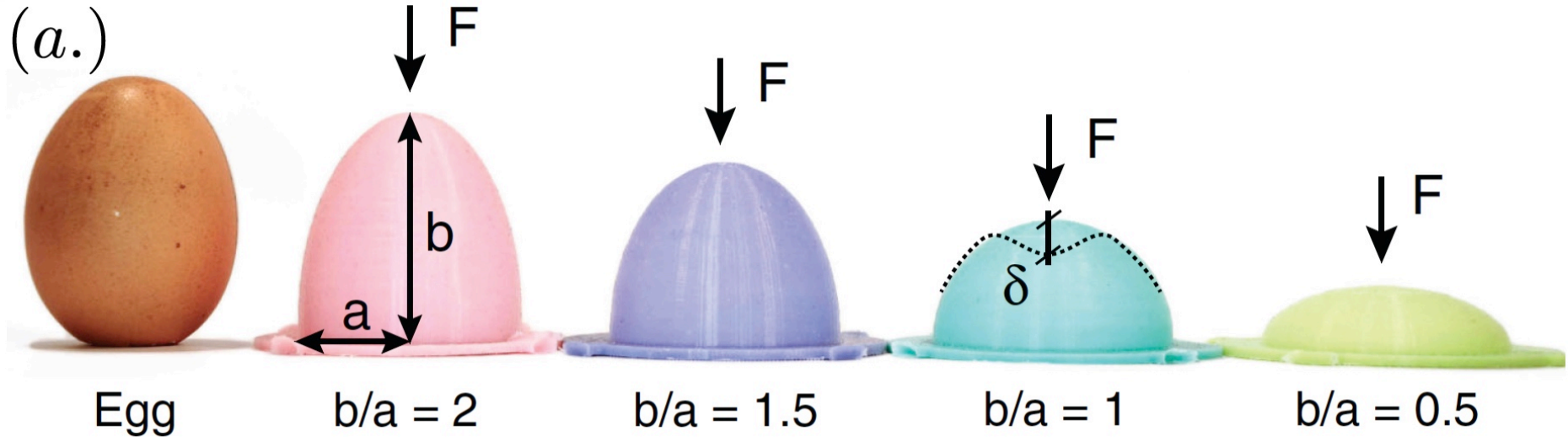
$$\tau = \frac{1}{4} p R^2 (EhB)^{-1/2}$$

In the Reissner limit of  $\tau \ll 1$

$$k_1 = \frac{8B}{\ell_b^2} = 8\sqrt{BEh\mathcal{K}}$$

The shell's **rigidity** is tied to its **Gaussian curvature** – **deforming** the shell requires in-plane **stretching**, which is energetically costly.

# Geometric Rigidity



In the Reissner limit of  $\tau \ll 1$

$$k_1 = \frac{8B}{\ell_b^2} = 8\sqrt{BEh\mathcal{K}}$$

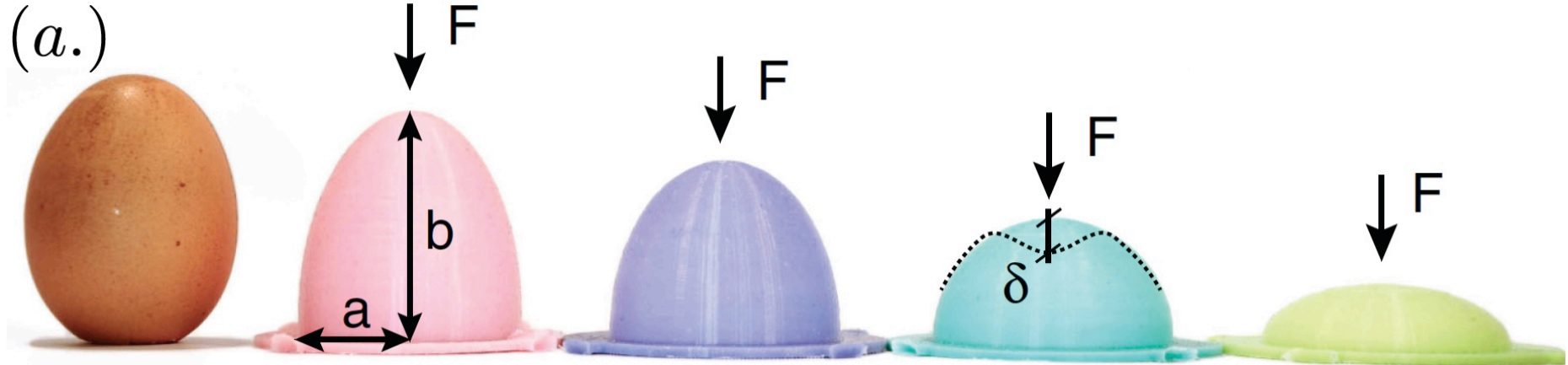
For an elliptical shell,  $R = a^2/b$

$$k_1 = \frac{b}{a} k_1^s$$

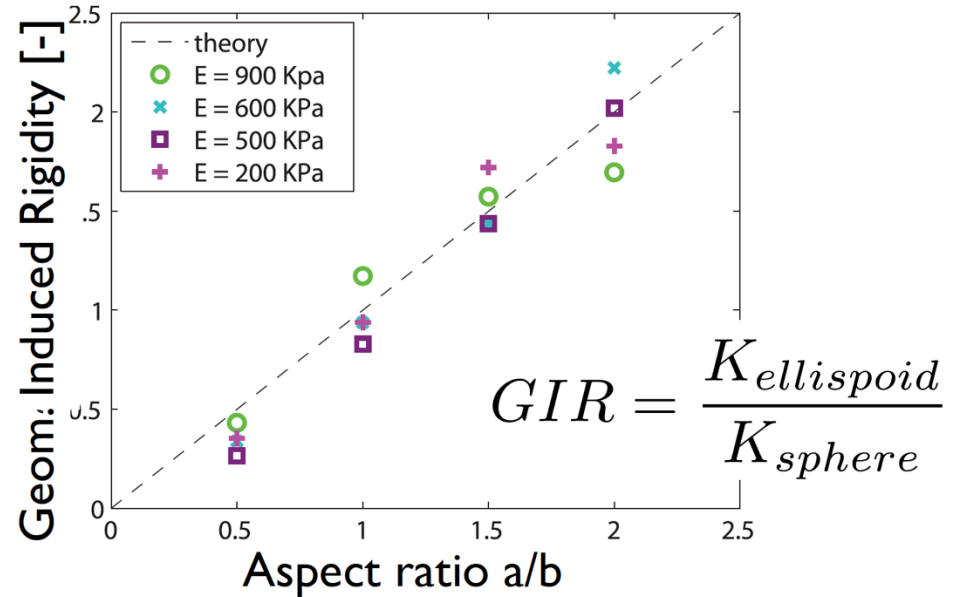
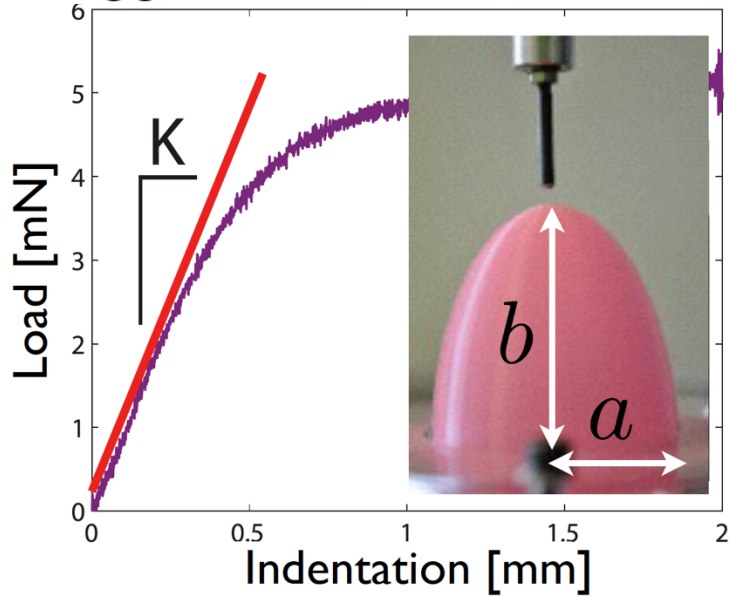
The shell's **rigidity** is tied to its **Gaussian curvature** – **deforming** the shell requires in-plane **stretching**, which is energetically costly.

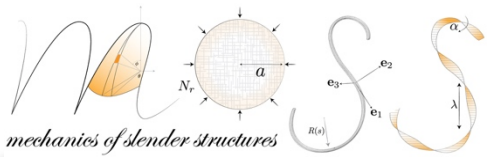
Enhancement of structural rigidity by changing the shell's aspect ratio.

# Geometric Rigidity



Egg       $b/a = 2$        $b/a = 1.5$        $b/a = 1$        $b/a = 0.5$





# Geometric Rigidity

Large deformations:  
(From experiments and numerics)

A second linear relationship  
between force and displacement:

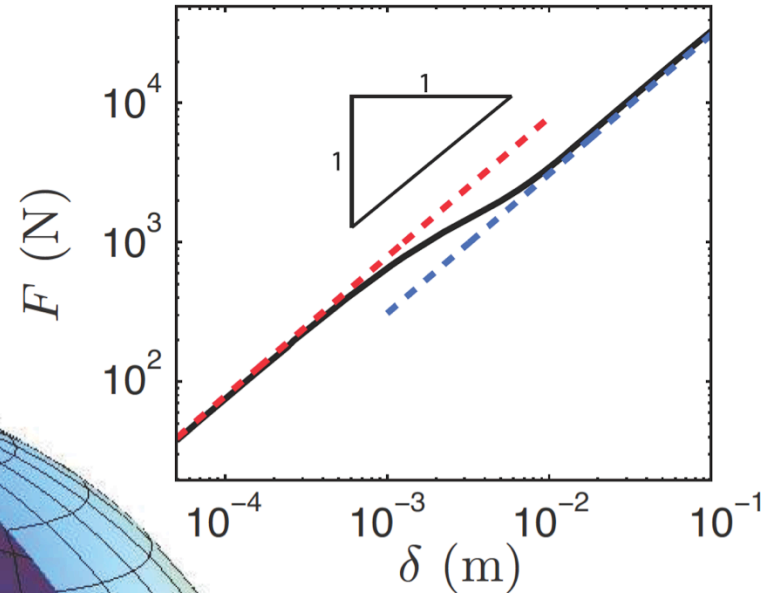
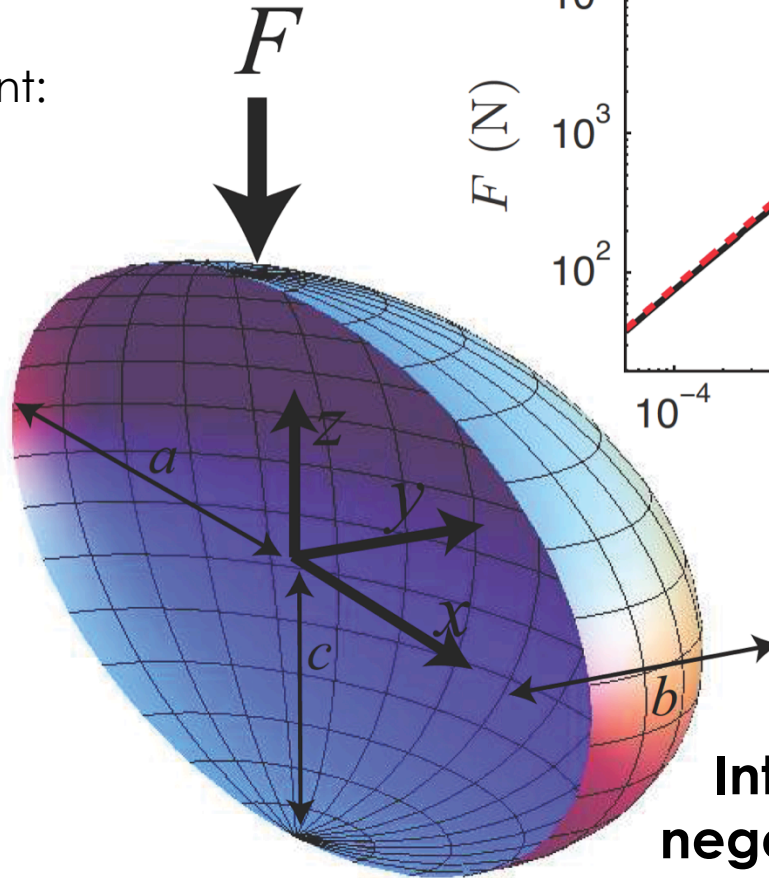
$$F = k_2 w_0$$

Considering the limit of large pressures, bending can be neglected:  
(Balance of in-plane stretching to shell stretching from internal pressure)

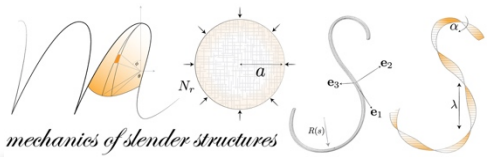
$$l_p \sim R \left( \frac{pR}{Eh} \right)^{1/2}$$

**Mean Curvature**  
governs shell rigidity

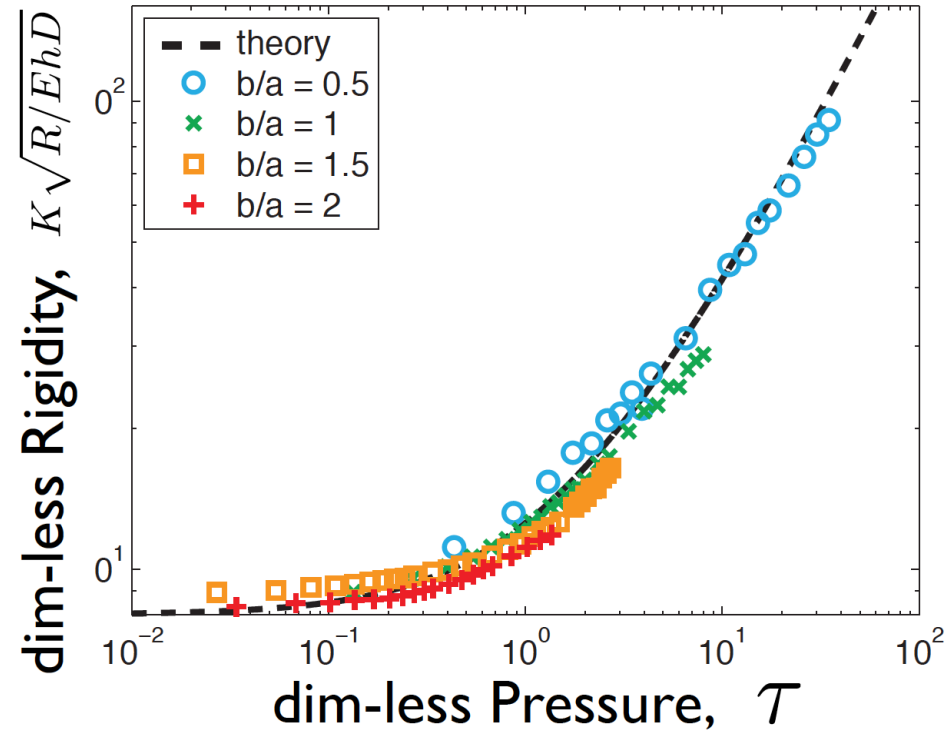
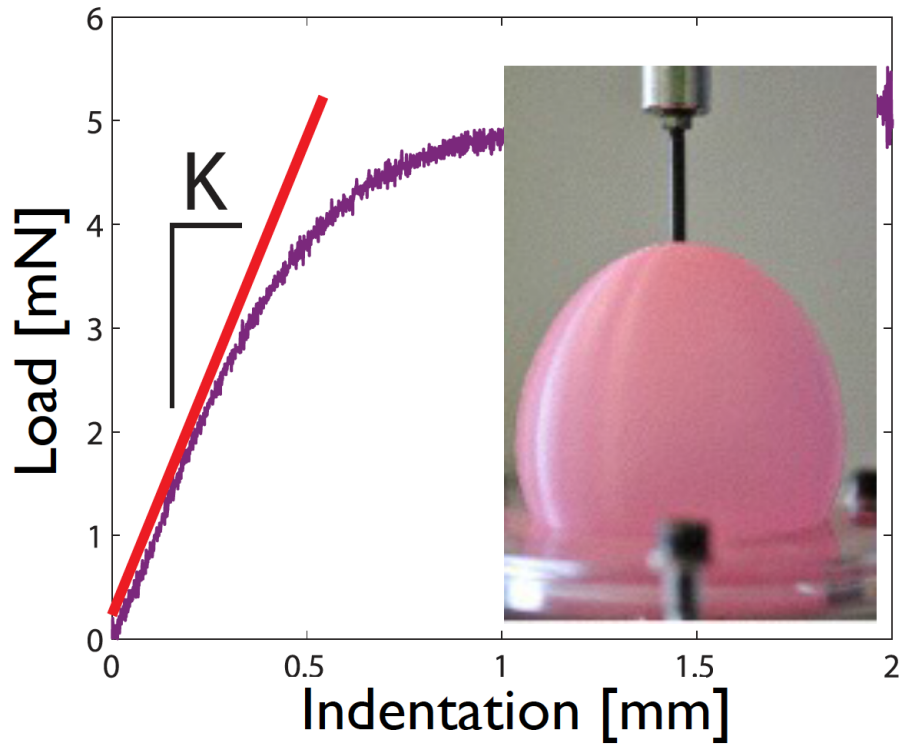
$$k_2 = \pi p \mathcal{H}$$

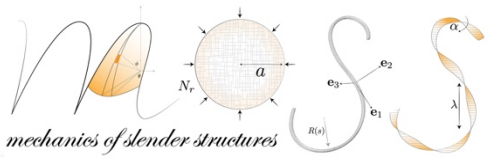


**Internal pressure  
negates the effect of  
geometric rigidity**



# Geometric Rigidity





# Geometric Rigidity

## Saccharomyces cerevisiae



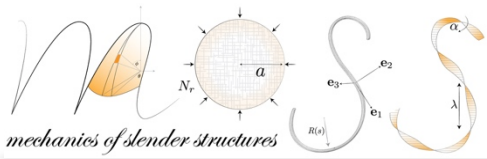
AFM experiments measured yeast cell stiffness as osmotic pressure was varied.

Deflections  $\sim$  thickness, low internal pressure.

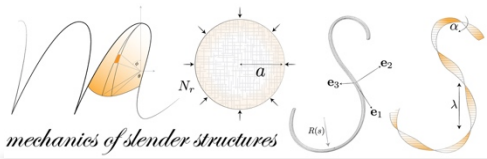
Turgor pressure estimated of 0.1 to 0.2 MPa, consistent with measurements from other techniques.

Same technique being used to measure **elastic** properties of **tomato fruit cells**, **plant tissues**, and **artificial biological microcapsules**.



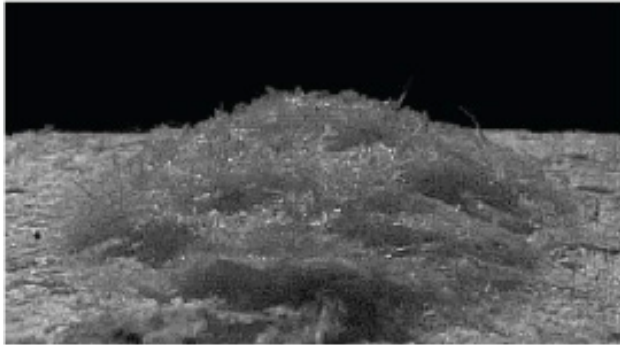
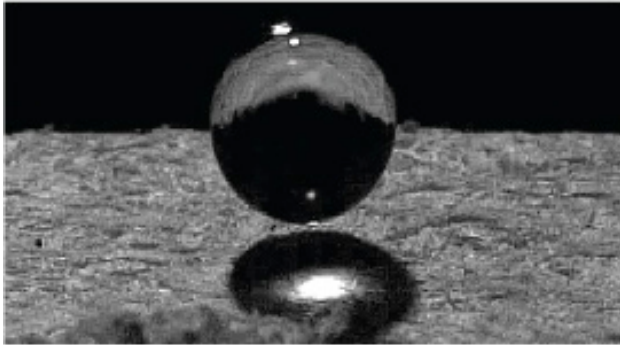


# Geometric Morphing



# Swelling & Growth

## Materials Science



Swelling of a sponge.

## Mechanics



An almond leaf which was attacked by *Taphrina Deformans*.



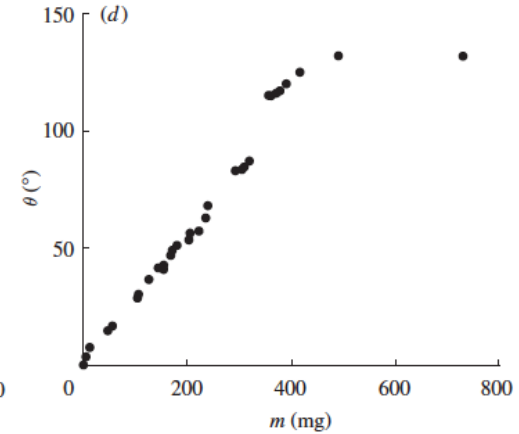
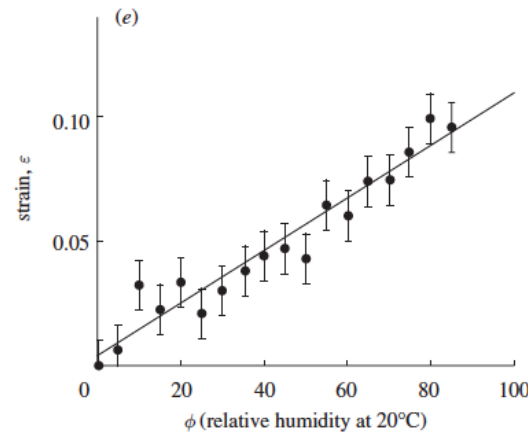
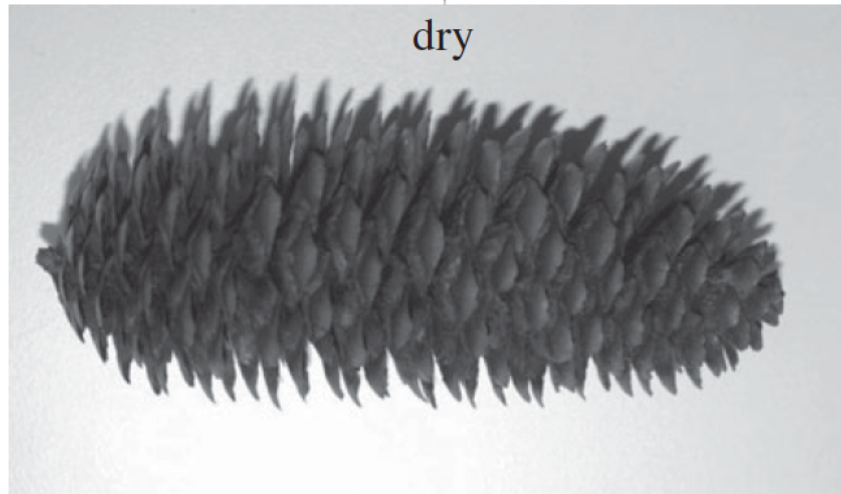
# Pine Cones



wet



dry

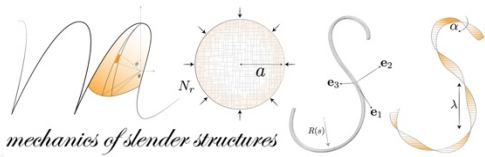


## Tree-bound pine cones:

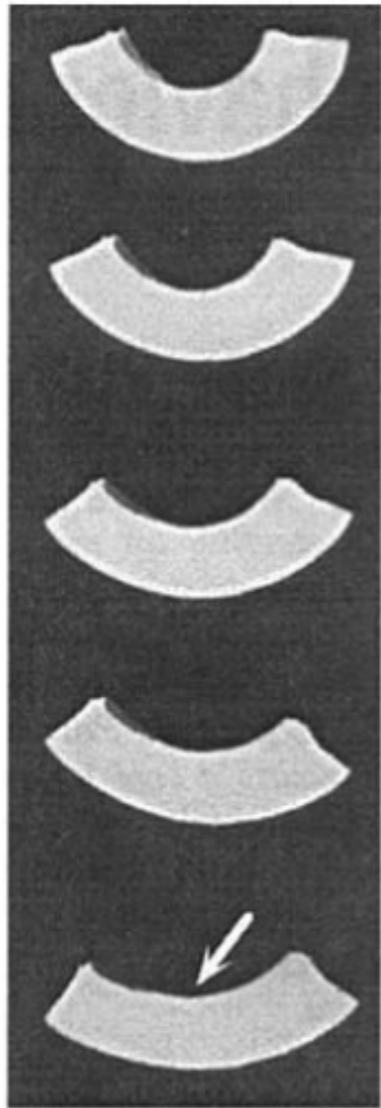
Hydrated & **closed**, protecting seeds

## Fallen pine cones:

Dried out & **opened**, releasing seeds



# Articular Cartilage



0.015M NaCl

Shape change caused by ion concentration.

0.05M NaCl

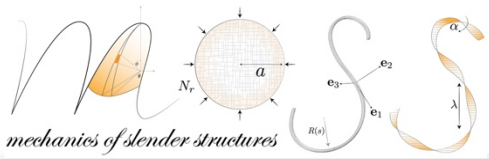
0.15 M NaCl

0.5 M NaCl

2M NaCl

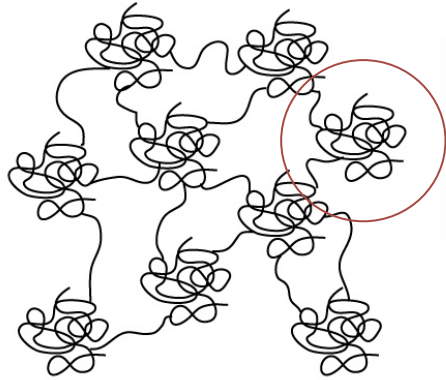
Residual strain at physiological conditions: **3-15%**

Tensile prestress in cartilage protective against frequent compresses forces.



# Swelling & Growth

Crosslinked Polymer



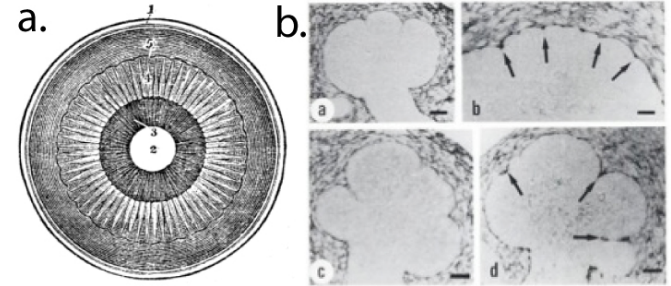
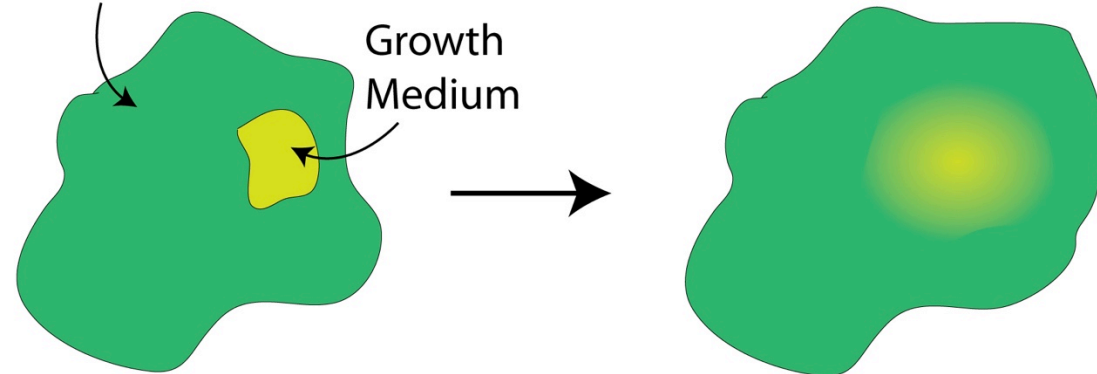
Good Solvent



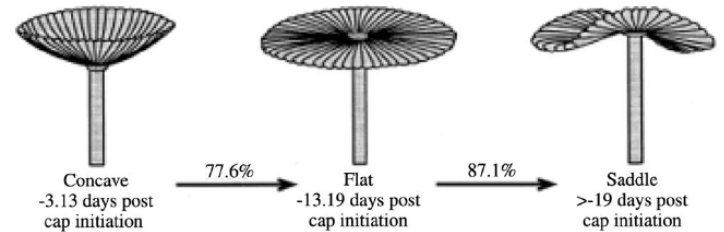
Bad Solvent

Soft Structure

Growth Medium



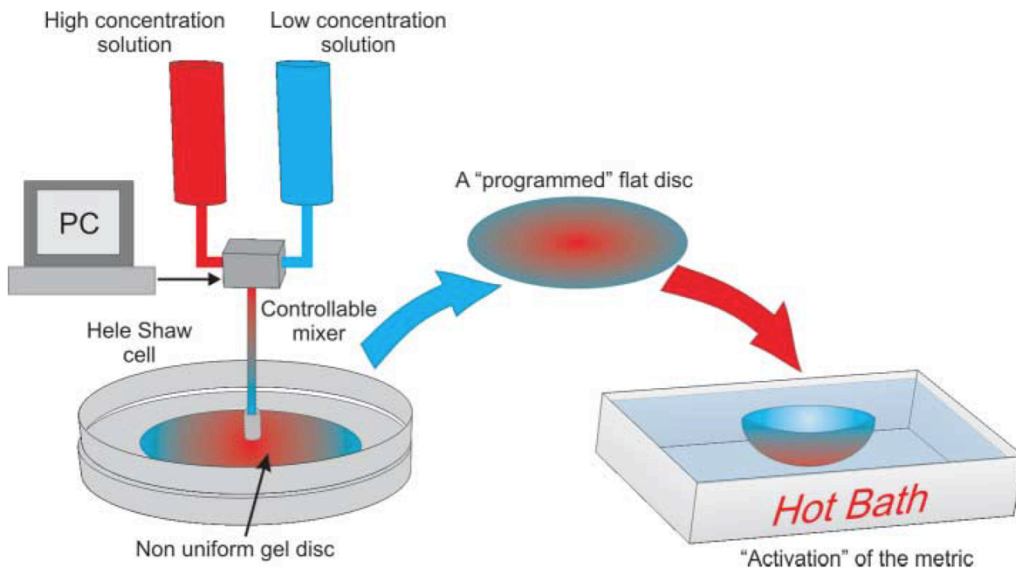
Deformation confined to surface



Deformation of entire structure

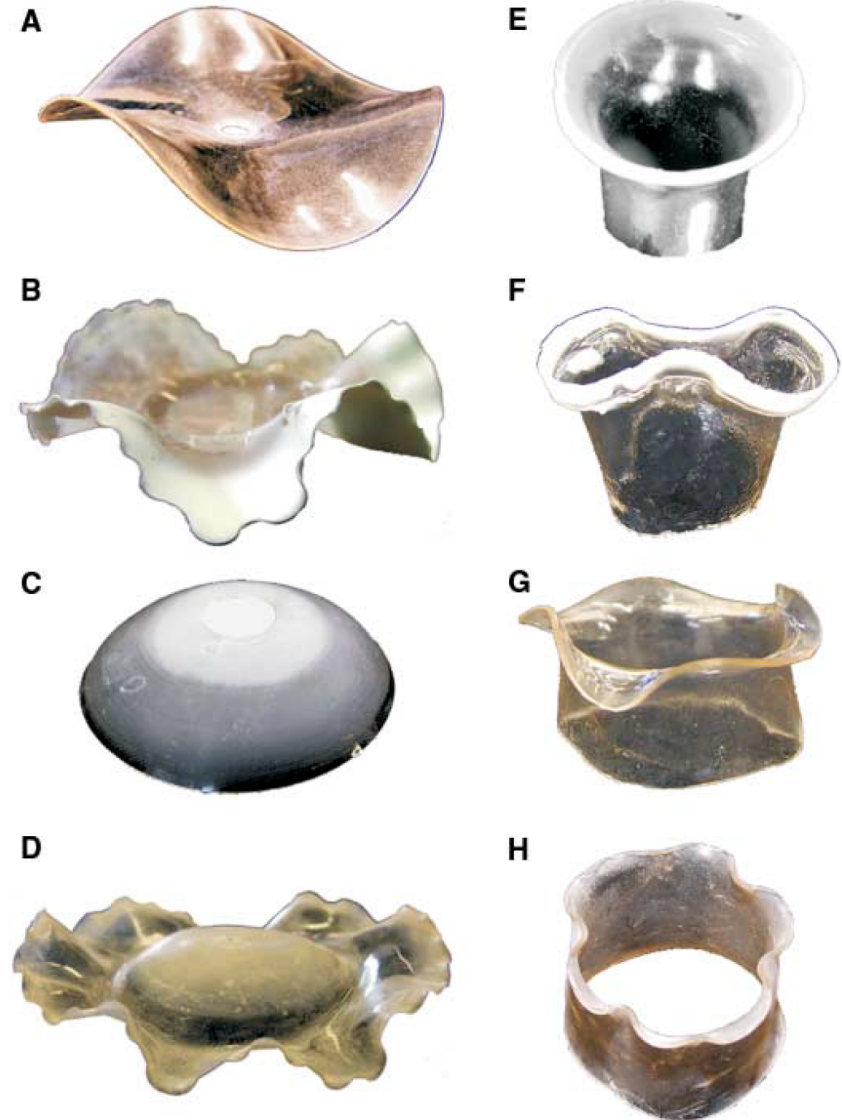
J. Bard. *Morphogenesis: The cellular and molecular processes of developmental anatomy*, Cambridge University Press, 1990.  
 J. Dervaux and M. Ben Amar. *Physical Review Letters*, **101**, 068101, 2008.

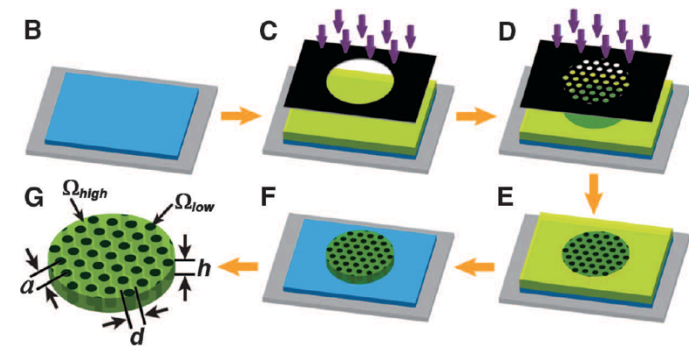




## Shaping elastic sheets by prescribing non-Euclidean metrics

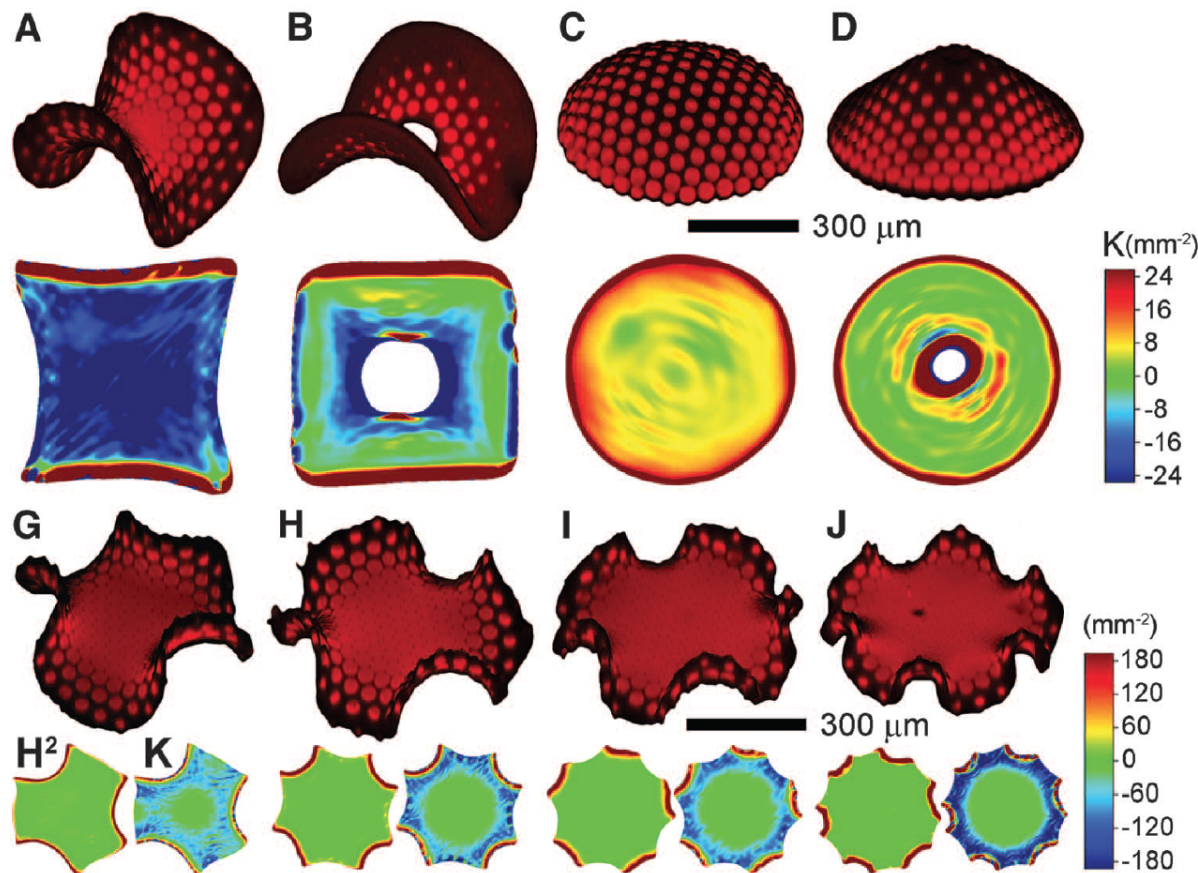
- Prepare gels that undergo nonuniform shrinkage.
- Buckling thin films based on chosen metrics.

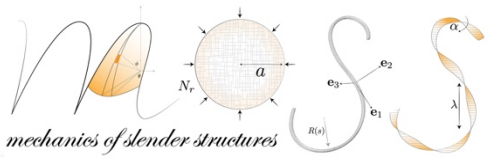




## Shaping elastic sheets by half-tone gel lithography

- Photopattern thin films.
- Thermal-actuated shape change.
- Swell to embedding based on prescribed metric.

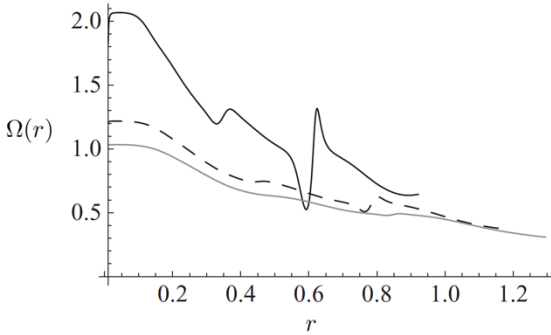




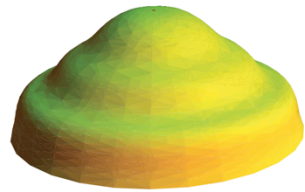
# Shaping Sheets

Programmed buckling by controlled lateral swelling in a thin elastic sheet

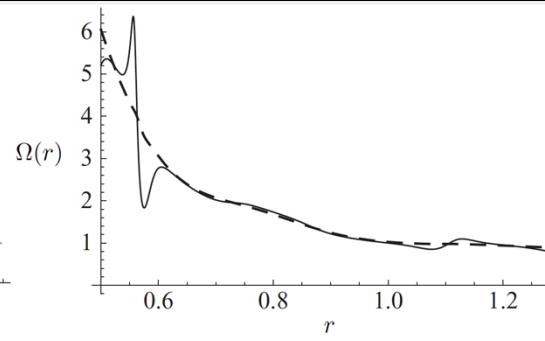
- How to prescribe a metric to produce a desired shape.
- Axisymmetric 3D structures.



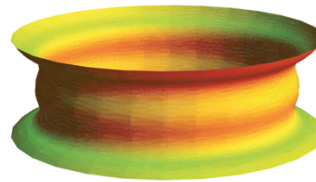
(a) Swelling factor



(b) Ziggurat



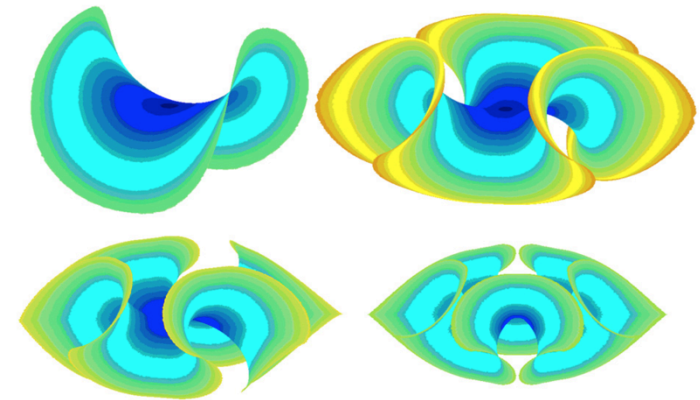
(a) Swelling factor



(b) Sheave

## Shape selection in non-Euclidean plates

- Existence of local isometries with waves that increase with radius.
- Energetically favorable to form lobes rather than saddles.



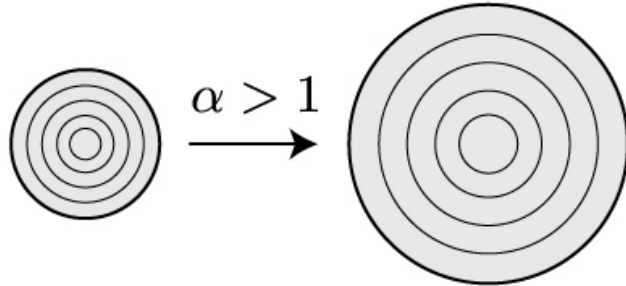




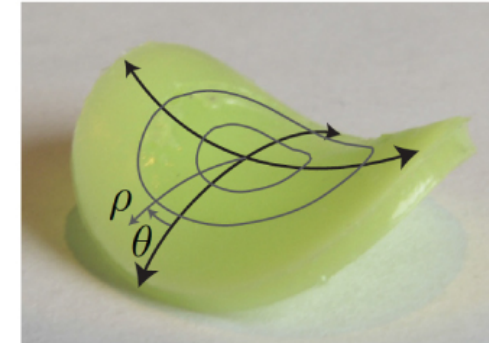
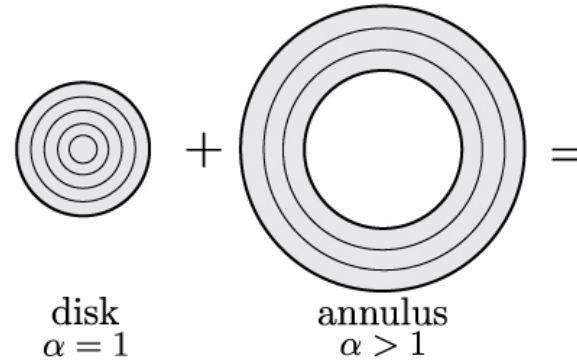
# Geometric Composite

## Independent Homothety

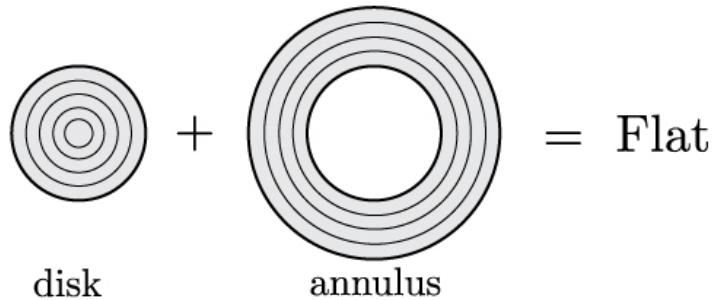
### Homothetic Transformation



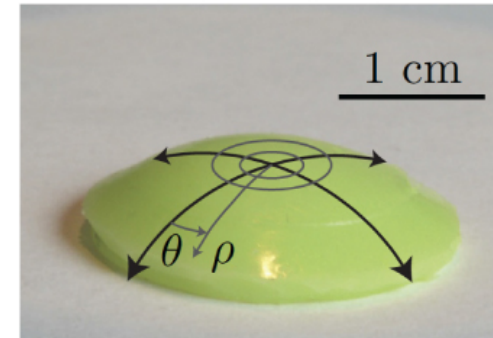
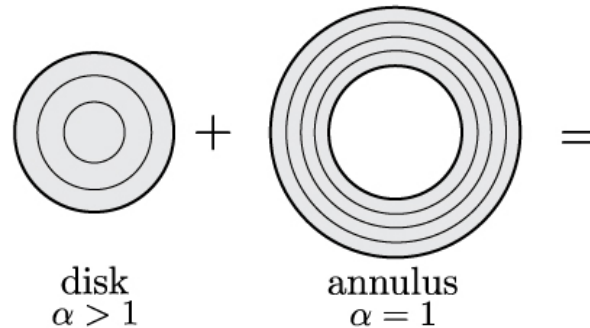
### Stretched Annulus



### Reference Configuration



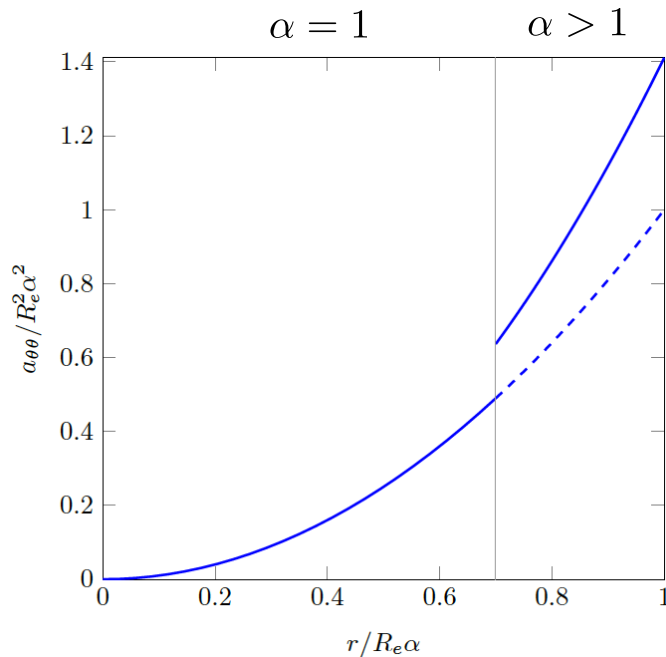
### Stretched Disk



**Goal:** Use swelling to predictably & permanently morph plates into shells



# Geometric Composite

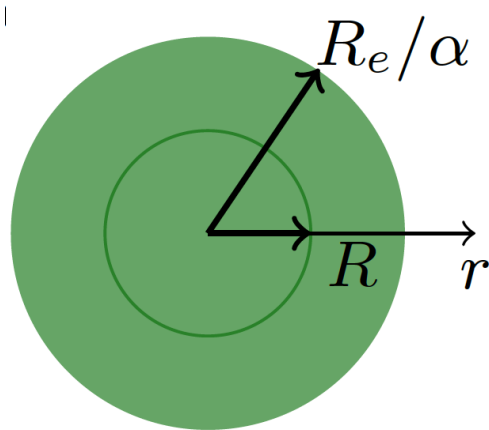


## Stretching Energy of the Plate (incompressible)

$$\mathcal{U}_s \simeq h \int_A E [\underbrace{\text{tr}^2(a - \bar{a})}_{\text{Realized metric}} + \underbrace{\text{tr}(a - \bar{a})^2}_{\text{Target metric}}] \sqrt{|\bar{a}|} dA$$

## Target Metric (Polar Coordinates)

$$\bar{a} = f^2(r) \begin{pmatrix} 1 & 0 \\ 0 & r^2 \end{pmatrix}, \quad f^2(r) = \begin{cases} 1, & r \leq R \\ \alpha^2, & r > R \end{cases}$$

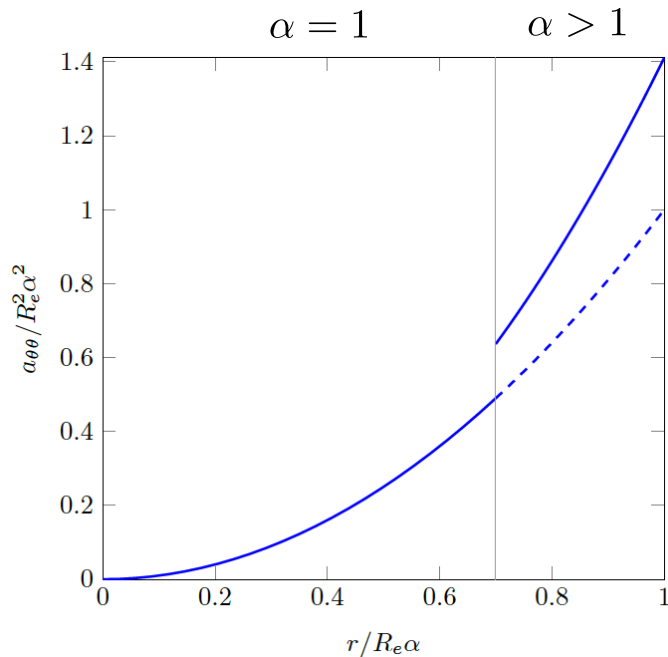


## Stretching Energy (Assume all strains zero, except: $a_{\theta\theta} - \bar{a}_{\theta\theta}$ )

$$\mathcal{U}_s \simeq Eh \int_0^R \frac{(a_{\theta\theta} - r^2)^2}{r^3} dr + Eh \int_R^{R_e/\alpha} \frac{(a_{\theta\theta} - \alpha^2 r^2)^2}{\alpha^2 r^3} dr$$



# Geometric Composite



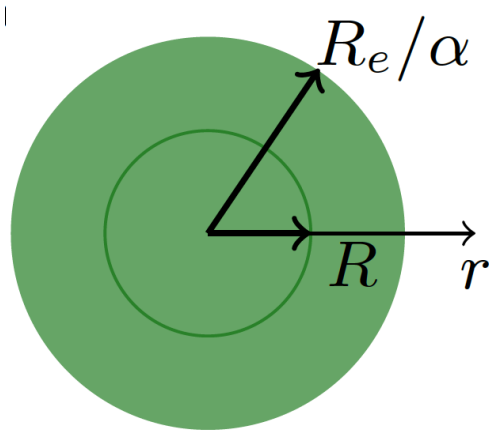
**Stretching Energy** (Assume all strains zero, except:  $a_{\theta\theta} - \bar{a}_{\theta\theta}$ )

$$\mathcal{U}_s \simeq Eh \int_0^R \frac{(a_{\theta\theta} - r^2)^2}{r^3} dr + Eh \int_R^{R_e/\alpha} \frac{(a_{\theta\theta} - \alpha^2 r^2)^2}{\alpha^2 r^3} dr$$

**Realized Metric:** Gauss Normal Coordinates

$$\rho(r) = \int_0^r \sqrt{a_{rr}(r')} dr'$$

$\rho(r)$  measures the arc length along radial geodesics

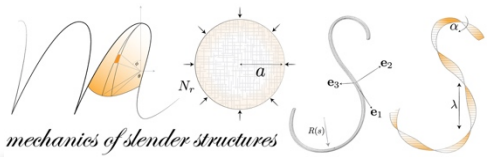


First Fundamental Form  
 $ds^2 = d\rho^2 + a_{\theta\theta}(\rho)d\theta^2$

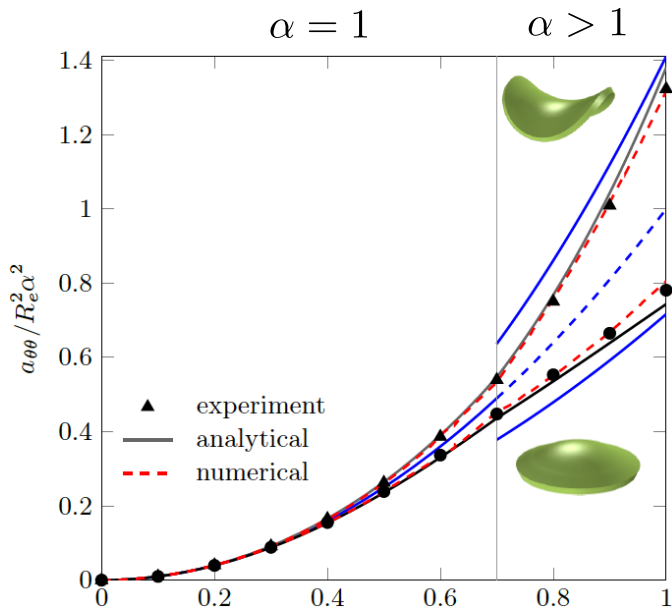
Gaussian curvature  
 $-\partial_{\rho\rho}\sqrt{a_{\theta\theta}}/\sqrt{a_{\theta\theta}}$

**Minimize Stretching Energy** (Constant  $K$  metric)

$$a_{\theta\theta}(\rho) = (\sin(\sqrt{K}\rho)/\sqrt{K})^2$$



# Geometric Composite



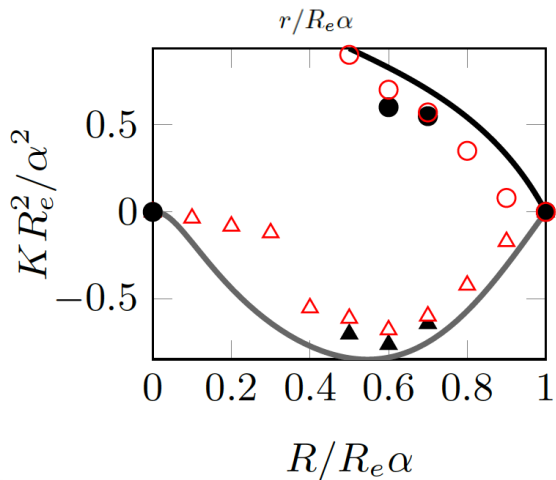
**Minimize Stretching Energy** (Constant  $K$  metric)

$$a_{\theta\theta}(\rho) = (\sin(\sqrt{K}\rho) / \sqrt{K})^2$$

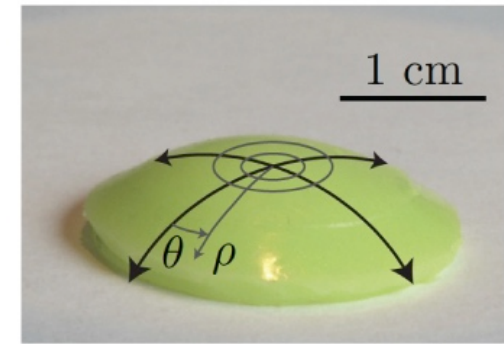
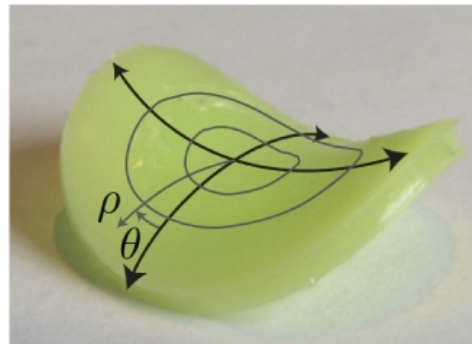
**Taylor Expand**  $a_{\theta\theta}(\rho)$  (Assume:  $|K| < \alpha^2 / R_e^2$ )

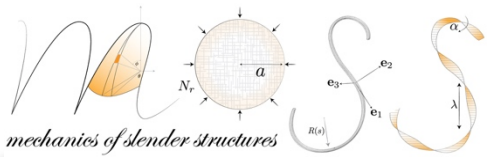
$$a_{\theta\theta}(\rho) = \rho^2 - \frac{K}{3}\rho^4 + \mathcal{O}(\rho^5)$$

$\uparrow$  Flat metric       $\uparrow$  Kind of non-Euclidean metric



**Experiments: Mechanical Strain**





# Residual Swelling

## Polyvinylsiloxane (PVS) – Zhermack Elite Double



$$E = 0.226 \text{ MPa}$$

$$\nu = 0.5$$

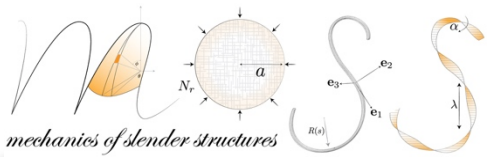


$$E = 0.963 \text{ MPa}$$

$$\nu = 0.5$$

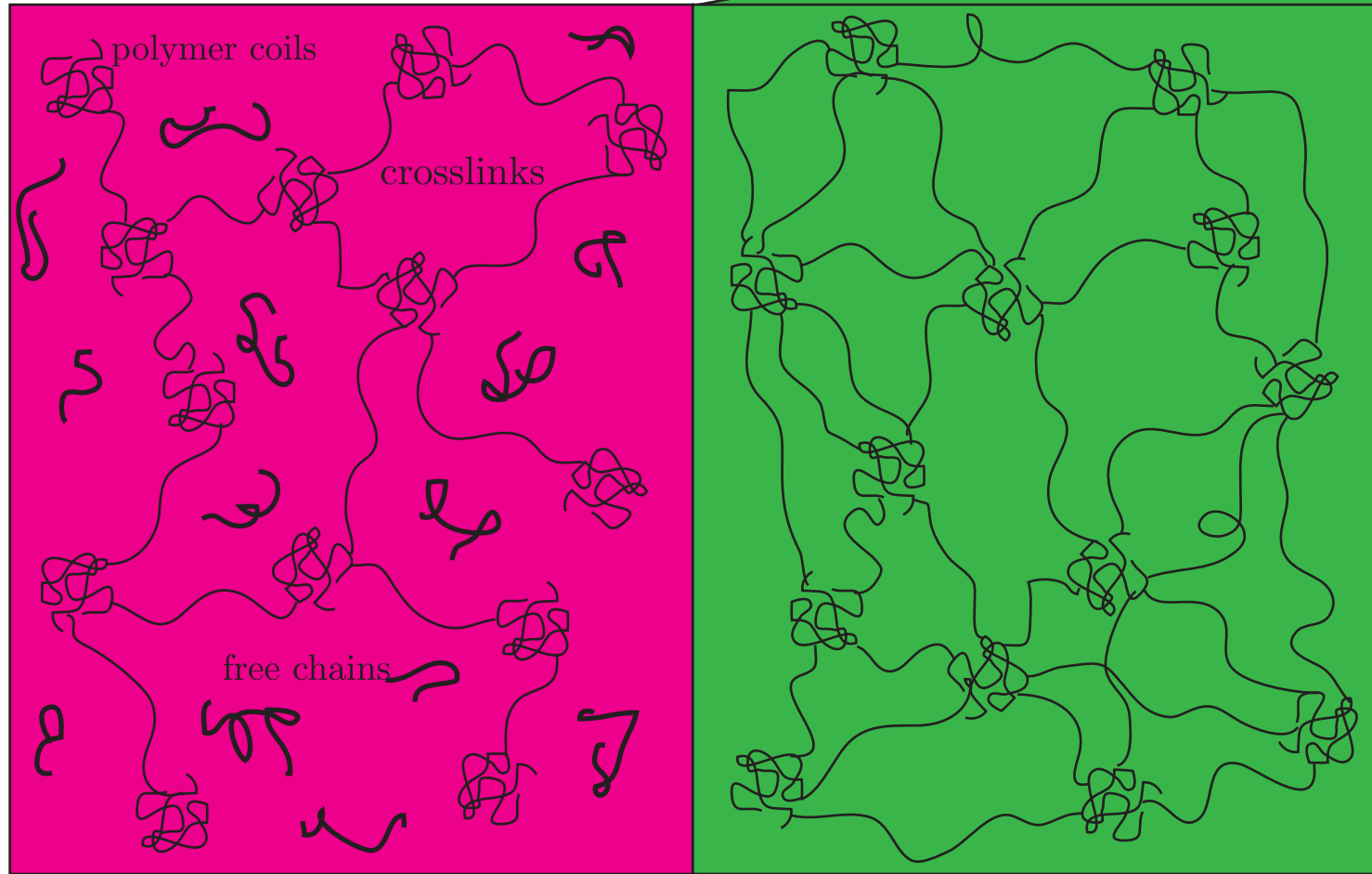
### Characteristics:

- Fast curing at room temperature.
- Easily vary elastic modulus
- The elastomer contains free, uncrosslinked polymer chains.



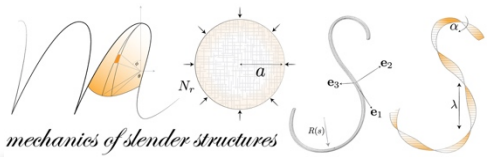
# Residual Swelling

The elastomer contains free, uncrosslinked polymer chains.

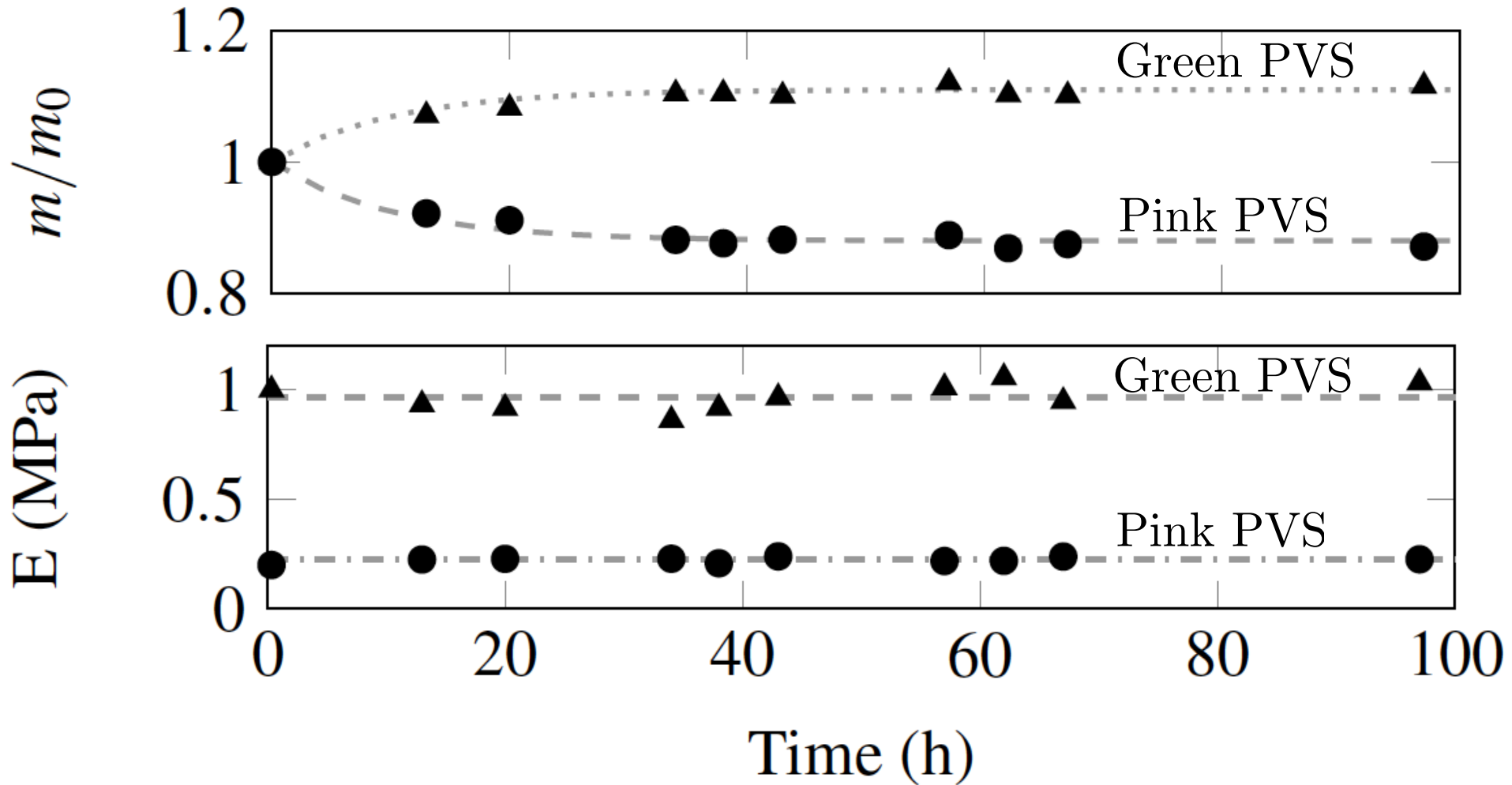


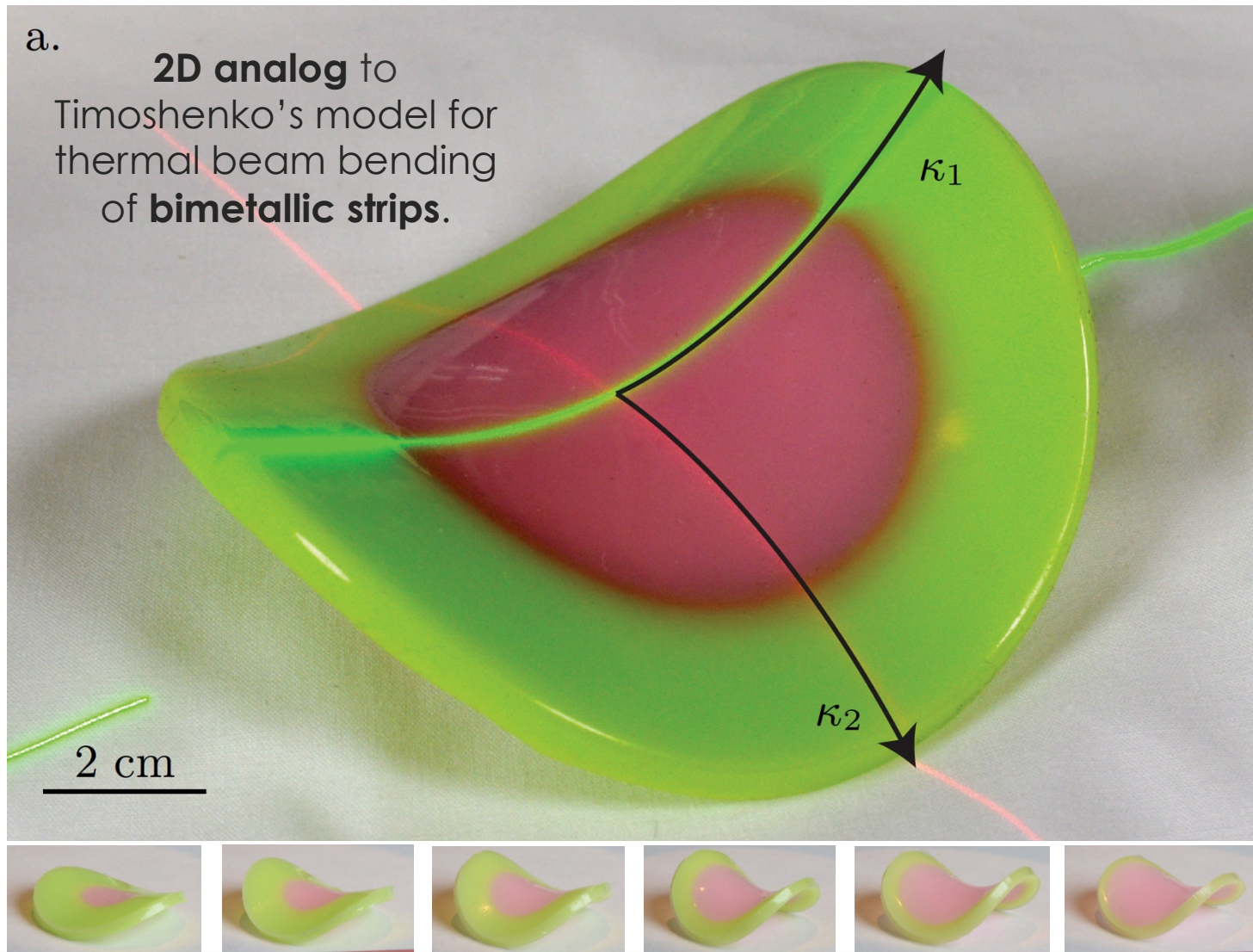
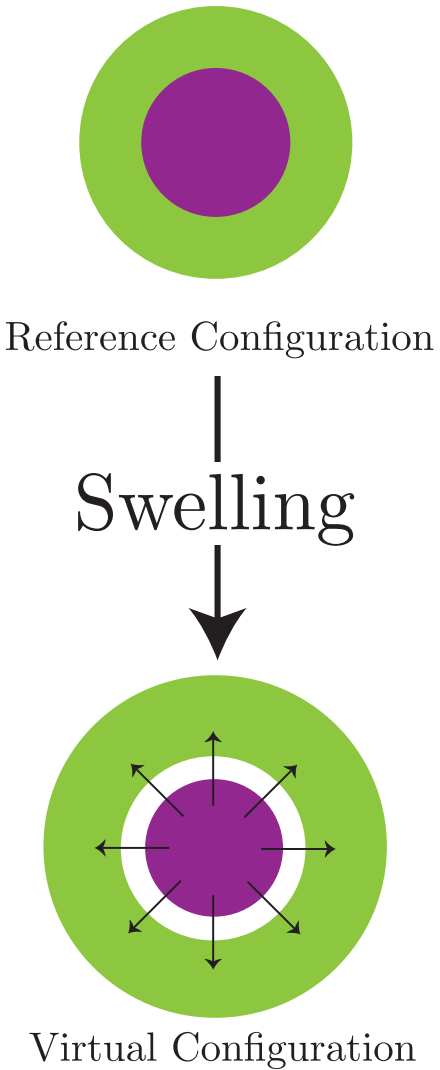
Elite Double 8  $E = 0.226 \text{ MPa}$   
 $\nu = 0.5$

Elite Double 32  $E = 0.963 \text{ MPa}$   
 $\nu = 0.5$



# Residual Swelling









# Residual Swelling

## Stretching Energy

$$\mathcal{U}_s \simeq \int_0^R \frac{(a_{\theta\theta} - \alpha^{-2}r^2)^2}{\alpha^{-2}r^3} dr + \frac{E_a}{E_d} \int_R^{R_e} \frac{(a_{\theta\theta} - \alpha^2r^2)^2}{\alpha^2r^3} dr$$

$\uparrow$   
 Modulus difference

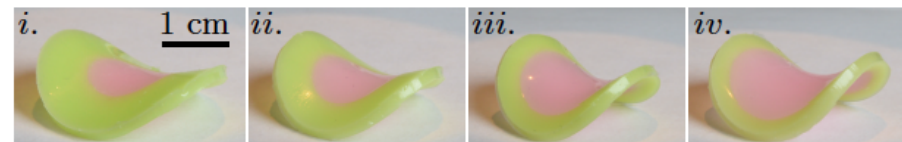
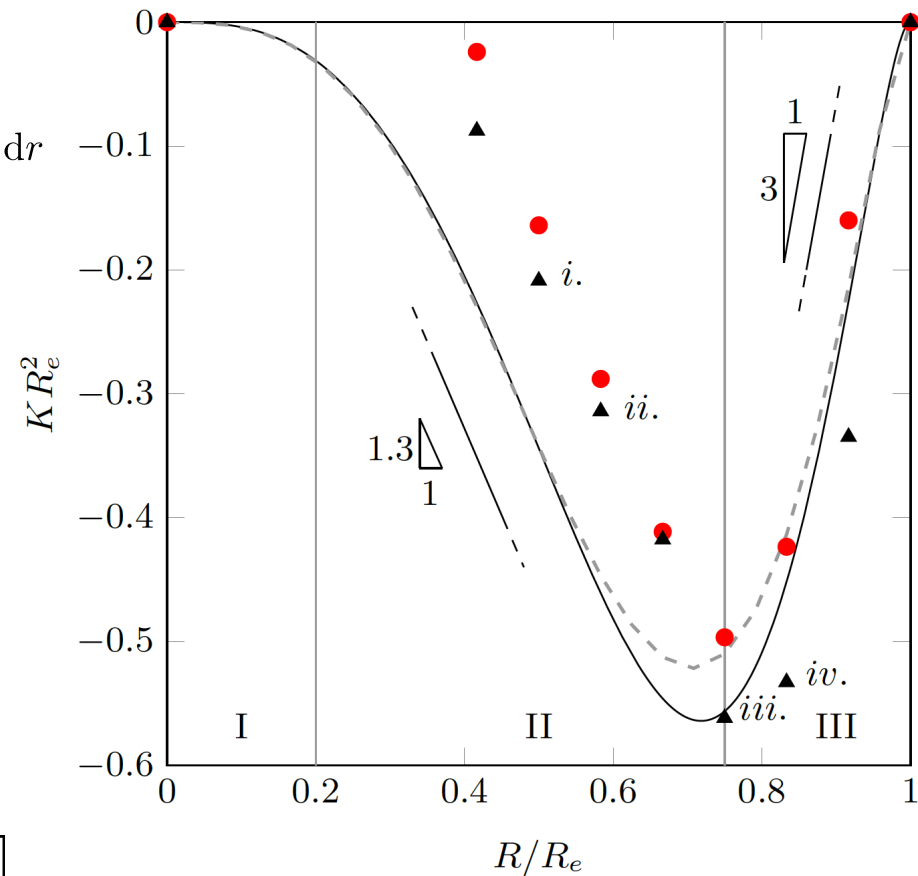
## Stretching ratio

- Depends on chemical and material properties.
- Should vary with  $R/R_e$
- Will depend on concentration gradient of free chains.

### ansatz

- from conservation of mass & proportional to mass uptake in annulus.)

$$\alpha = 1 + \eta(c_d - c_a) \left( \frac{R}{R_e} \right)^2 \left[ 1 - \left( \frac{R}{R_e} \right)^2 \right]$$





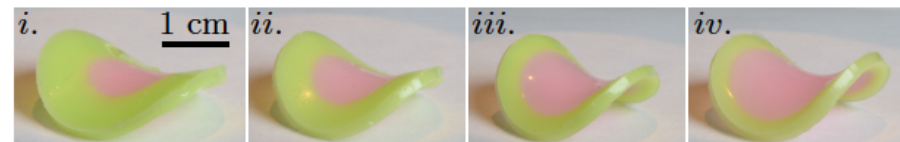
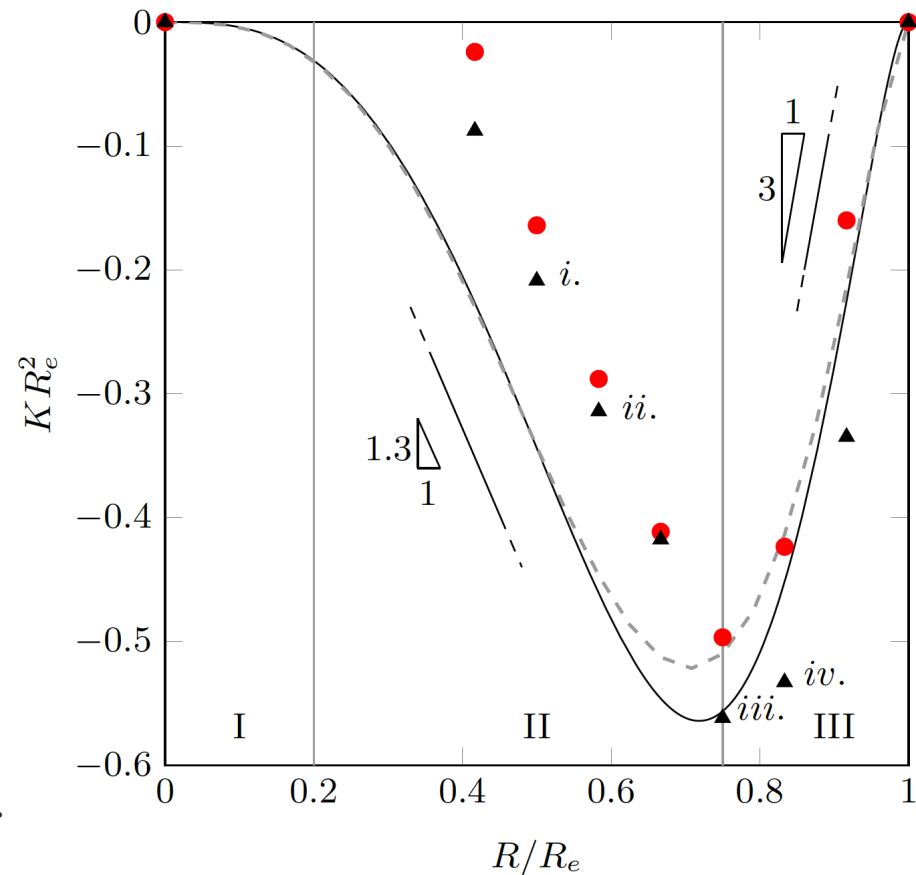
# Residual Swelling

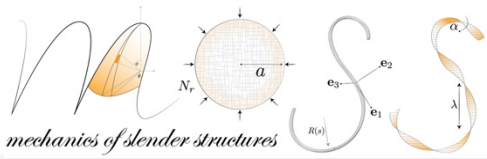
## Approximate Analytical Solution

- Taylor expand about  $(\alpha - 1)$
- $\bar{E} = E_a/E_d$  and  $\bar{R} = R/R_e$

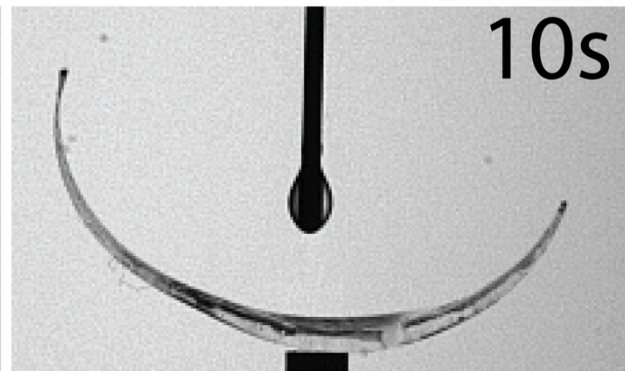
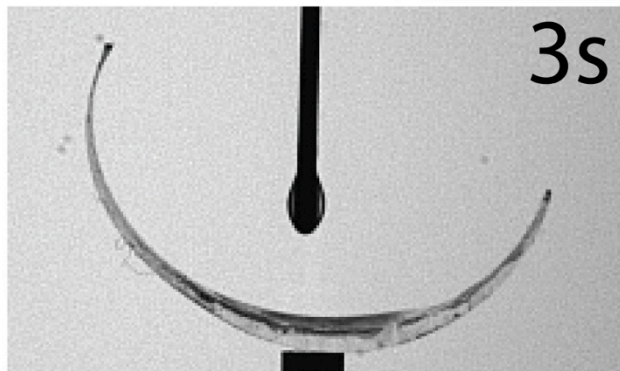
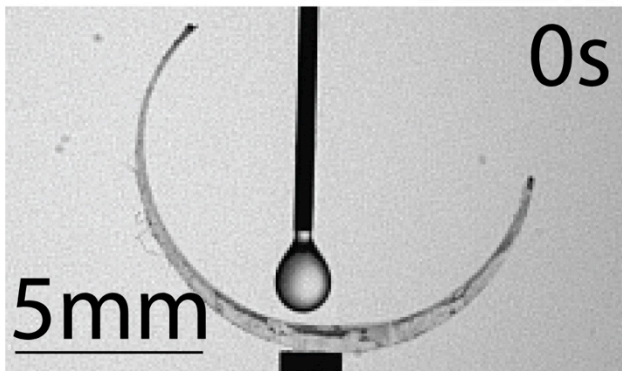
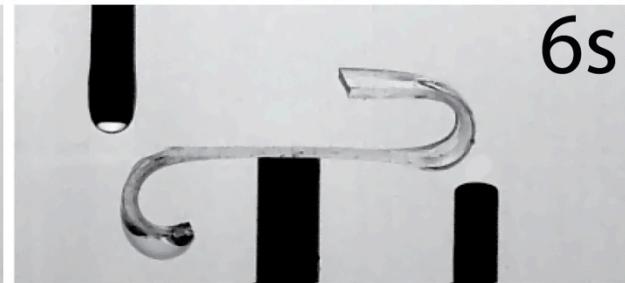
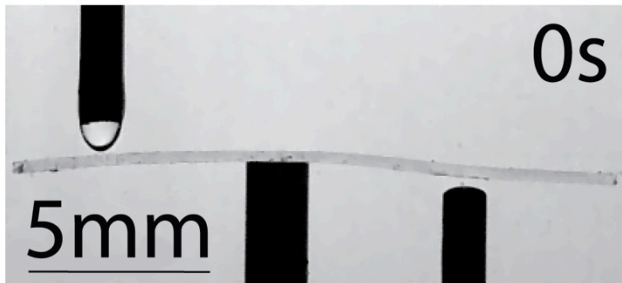
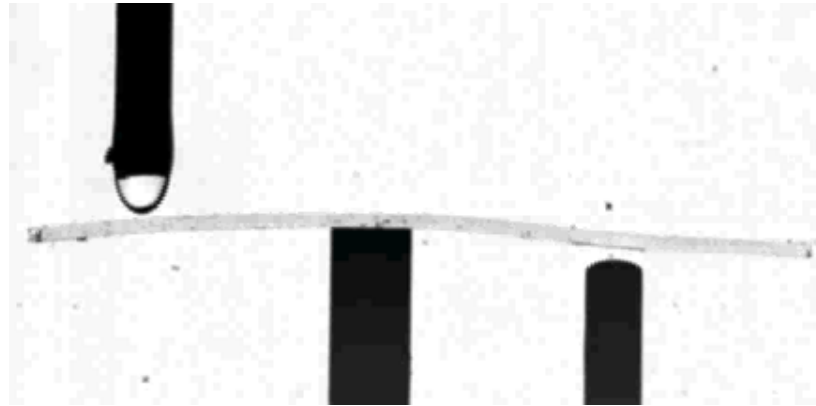
$$KR_e^2 \simeq 96(1 - \alpha_{\max})\bar{E}\bar{R}^3 \frac{(1 - \bar{R}^2)(1 - \bar{R}^3)}{\bar{R}^6(1 - \bar{E}) + \bar{E}}$$

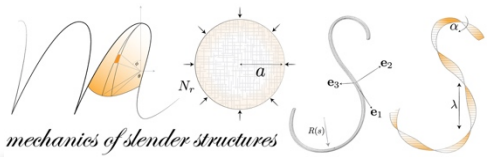
**2D analog** to Timoshenko's model for thermal beam bending of **bimetallic strips**.





# Controlling Shape





## Elastic Instabilities for Form and Function Buckling, Wrinkling, Folding, and Snapping

### **Geometry and Mechanics:**

- Fundamental equations, geometric rigidity, morphing.

### **Buckling & Wrinkling:**

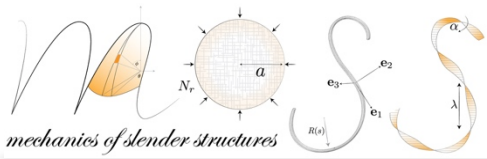
- Stability, wavelength, flexible electronics, mechanical metamaterials, adhesion.

### **Stress Focusing – Folding & Creasing:**

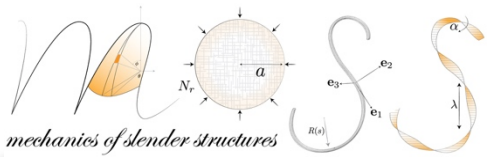
- Wrinkle-to-fold, origami.

### **Snapping:**

- Snapping surfaces.

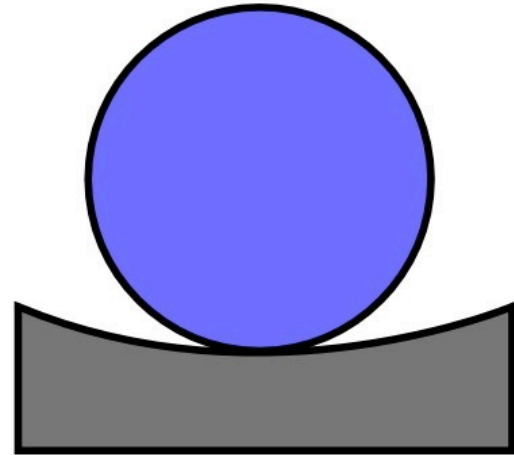


# Buckling



# Stability

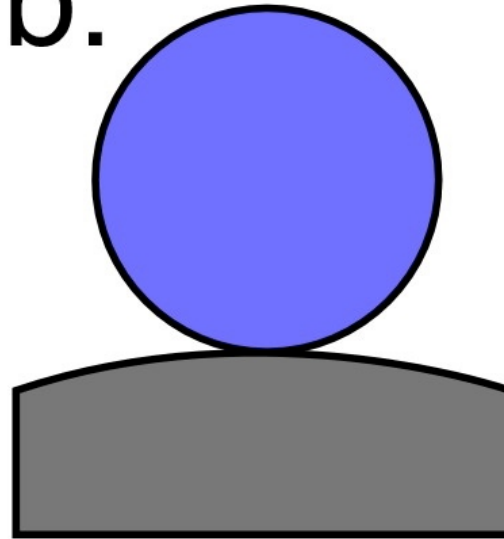
a.



Stable Equilibrium

- Lateral displacement **raises** the ball's center of gravity.
- **Increases** the potential energy.

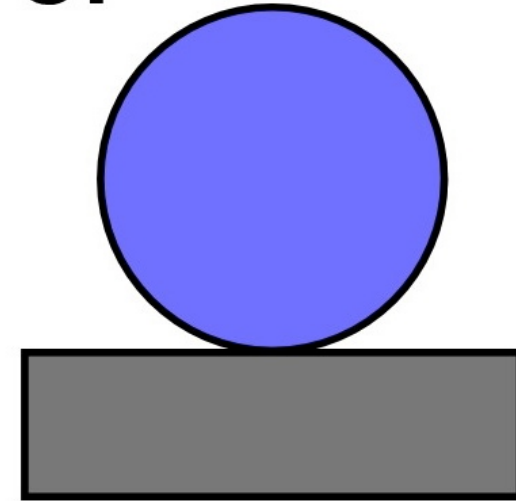
b.



Unstable Equilibrium

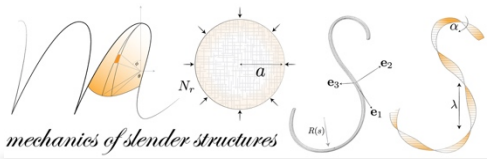
- Lateral displacement **lowers** the ball's center of gravity.
- **Decreases** the potential energy.

c.



Neutral Equilibrium

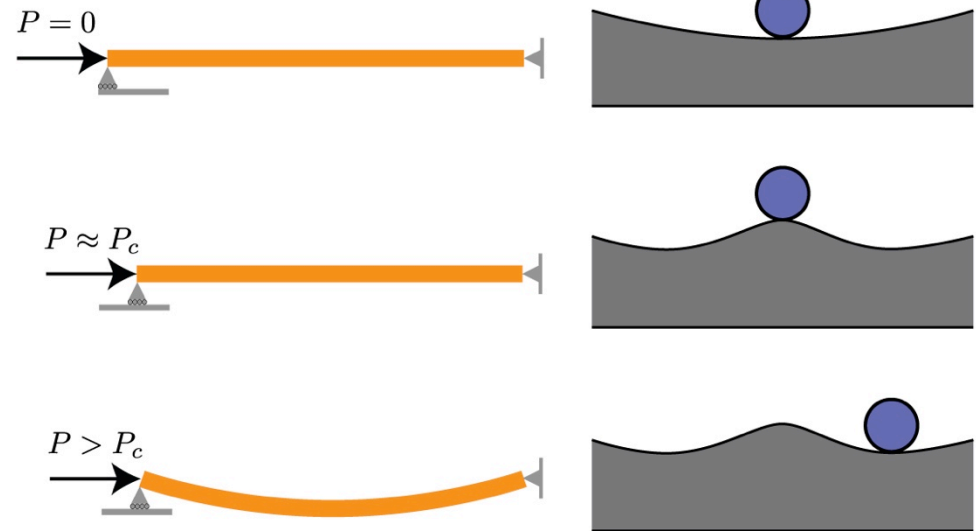
- Lateral displacement
  - no change in potential energy.

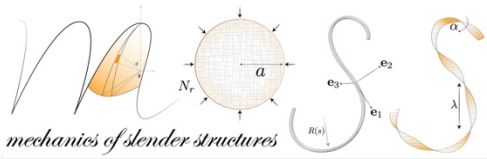


## Euler Buckling



“Stable-Symmetric Bifurcation”





$$\bar{\nabla}_\alpha \bar{\nabla}_\beta \mu^{\alpha\beta} - (\bar{b}_{\alpha\beta} + \bar{\nabla}_\alpha \bar{\nabla}_\beta w) \sigma^{\alpha\beta} - p = 0$$





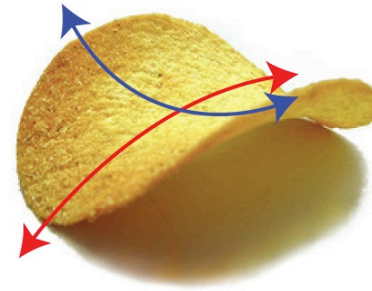
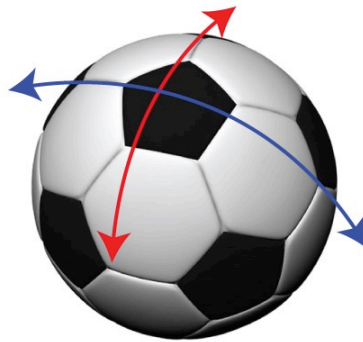
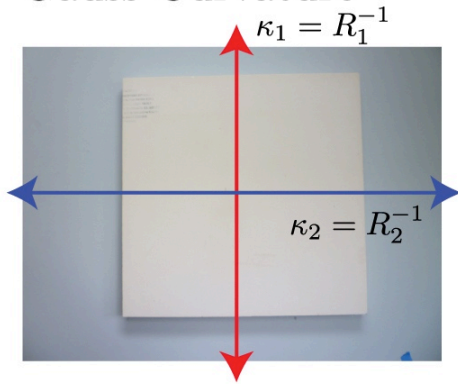
# Buckling into Arches

## Föppl-von Kármán Plate Equations

$$\underbrace{B\nabla^4 w}_{\text{bending}} - h \underbrace{\diamond^4[\phi, w]}_{\text{nonlinear stress \& curvature}} = 0$$

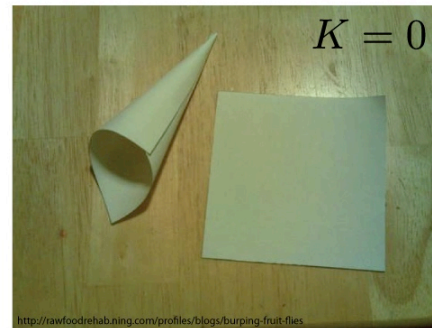
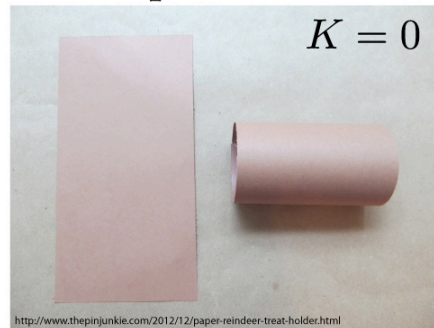
$$\underbrace{\nabla^4 \phi}_{\text{stress}} + \frac{E}{2} \underbrace{\diamond^4[w, w]}_{\text{geometry}} = 0$$

Gauss Curvature



$$K = \kappa_1 \kappa_2 = \diamond^4[w, w, ] = 0 \quad K = \kappa_1 \kappa_2 = \diamond^4[w, w, ] > 0 \quad K = \kappa_1 \kappa_2 = \diamond^4[w, w, ] < 0$$

Developable





# Buckling into Arches

## Föppl-von Kármán Plate Equations

$$\underbrace{B\nabla^4 w}_{\text{bending}} - h \underbrace{\diamond^4[\phi, w]}_{\text{nonlinear stress \& curvature}} = 0$$

$$\underbrace{\nabla^4 \phi}_{\text{stress}} + \frac{E}{2} \underbrace{\diamond^4[w]}_{\text{geometry}} = 0$$

$$\frac{\partial^4 \phi}{\partial x^4} = 0$$



- Stress independent of  $y$
- Clamped boundary conditions
- In-plane stress is constant
- Bending equation becomes ODE

$$w(x) = \frac{w_0}{2} \left( 1 + \cos \frac{\pi x}{L} \right)$$

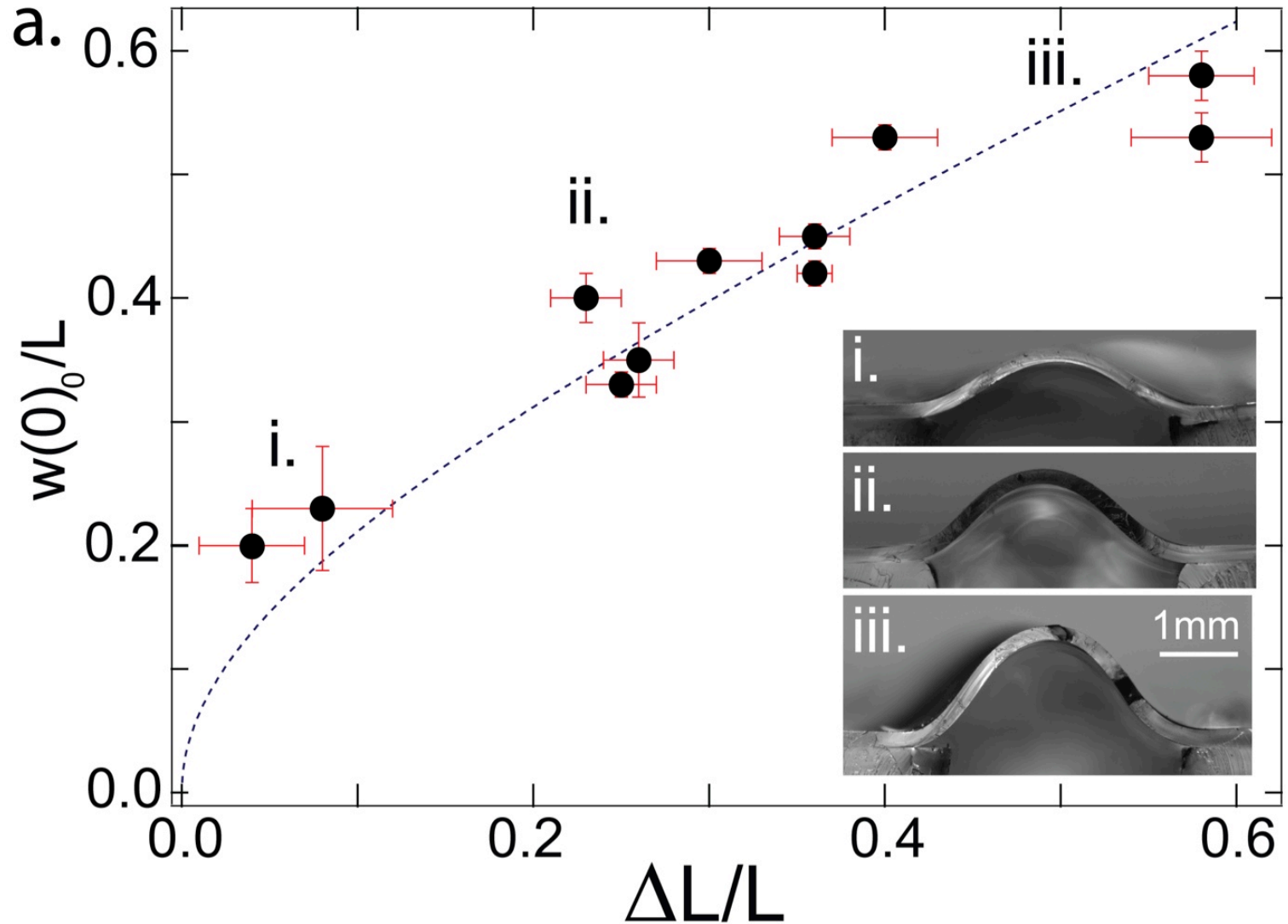
$$w_0 = \pm \frac{2h}{\sqrt{3}} \left( \frac{\sigma}{\sigma_{Eu}} - 1 \right)^{1/2}$$

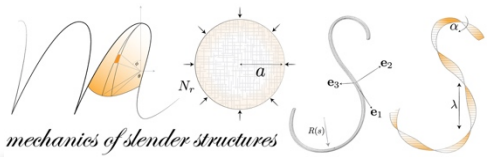
Euler Buckling Stress:

$$\sigma_{Eu} = \frac{\pi^2 E}{12(1 - \nu^2)} \left( \frac{h}{L} \right)^2$$

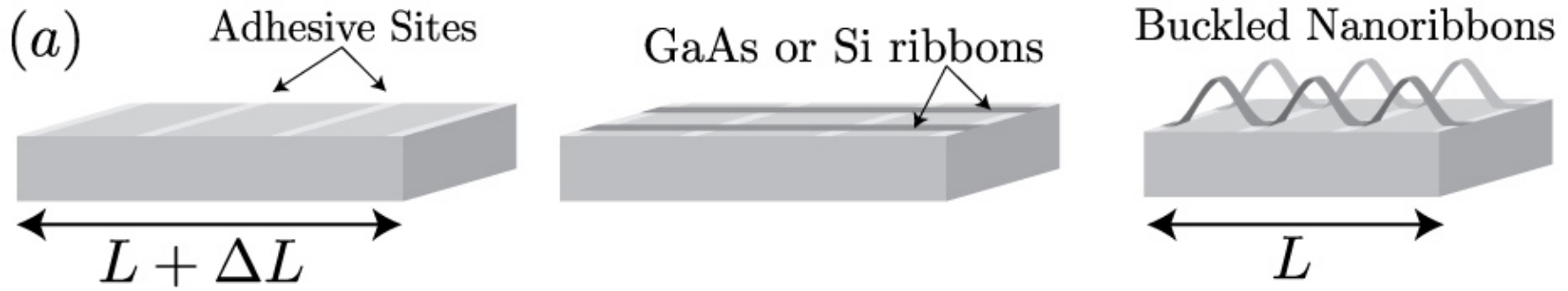


# Buckling into Arches



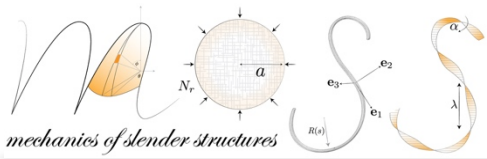


# Buckling into Arches

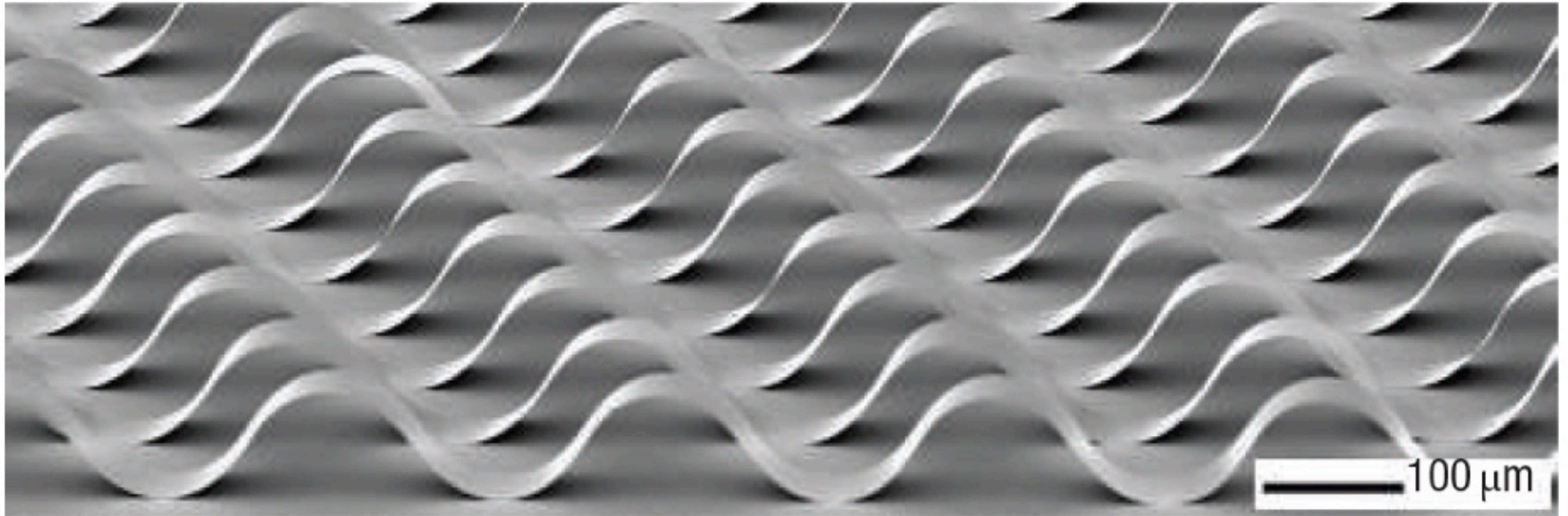
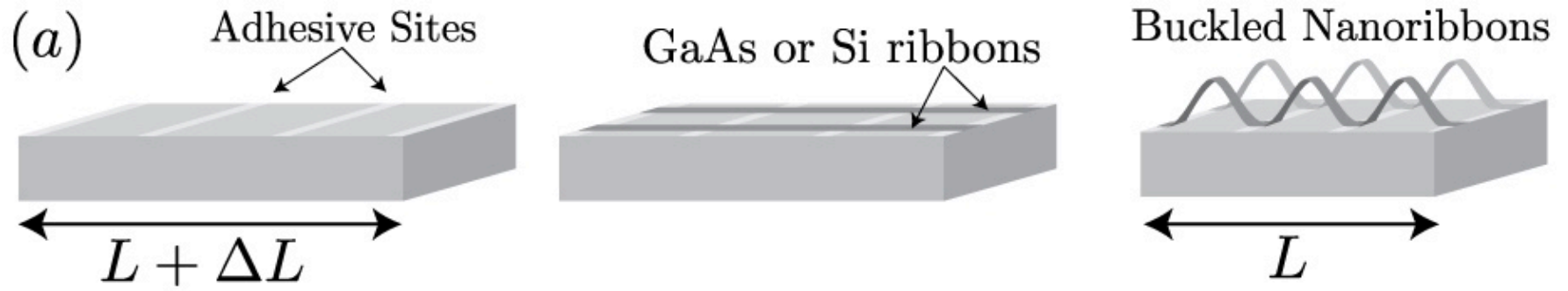


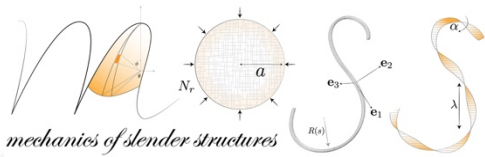
## Buckling of microscale elastic plates

- Fabrication of semiconductor **nanoribbons** for **stretchable electronics**.
- Flexible substrate: PDMS, chemically altered with UV light.
- Uniaxially strained PDMS, strips of single-crystal **GaAs** (gallium arsenide) **bonded** to the **adhesive sites**.
- Release of the prestrain causes a uniaxial compressive stress that buckles the nanoribbons into arches.
- Adhesive boundaries (as opposed to clamped) yields a variation on the classical elastica problem – the **sticky elastica**.

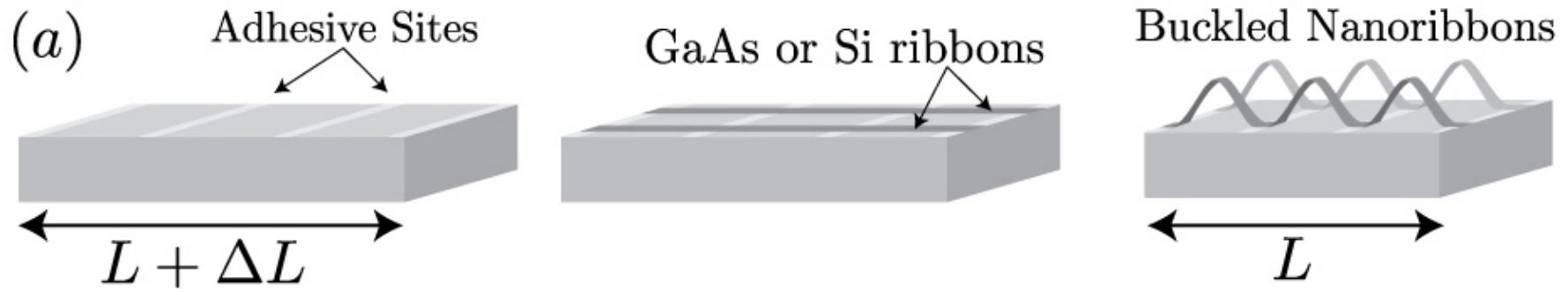


# Buckling into Arches



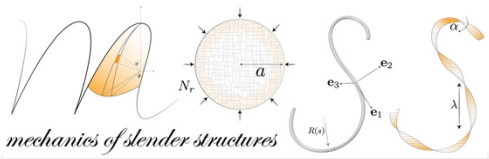


# Buckling into Arches

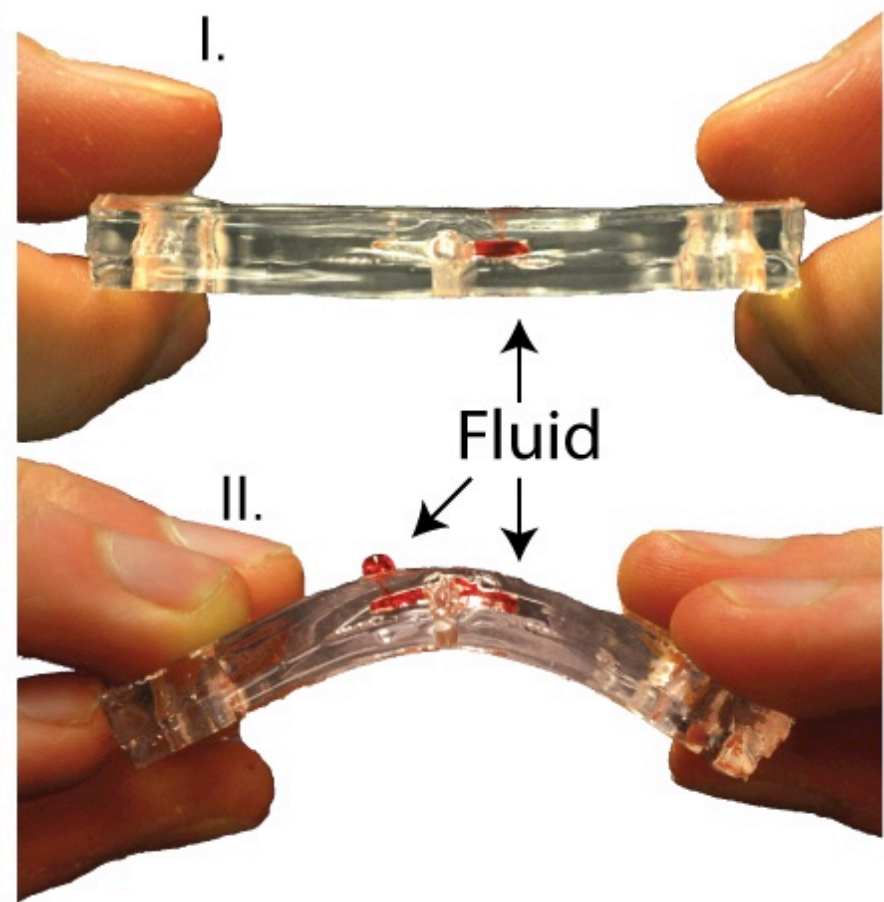
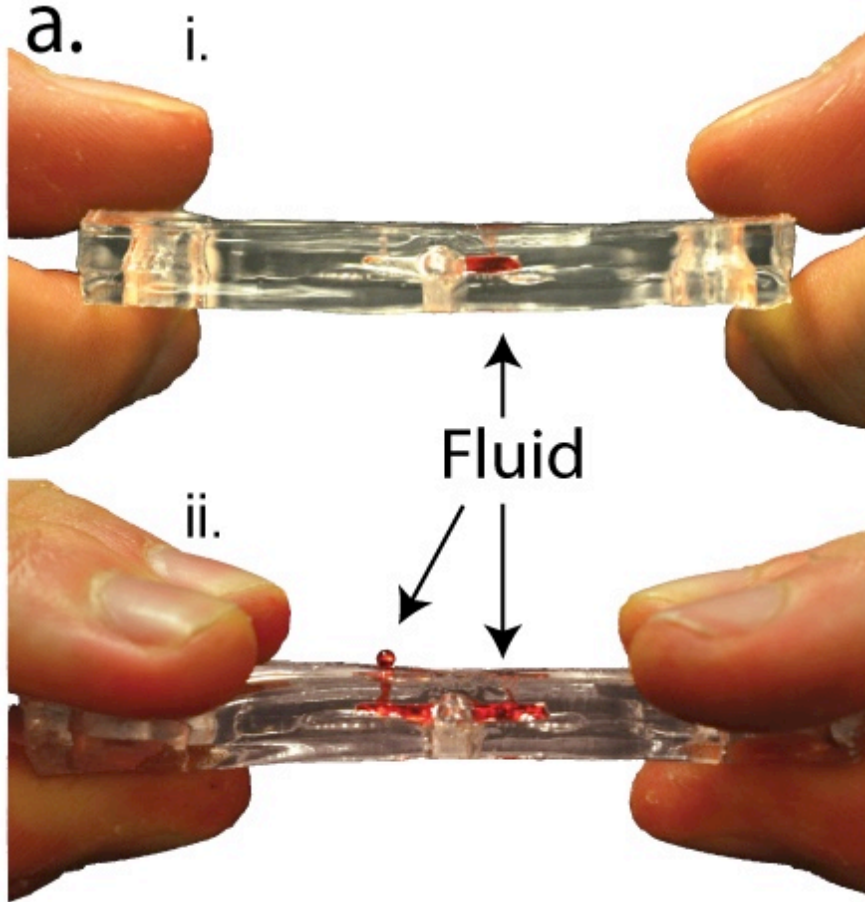
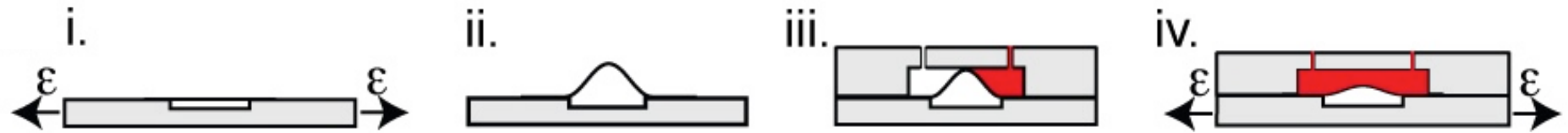


## Buckling of microscale elastic plates

- Nanoribbons with prestrains exceeding 50% were fabricated.
- Deposition of gold onto the arches makes them functional electrodes.
- Example device: Metal-semiconductor-metal photodetectors
  - Photosensor functional up to **51.4% strain in tension** and -18.3% in compression.
- Technique has been extended to form: single-wall carbon nanotube arches, MOSFETs, piezoelectric energy harvesters.

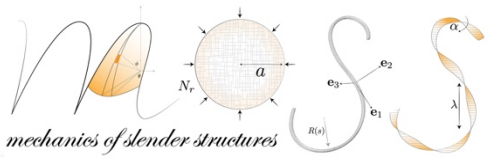


# Buckling into Arches



D.P. Holmes, B. Tavakol, G. Froehlicher, and H.A. Stone, "Control and Manipulation of Microfluidic Fluid Flow via Elastic Deformations", *Soft Matter*, **9**(29), 7049, (2013).  
 B. Tavakol, D.P. Holmes, and H.A. Stone, "Extended Lubrication Theory", *Under Review*, (2015).

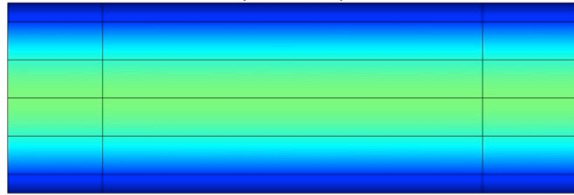




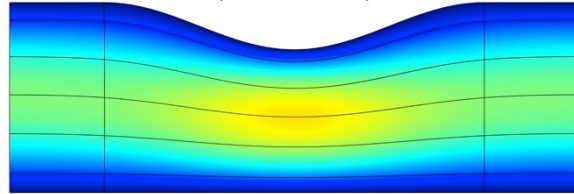
# Buckling into Arches

**Fluid flow** through channels with variable **geometry**

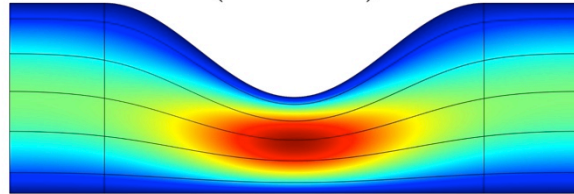
a. ( $\lambda = 0$ )



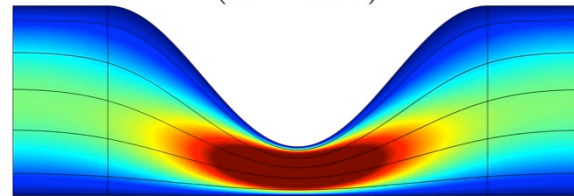
b. ( $\lambda = 0.25$ )



c. ( $\lambda = 0.50$ )



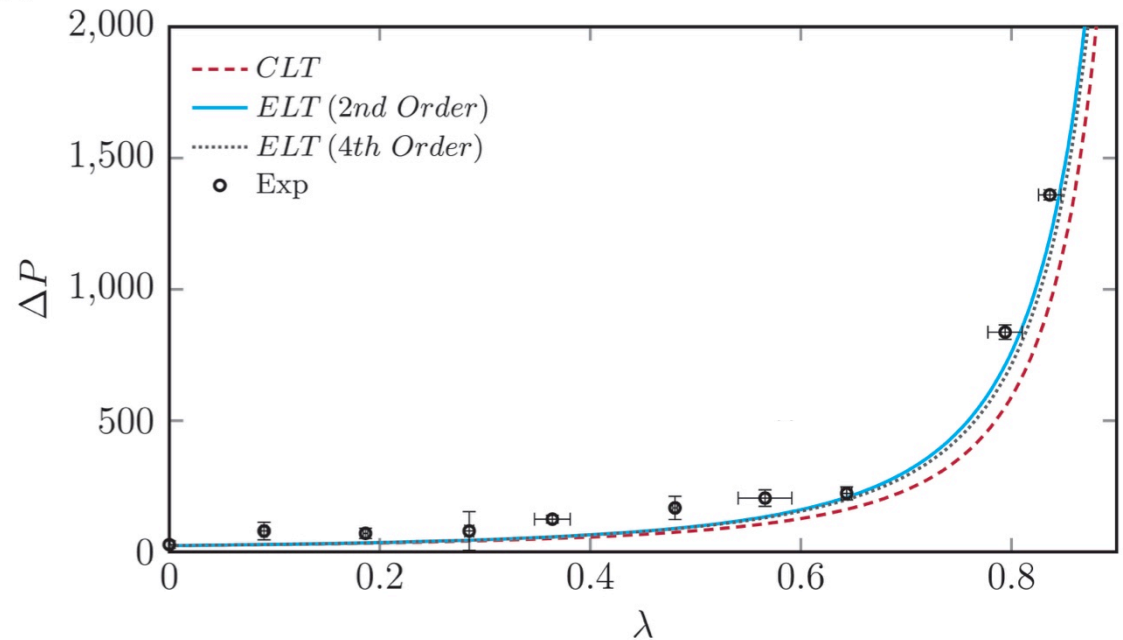
d. ( $\lambda = 0.75$ )



Extended Lubrication Theory

$$\Delta P = \underbrace{\Delta P_0}_{\text{Lubrication Theory}} \left( 1 + \underbrace{\frac{4}{5} \lambda^2 \delta^2 - \frac{64}{225} \lambda^4 \delta^4 + \mathcal{O}(\delta^6)}_{\text{Perturbation to higher orders}} \right)$$

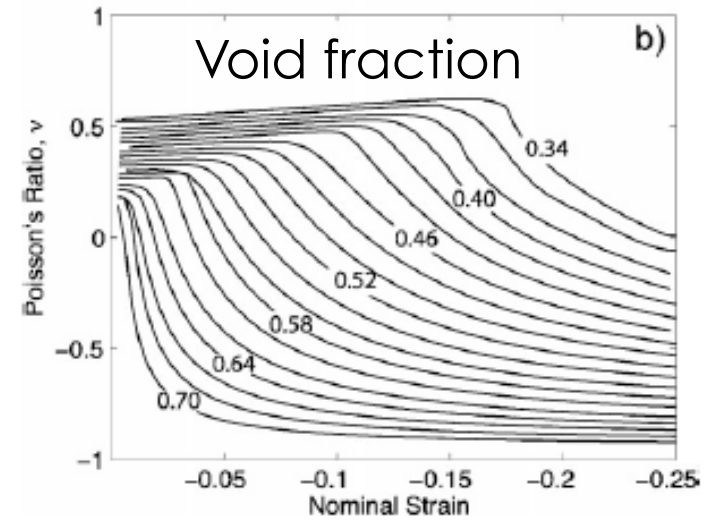
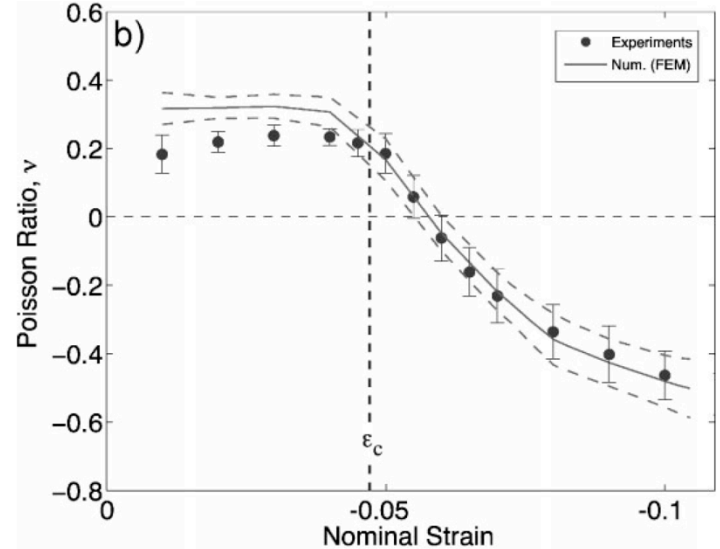
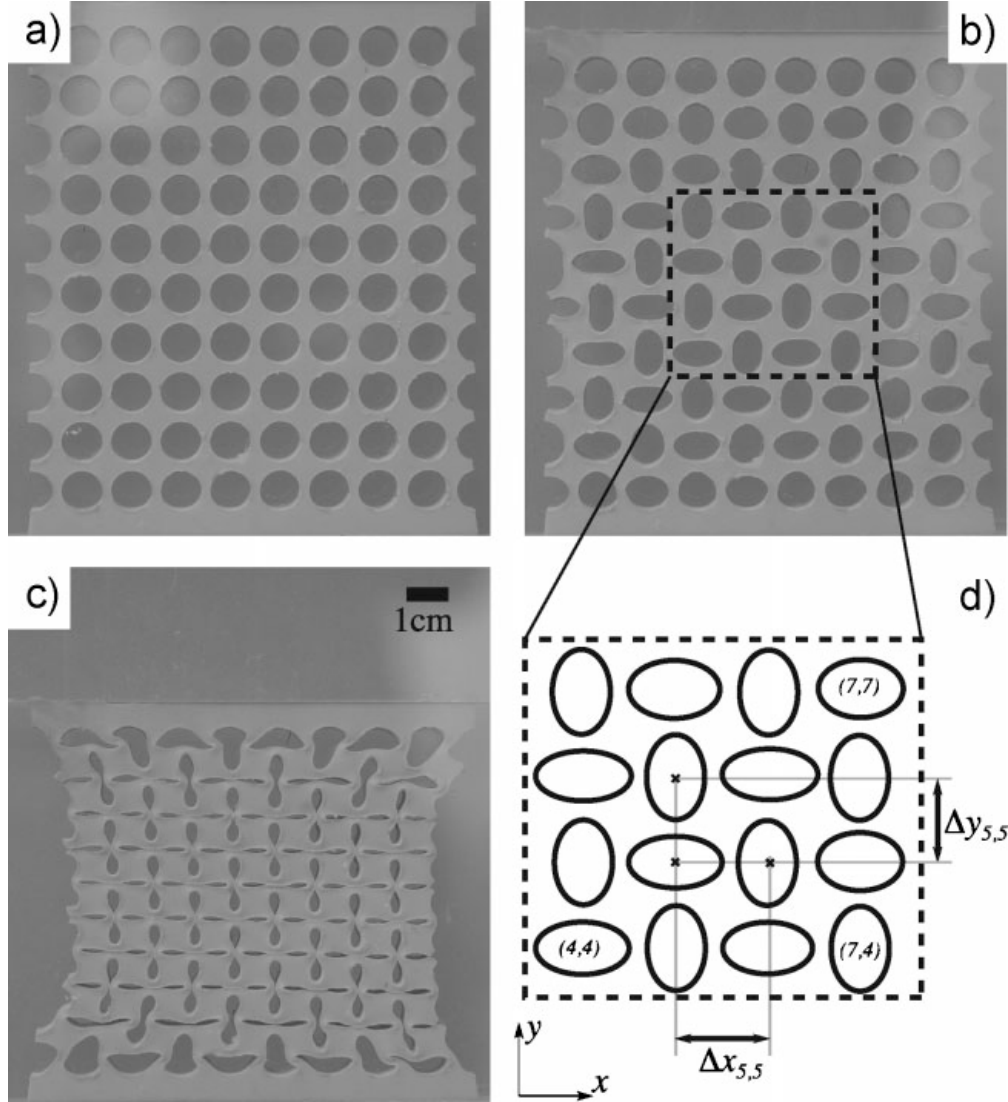
b.

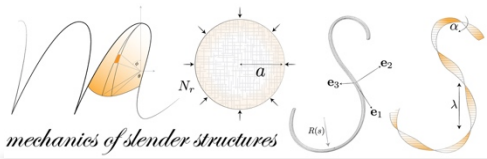




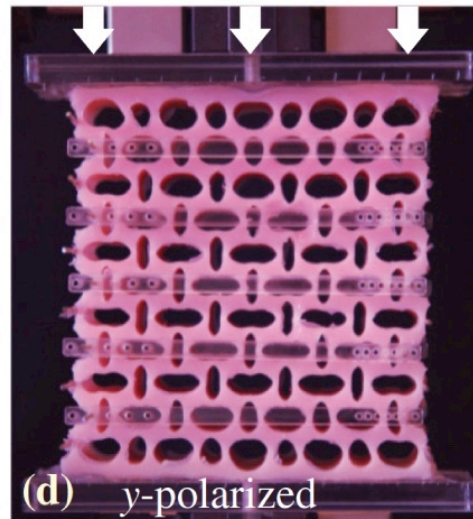
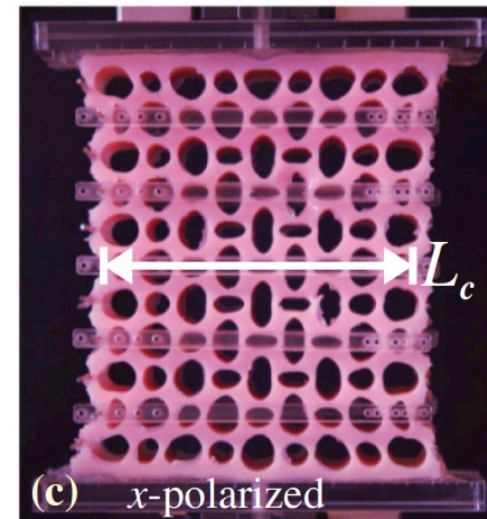
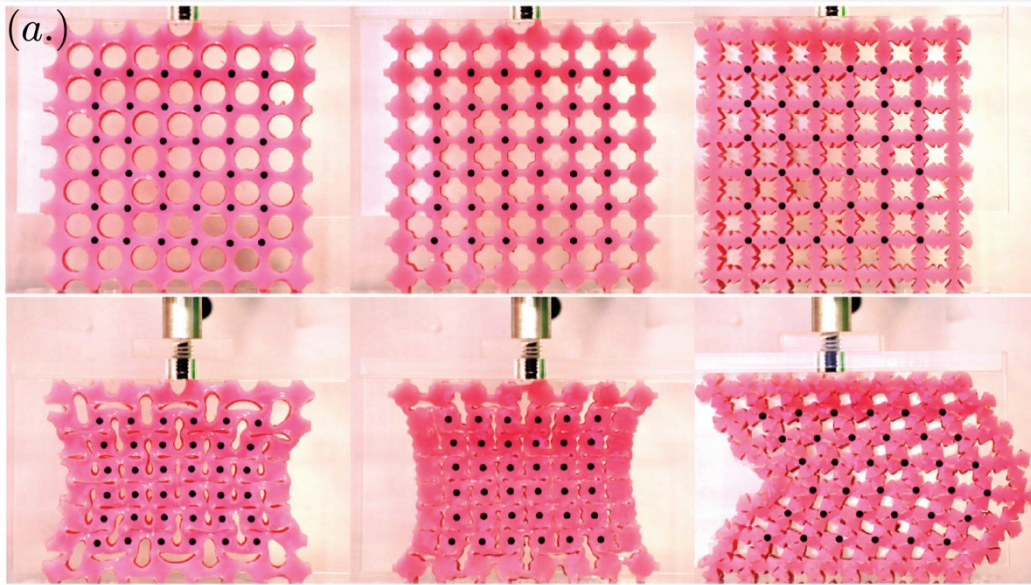


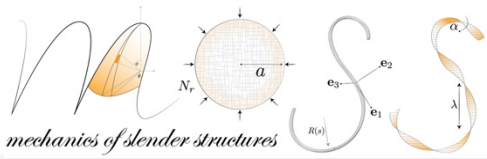
# Mech. Metamaterials





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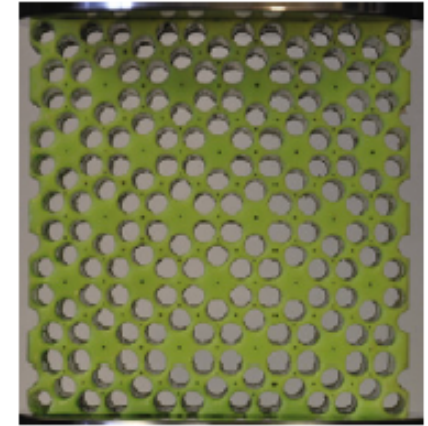
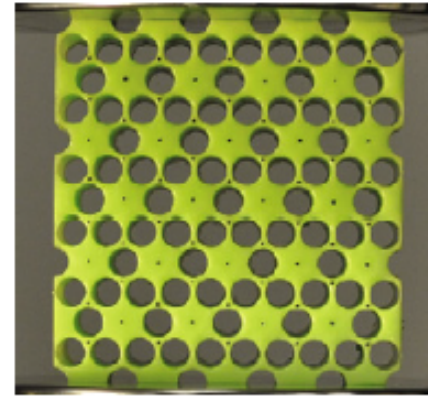
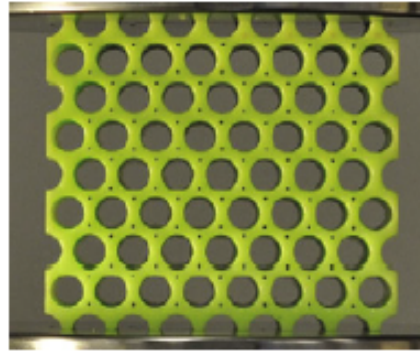
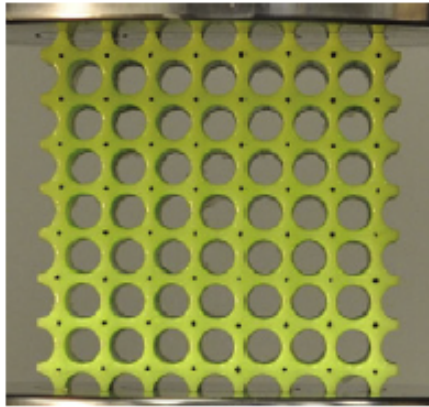
**(A) 4.4.4.4**

**(B) 3.3.3.3.3.3**

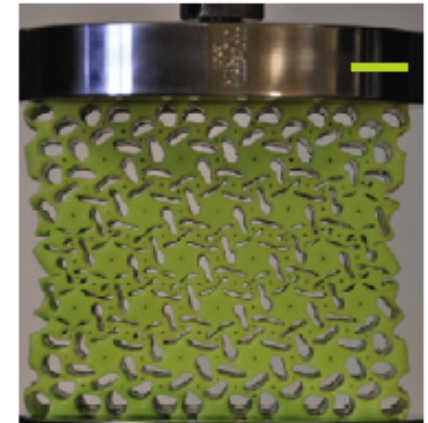
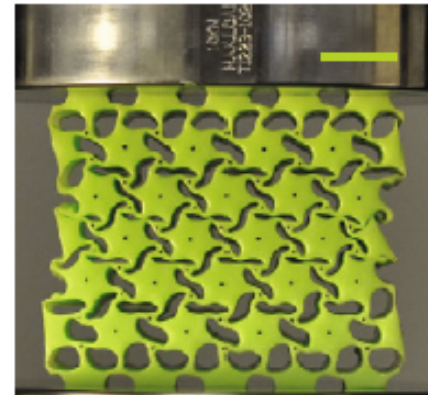
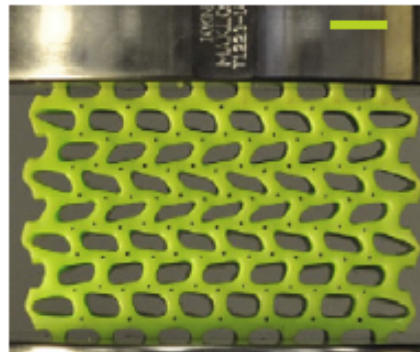
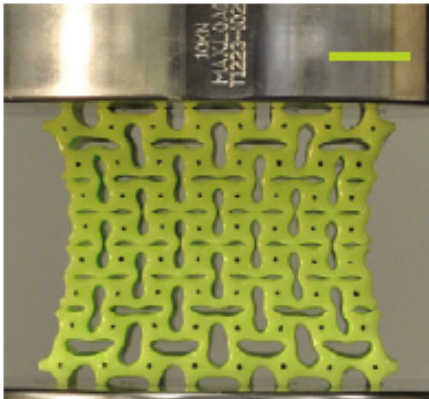
**(C) 3.6.3.6**

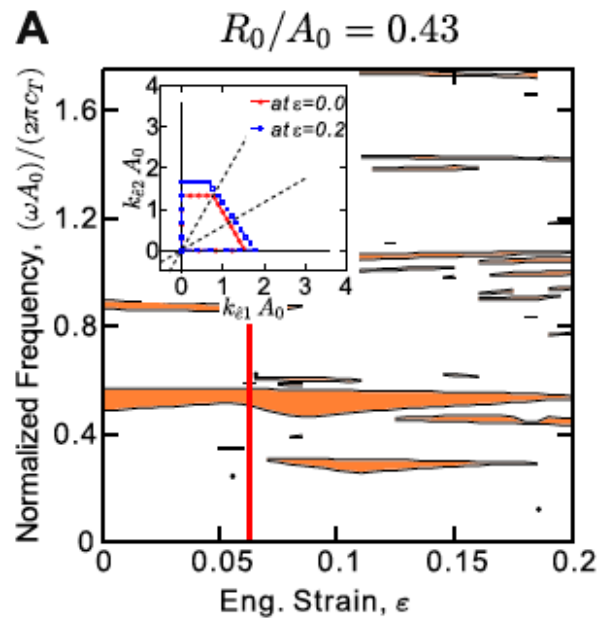
**(D) 3.4.6.4**

$\varepsilon = 0.00$



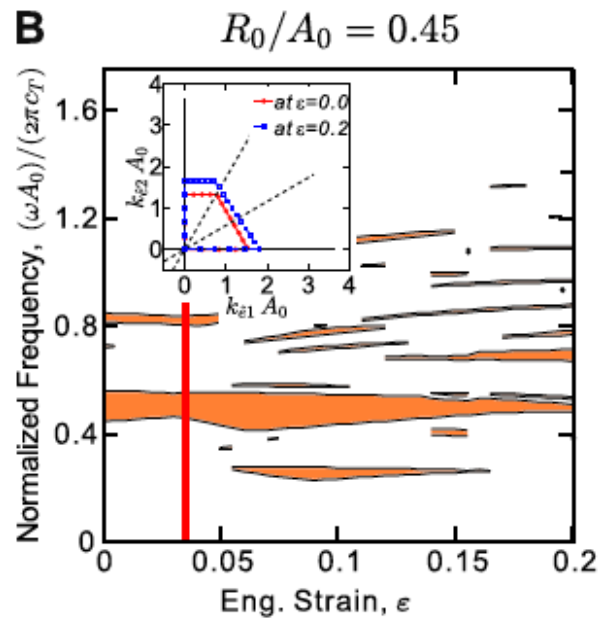
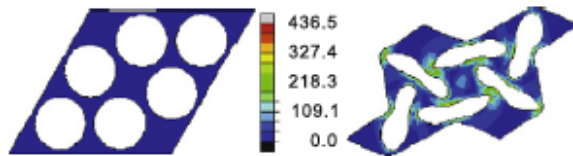
$\varepsilon = 0.21$





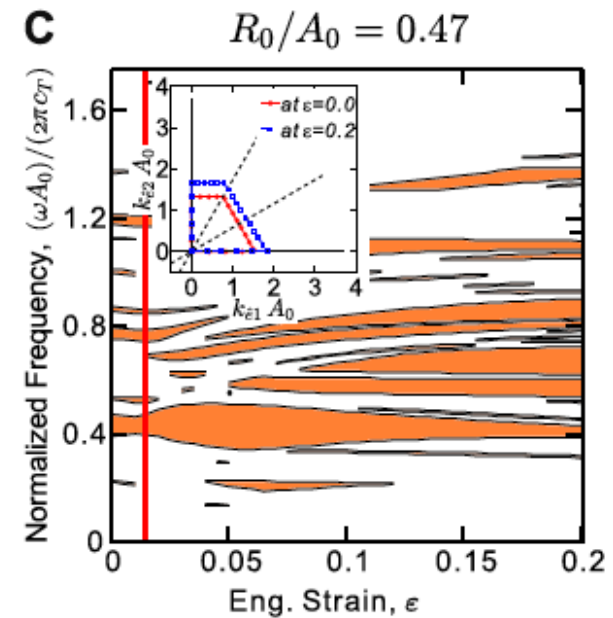
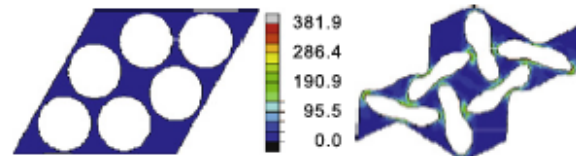
at  $\epsilon=0.0$

at  $\epsilon=0.2$



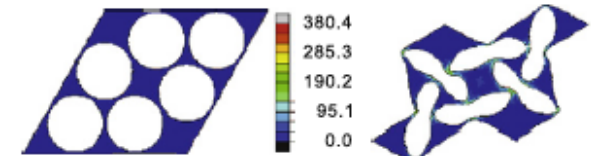
at  $\epsilon=0.0$

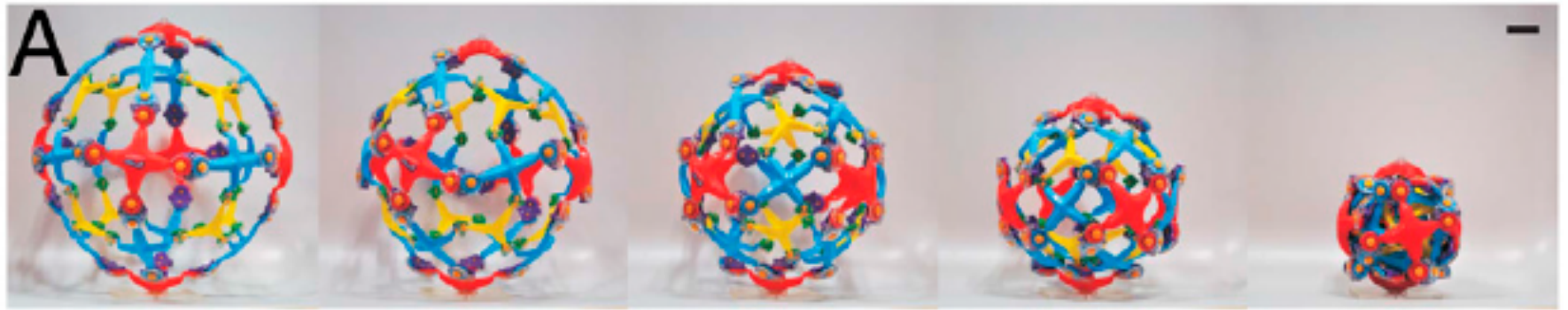
at  $\epsilon=0.2$



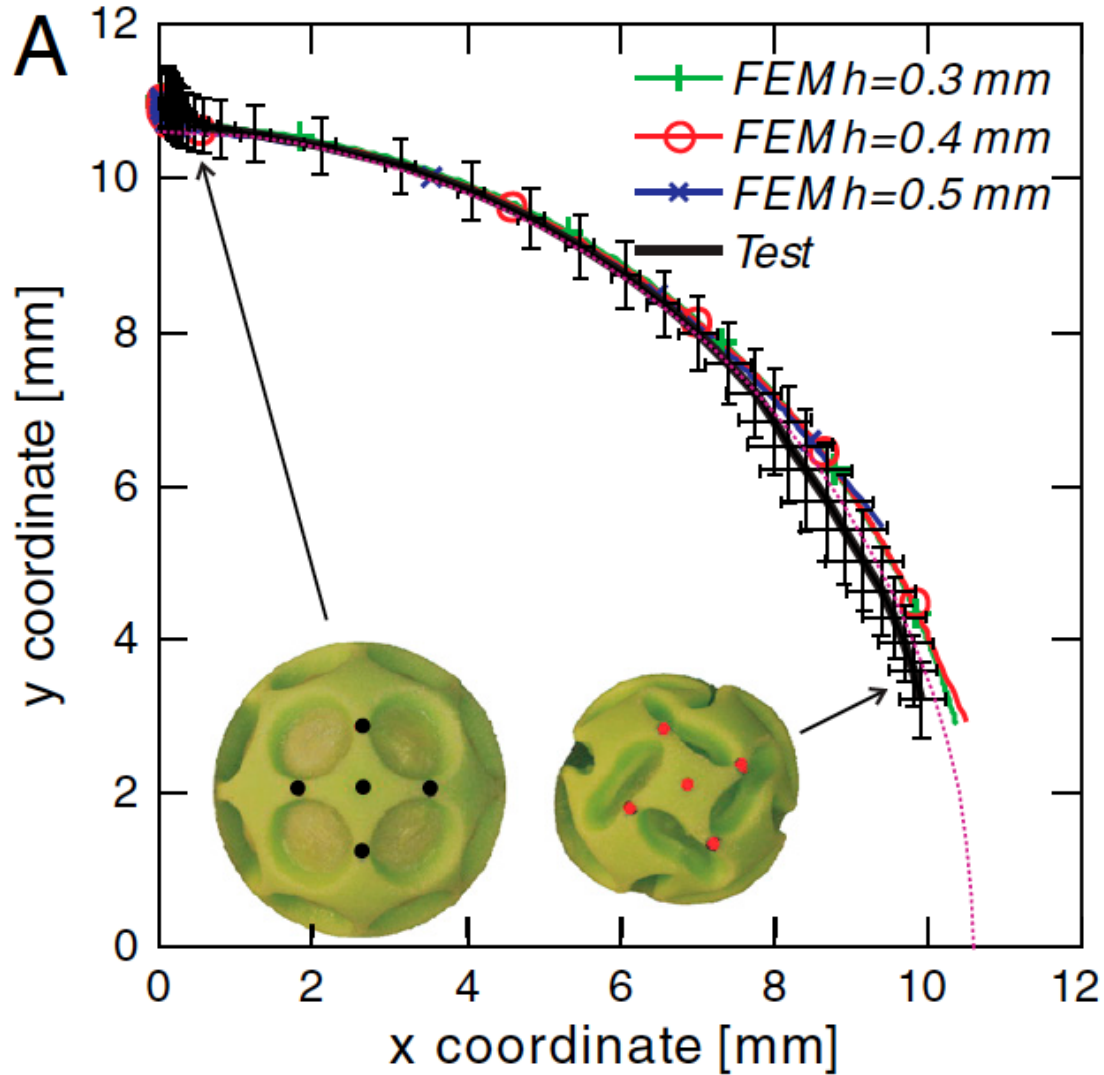
at  $\epsilon=0.0$

at  $\epsilon=0.2$



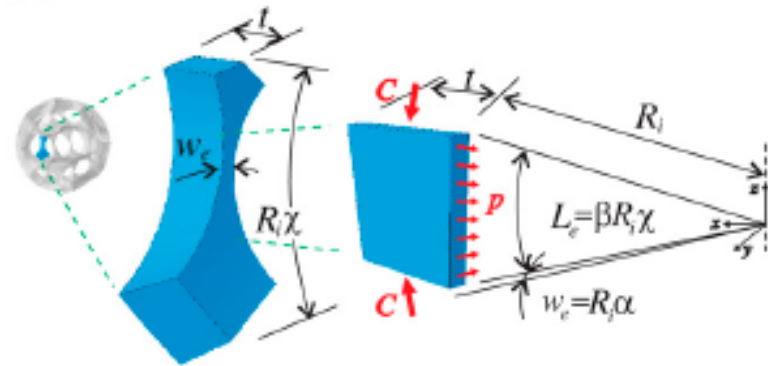


# Mech. Metamaterials

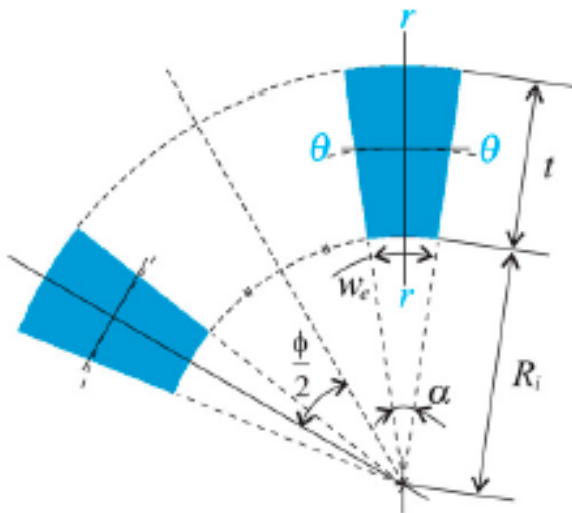


Expanded		Folded	
A	B	C	D
cuboctahedron (6 Holes)		octahedron	
rhombicuboctahedron (12 Holes)		cuboctahedron	
nonflat hole surfaces (24 Holes)		rhombicuboctahedron	
rhombicosidodecahedron (30 Holes)		icosidodecahedron	
nonflat hole surfaces (60 Holes)		rhombicosidodecahedron	

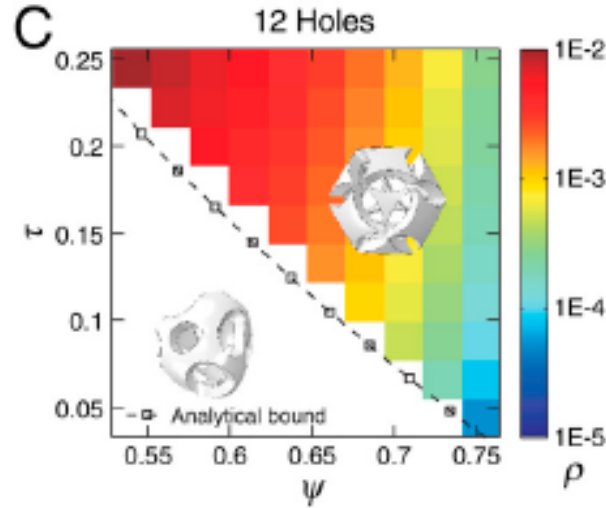
A



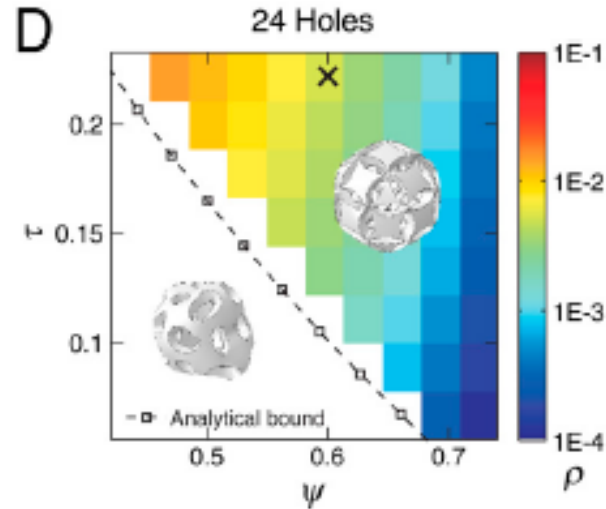
B



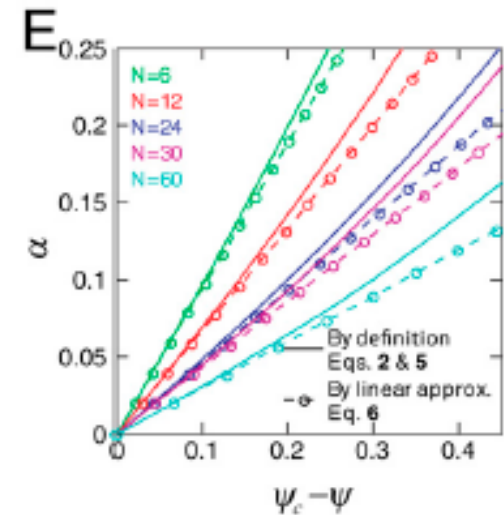
C



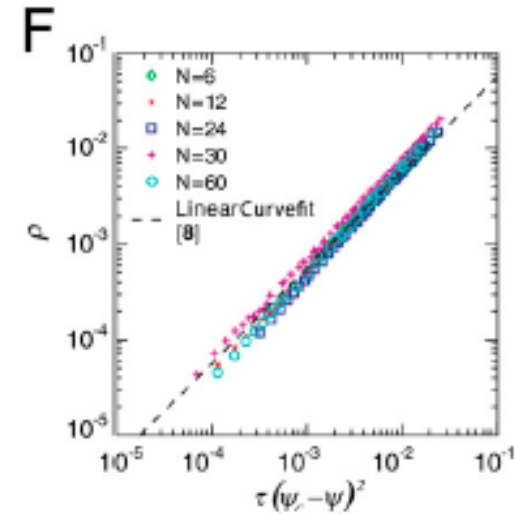
D

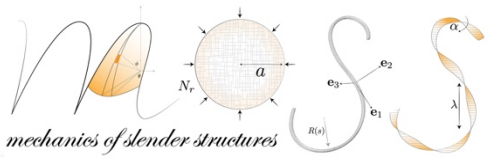


E



F





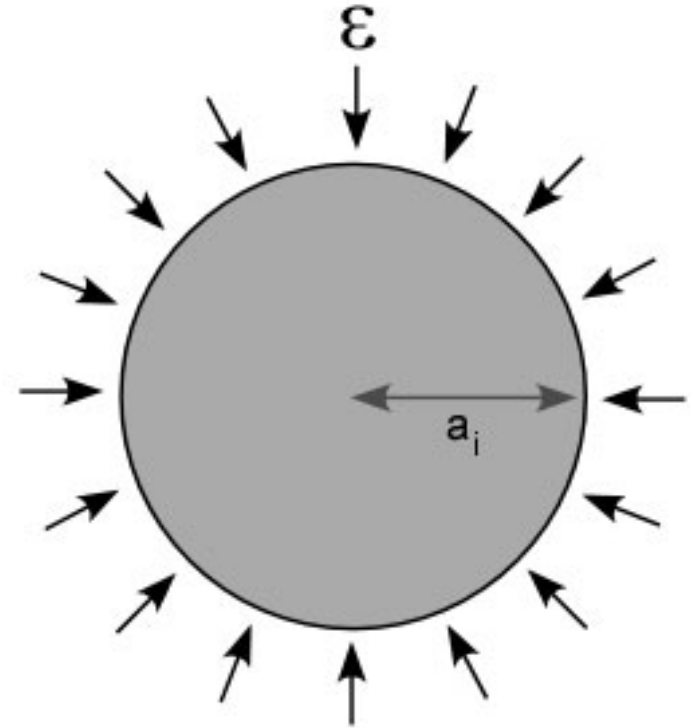
# Buckling into Shells

Critical Buckling Stress

$$\sigma_c = \frac{k^2 E}{12 (1 - \nu^2)} \left( \frac{h}{a_i} \right)^2$$

Critical Buckling Strain

$$\varepsilon_c = \frac{k^2}{12 (1 + \nu)} \left( \frac{h}{a_i} \right)^2$$



Equilibrium Equation Circular Plate (cylindrical coordinates):

$$r^2 \frac{d^2 \varphi}{dr^2} + r \frac{d\varphi}{dr} + \left( \frac{P r^2}{D} - 1 \right) \varphi = 0$$

General Solution:  $\varphi = A_1 J_1(k) + A_2 Y_1(k)$

Solution:

$$k J_0(k) - (1 - \nu) J_1(k) = 0$$

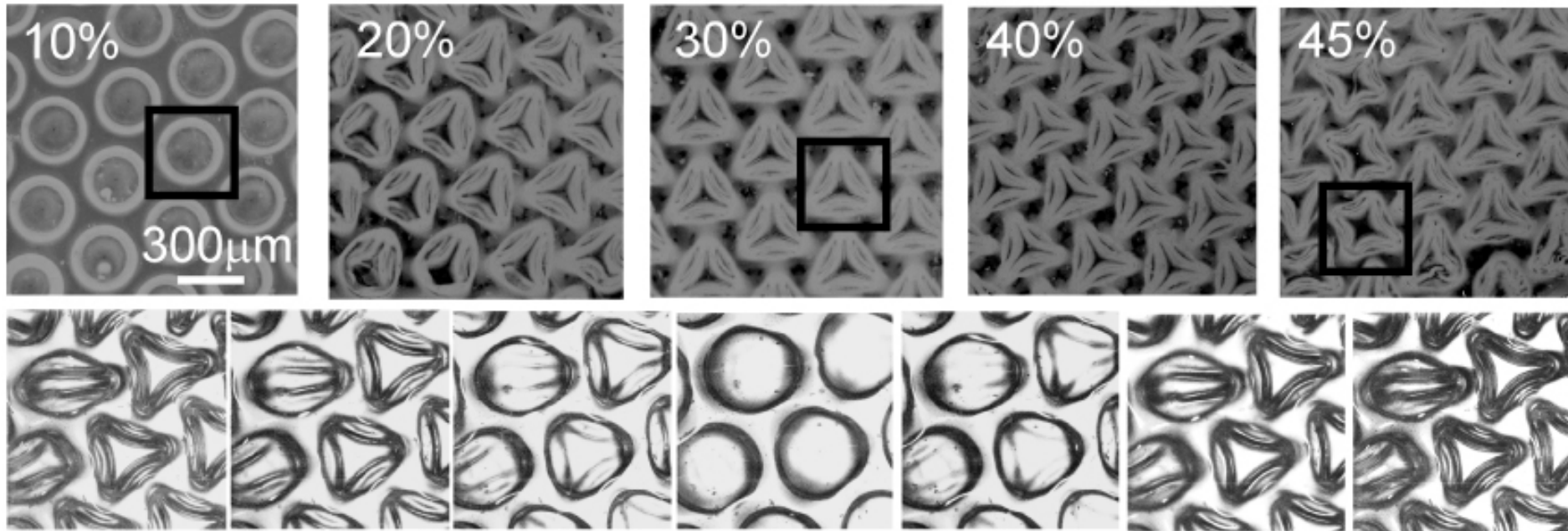
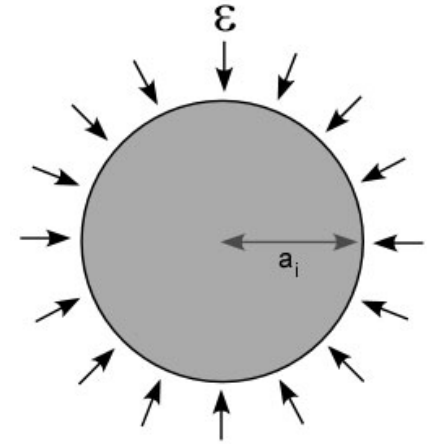
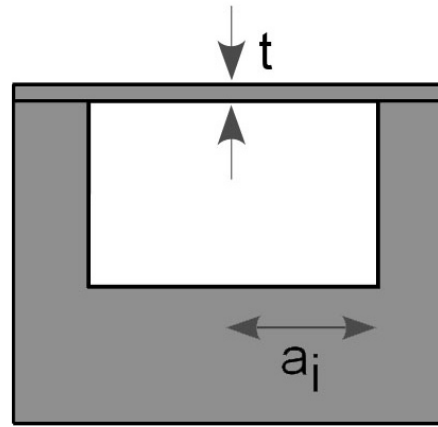
**First buckling mode:  $k = 2.16$**   
(simply supported B.C.'s)





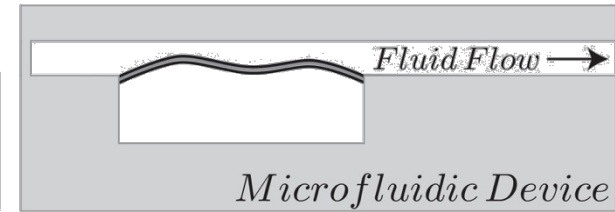
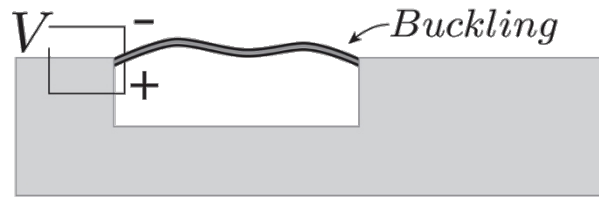
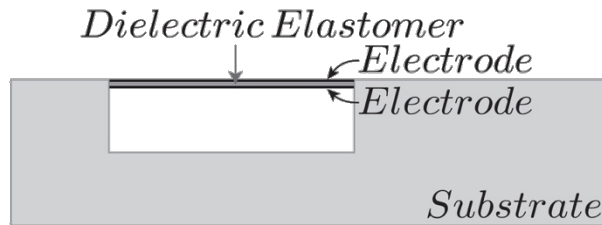
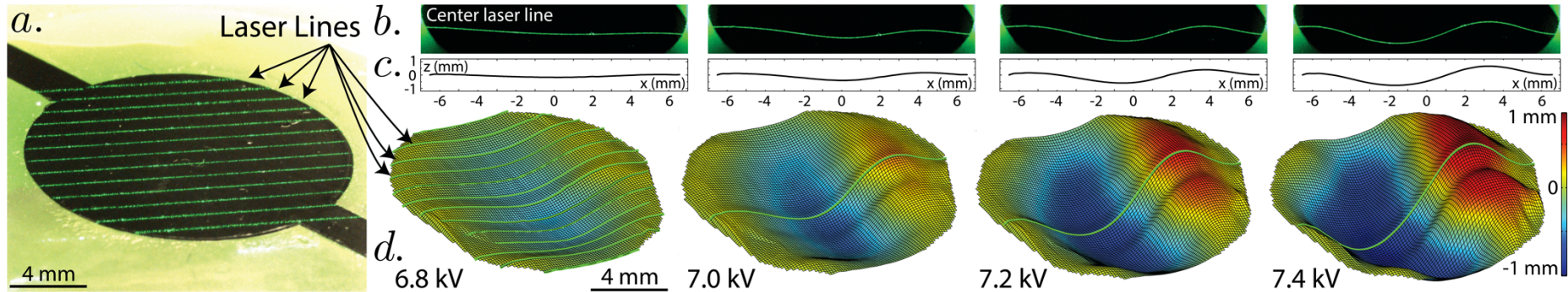
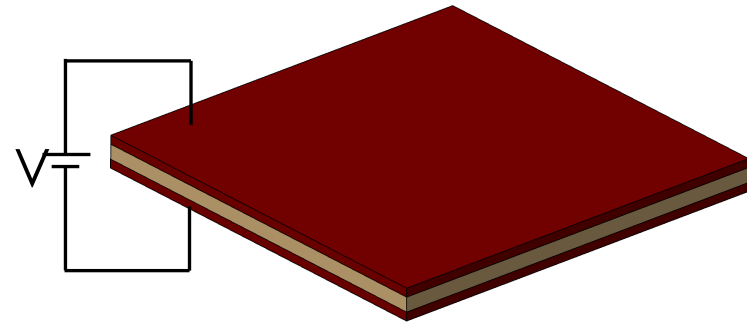
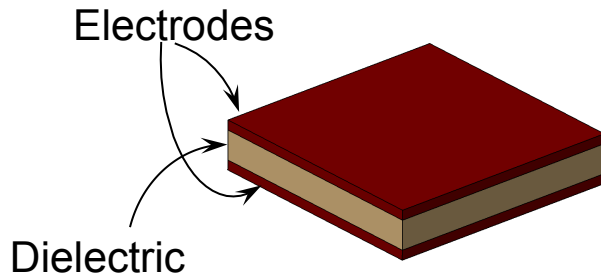
# Buckling into Shells

## Buckling of microscale elastic plates



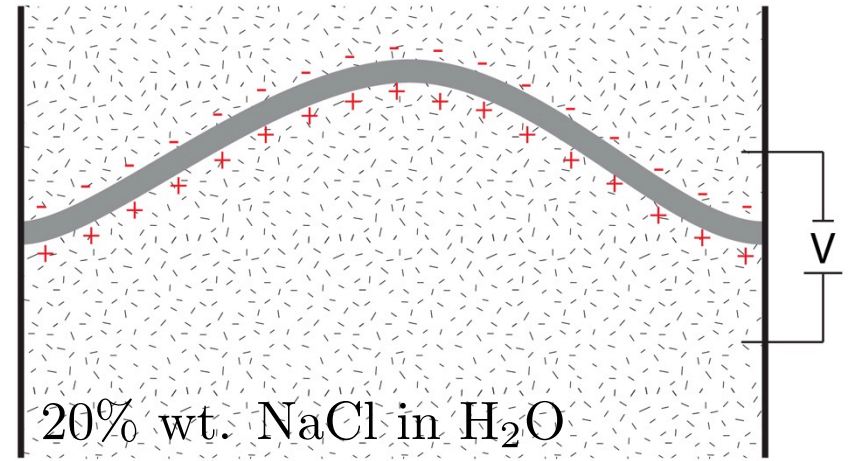
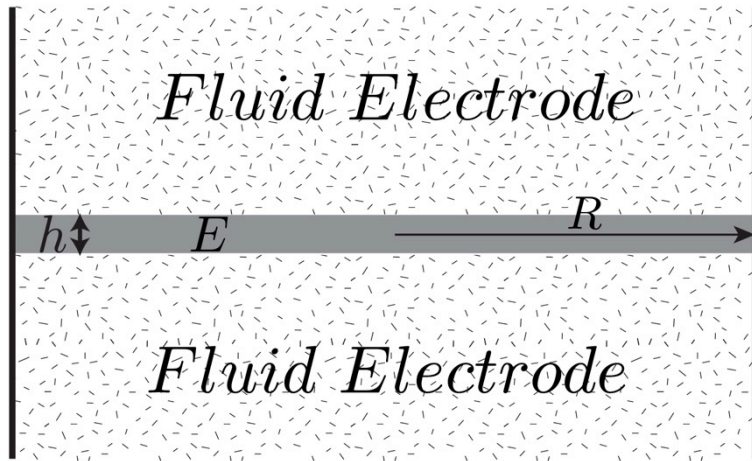


# Buckling into Shells





# Buckling into Shells



At what **voltage** will the plate **buckle**?

$$\epsilon_r \approx \frac{\epsilon_0 \epsilon}{2E} \left( \frac{V}{h} \right)^2 \rightarrow \sigma_r \approx \frac{\epsilon_0 \epsilon}{2(1 - \nu^2)} \left( \frac{V}{h} \right)^2 \rightarrow \sigma_c = \frac{kEh^2}{12(1 - \nu^2)R^2}$$

Radial strain in a dielectric elastomer

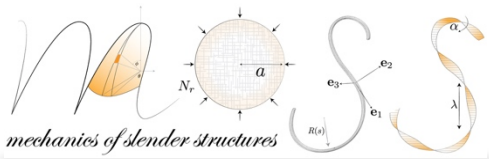
Plane stress relation

Linear stability analysis for a buckling plate

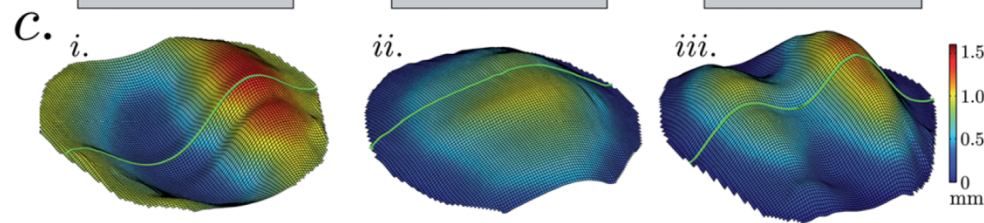
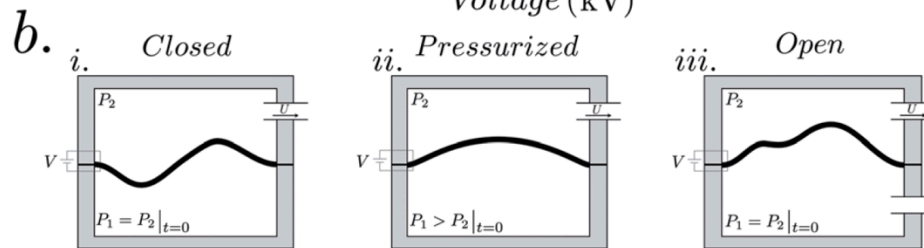
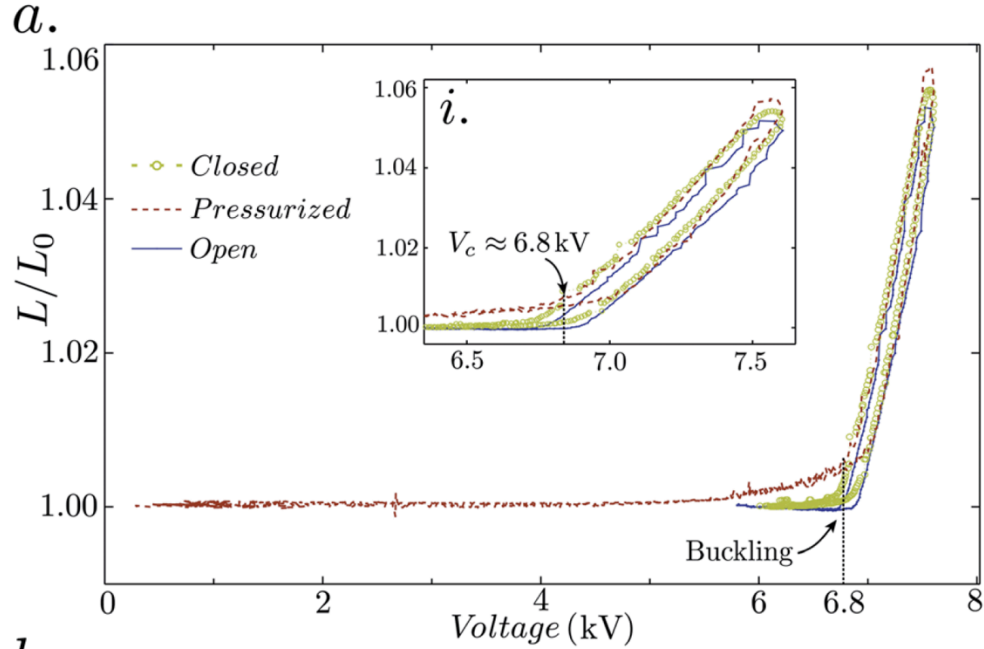
$$V_c \approx h^2 \left( \frac{kE}{6\epsilon_0 \epsilon R^2} \right)^{1/2}$$

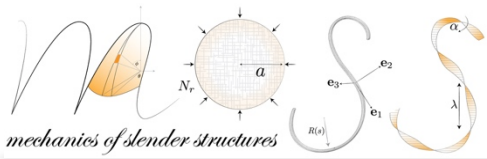
B. Tavakol, M. Bozlar, G. Froehlicher, H. A. Stone, I. Aksay, and D. P. Holmes, "Buckling Instabilities of Dielectric Elastomeric Plates for Flexible Microfluidic Pumps," *Soft Matter*, 10(27), 4789–4794, (2014).

B. Tavakol, A. Chawan, and D. P. Holmes, "Buckling Instability of Thin Films as a Means to Control or Enhance Fluid Flow within Microchannels," in preparation, (2015).

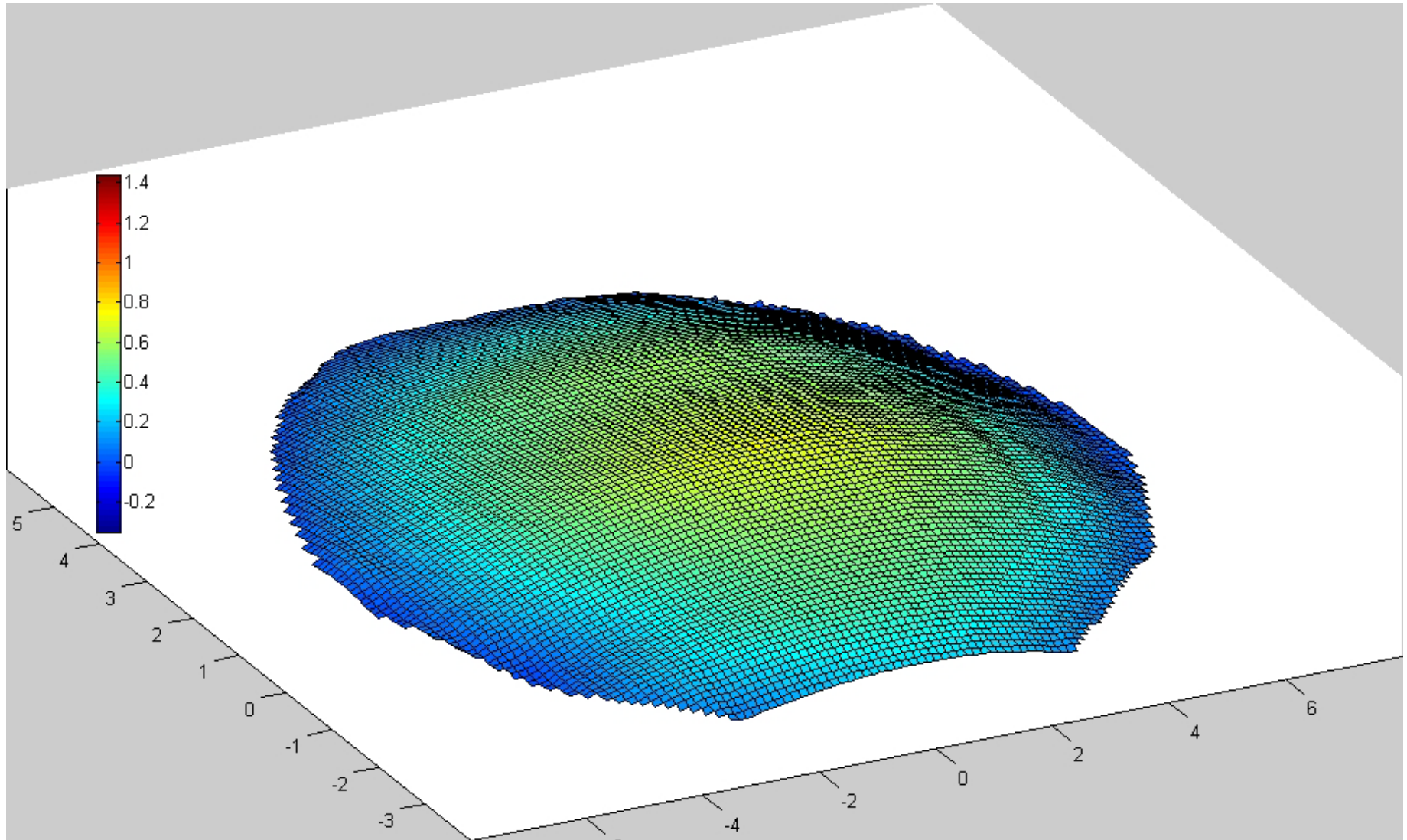


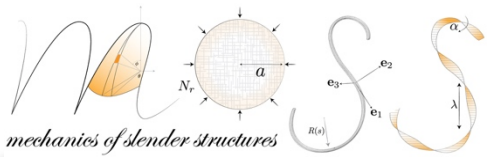
# Buckling into Shells



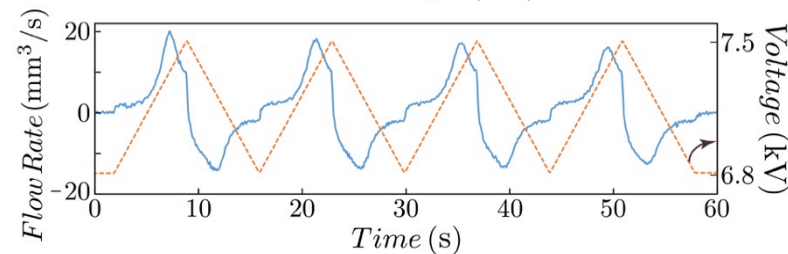
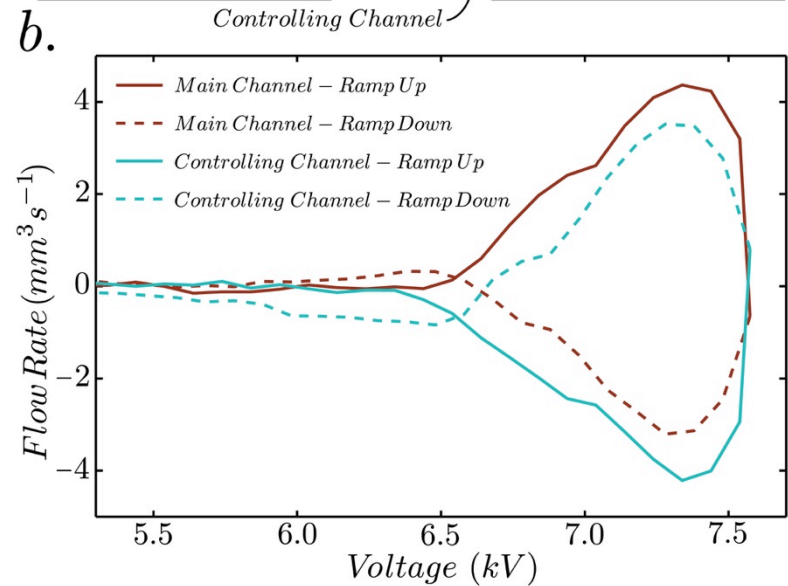
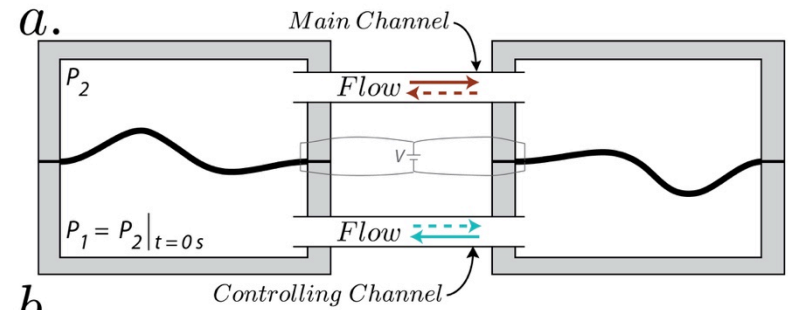
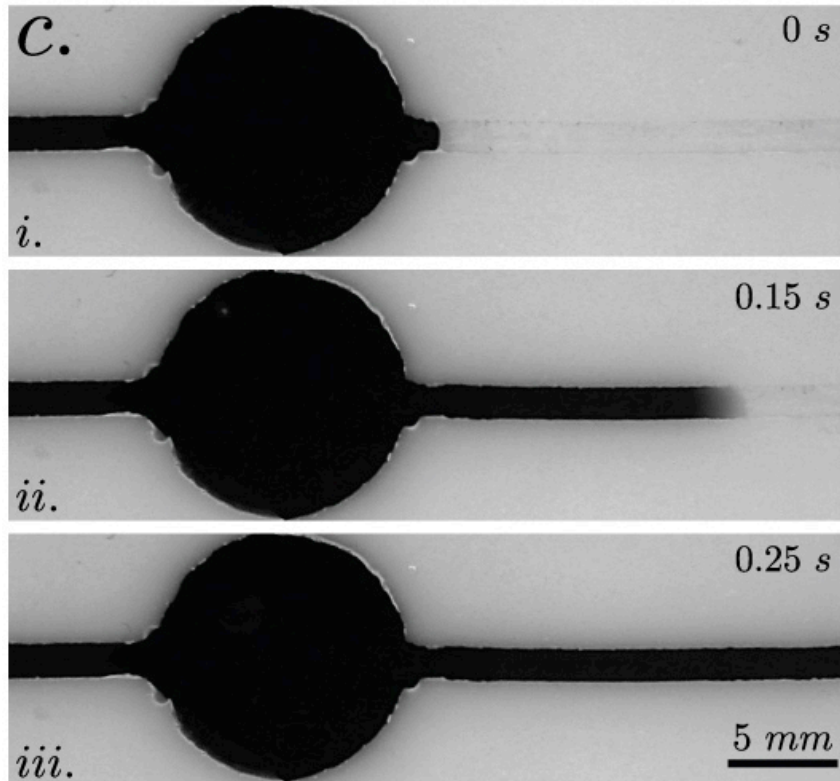
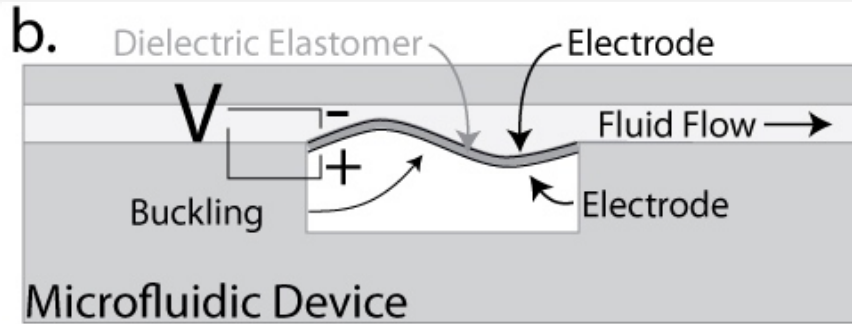


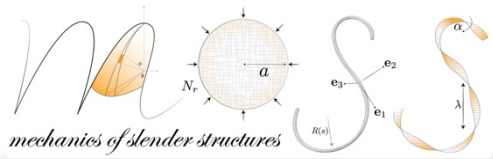
# Buckling into Shells



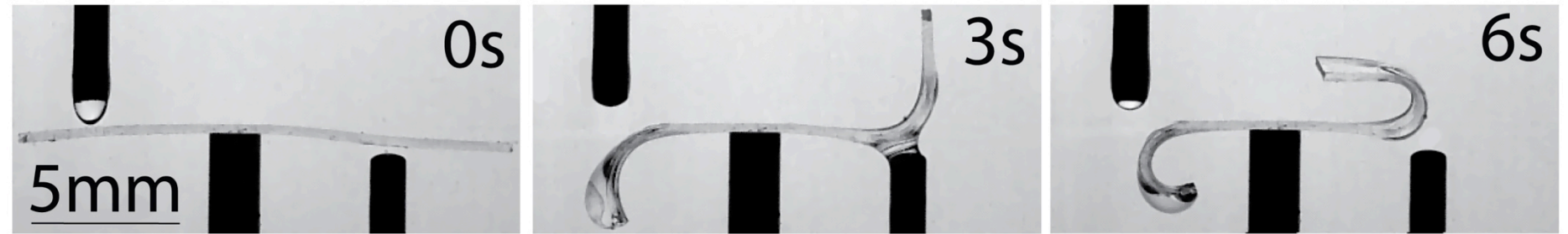
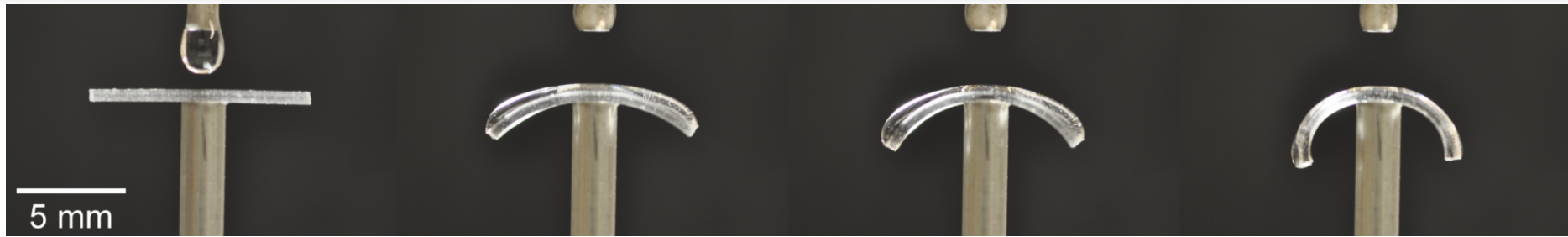


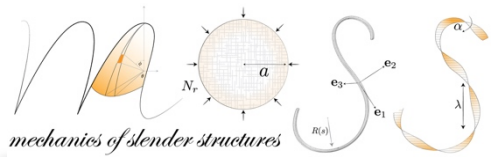
# Buckling into Shells



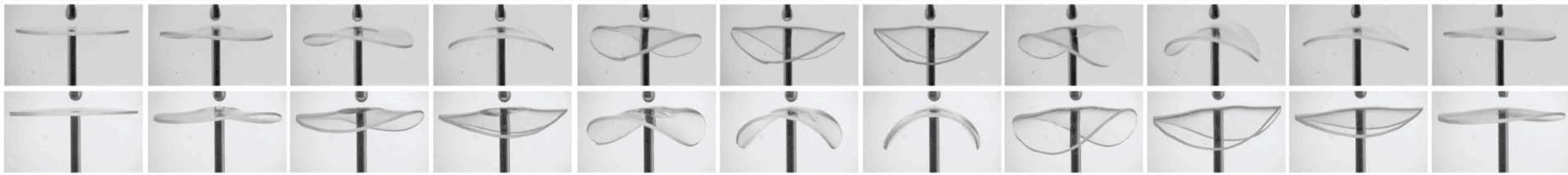


# Swelling & Buckling





# Plate Shape



- Calculate & **minimize** the **plate's energy** as a function of time.

$$\mathcal{U}_m = \frac{h}{2E} \int_A \left[ (\nabla^2 \varphi)^2 - (1 + \nu) \diamond^4[\varphi, \varphi] \right] dA,$$

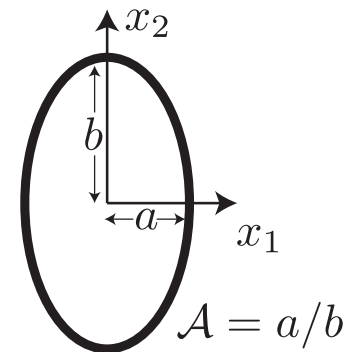
$$\mathcal{U}_b = \frac{B}{2} \int_A \left[ (\nabla^2 w)^2 - (1 - \nu) \diamond^4[w, w] \right] dA$$

- Satisfy compatibility between out-of-plane **curvature** & in-plane **stretching**.

$$-\Delta K = \epsilon_{\alpha\gamma} \epsilon_{\beta\delta} \epsilon_{\alpha\beta, \gamma\delta} = \frac{1}{E} \nabla^4 \varphi$$

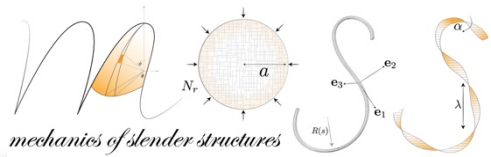
- Assume a form for the Airy stress function

$$\varphi = \frac{E \Delta K}{64} (x_1^2 + x_2^2) (a^2 + b^2 - x_1^2 - x_2^2)$$

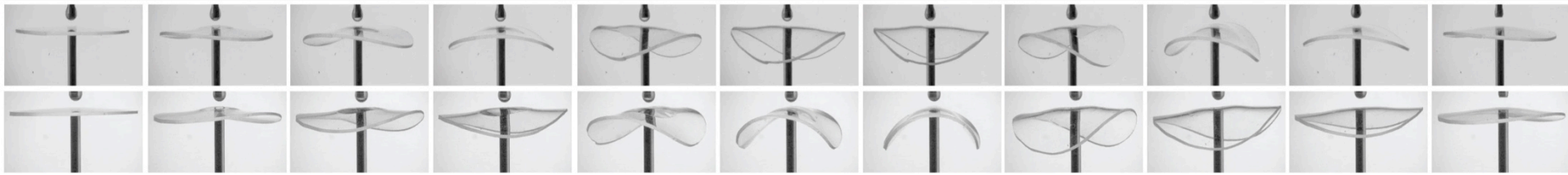


- Minimize the energy with respect to the **principal curvatures**.





# Plate Shape



- Calculate & **minimize** the **plate's energy** as a function of time.

$$\mathcal{U}_m = \frac{h}{2E} \int_A \left[ (\nabla^2 \varphi)^2 - (1 + \nu) \diamond^4[\varphi, \varphi] \right] dA,$$

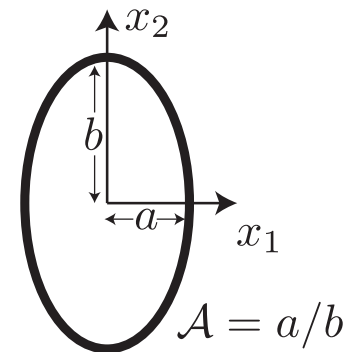
$$\mathcal{U}_b = \frac{B}{2} \int_A \left[ (\nabla^2 w)^2 - (1 - \nu) \diamond^4[w, w] \right] dA$$

- Satisfy compatibility between out-of-plane **curvature** & in-plane **stretching**.

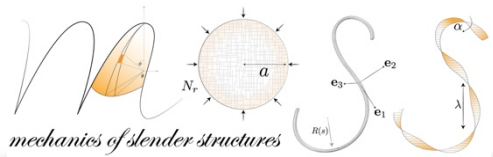
$$-\Delta K = \epsilon_{\alpha\gamma} \epsilon_{\beta\delta} \epsilon_{\alpha\beta, \gamma\delta} = \frac{1}{E} \nabla^4 \varphi$$

- Assume a form for the Airy stress function

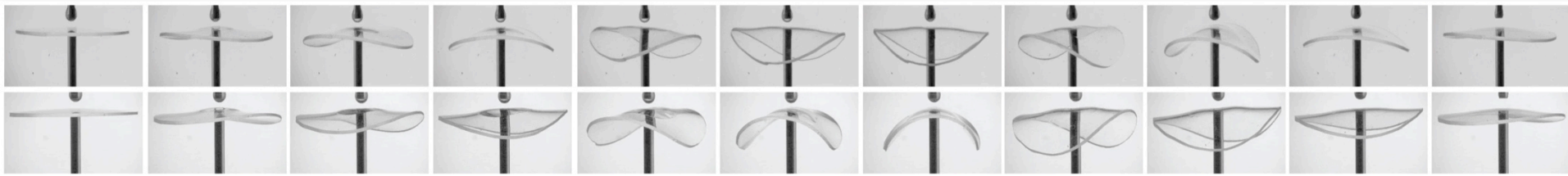
$$\varphi = \frac{E \Delta K}{64} (x_1^2 + x_2^2) (a^2 + b^2 - x_1^2 - x_2^2)$$



- Minimize the energy with respect to the **principal curvatures**.



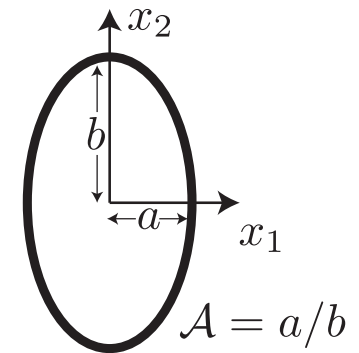
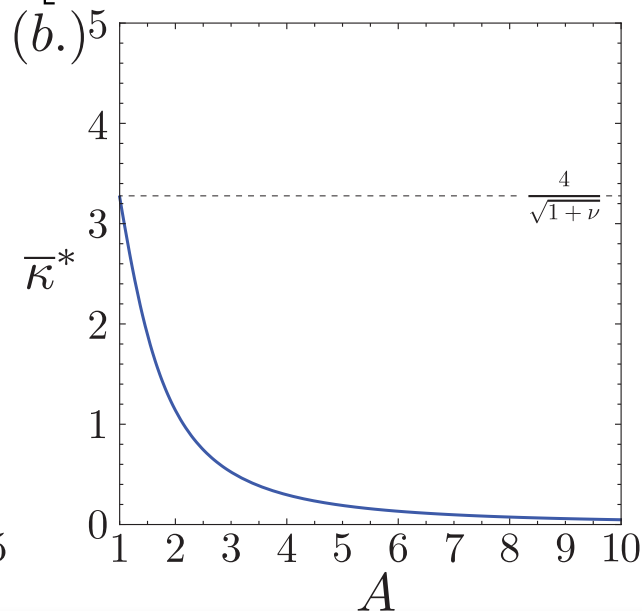
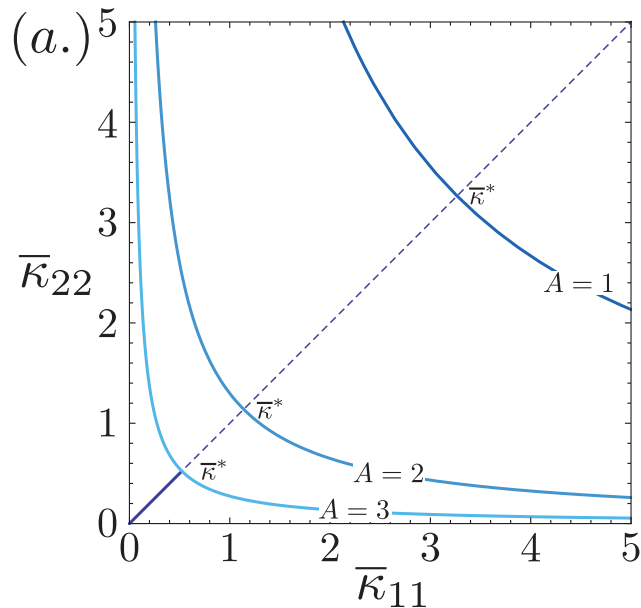
# Plate Shape

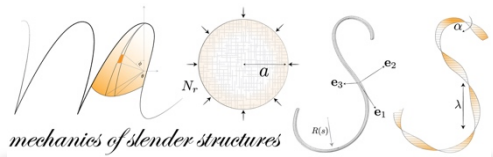


Minimization of the total strain energy with respect to the unknown curvatures:

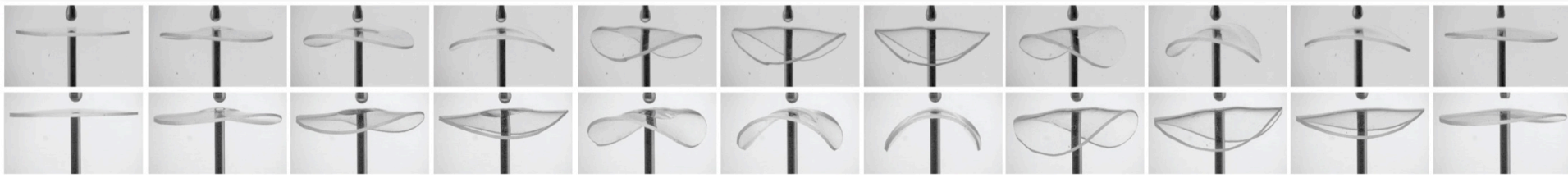
$$\bar{\kappa}_{11} = \bar{\kappa}_{22}, \text{ or}$$

$$\bar{\kappa}_{11} = -\frac{256}{\bar{\kappa}_{22} (1 + \nu) \left[ -9 + 2A^2 - 9A^4 + 3\nu (A^2 - 1)^2 \right]}$$





# Dynamic Plate Shape



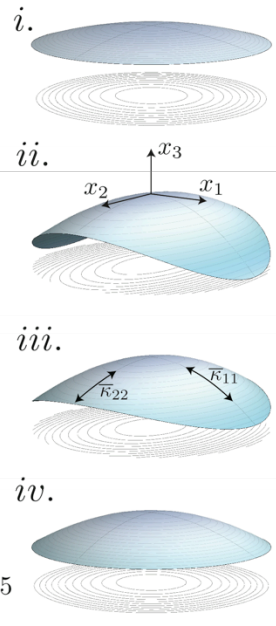
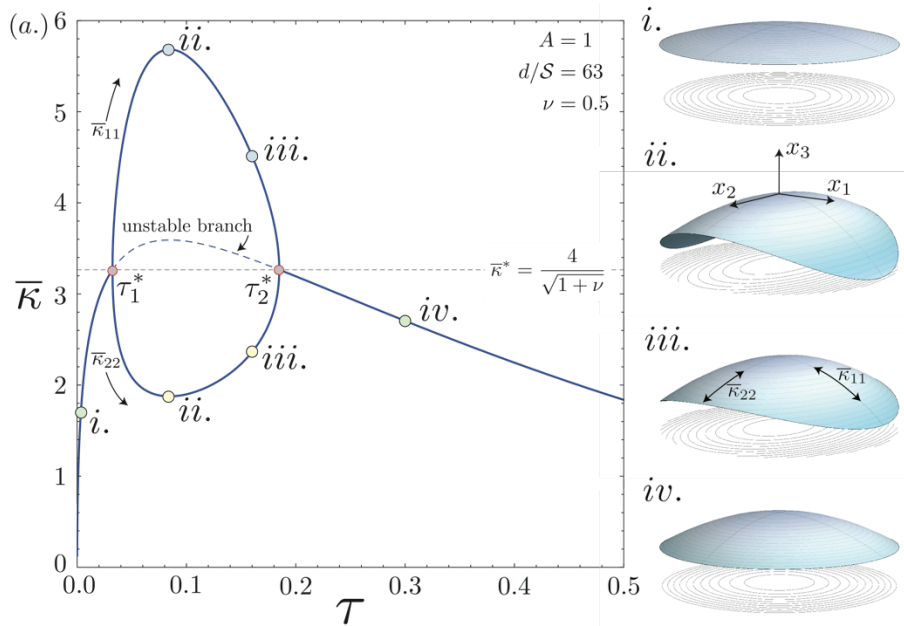
Diffusive dynamics:

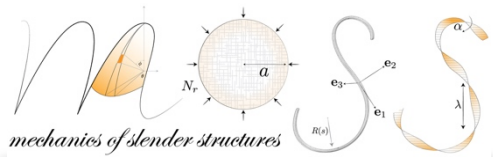
$$\psi(\bar{x}_3, \tau) = 1 + \sum_{n=0}^{\infty} \mathcal{A}(n) \sin[\lambda(n)\bar{x}_3] e^{-\lambda(n)^2 \tau}$$

Bending dynamics:

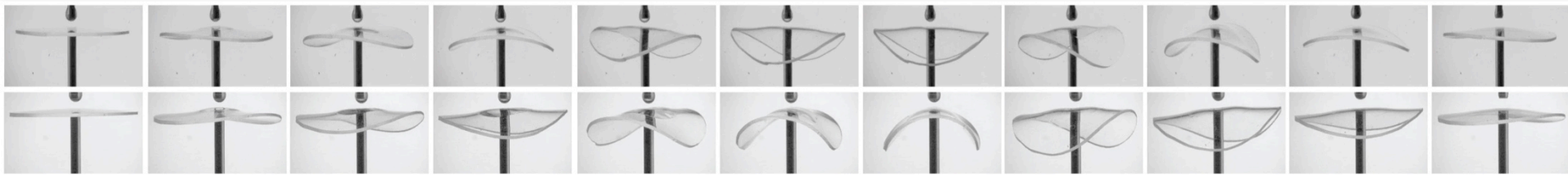
$$\bar{\kappa}_\varepsilon(\tau) = \frac{d}{S} \int_{-1/2}^{1/2} \psi(\bar{x}_3, \tau) \bar{x}_3 \, d\bar{x}_3$$

$$d/S = 12\varepsilon_{eq} L^2 / h^2$$





# Dynamic Plate Shape

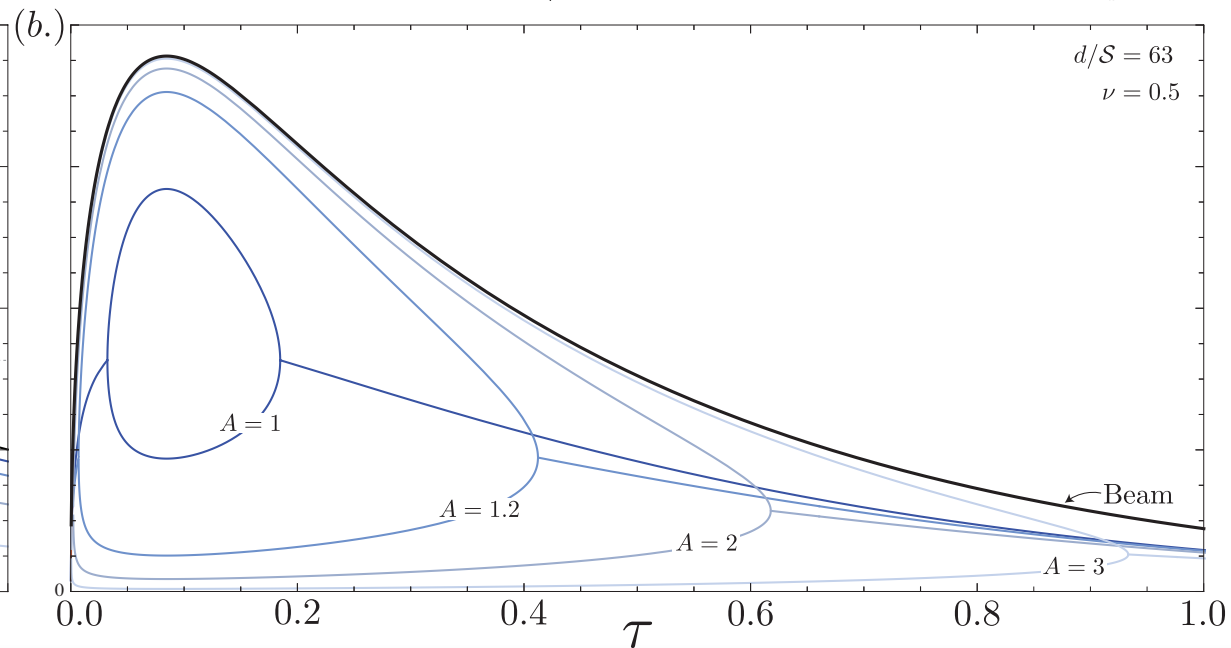
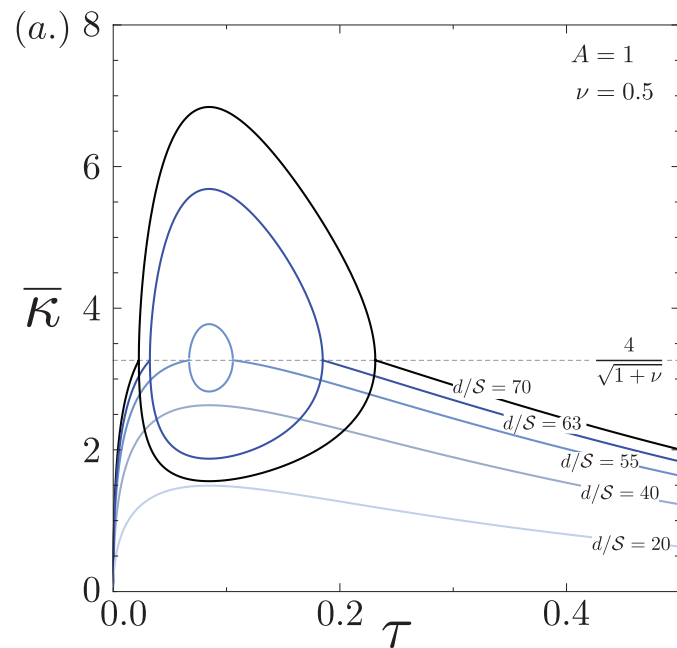


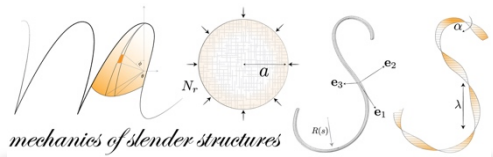
Diffusive dynamics:

$$\psi(\bar{x}_3, \tau) = 1 + \sum_{n=0}^{\infty} \mathcal{A}(n) \sin[\lambda(n)\bar{x}_3] e^{-\lambda(n)^2 \tau}$$

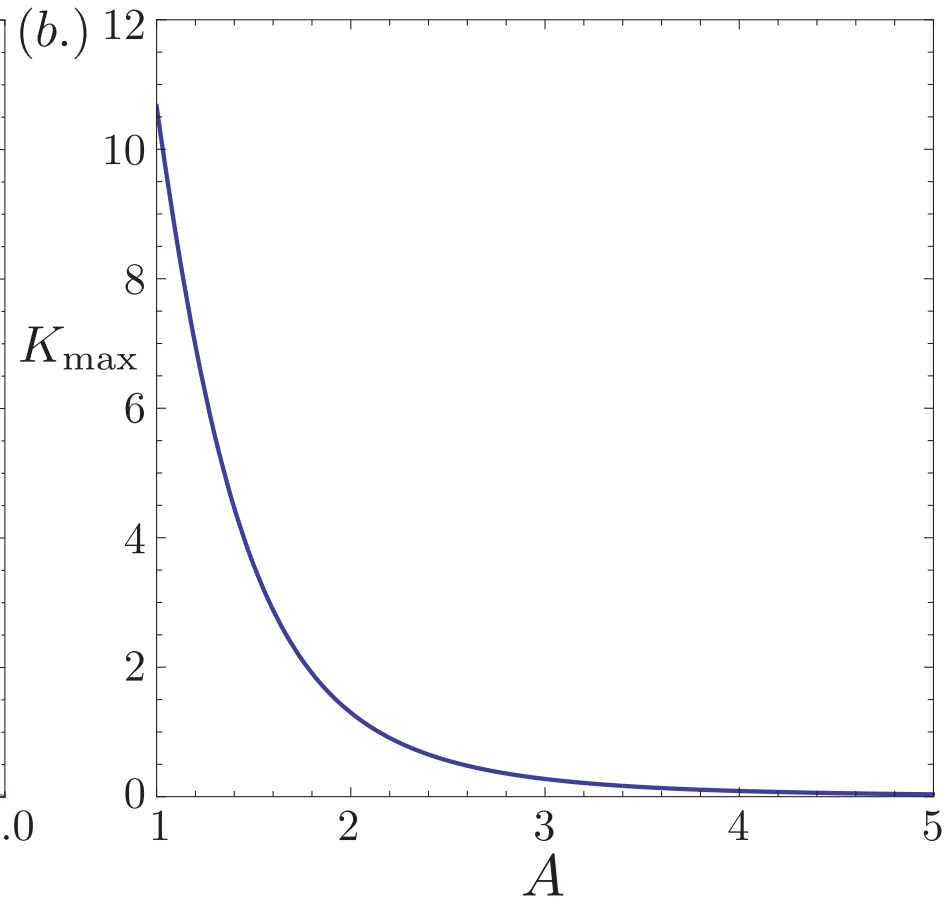
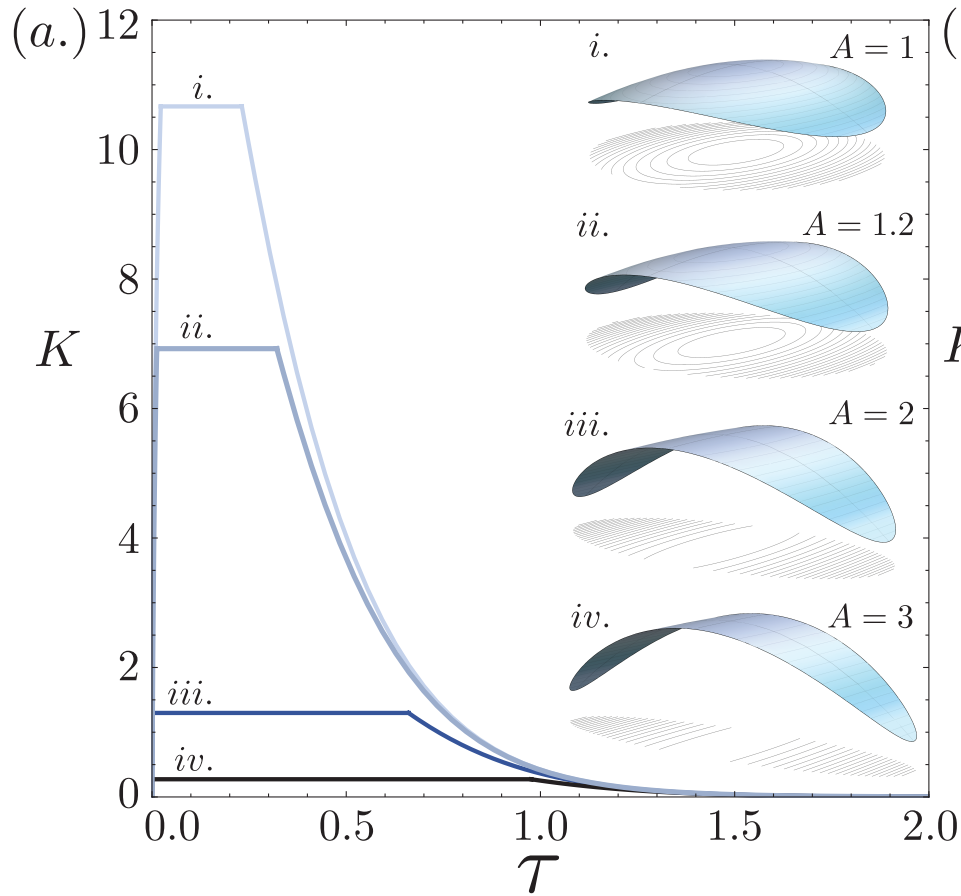
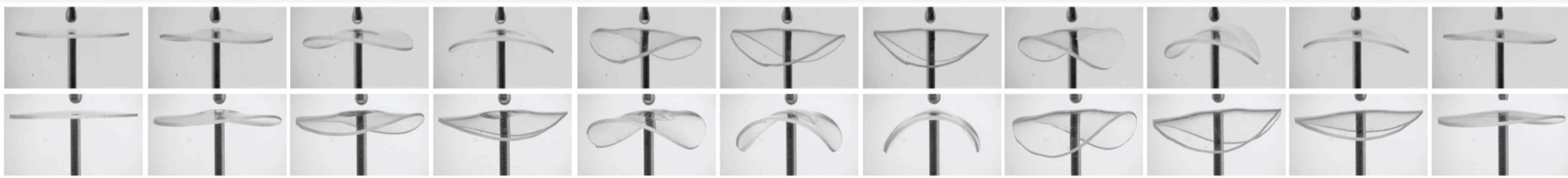
Bending dynamics:

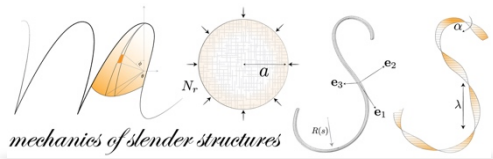
$$\bar{\kappa}_\varepsilon(\tau) = \frac{d}{S} \int_{-1/2}^{1/2} \psi(\bar{x}_3, \tau) \bar{x}_3 \, d\bar{x}_3 \quad d/S = 12\varepsilon_{eq} L^2 / h^2$$



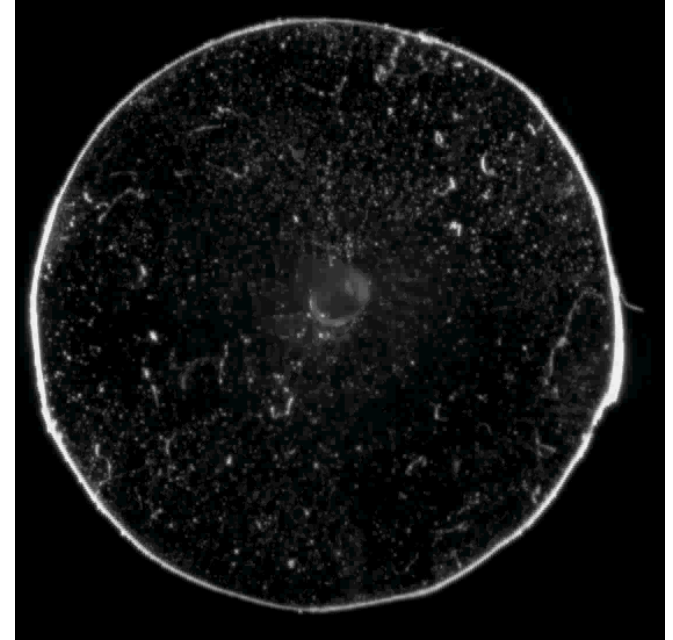


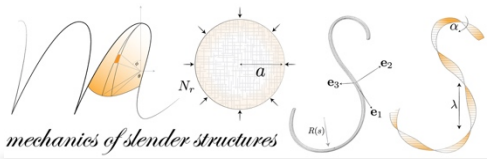
# Plate Shape



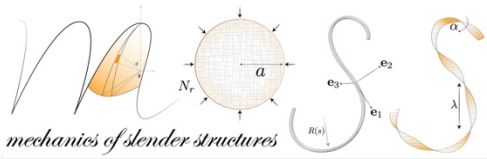


# Dynamics: Twisting





# Wrinkling



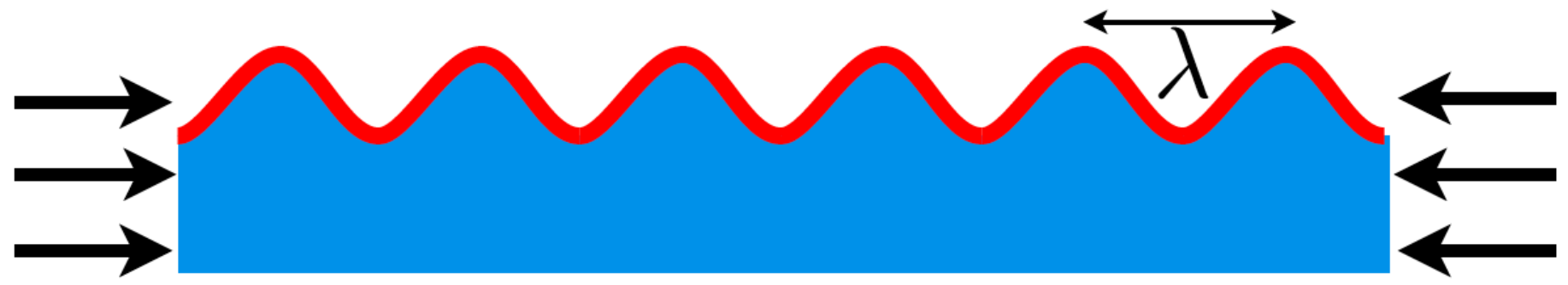
## Uniaxial compression:

Thin stiff film



$E_f$

$E_s$





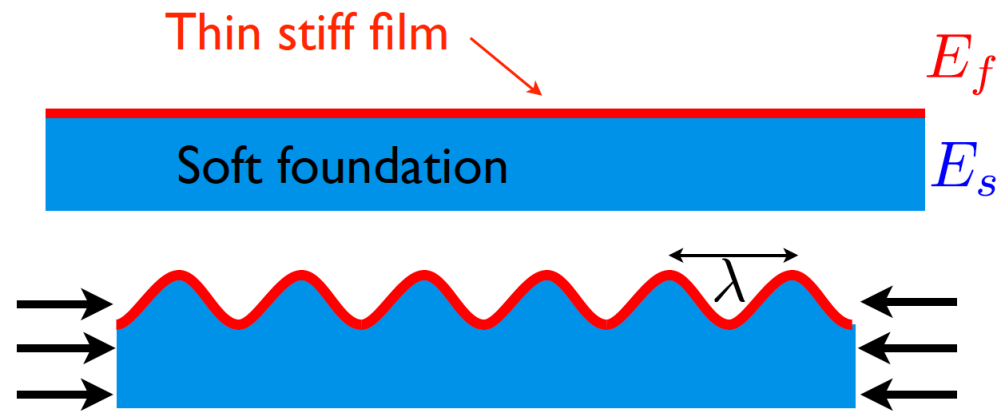


# Wrinkling

## Wrinkle Wavelength:

- Balance of the **plate's bending energy**, and the energy required to **deform** the underlying **substrate**.
- **Bending** penalizes **short wavelengths**.
- Deforming the **elastic foundation** penalizes **long wavelengths**.
- **Intermediate wavelength** emerges when the reaction of the underlying layer is proportional to the deflection of the plate.

## Uniaxial compression:



## Equilibrium Plate Eq. (FvK)

$$B \nabla^4 w - h \sigma_{\alpha\beta} \partial_{\alpha\beta}^2 w = 0$$

## 1D wrinkles in x-direction:

$$B \frac{\partial^4 w}{\partial x^4} - h \sigma_{xx} \frac{\partial^2 w}{\partial x^2} + K w = 0$$



# Wrinkling

1D wrinkles in x-direction:

$$B \frac{\partial^4 w}{\partial x^4} - h \sigma_{xx} \frac{\partial^2 w}{\partial x^2} + K w = 0$$

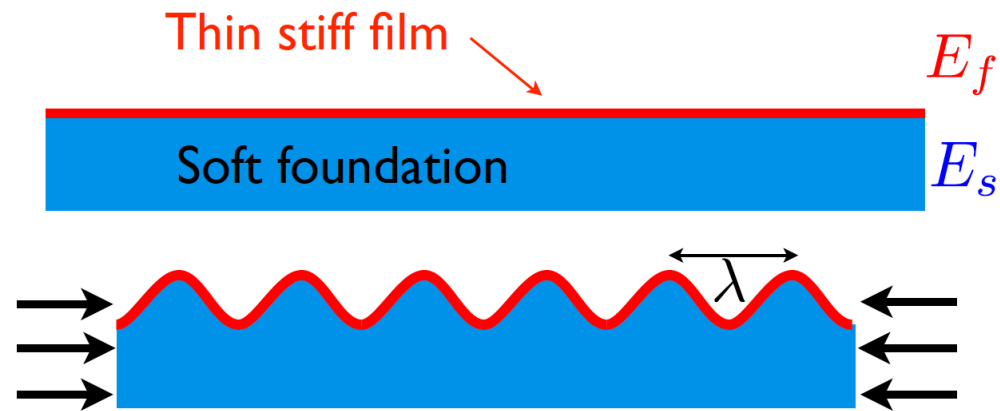
Linearizing this equation & disregarding any stretching of the mid-plane due to curvature, i.e. second term is zero

$$B \frac{w}{L^4} \sim K w$$

Scaling sets the wavelength:

$$L \rightarrow \lambda \sim \left( \frac{B}{K} \right)^{1/4}$$

Uniaxial compression:



In the limit of a deep substrate:

$$h \ll \lambda \ll H_s$$

$$\lambda \sim h \left( \frac{E}{E_s} \right)^{1/3}$$



# Wrinkling

1D wrinkles in x-direction:

$$B \frac{\partial^4 w}{\partial x^4} - h \sigma_{xx} \frac{\partial^2 w}{\partial x^2} + K w = 0$$

Linearizing this equation about the flat, unbuckled state:

$$u_\alpha = w = 0$$

Linearized strains:

$$\varepsilon_{\alpha\beta} \approx 1/2(u_{\alpha,\beta} + u_{\beta,\alpha})$$

Perform linear stability analysis. Linearized equations below permit periodic solutions:

$$B \nabla^4 w - h \sigma_0 \nabla^2 w = -p \quad \text{and} \quad \nabla^4 \phi = 0$$

$$\sigma_c = E^* \left( \frac{3E_s^*}{2E^*} \right)^{2/3} \quad \lambda = \pi h \left( \frac{2E^*}{3E_s^*} \right)^{1/3}$$

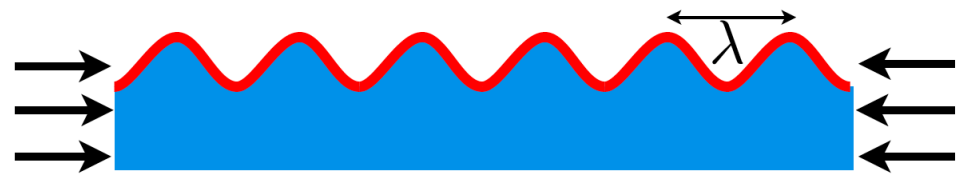
Uniaxial compression:

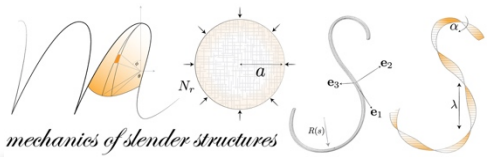
Thin stiff film

$E_f$

Soft foundation

$E_s$



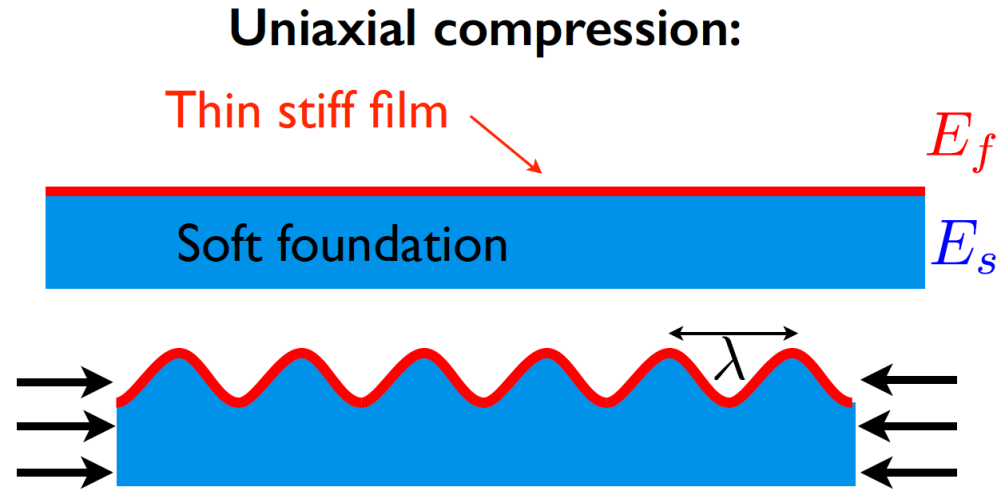


# Wrinkling

1D wrinkles in x-direction:

$$\sigma_c = E^* \left( \frac{3E_s^*}{2E^*} \right)^{2/3}$$

$$\lambda = \pi h \left( \frac{2E^*}{3E_s^*} \right)^{1/3}$$



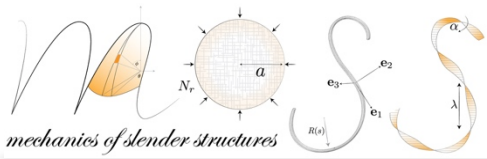
Effective or Reduced moduli:

Plate:

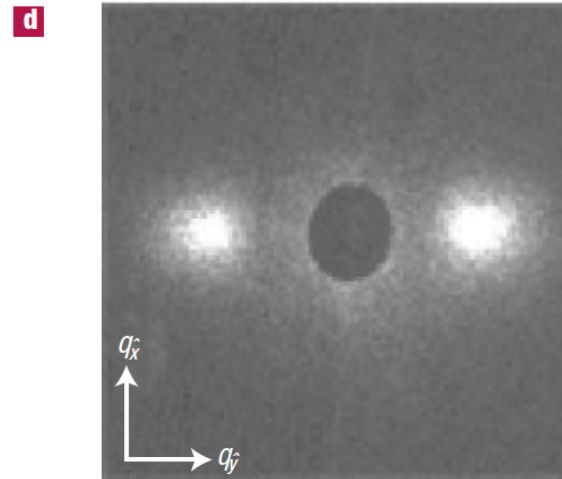
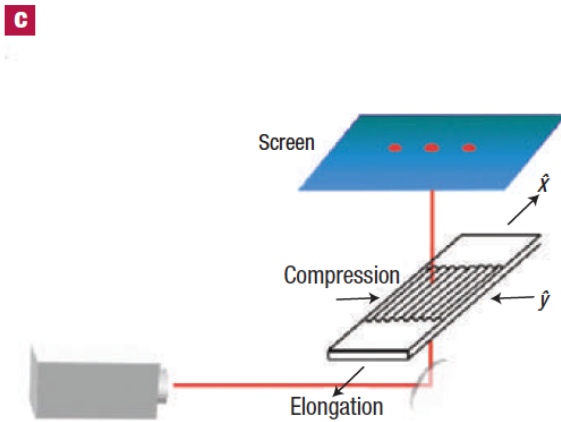
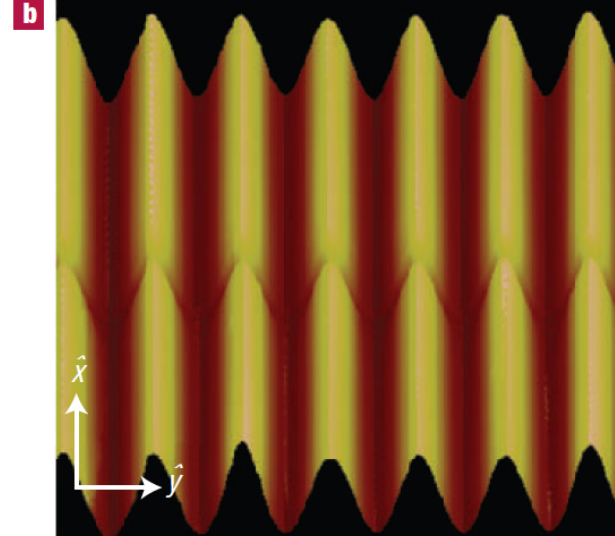
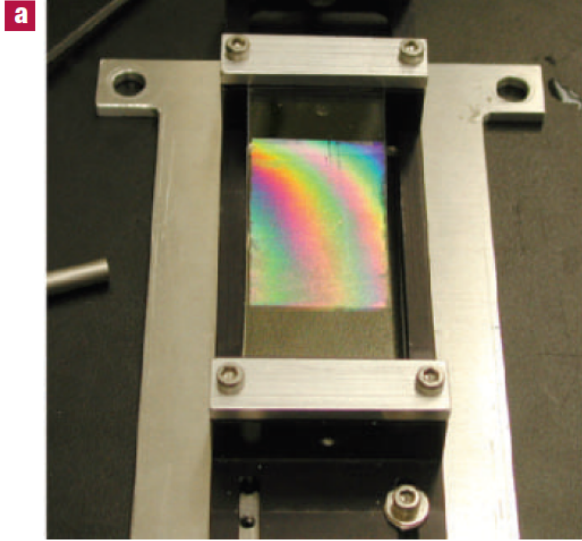
$$E^* = \frac{E}{1 - \nu^2}$$

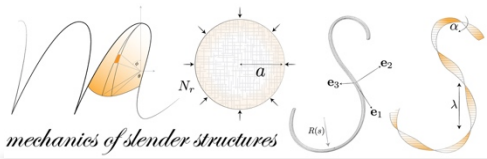
Substrate:

$$E_s^* = \frac{E_s(1 - \nu_s)}{(1 + \nu_s)(3 - 4\nu_s)}$$



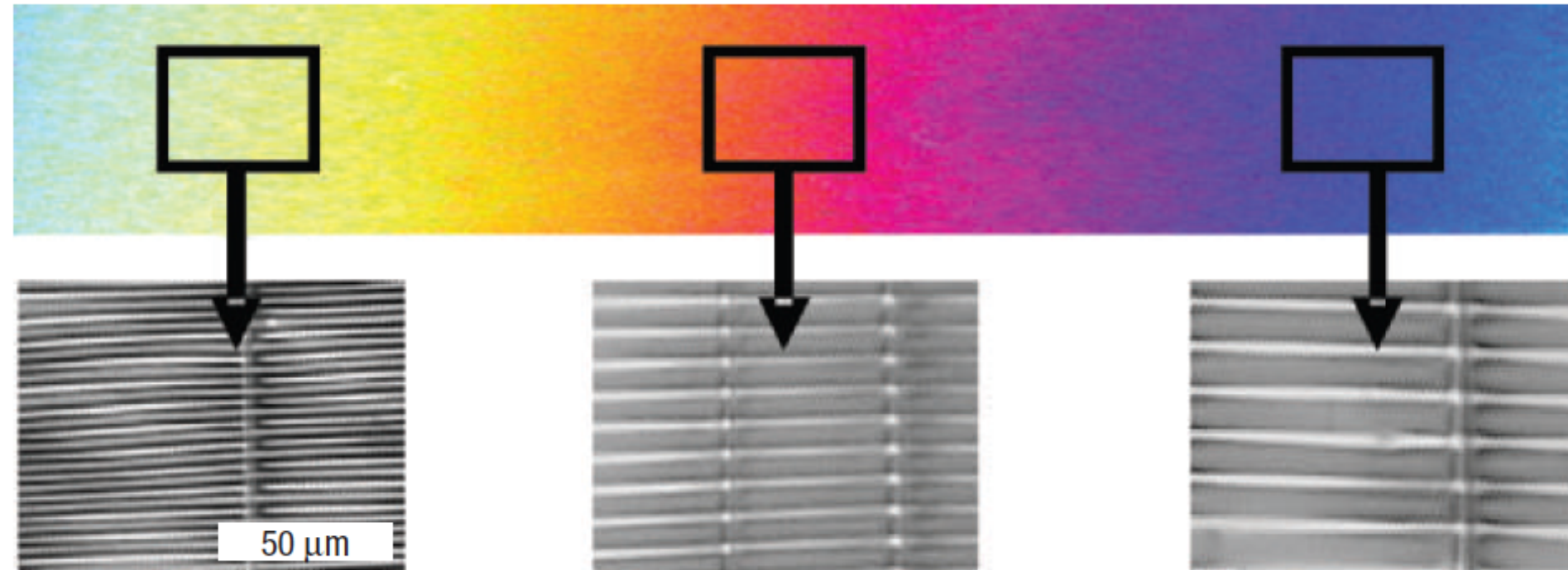
# Wrinkling

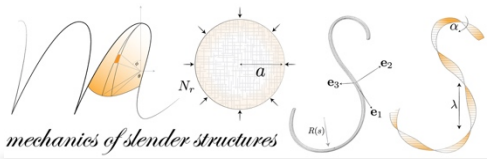




# Wrinkling

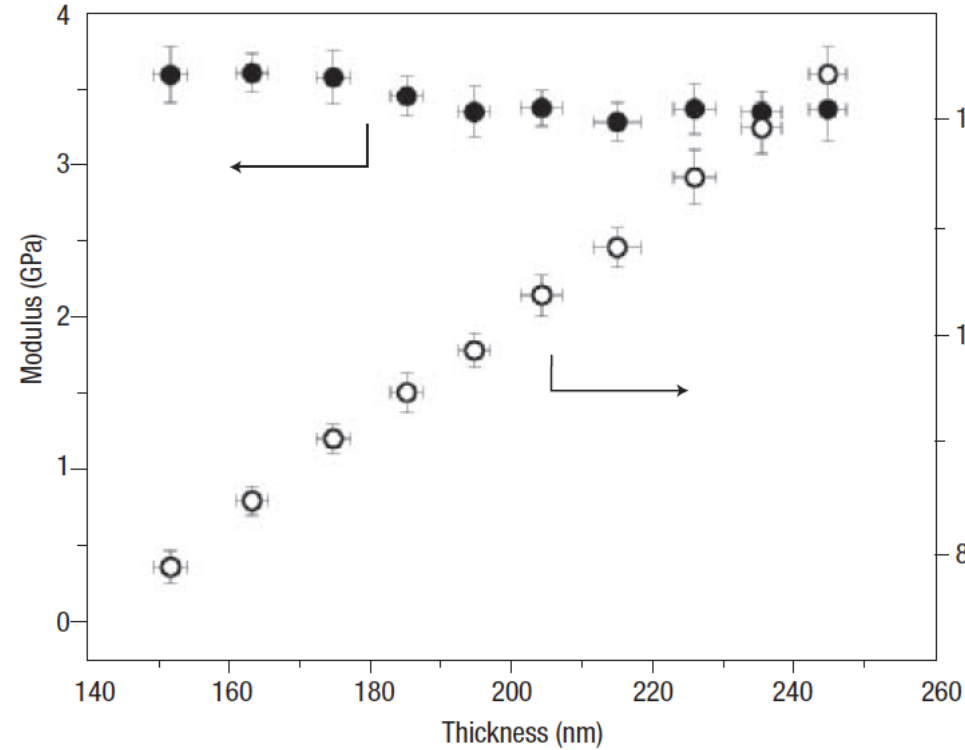
Thickness gradient:



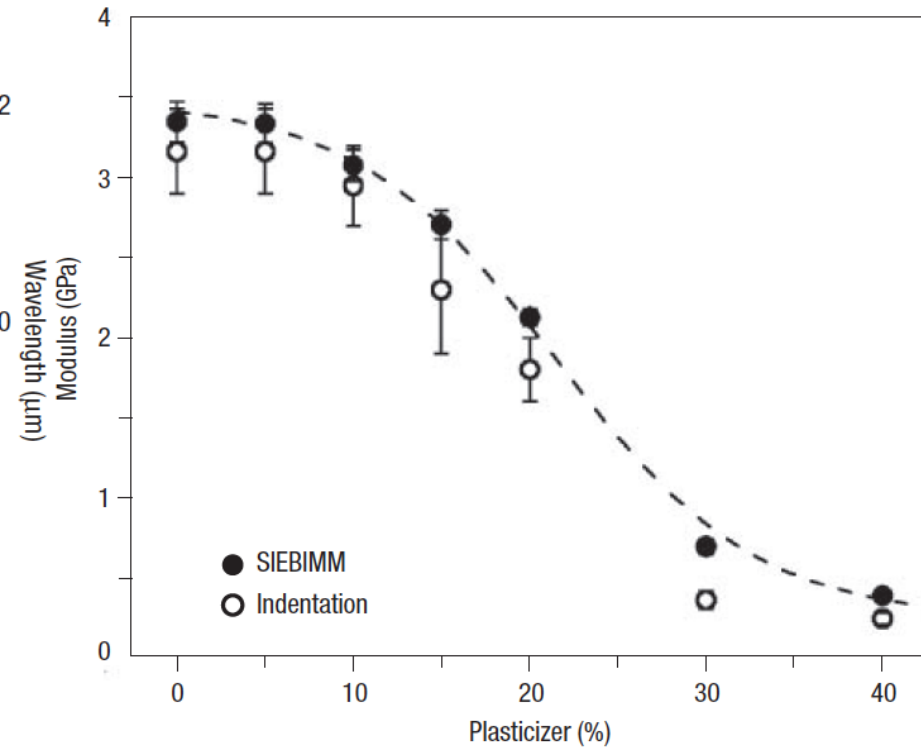


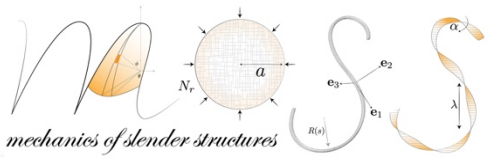
# Wrinkling

Polystyrene films:  $E=3.4$  GPa



Effect of plasticizing agent



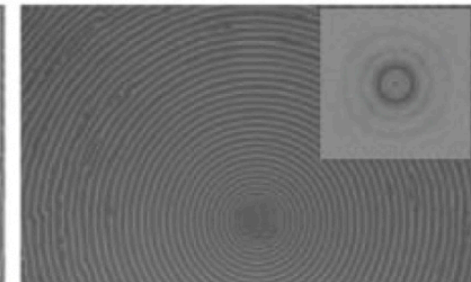
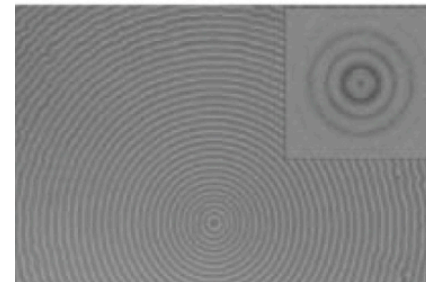
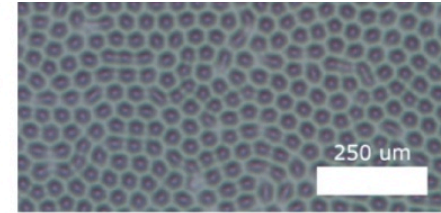
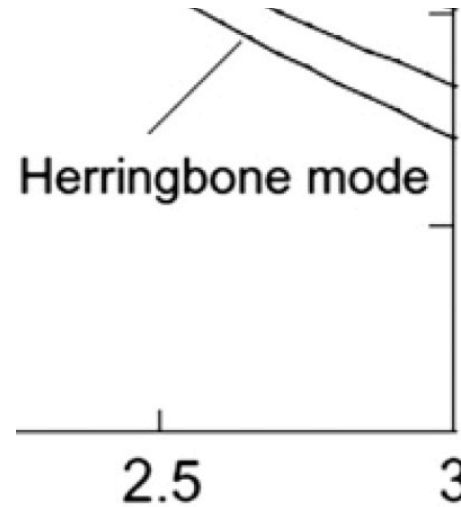


# Wrinkling

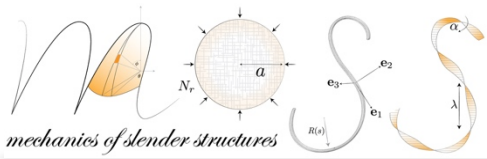
## Wrinkled adhesion:

- Maximum separation for a single wrinkle scales with its **perimeter** and the adhesion energy of the **interface**.
- Smaller wrinkles with a short persistence length will enhance adhesion.
- Low amplitude wrinkles enhance adhesion by increasing the contact line during separation.

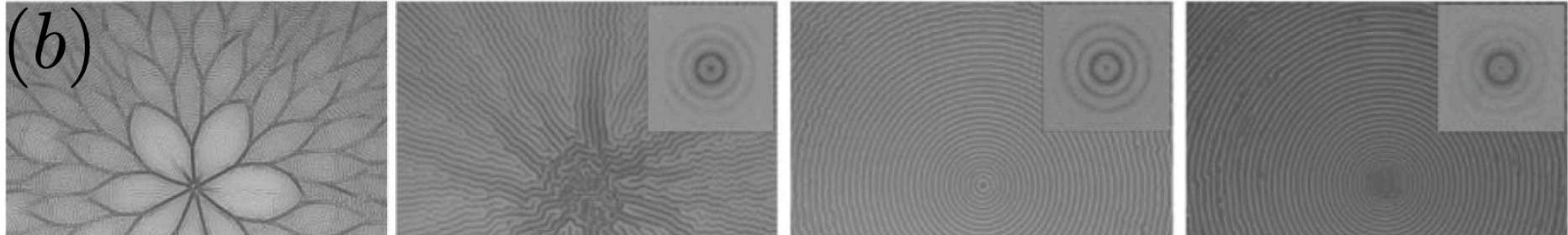
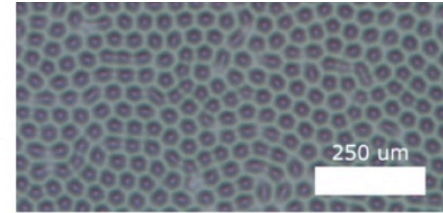
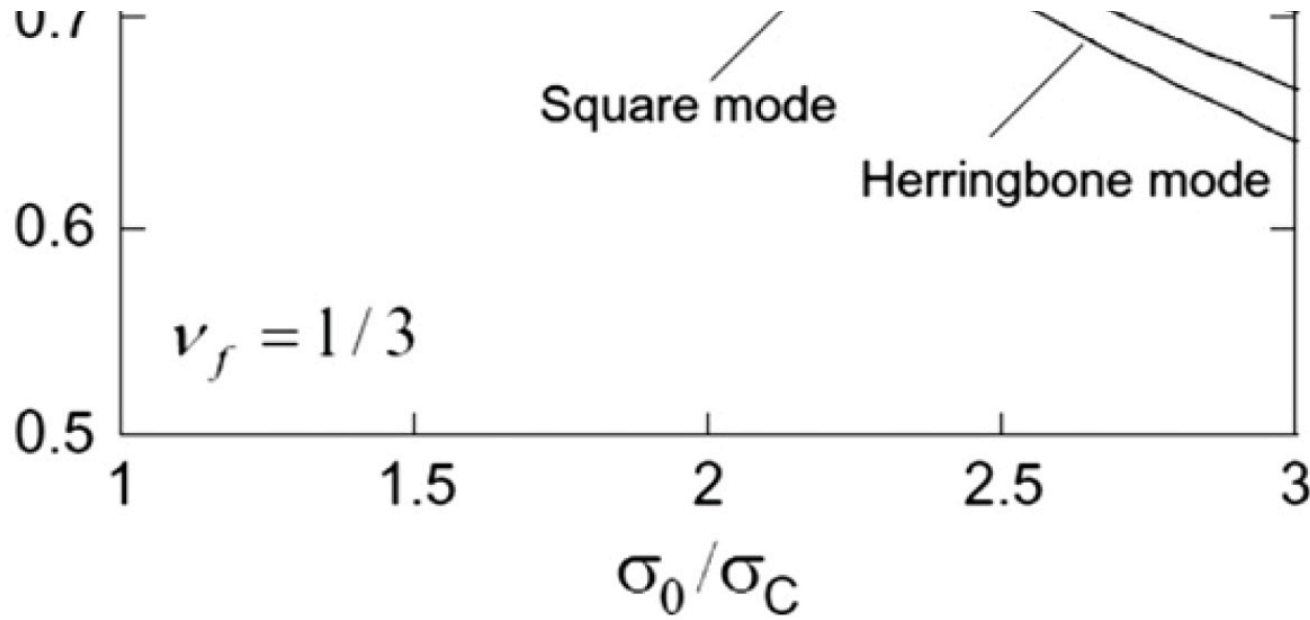
$$\frac{P_s}{P} \sim \underbrace{\left( \frac{G_c}{E^* A} \right)^{1/4}}_{\text{materials}} \underbrace{\left( \frac{R}{A} \right)^{1/4}}_{\text{geometry}}$$







# Wrinkling



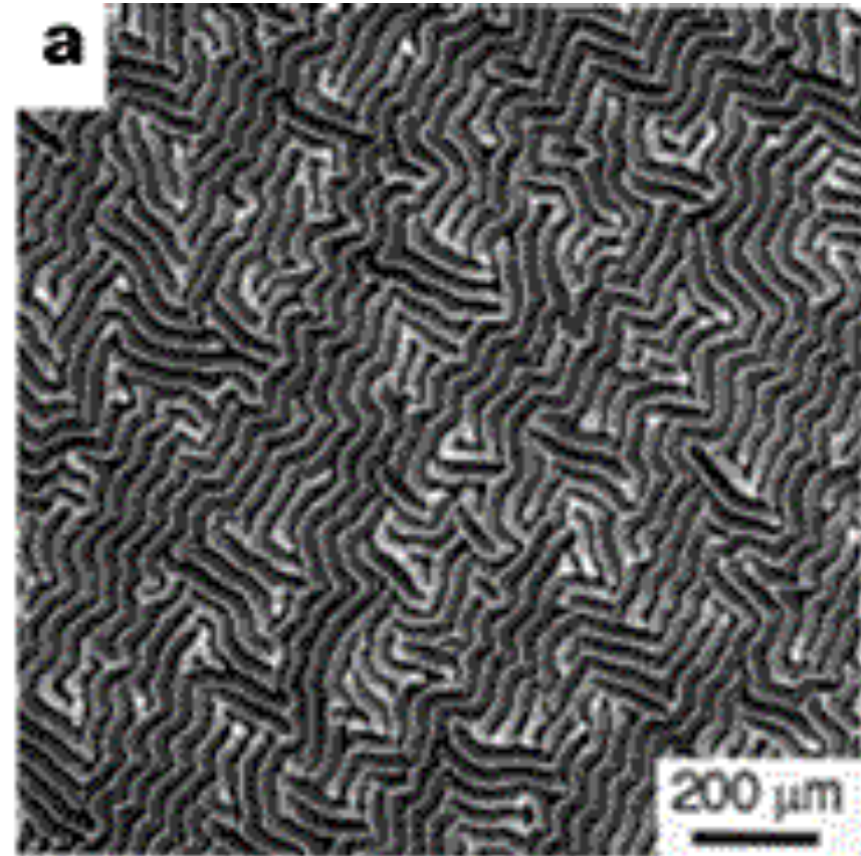


# Wrinkling

## Biaxial Compression

- Early examples of wrinkling appeared when **thin metallic films** (50 nm) were **deposited** by electron beam evaporation **onto PDMS**.
- Deposition locally heats the PDMS, upon cooling the compressive stress buckles the bilayer into a herringbone pattern.

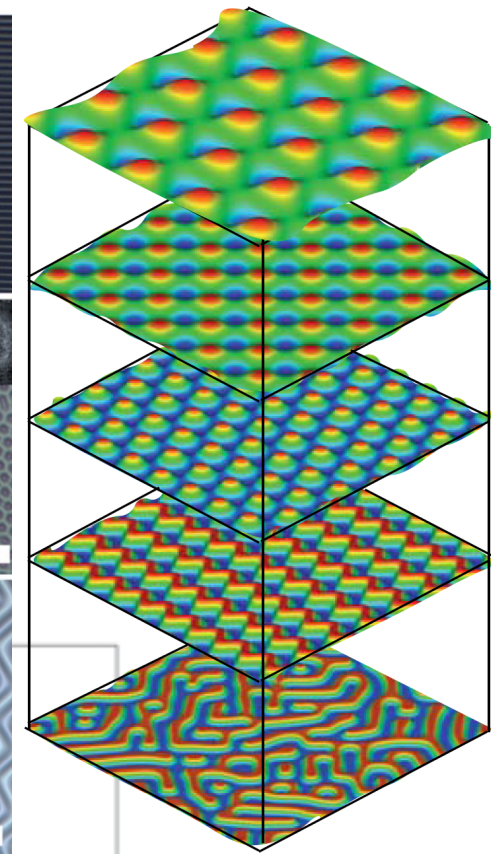
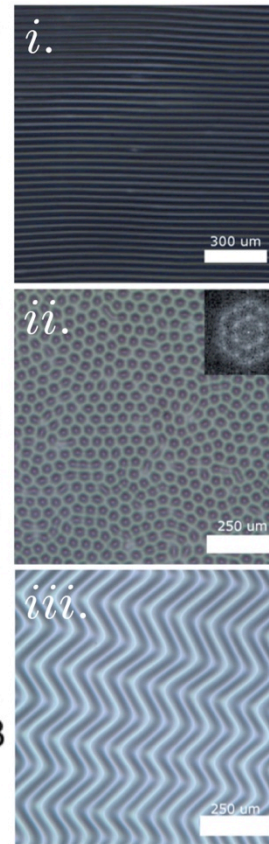
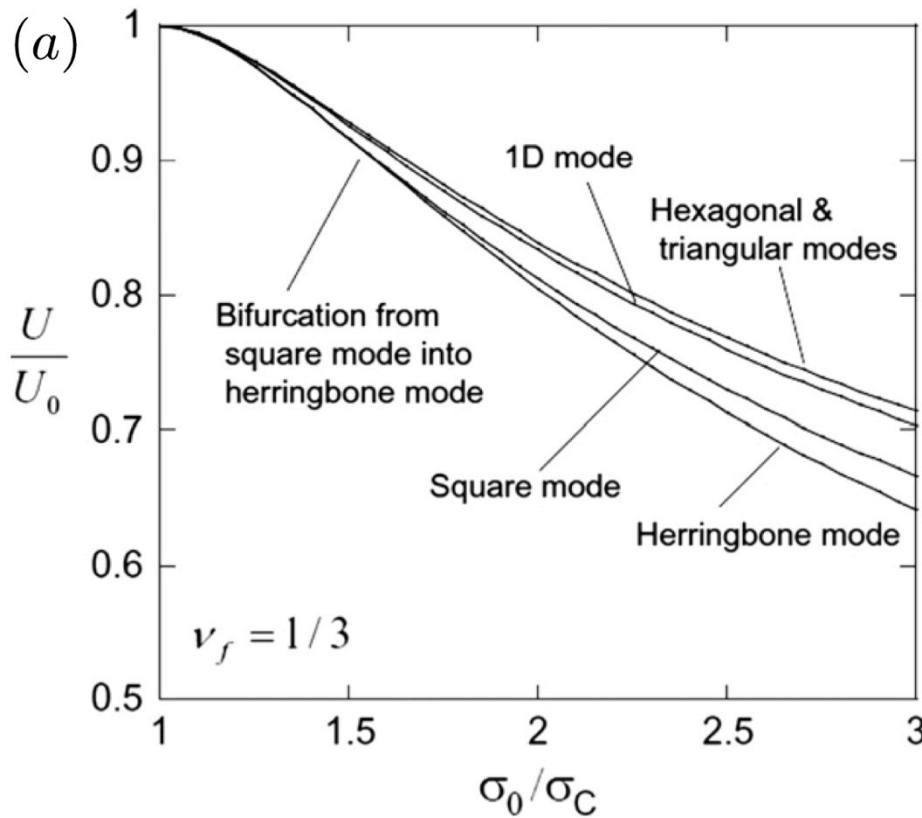
$$20\mu m \leq \lambda \leq 50\mu m$$










# Wrinkling

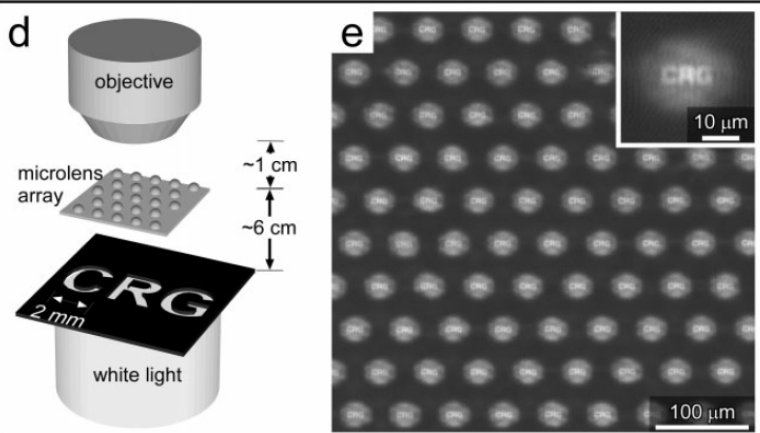
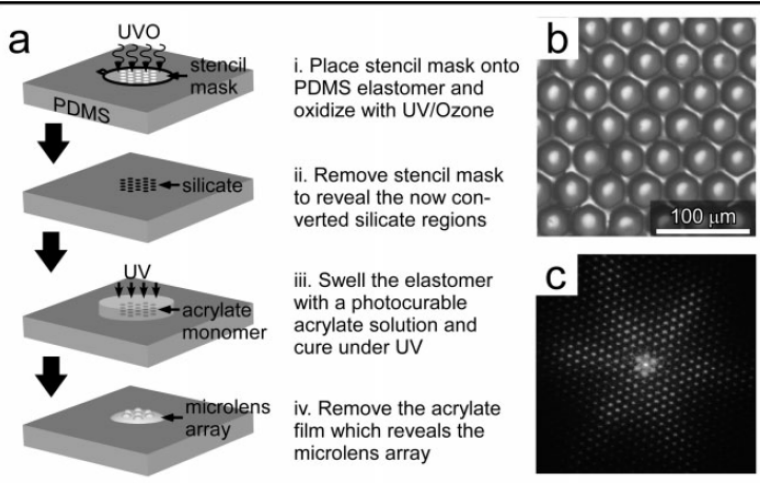
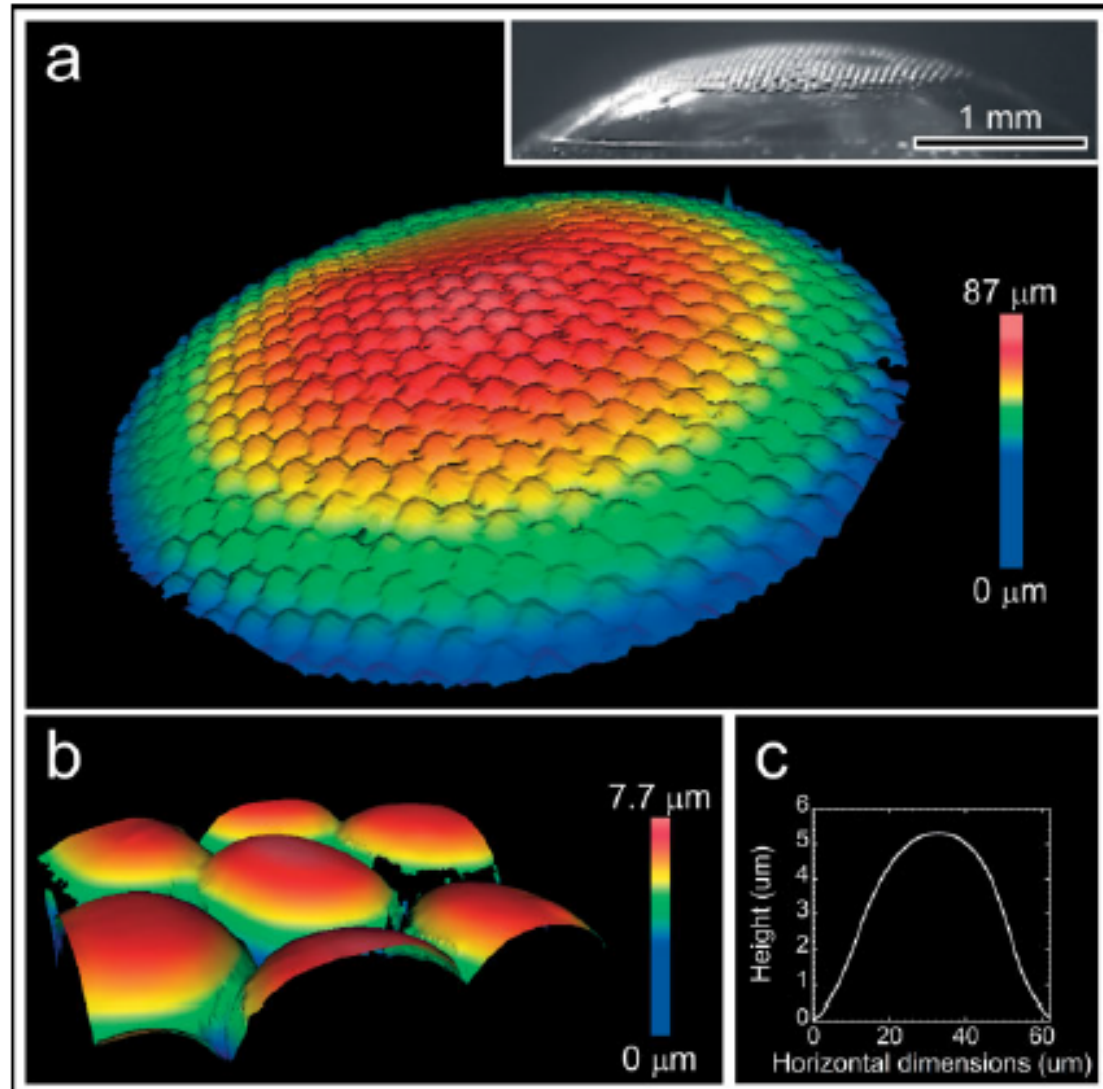
## Biaxial Compression

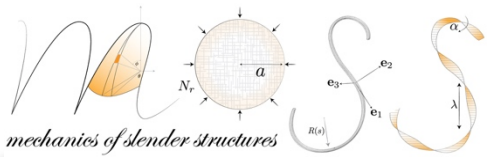


-  triangular
-  checkerboard
-  hexagonal
-  herringbone
-  labyrinthine

Obtained by linear stability of the cylindrical wrinkling pattern

## Biaxial Compression





## Elastic Instabilities for Form and Function Buckling, Wrinkling, Folding, and Snapping

### **Geometry and Mechanics:**

- Fundamental equations, geometric rigidity, morphing.

### **Buckling & Wrinkling:**

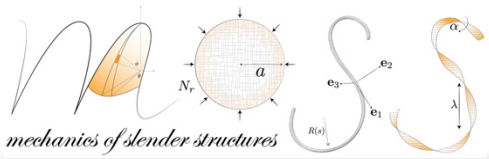
- Stability, wavelength, flexible electronics, mechanical metamaterials, adhesion.

### **Stress Focusing – Folding & Creasing:**

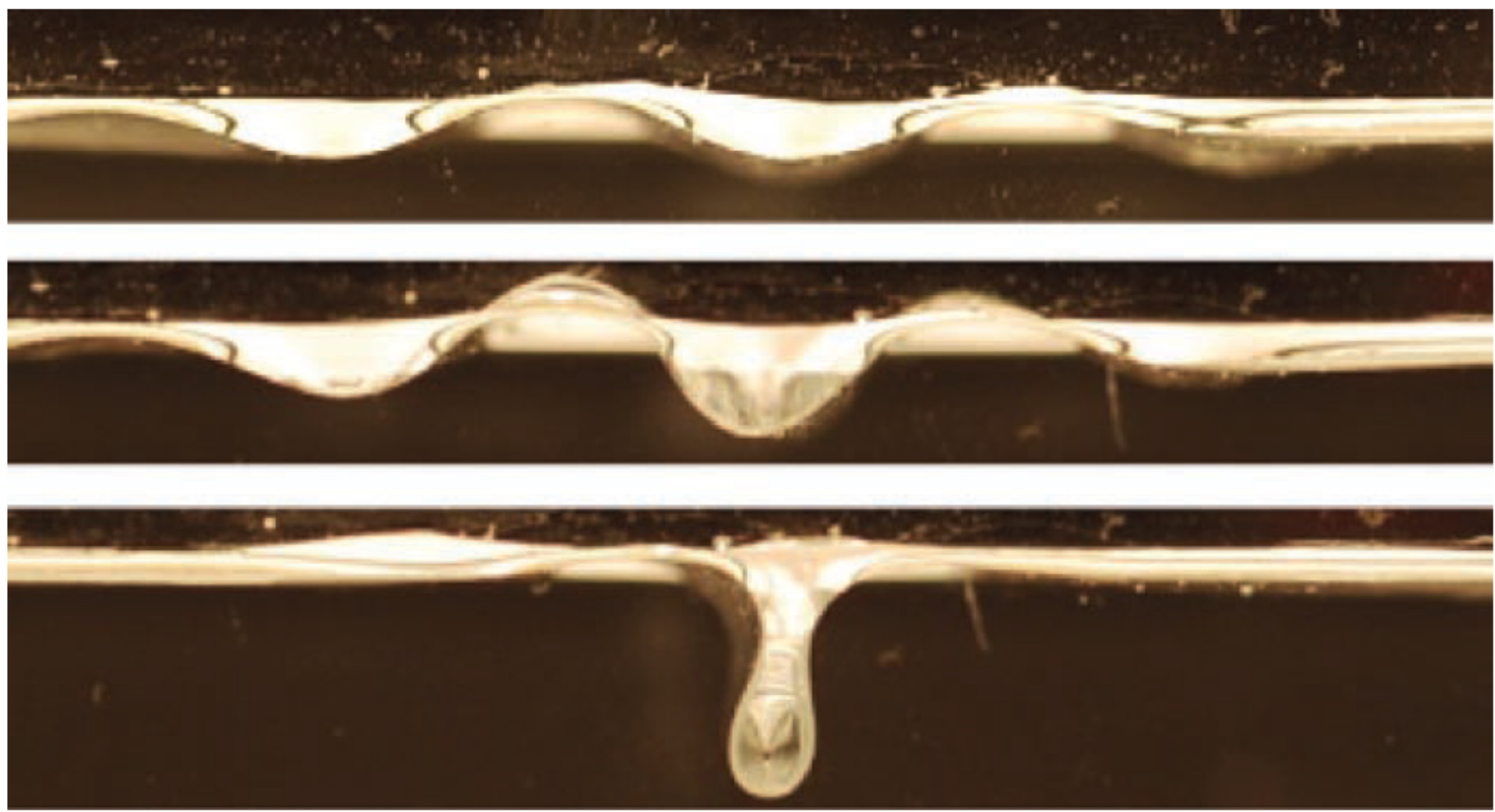
- Wrinkle-to-fold, origami.

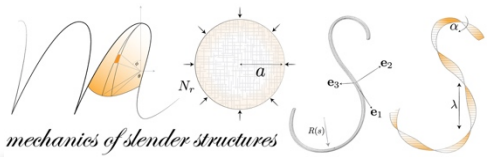
### **Snapping:**

- Snapping surfaces.



# Folding





# Folding

## Wrinkling to Fold

Bending

$$\mathcal{U}_b \approx \frac{B}{2} \int_0^L dl \partial_x^2 w \sim BL \left( \frac{A}{\lambda^2} \right)^2$$

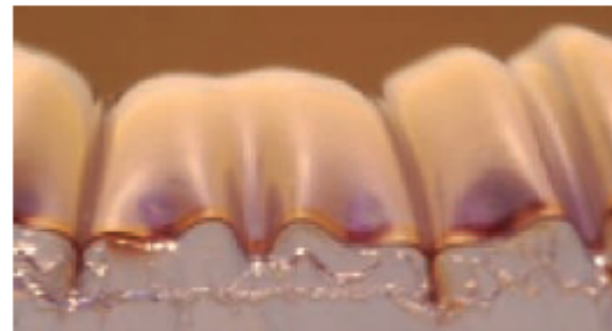
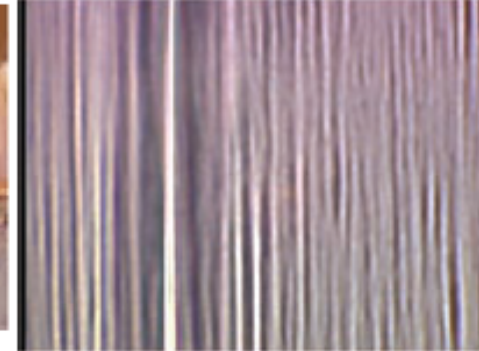
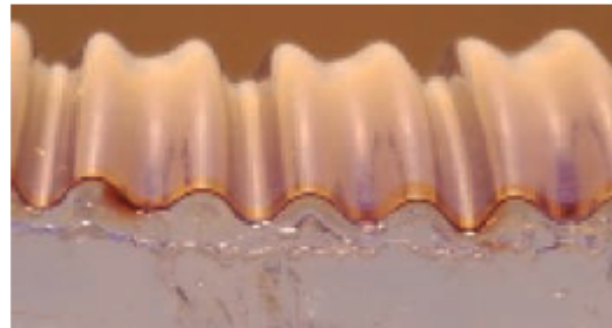
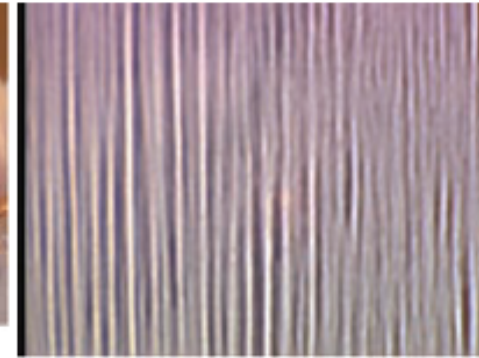
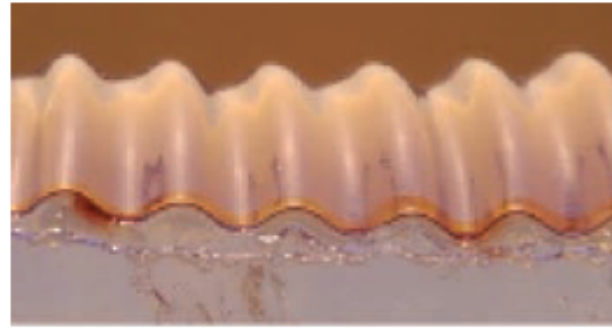
Resistance of substrate

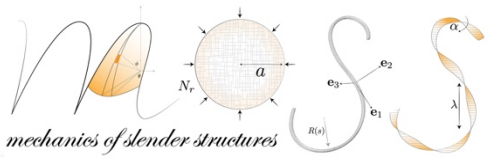
$$\mathcal{U}_K \approx \frac{K}{2} \int_0^L dl \partial_x^2 w \sim KLA^2$$

Inextensibility

$$\Delta \approx \frac{K}{2} \int_0^L dl \partial_x^2 w \sim L \left( \frac{A}{\lambda} \right)^2$$

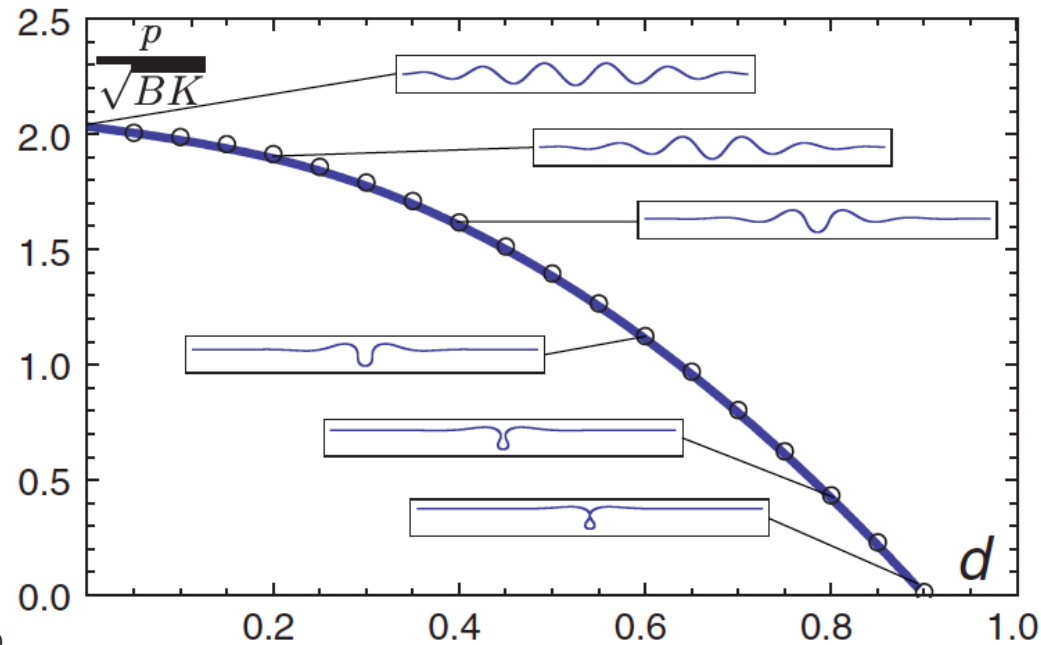
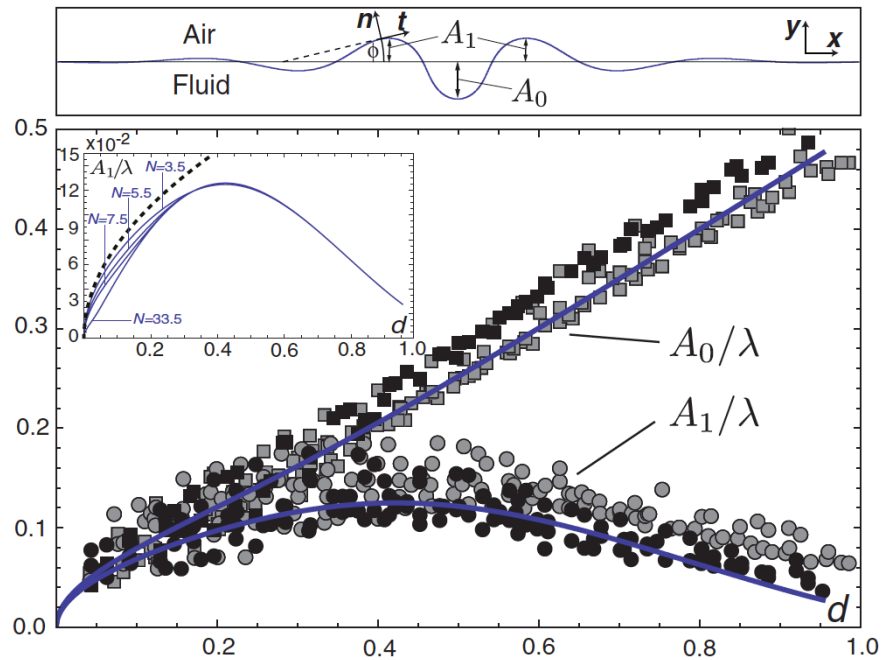
**Fold:**  $\frac{\Delta}{\lambda} \approx 0.3$



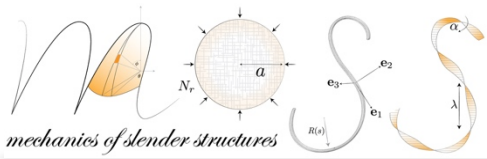


# Folding

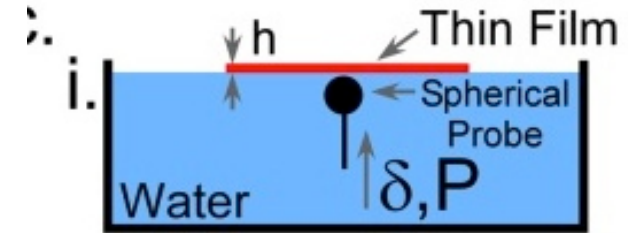
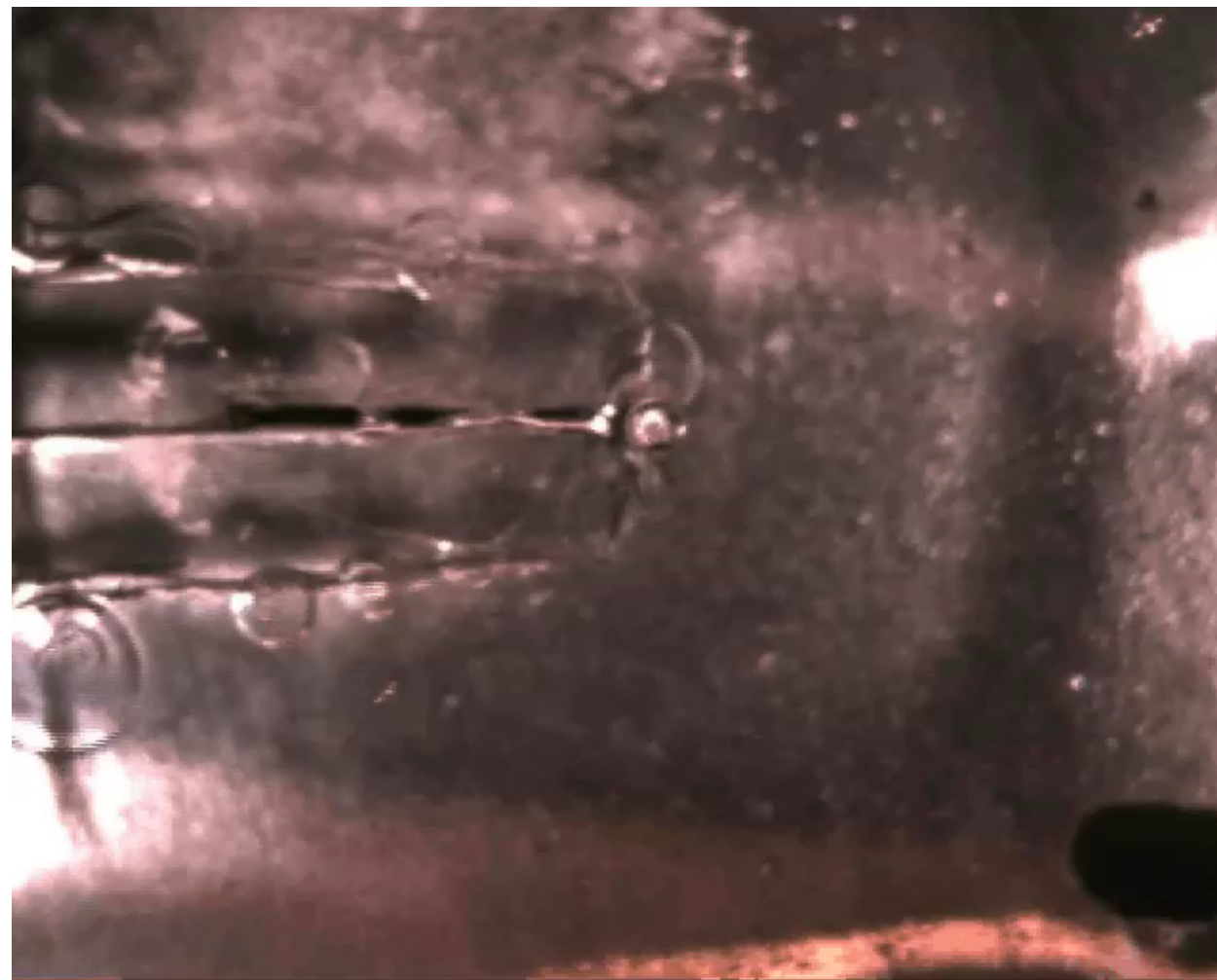
## Wrinkling to Fold







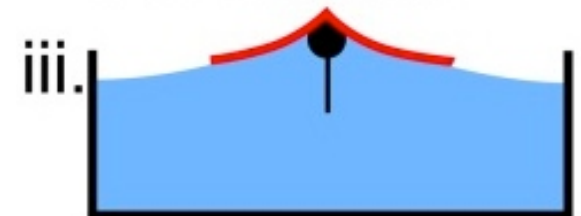
# Wrinkling & Folding



Wrinkle Formation



Strain Localization





# Wrinkling & Folding

## Wrinkling – Point Load

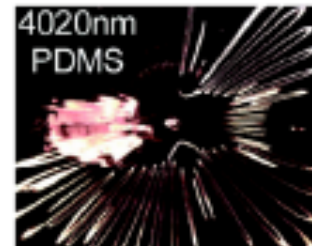
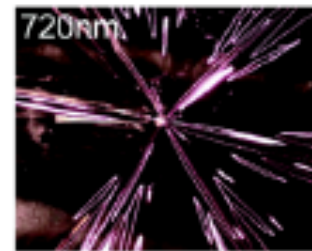
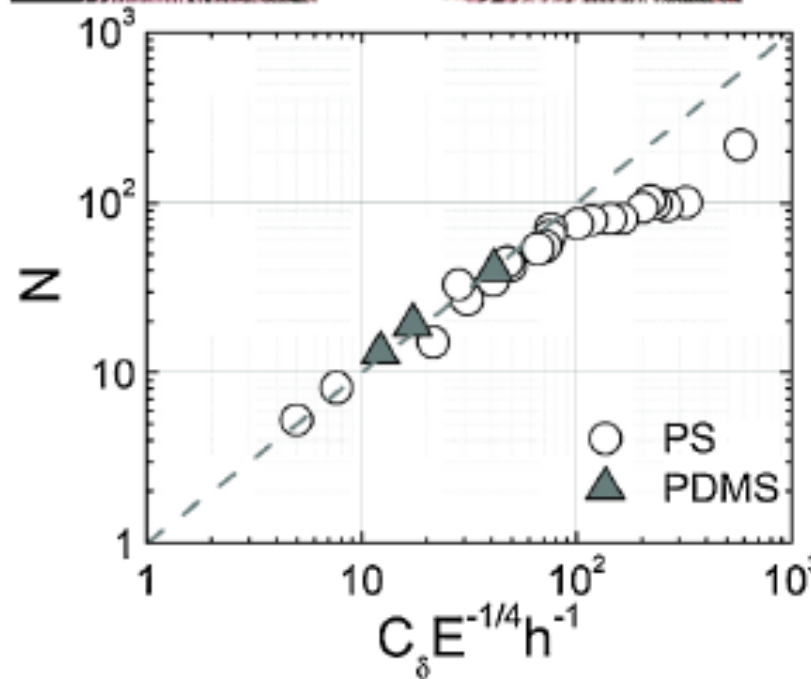
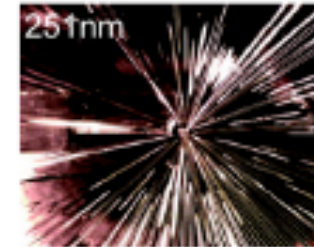
- Stress distribution:

$$\sigma_{rr}^w \sim \rho g \delta \left( \frac{L}{h} \right)^2$$

- Resistance of fluid substrate proportional to  $\rho g$

- Number of radial wrinkles:**

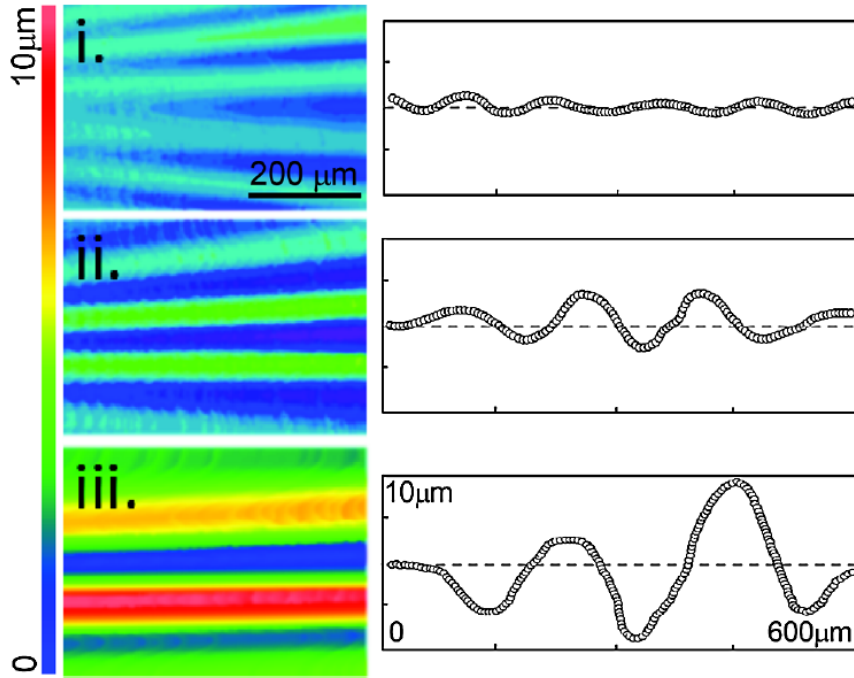
$$N \sim \left( \frac{\sigma_{rr} a^2}{\bar{E} h^2} \right)^{1/4} \sim \frac{1}{h} \left( \frac{\rho g \delta L^2 a^2}{\bar{E}} \right)^{1/4}$$



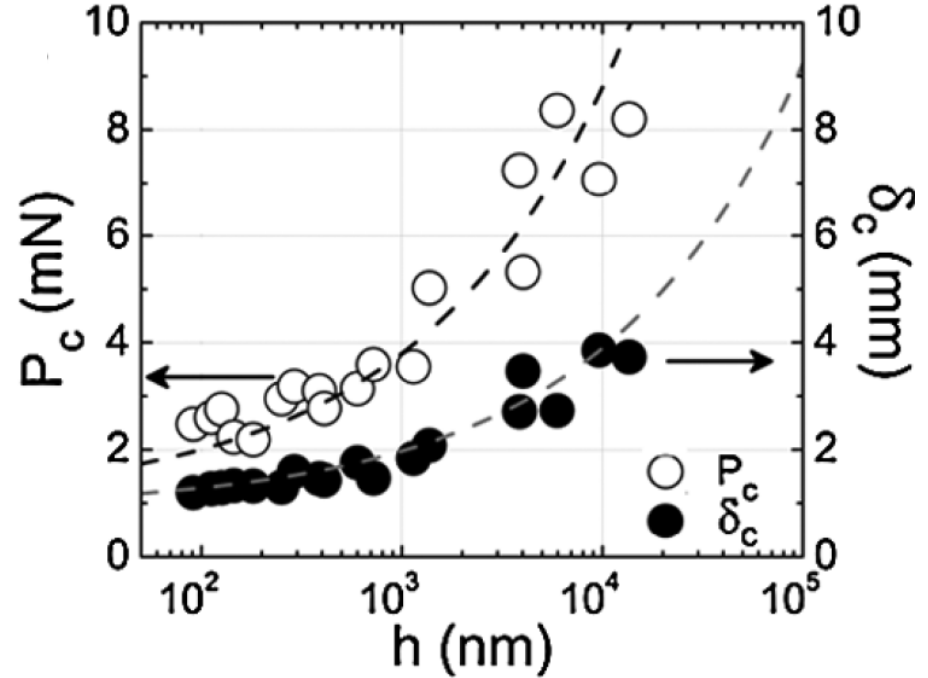


# Wrinkling & Folding

## Wrinkling – Point Load

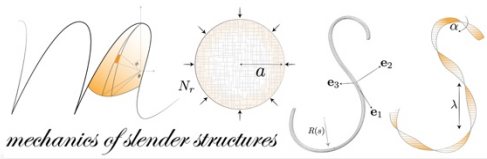


$$\delta_c \sim P_c \sim h^{1/2}$$

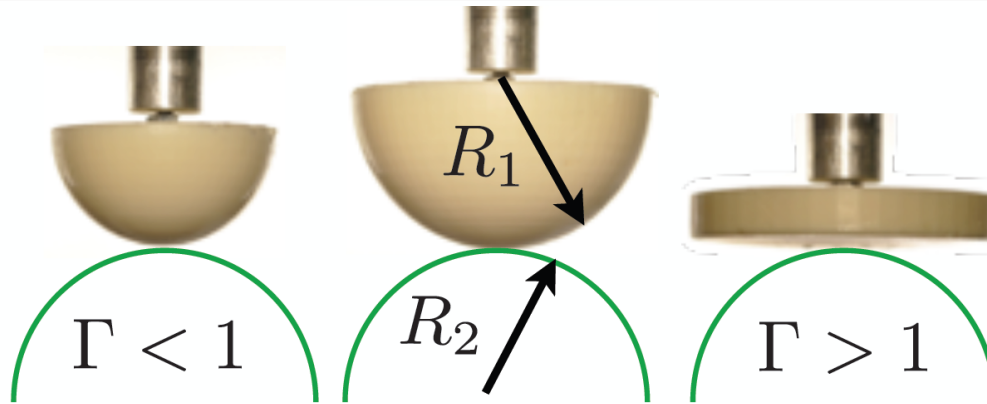


## Folding at large deformation – **asymptotic isometry**

- Thin elastic film is forced into a curved, nondevelopable shape

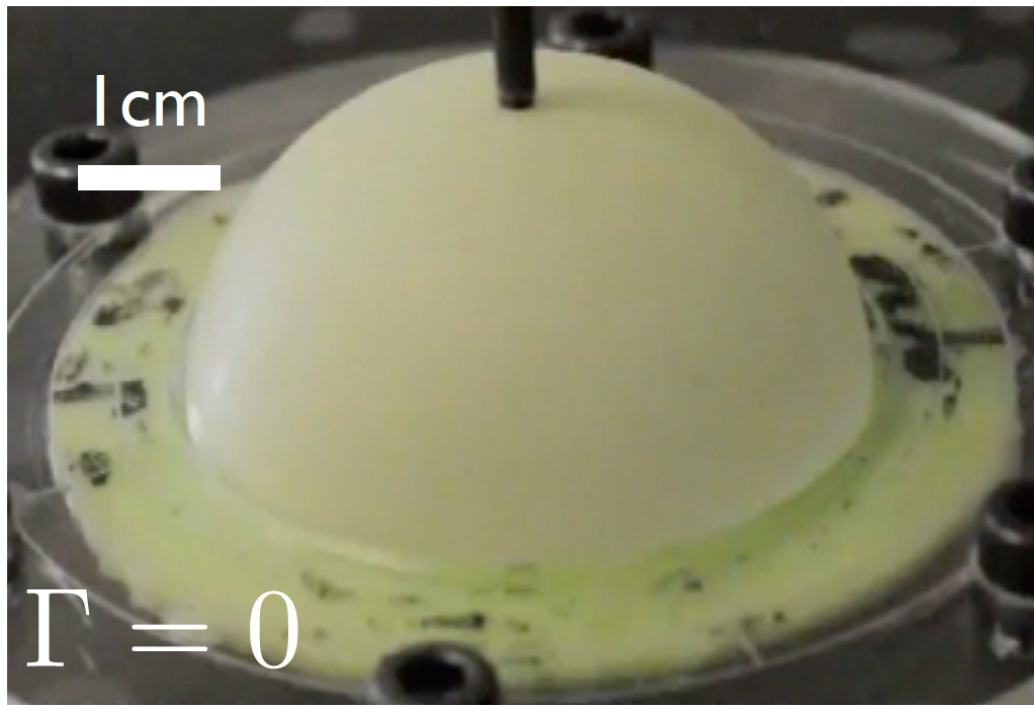


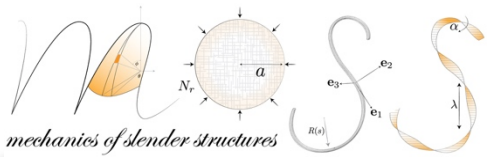
# Localization



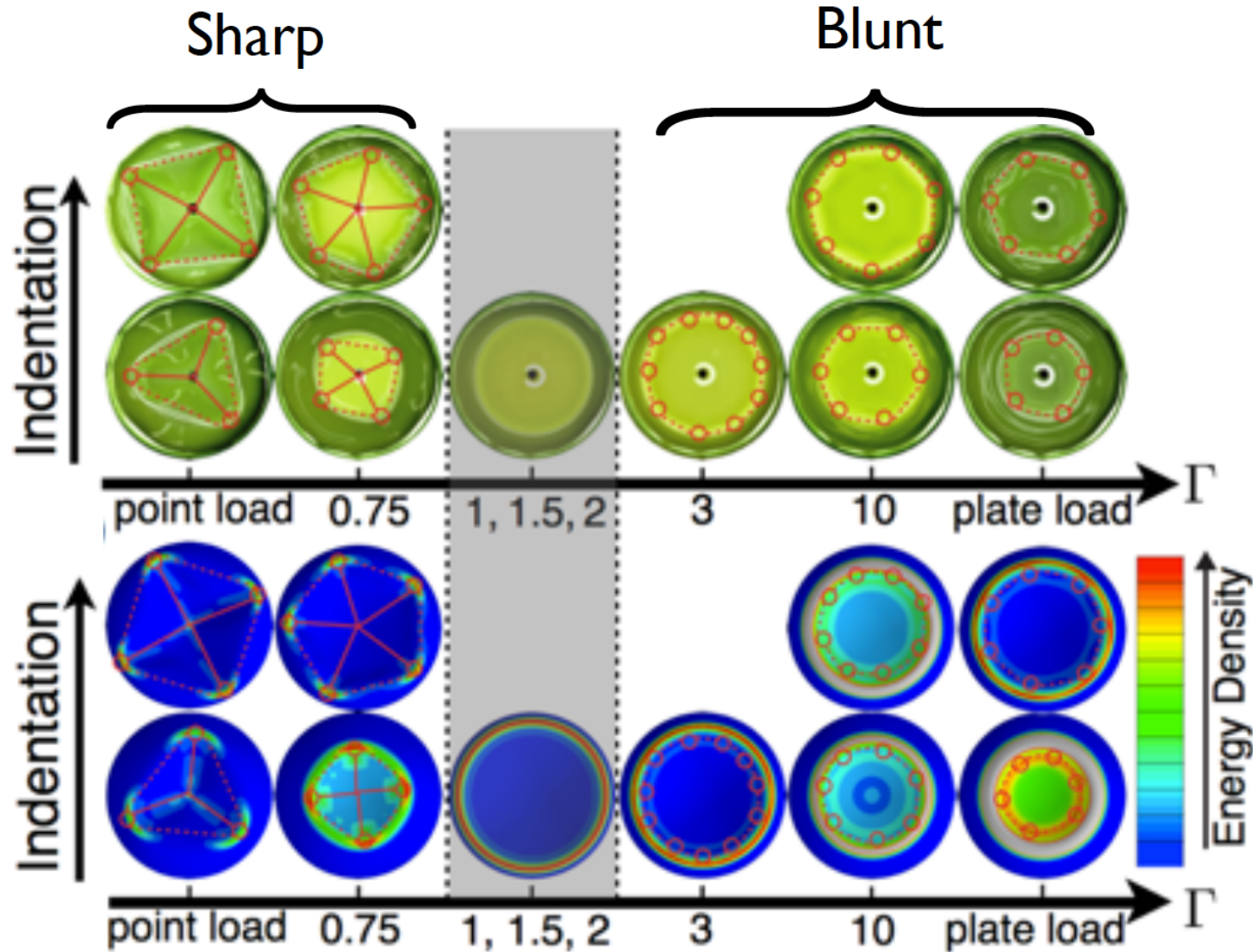
Indenter-shell  
aspect ratio:

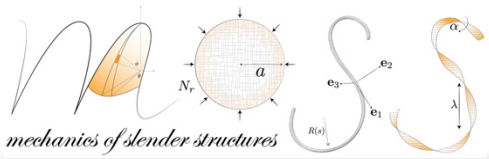
$$\Gamma = \frac{R_1}{R_2}$$



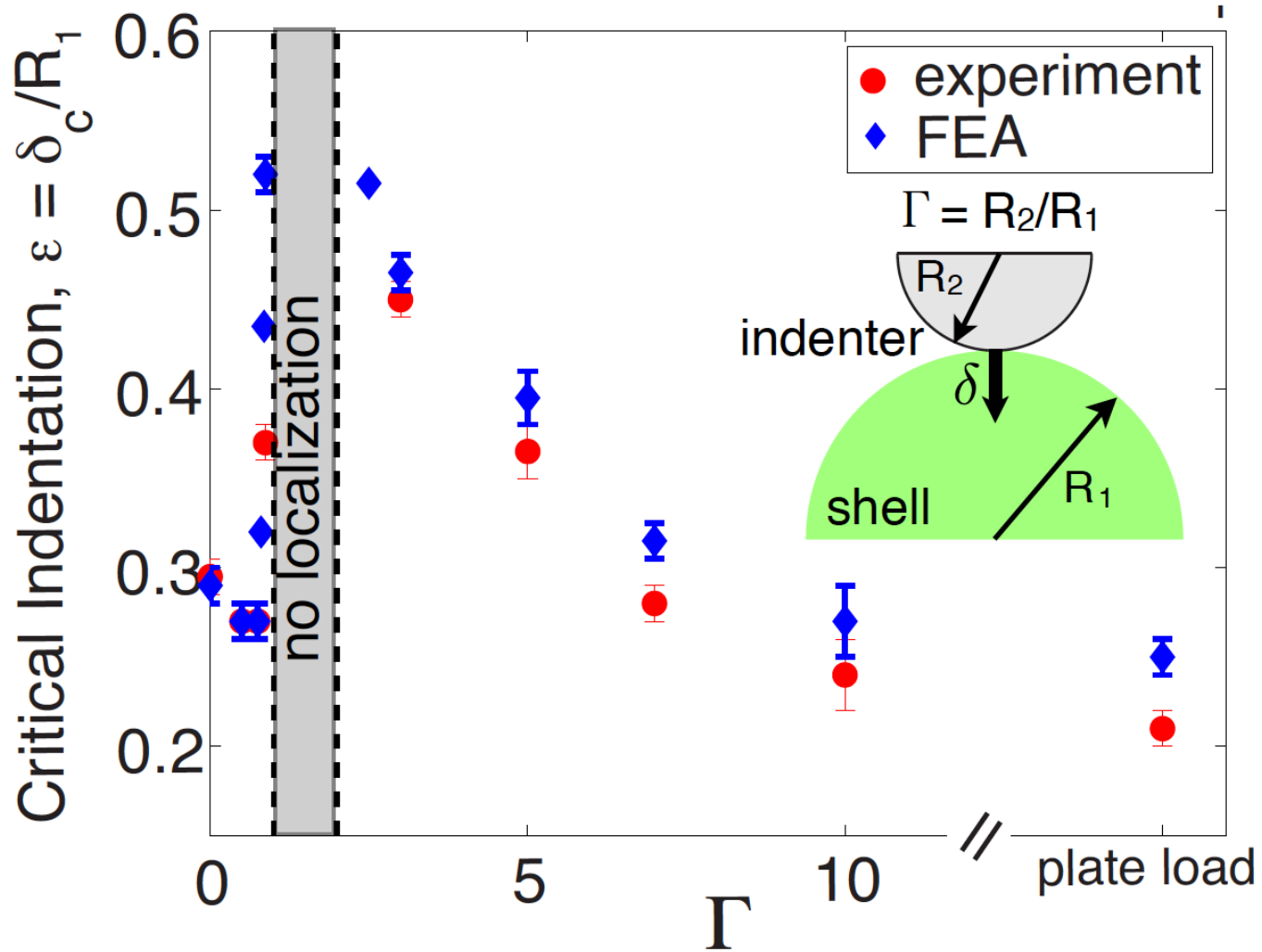


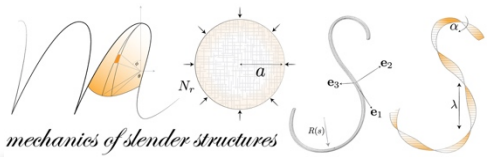
# Localization





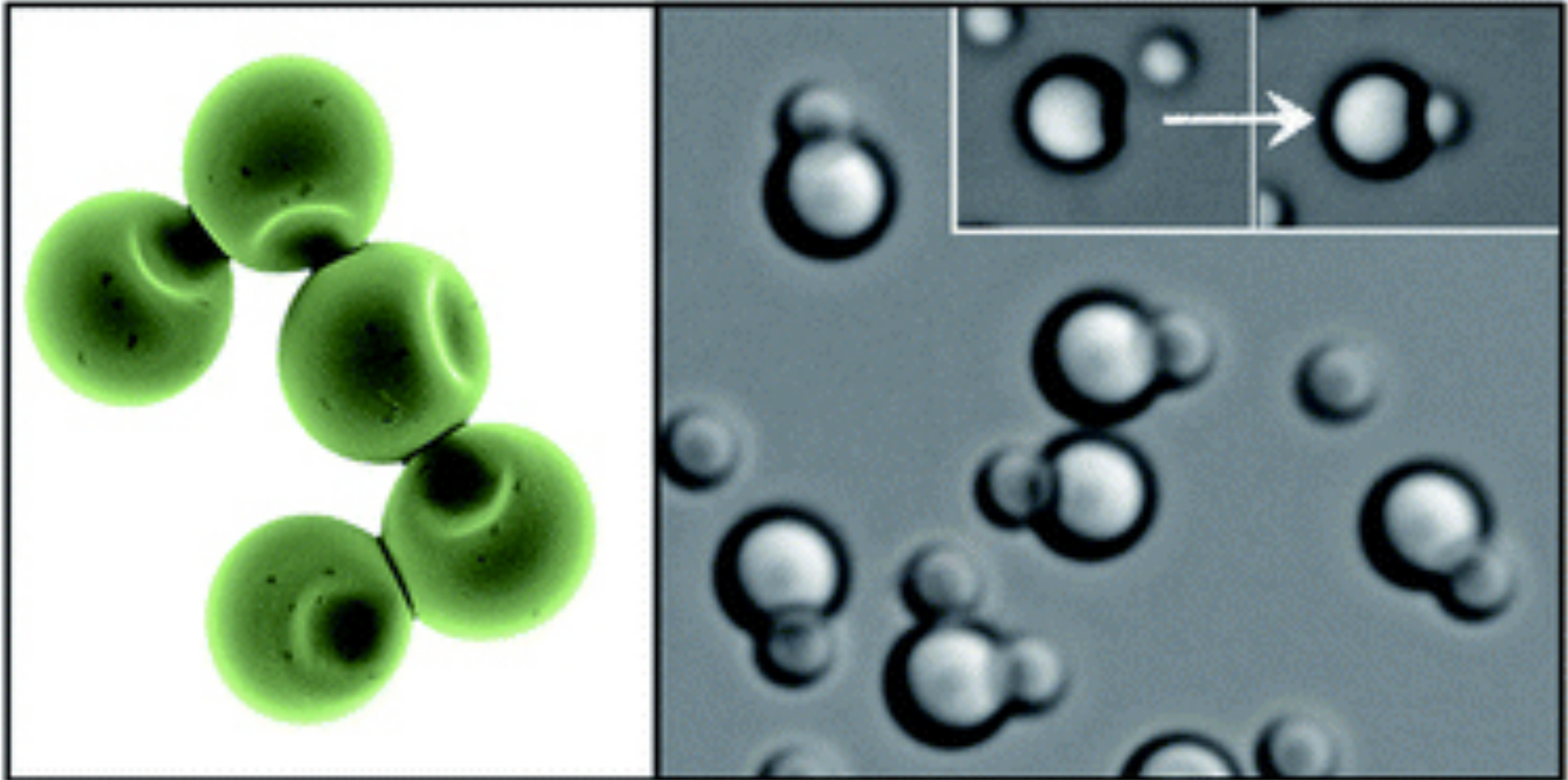
# Localization

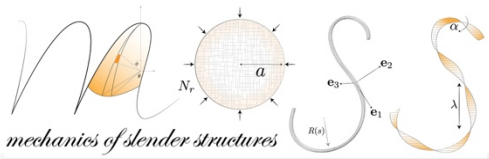




# Localization

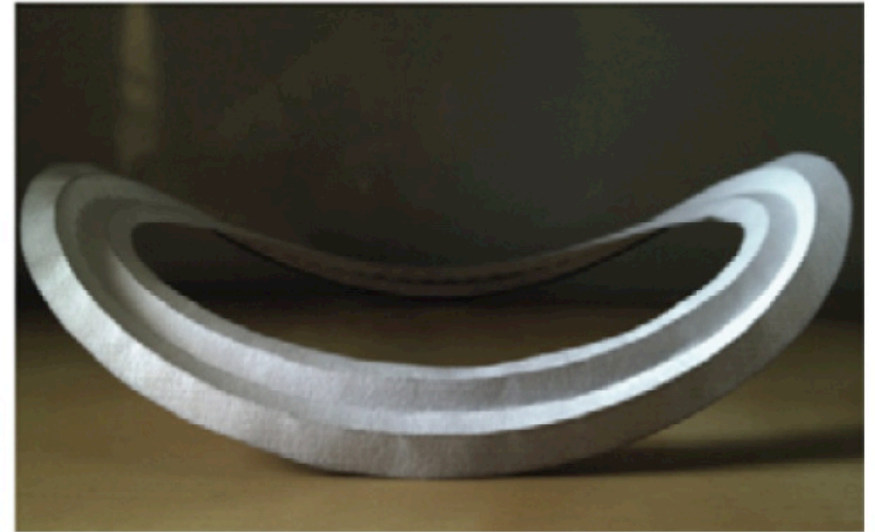
Lock & Key Colloids – Polymerization induced buckling.



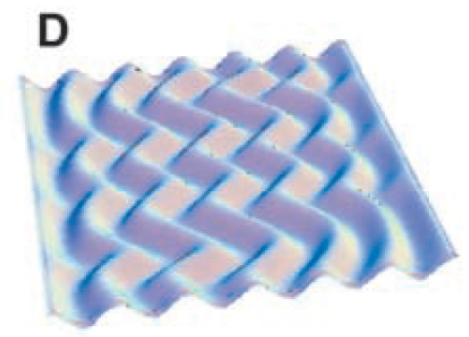
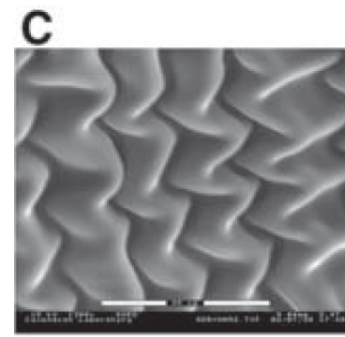
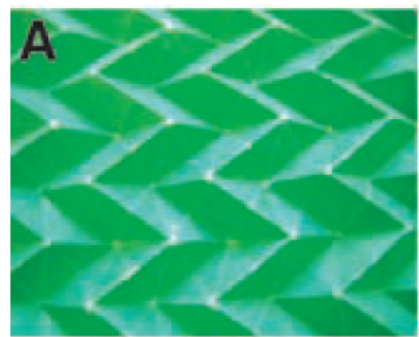


# Folding

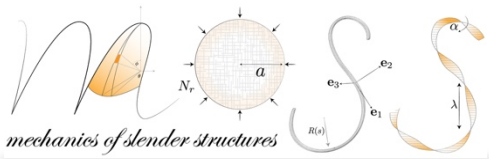
## Curved Crease



## Origami – Miura Ori

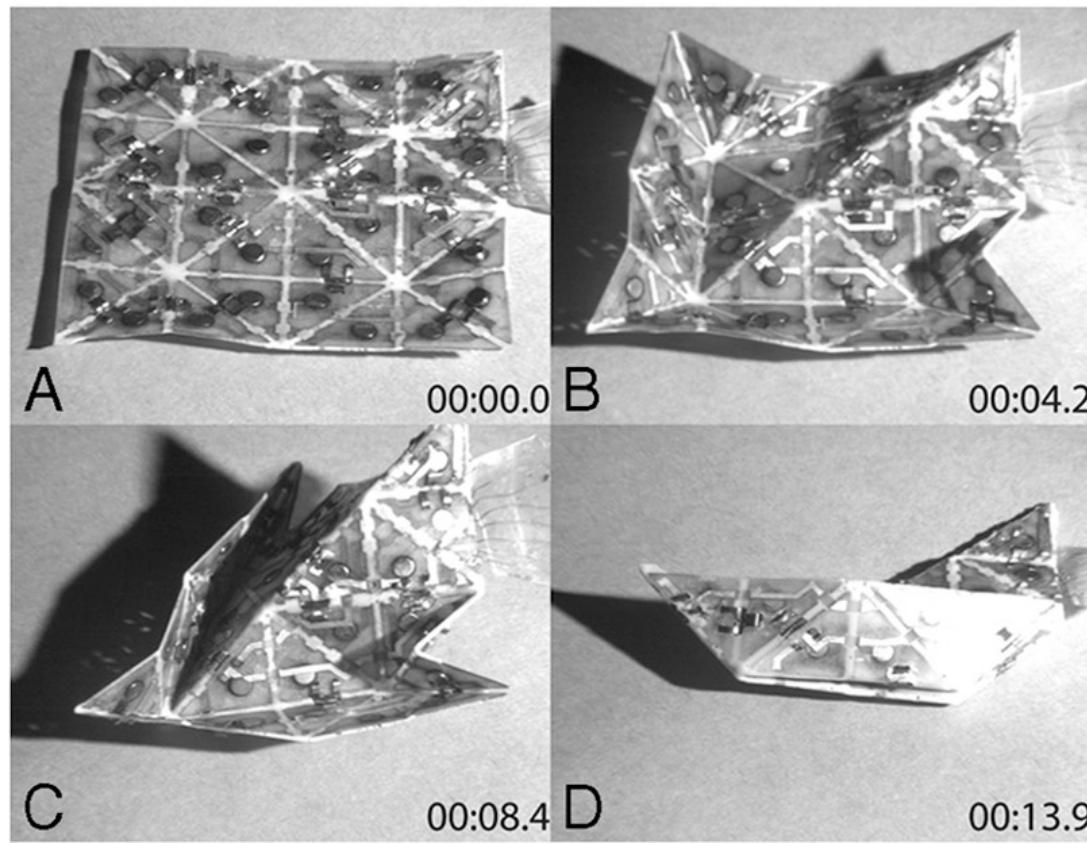
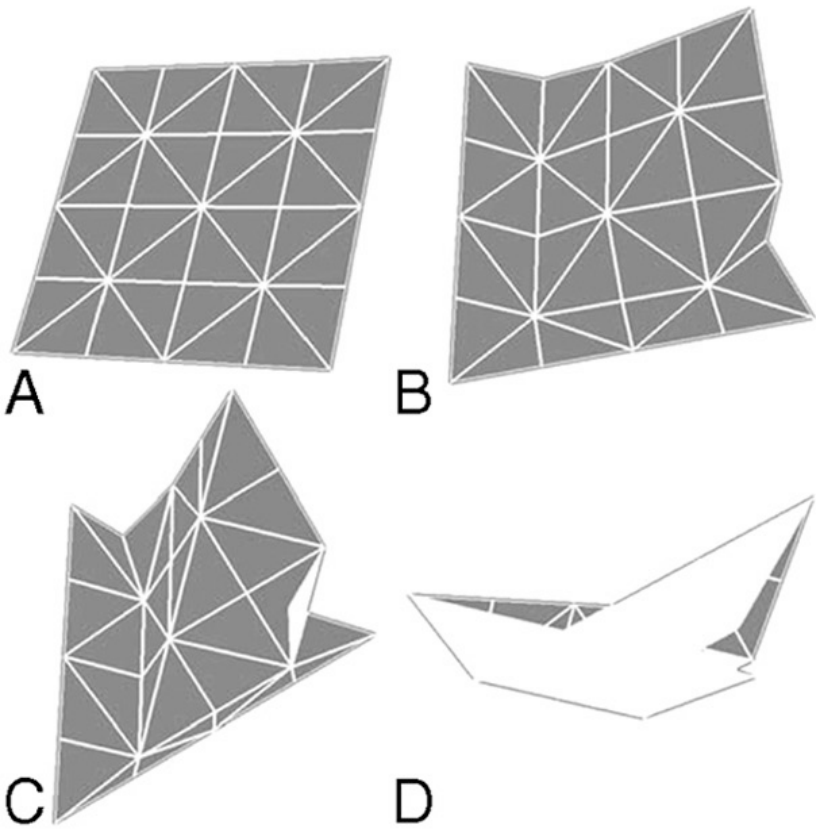






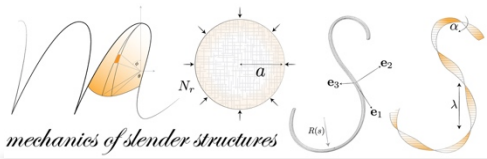
# Folding

## Nitinol Shape Memory Alloy



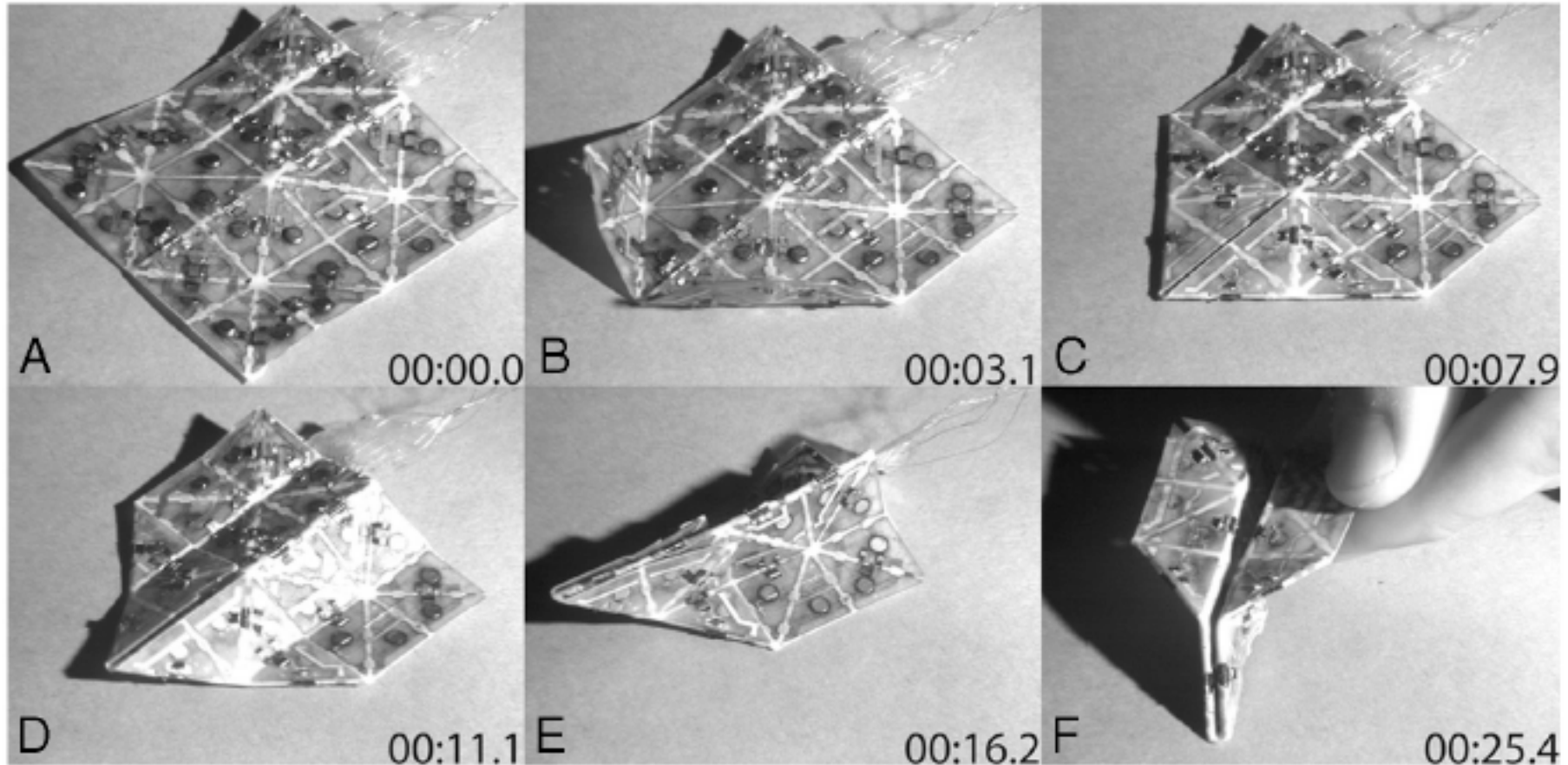
Hawkes et al. "Programmable matter by folding," PNAS, 107(28), 12441-12445, (2010).





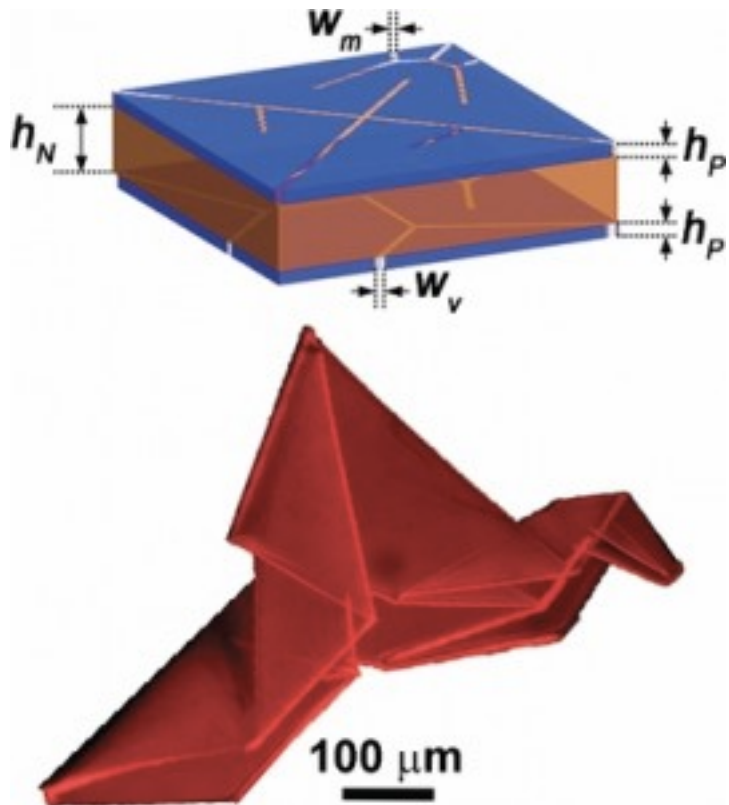
# Folding

## Nitinol Shape Memory Alloy

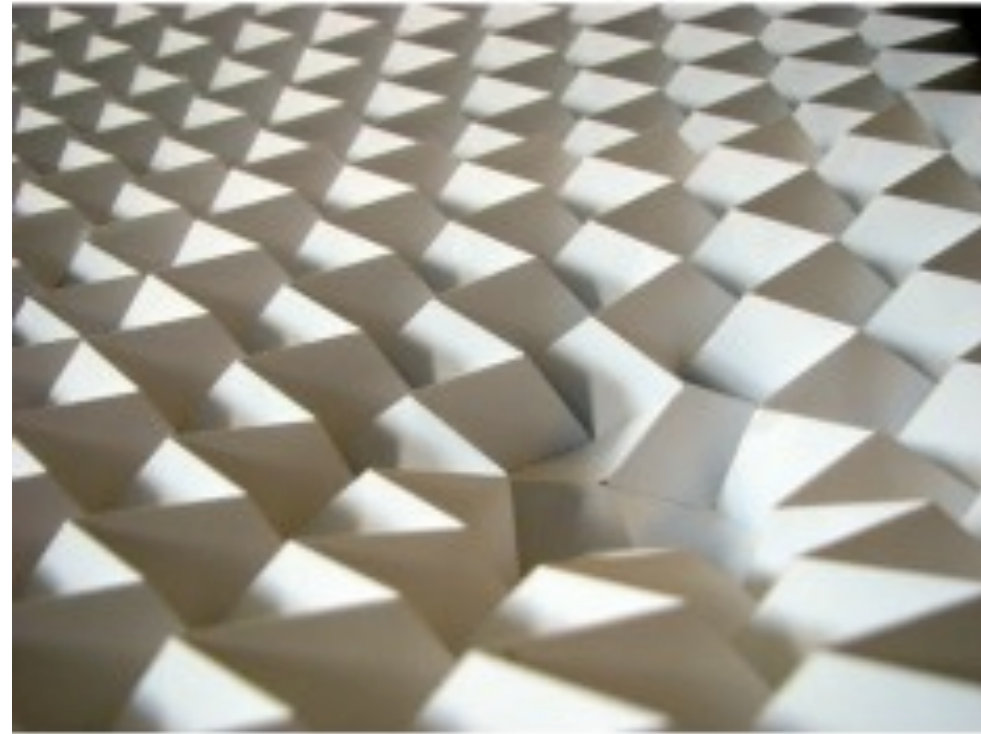


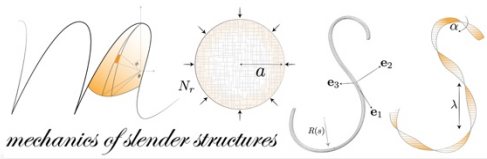
Hawkes et al. "Programmable matter by folding," PNAS, 107(28), 12441-12445, (2010).

## Self-folding Origami Bird

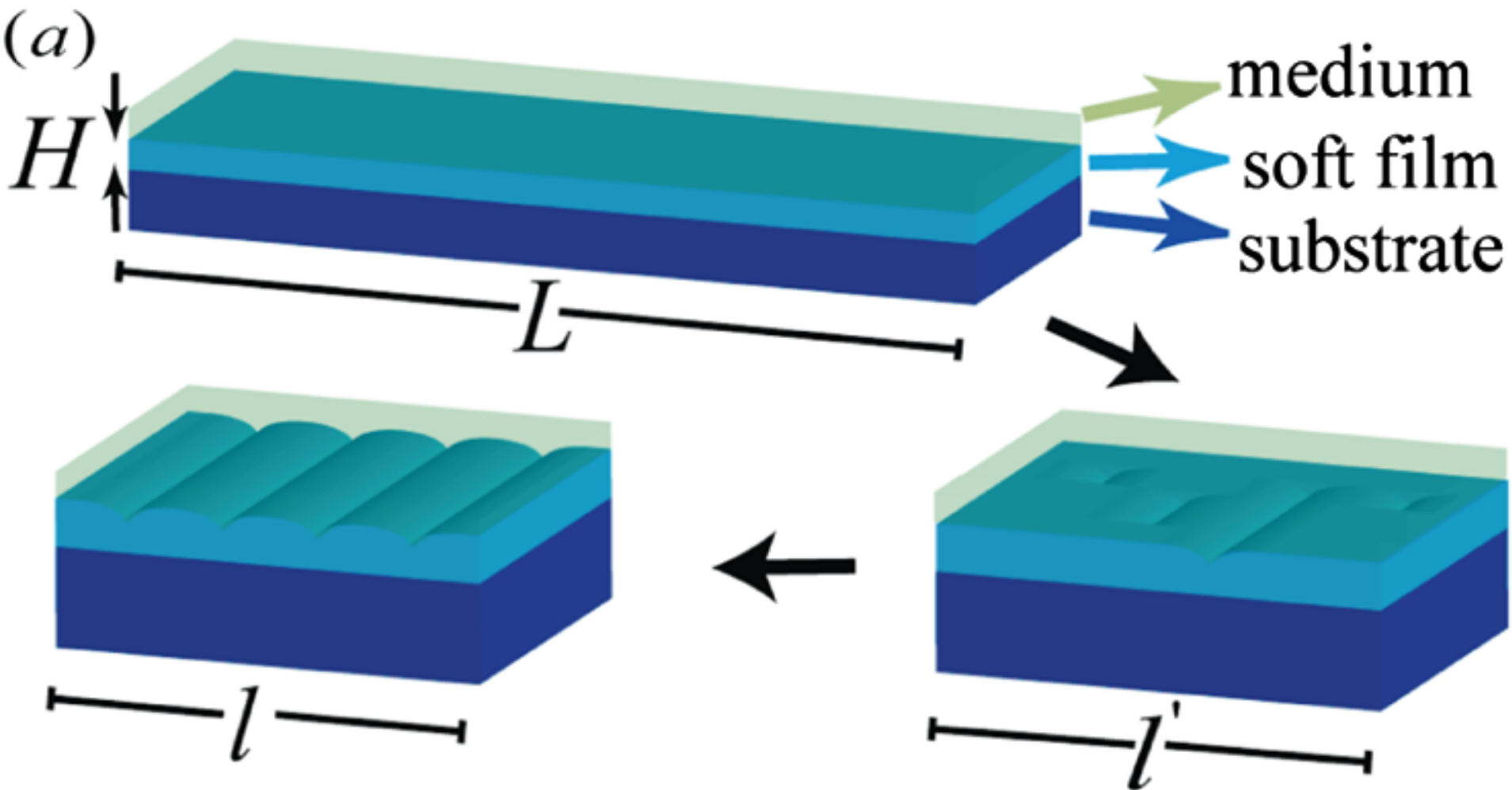


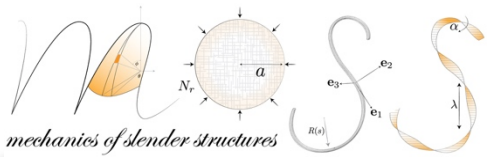
## Origami & Mechanical Metamaterials





# Creasing

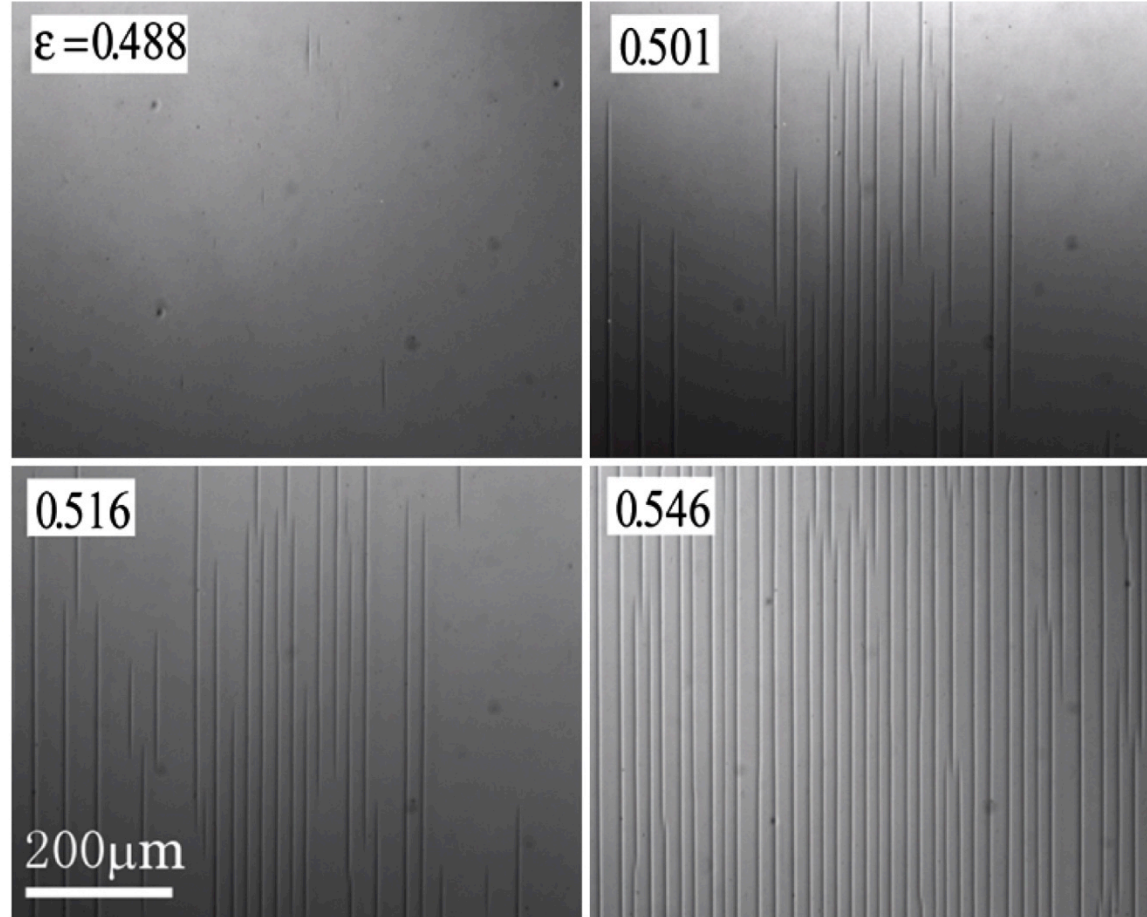




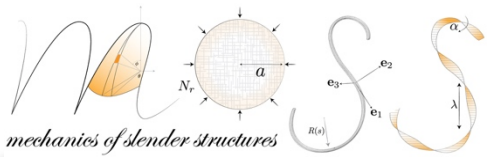
# Creasing

## Crease vs. Wrinkle

- Fundamentally different type of instability.
- Both are **bifurcations** from a state of homogenous compression.
- Wrinkles bifurcate by a field of **strain small in amplitude**, and **nonlocal in space**.
- Creases bifurcate by a field of **strain large in amplitude** and **localized in space**.
- Wrinkles are predicted theoretically, they are **preceded by creases**.



Wrinkles form by a **linear perturbation**, creases form by **nucleation and growth**.



# Creasing

## Crease vs. Wrinkle

- Nucleation and growth of creases are analogous to classical nucleation theory for a **thermodynamic phase transition**.
- Forming a crease reduces the elastic energy, yet increases the surface area – barrier to nucleation.
- Absence of surface energy, crease forms at fixed strain:

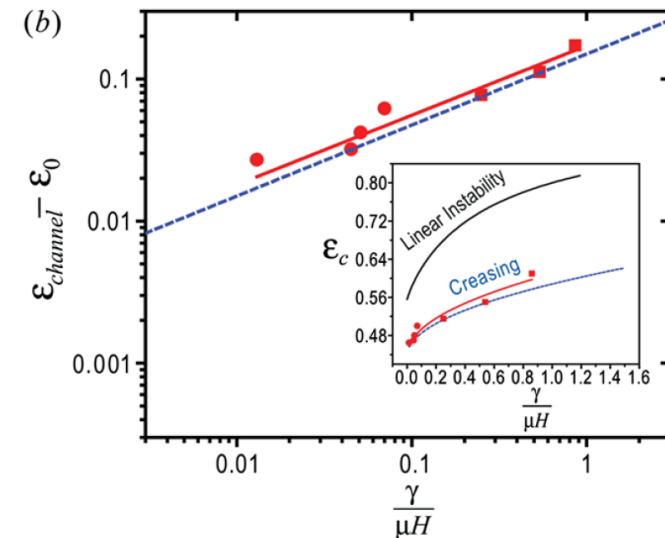
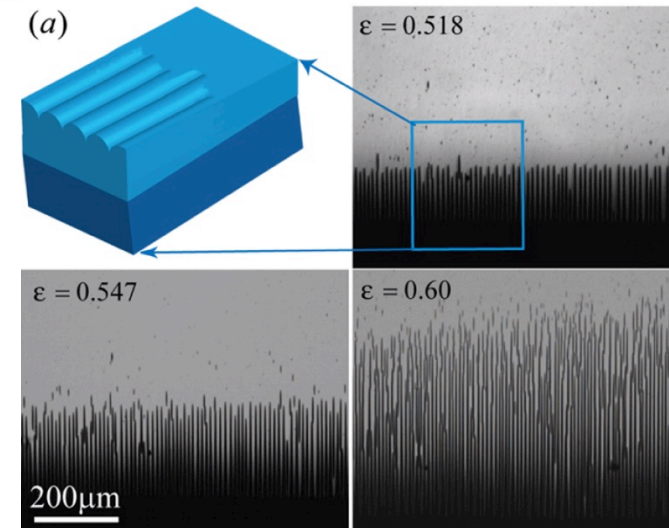
$$\epsilon_0 = 0.438$$

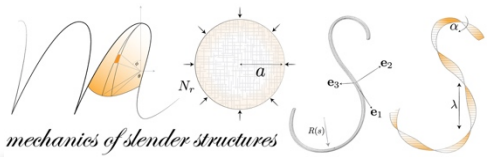
- Nucleation size:

$$a_{\text{nuc}} \approx \frac{1}{2} \frac{A}{B} \frac{\gamma}{\mu(\epsilon - \epsilon_0)}$$

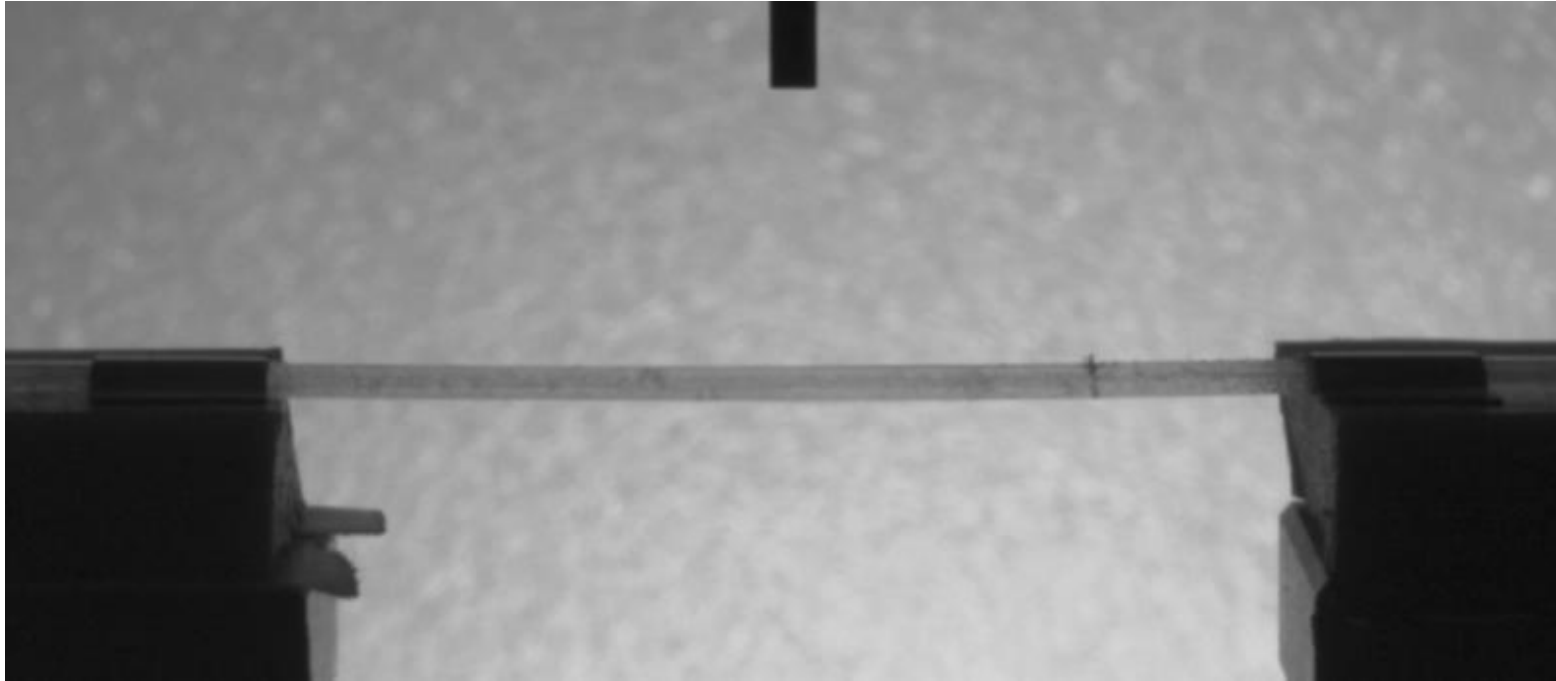
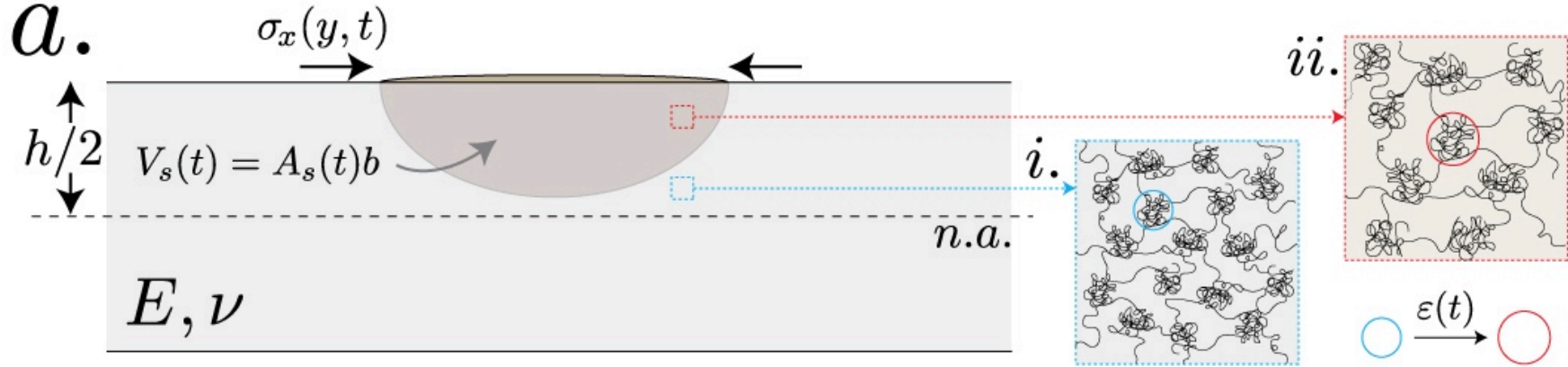
- Overstrain:

$$\epsilon - \epsilon_0 \sim \underbrace{\left( \frac{\gamma}{\mu H} \right)^{1/2}}_{\text{elastocapillary}}$$

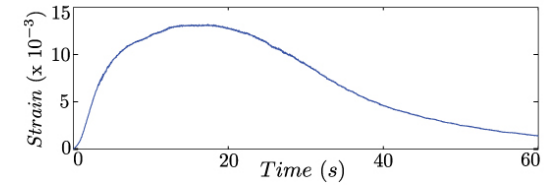
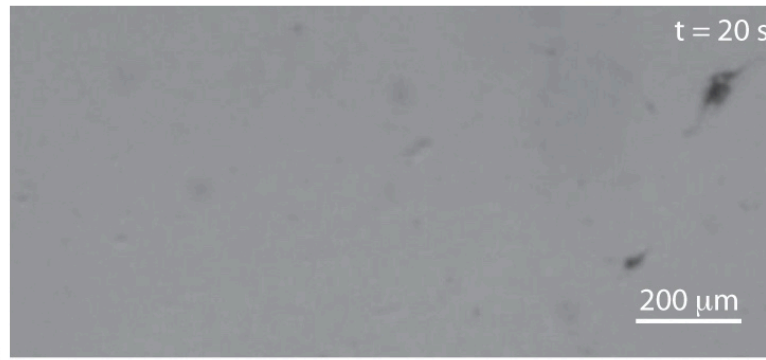
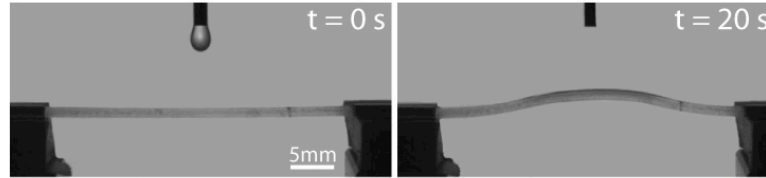
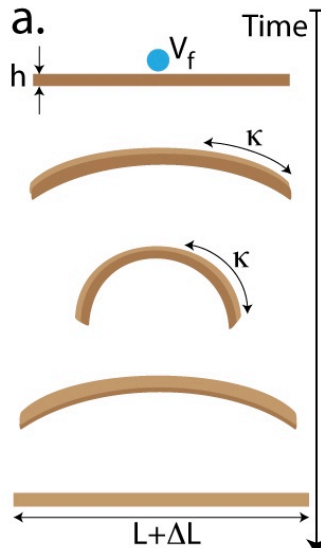




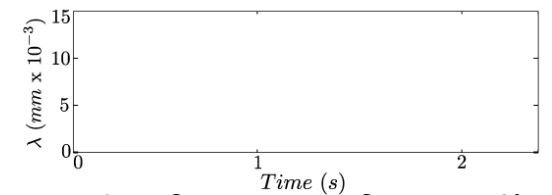
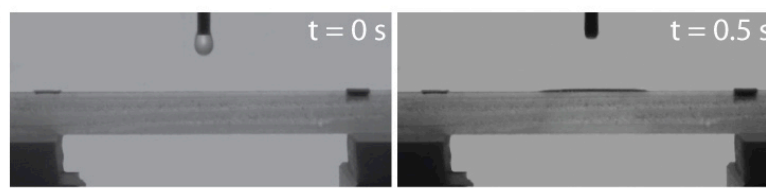
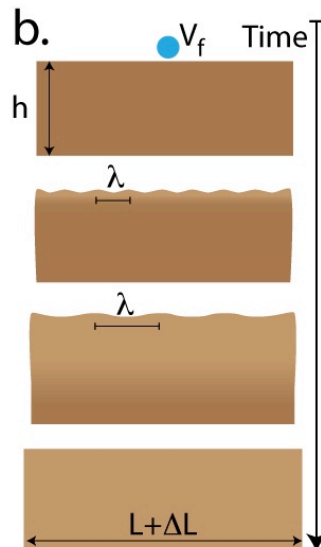
# Creasing



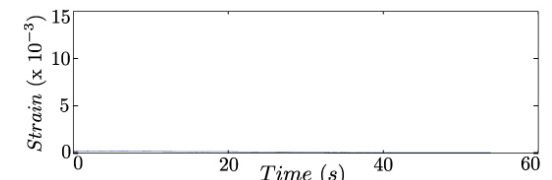
# Creasing



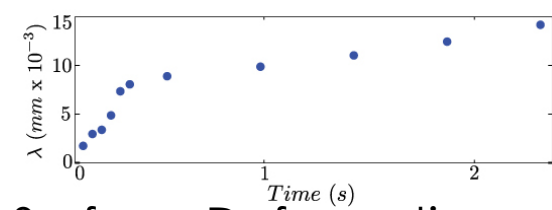
Structural Deformation



No Surface Deformation

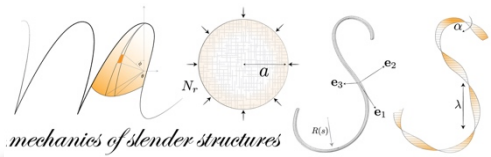


No Structural Deformation

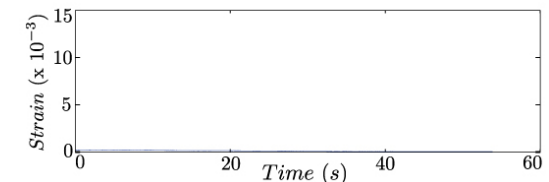
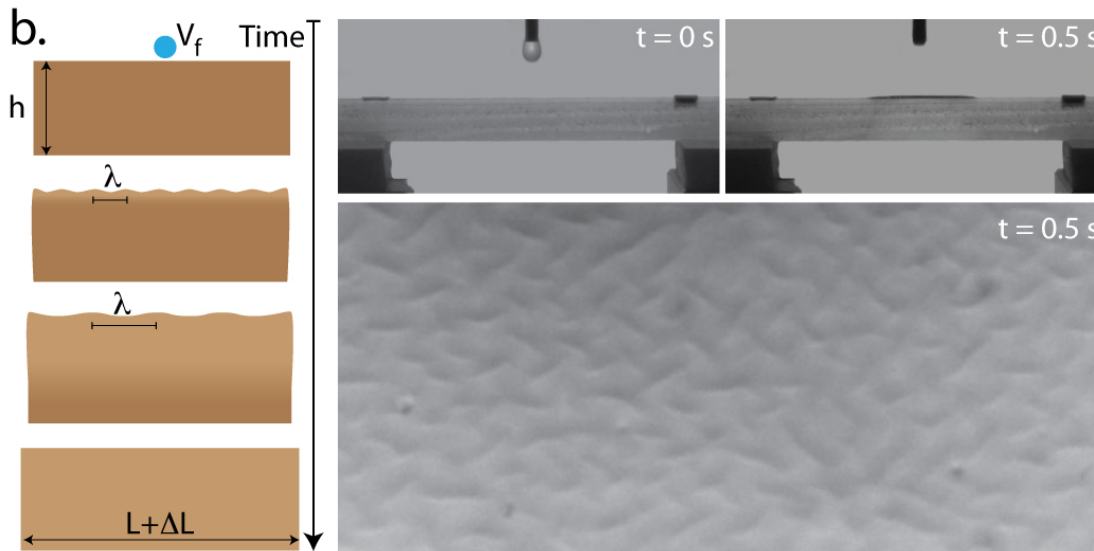
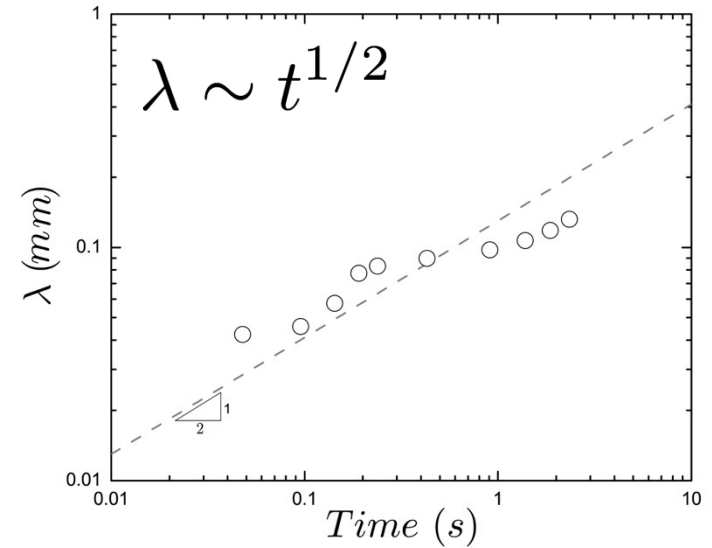
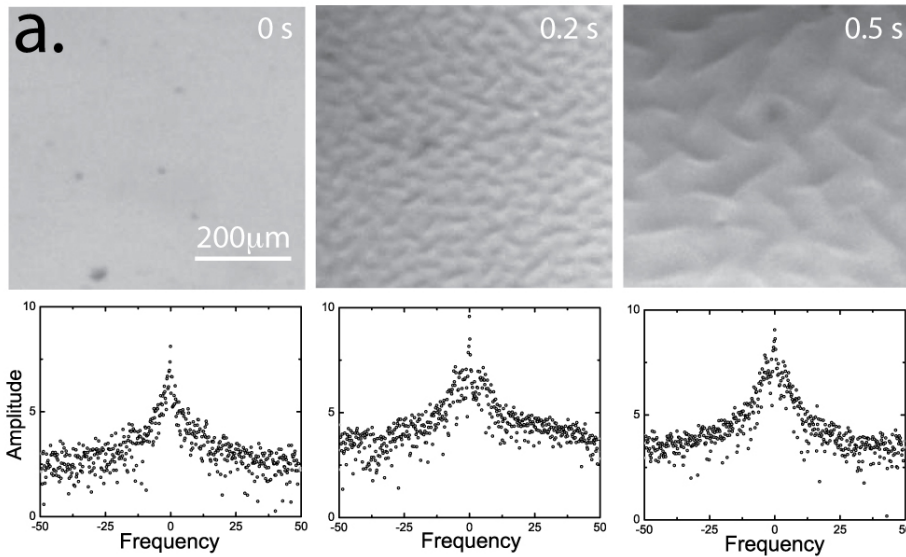


Surface Deformation

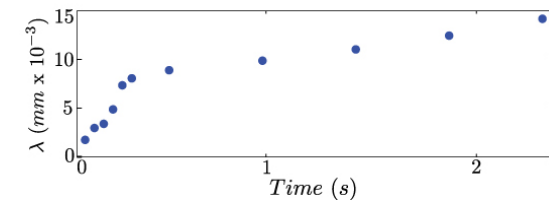




# Creasing



No Structural Deformation



Surface Deformation



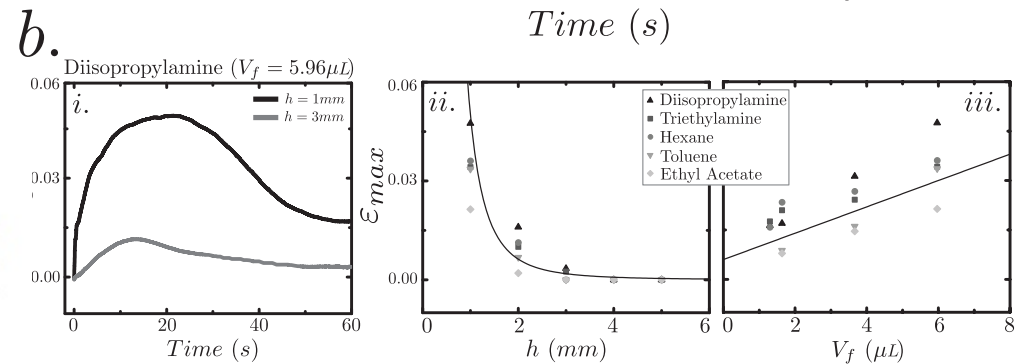
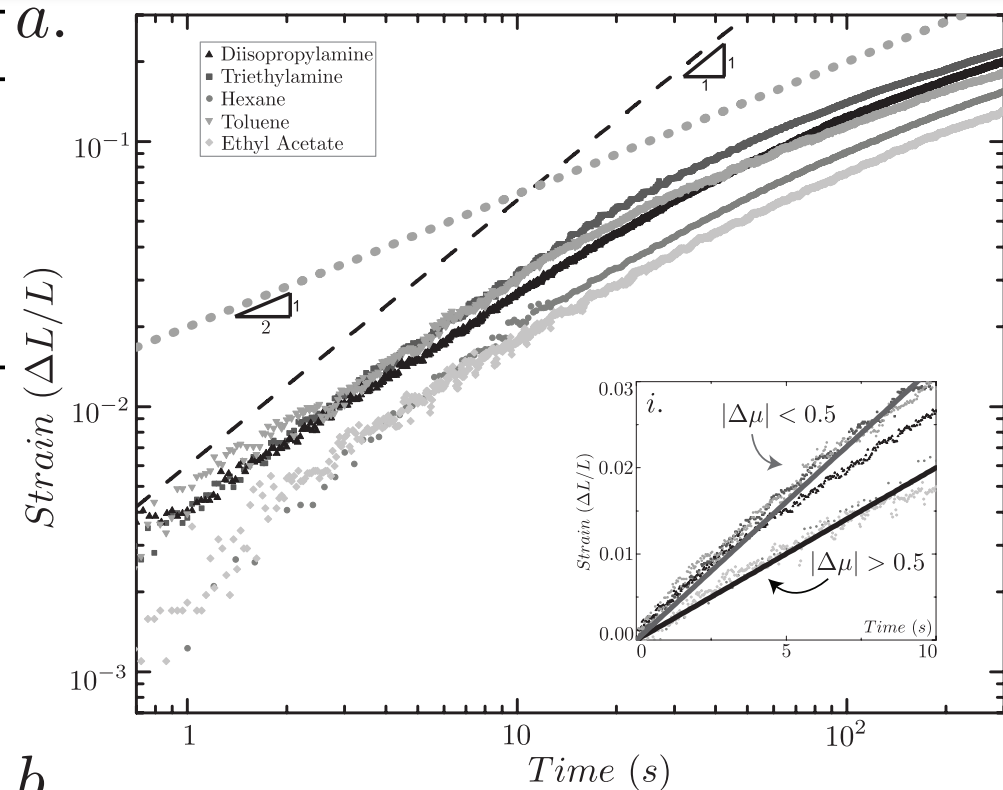
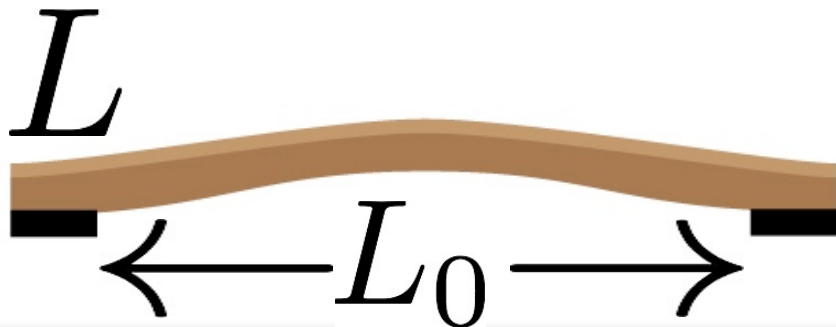
# Creasing

Material	$\delta_s$ ( $\text{cal}^{1/2}\text{cm}^{-3/2}$ )	$\mu$ (D)	$\epsilon_{eq}$
PDMS	7.3	0.6-0.9	–
Diisopropylamine	7.3	1.2	<b>1.13</b>
Triethylamine	7.5	0.7	<b>0.58</b>
Hexanes	7.3	0.0	<b>0.35</b>
Toluene	8.9	0.4	<b>0.31</b>
Ethyl acetate	9.0	1.8	<b>0.18</b>

$\delta_s$  = Solubility parameter

$\mu$  = Solvent polarity

$\epsilon_{eq}$  = Strain at equilibrium



## Can the fluid bend the structure?

Bending

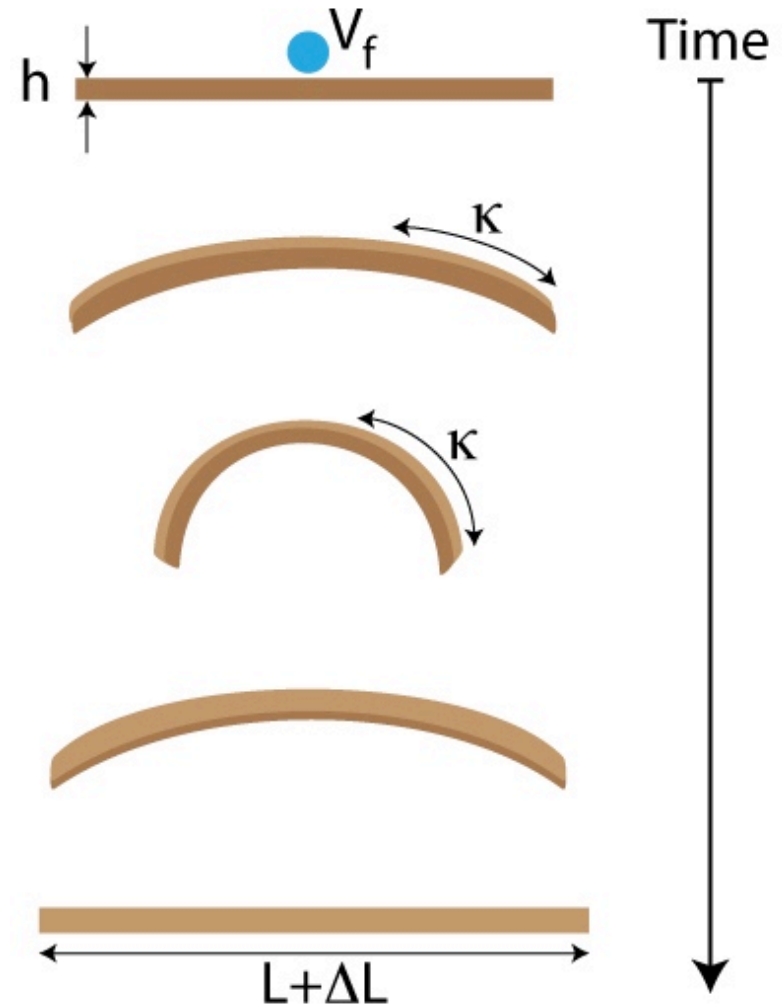
$$U_b = \frac{B}{2} \int_L \theta'(s)^2 ds \sim \bar{E} h^3$$

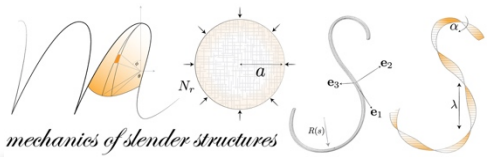
Swelling

$$U_s = \int_{V_f} \sigma \varepsilon_{eq} dV_f \sim E \varepsilon_{eq}^2 V_f$$

Length scale:

$$l_{es} \sim (\varepsilon_{eq}^2 V_f)^{1/3}$$

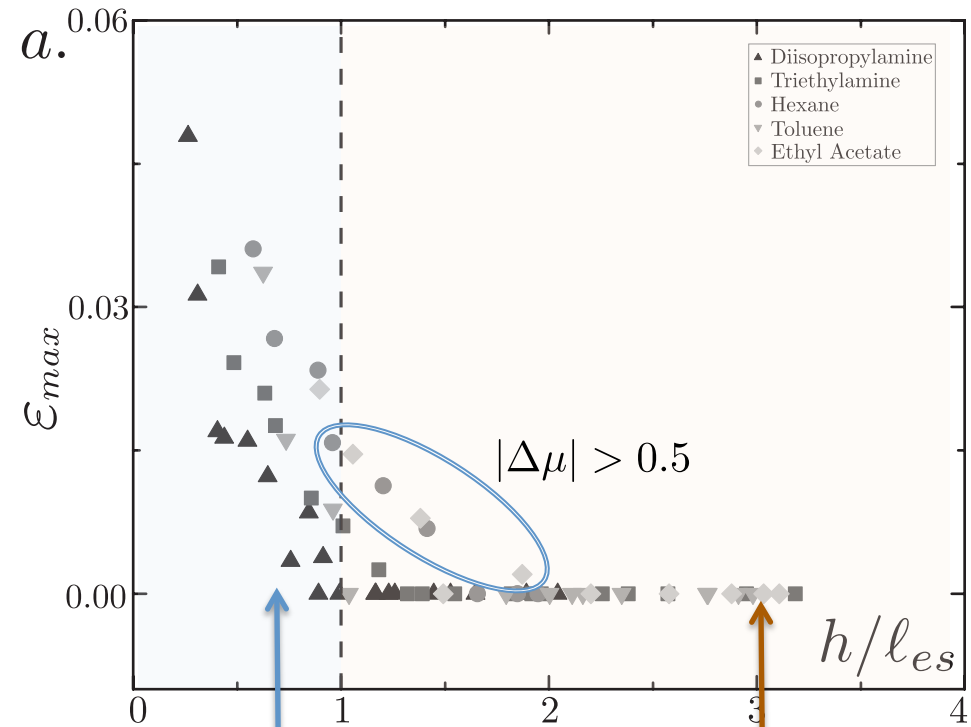
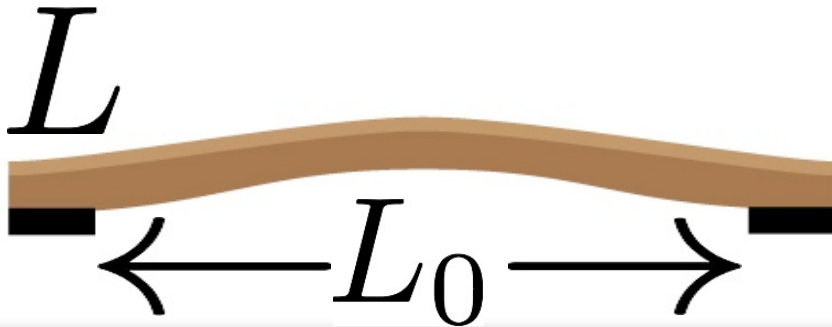




# Creasing

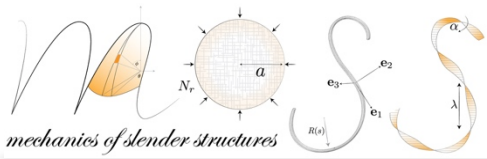
Material	$\delta_s$ ( $\text{cal}^{1/2}\text{cm}^{-3/2}$ )	$\mu$ (D)	$\epsilon_{eq}$
PDMS	7.3	0.6-0.9	-
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Hexanes	7.3	0.0	<b>0.35</b>
Toluene	8.9	0.4	<b>0.31</b>
Ethyl acetate	9.0	1.8	<b>0.18</b>

$$l_{es} \sim (\epsilon_{eq}^2 V_f)^{1/3}$$

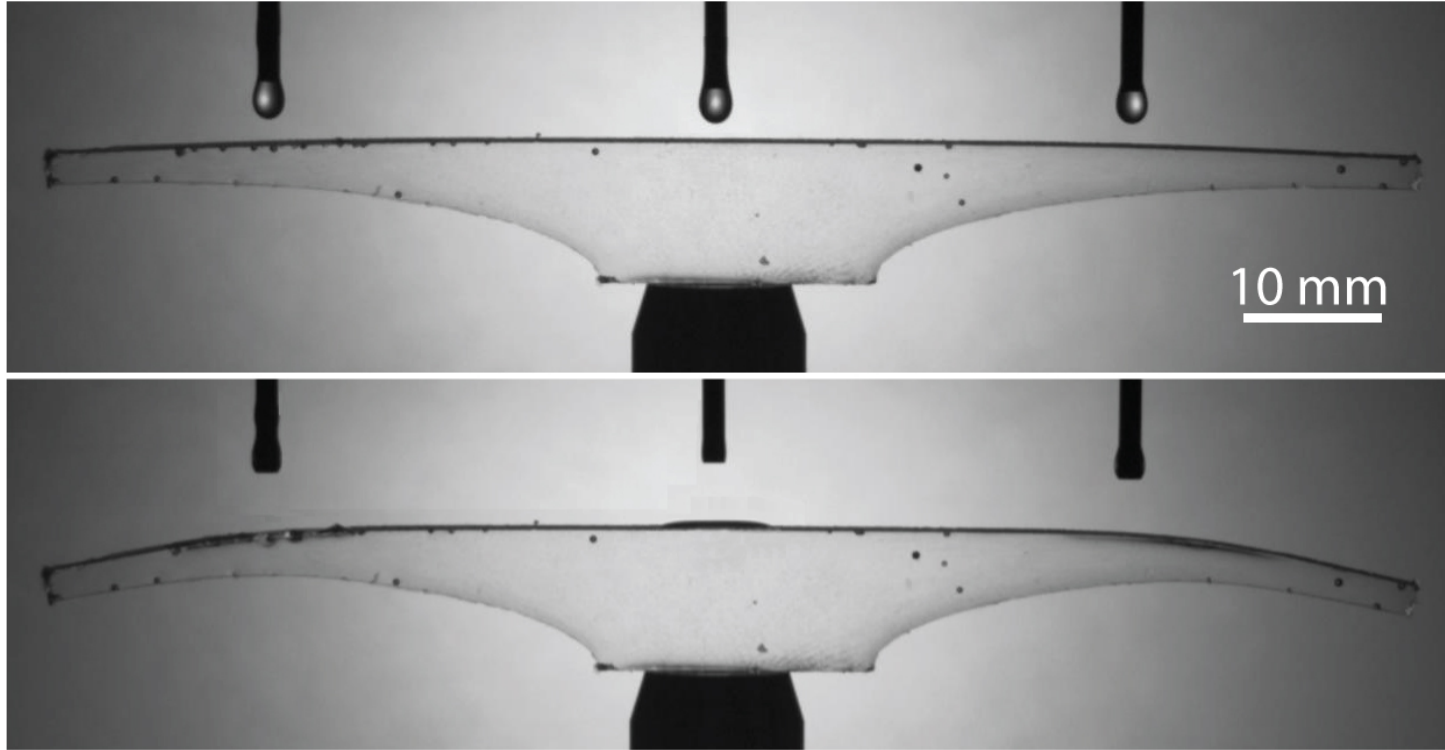


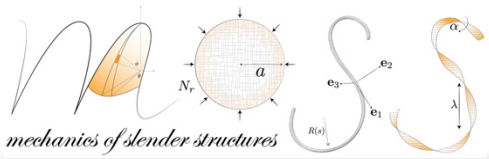
Structural  
Deformation  
(Bending/Buckling)

Surface  
Deformation  
(Creasing)

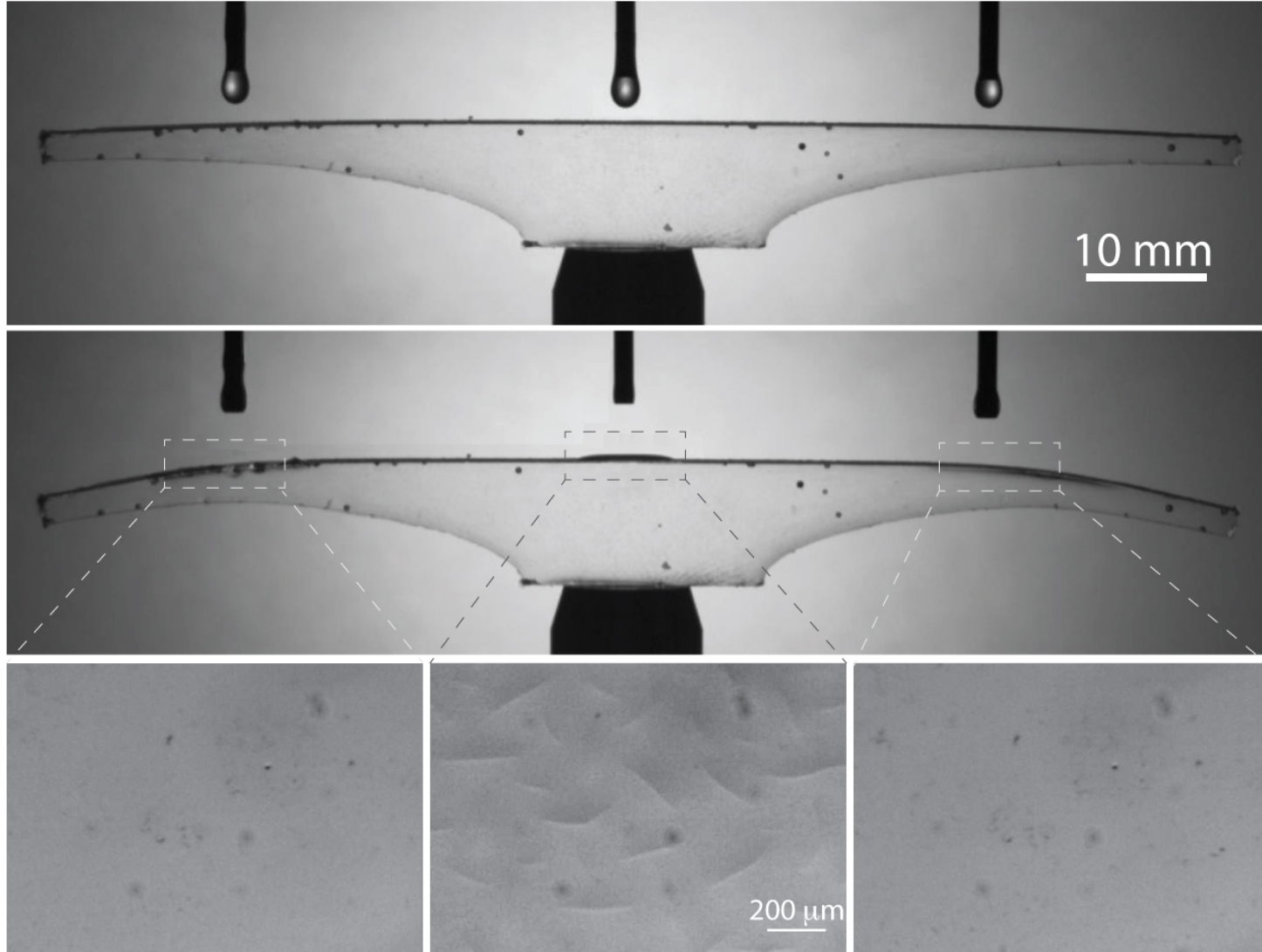


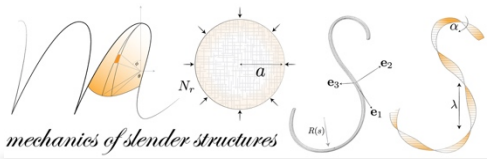
# Creasing



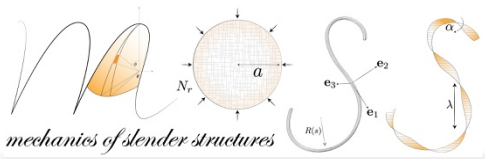


# Creasing





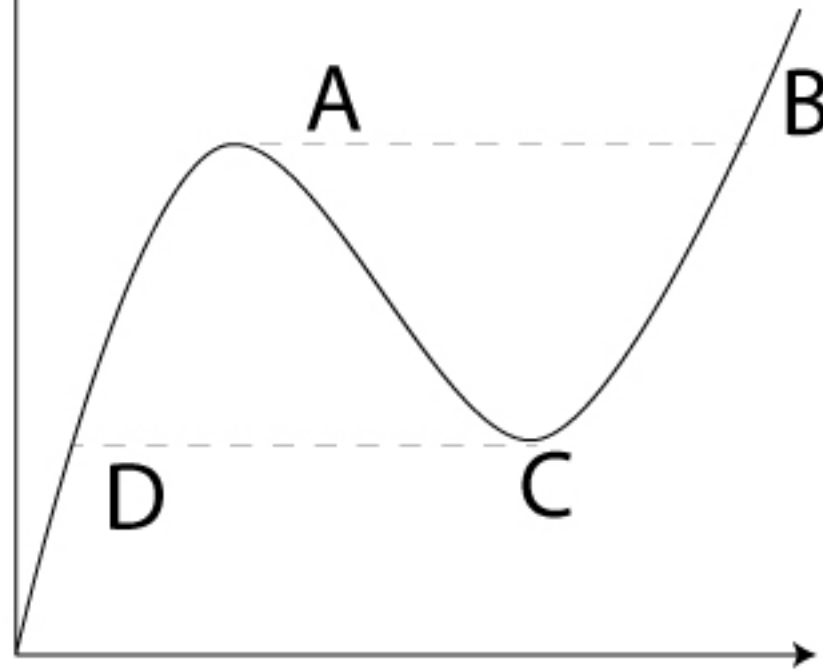
# Snapping



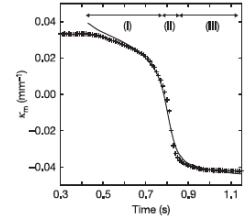
# Snappy Functionality

Limit Point Instability

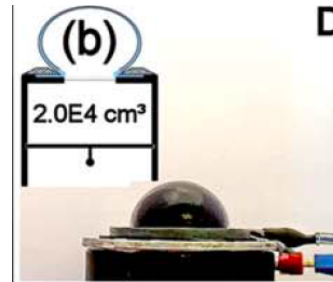
Rapid Jump From A to B



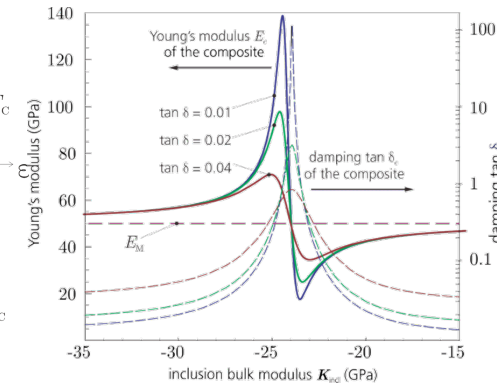
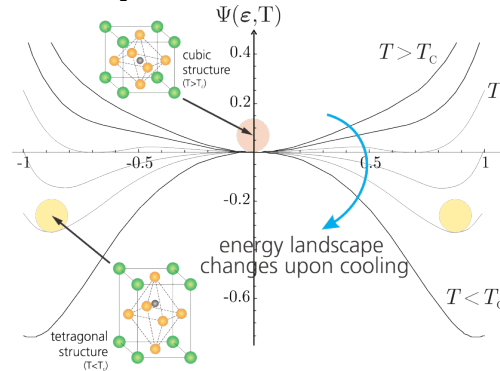
## Venus Flytrap



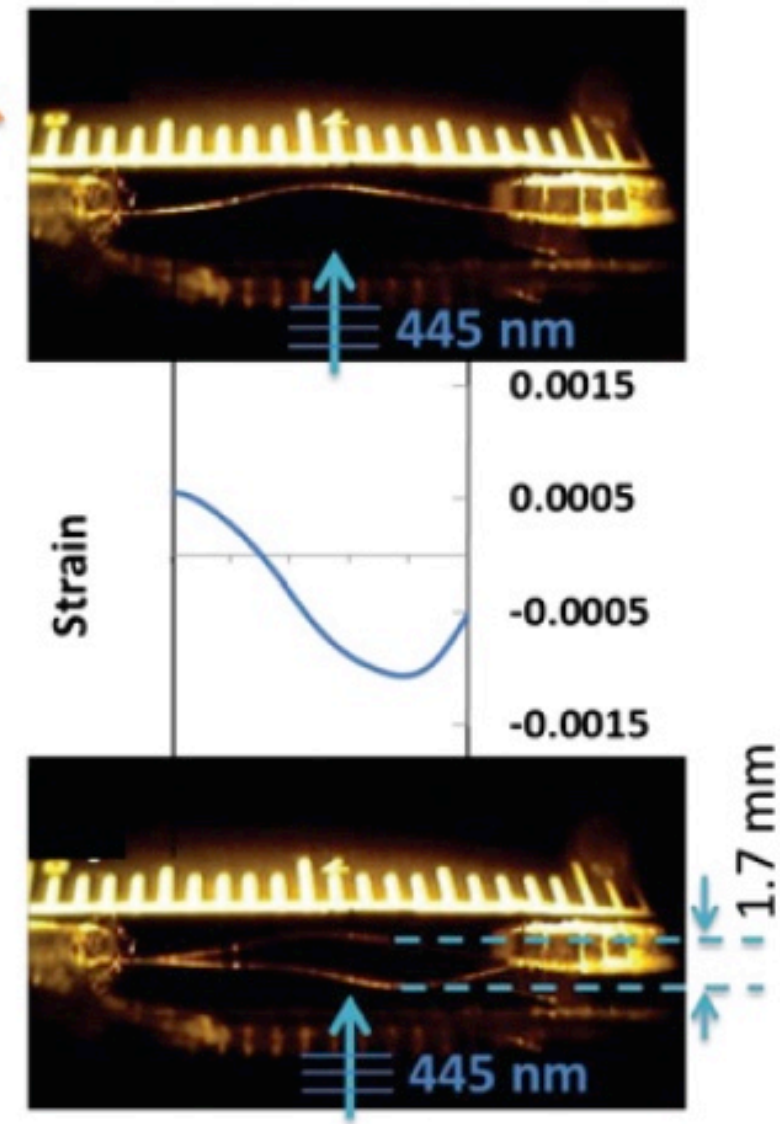
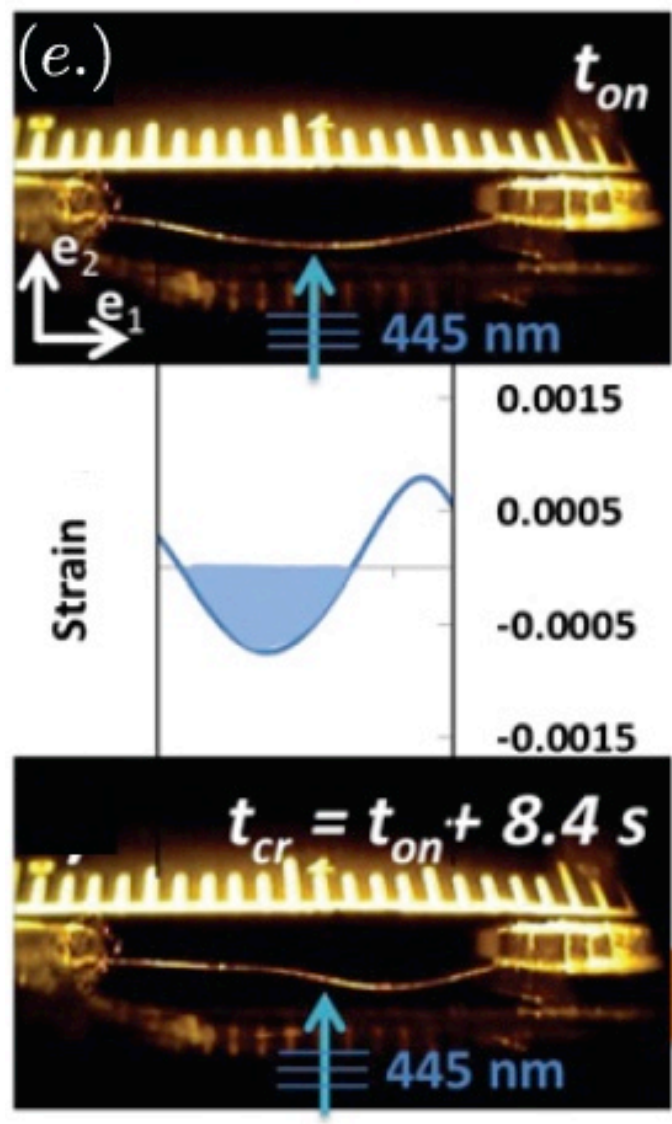
## Dielectric Elastomer

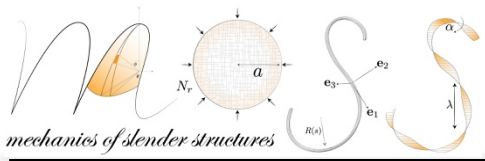


## Composite Materials



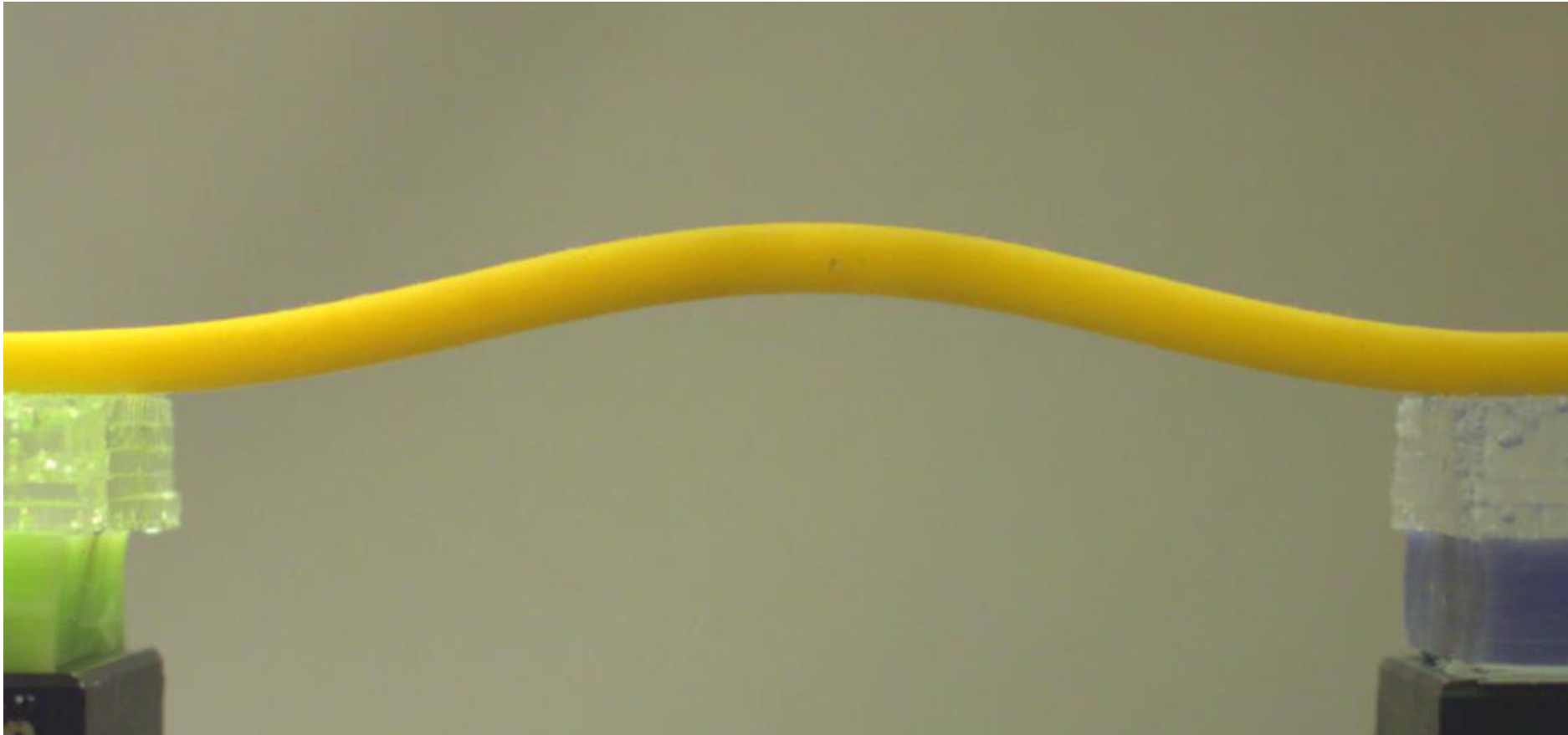


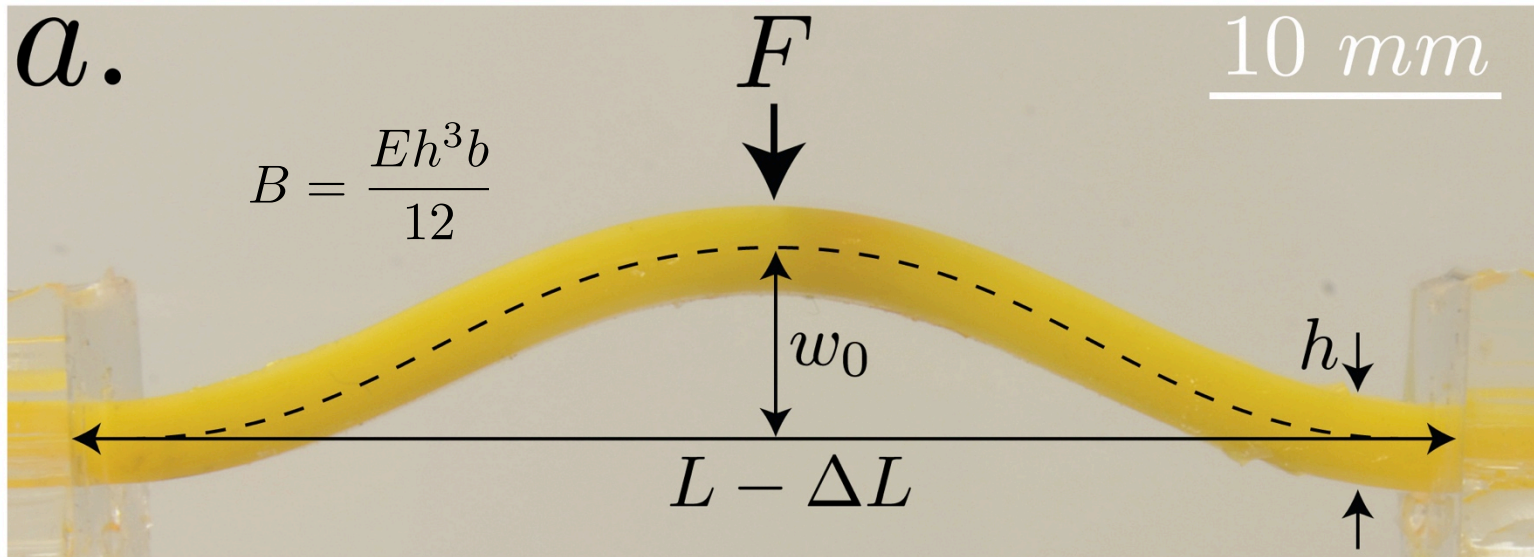




*mechanics of slender structures*

# Snapping





## Two Geometric Quantities:

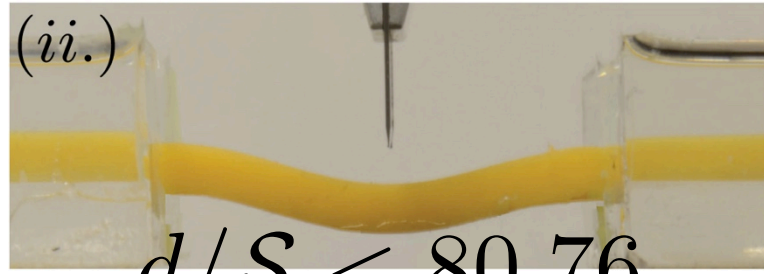
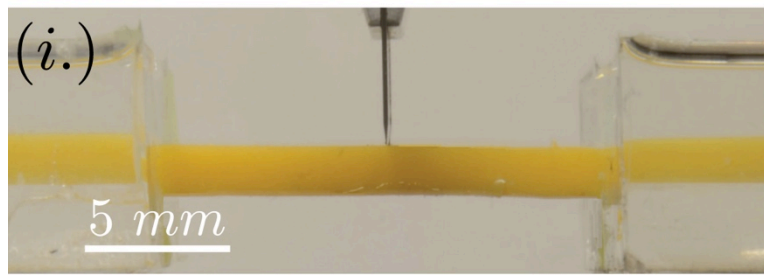
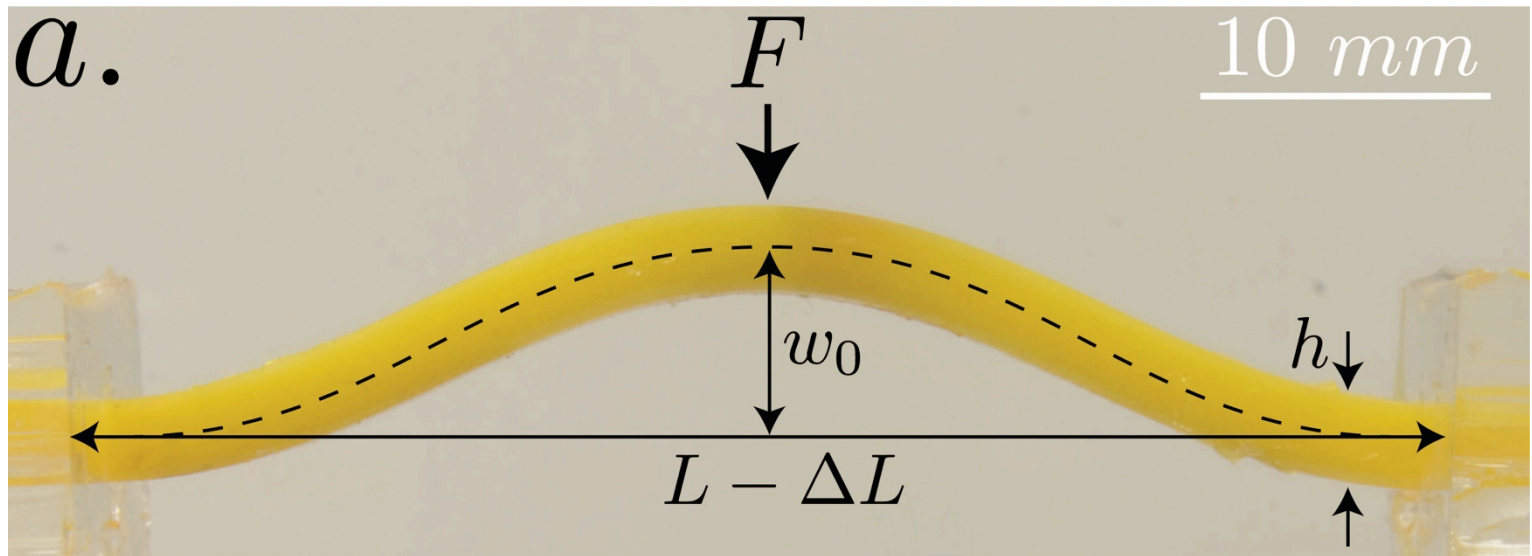
End Shortening:

$$d = \frac{\Delta L}{L}$$

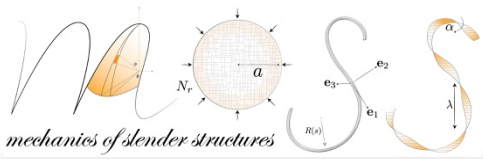
Stretchability:

$$S = \frac{B}{EhL^2b} = \frac{h^2}{12L^2}$$

$$\boxed{d/S}$$



$$d/S < 80.76$$



$$\underbrace{\rho_s h \frac{\partial^2 w}{\partial t^2}}_{\text{dynamics}} + \underbrace{B \frac{\partial^4 w}{\partial x^4} + T \frac{\partial^2 w}{\partial x^2}}_{\text{potential energy}} = \underbrace{F \delta(x)}_{\text{indentation}}$$

von Kármán strains:

$$\varepsilon_{xx} = \partial_x u + \frac{1}{2} (\partial_x w)^2 = -S\tau^2$$

Dimensionless tension:

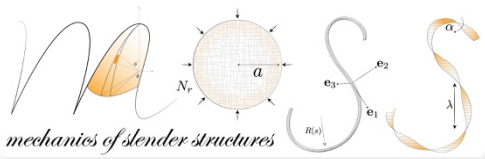
$$\tau^2 = \frac{TL^2}{B}$$

Dimensionless indentation:

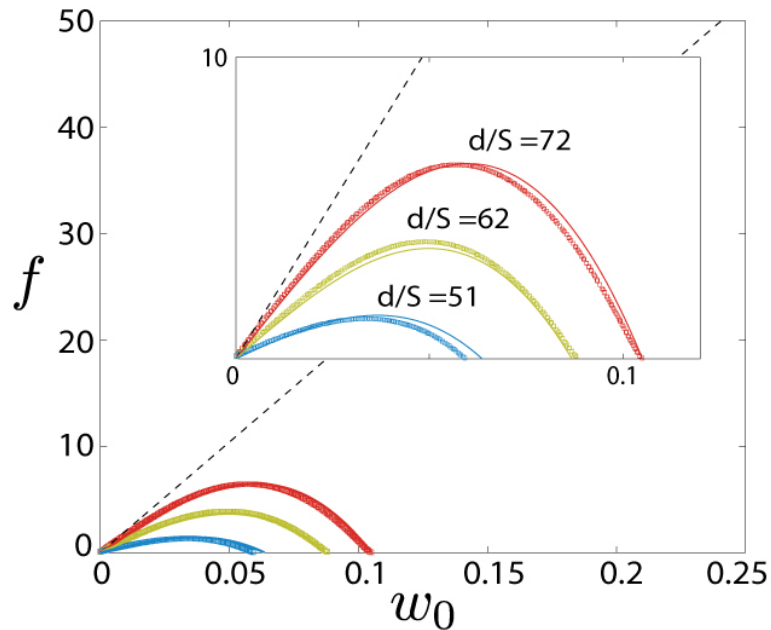
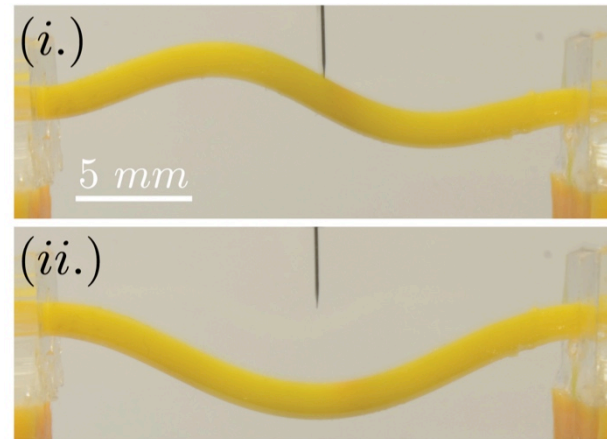
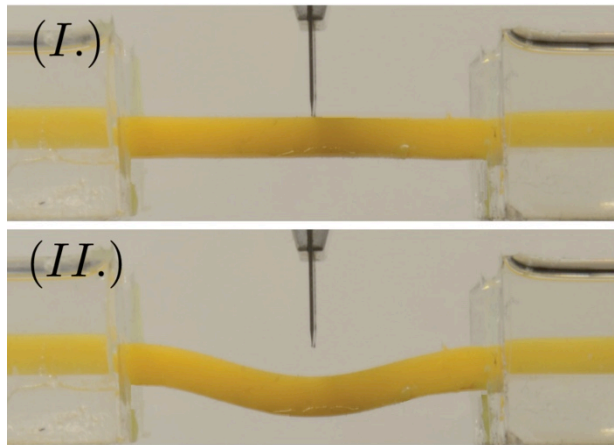
$$f = \frac{FL^2}{B}$$

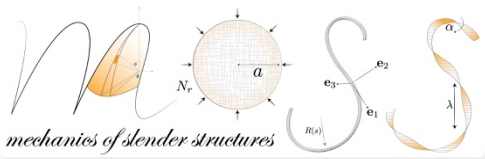
Transition from Symmetric to Asymmetric shape: **Minimize Bending Energy**

$$U_B = \frac{1}{2} w_0 f + \tau^2 (d - S\tau^2)$$

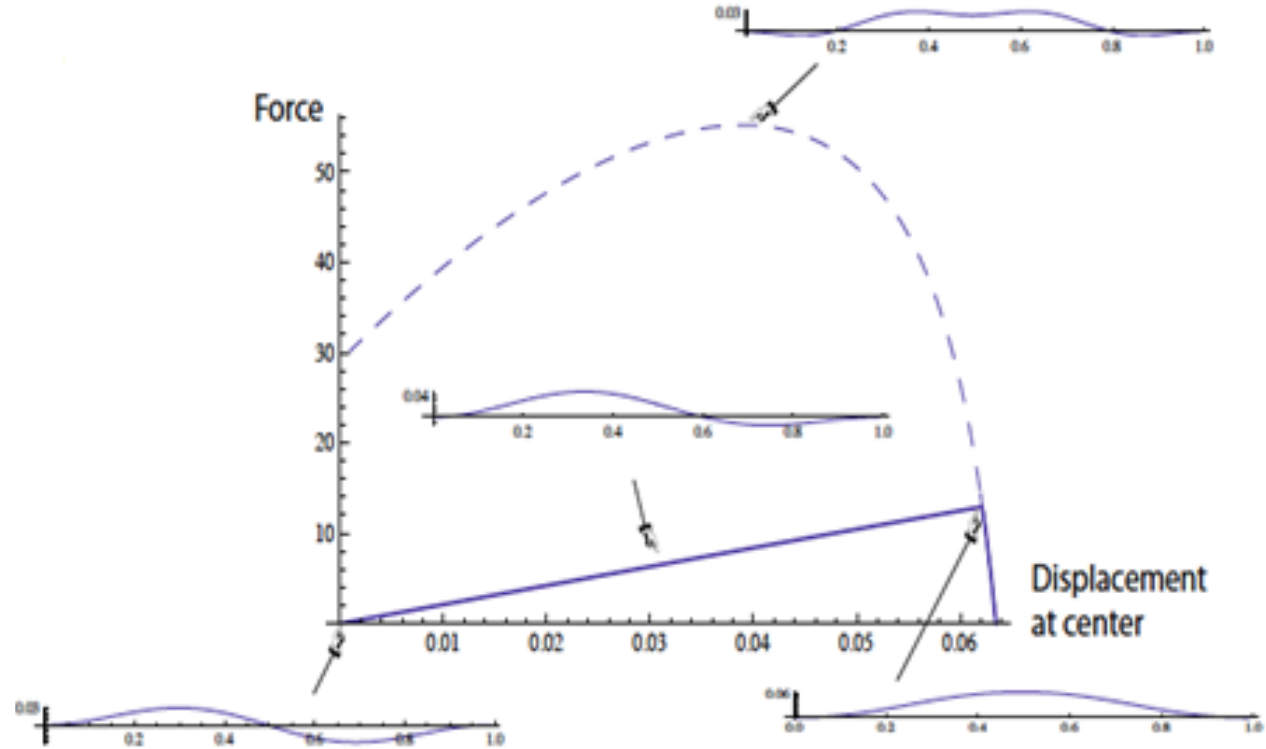
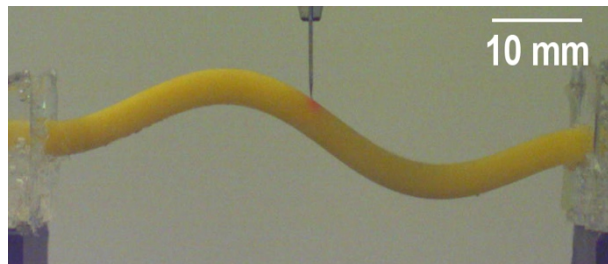
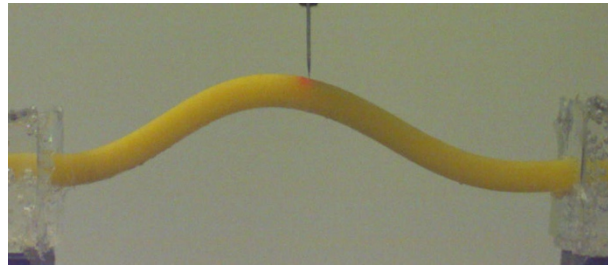
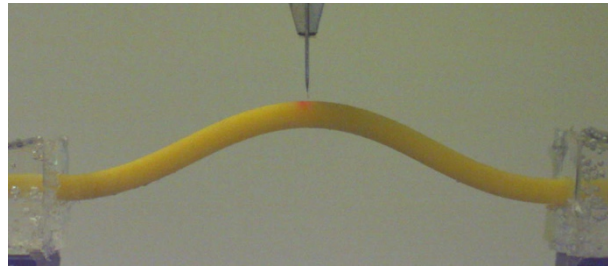


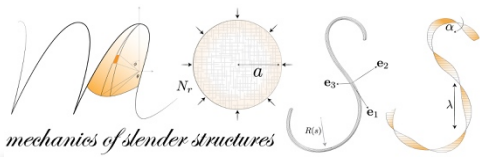
# Force vs. Displacement





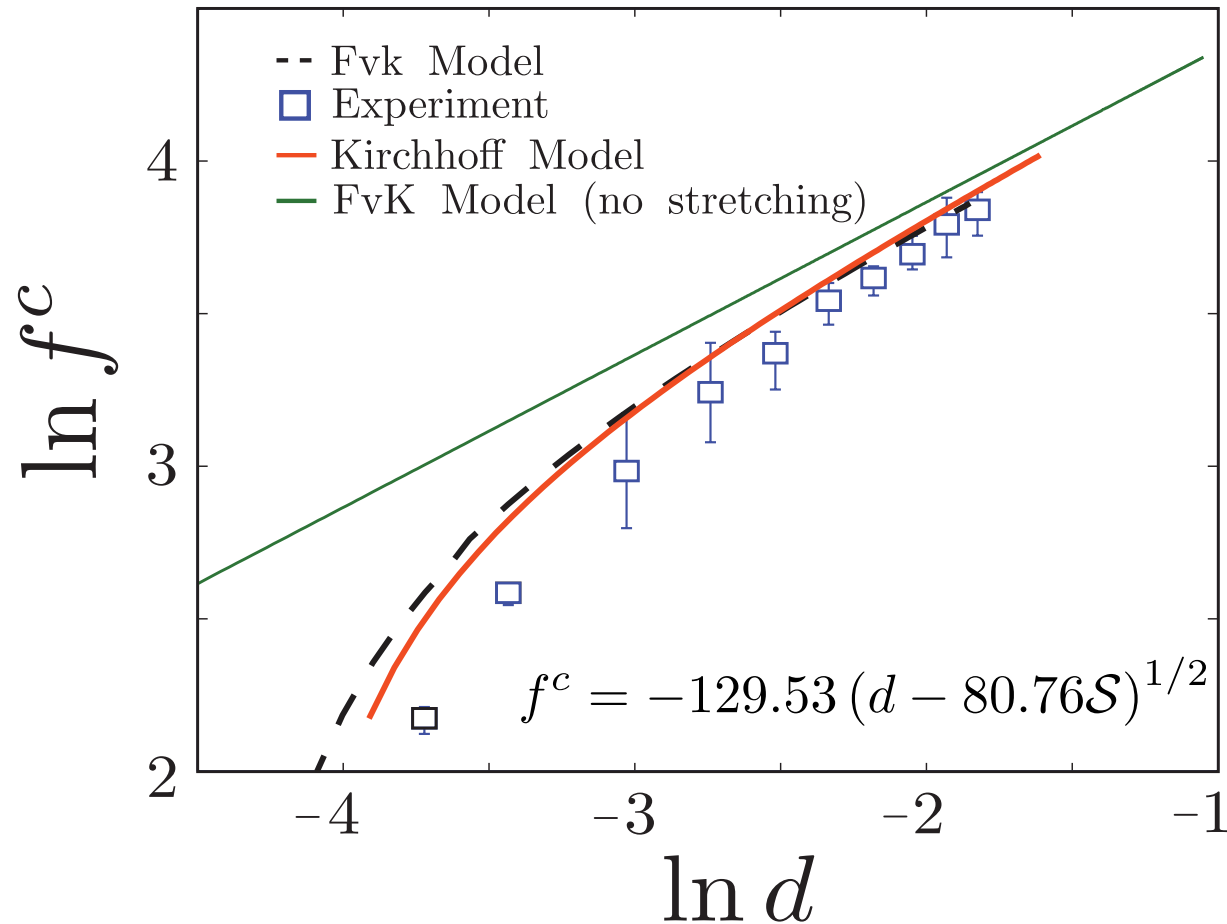
# Force vs. Displacement



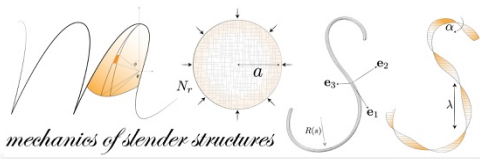


Transition from Symmetric to Asymmetric shape: **Minimize Bending Energy**

$$U_B = \frac{1}{2} w_0 f + \tau^2 (d - \mathcal{S} \tau^2)$$

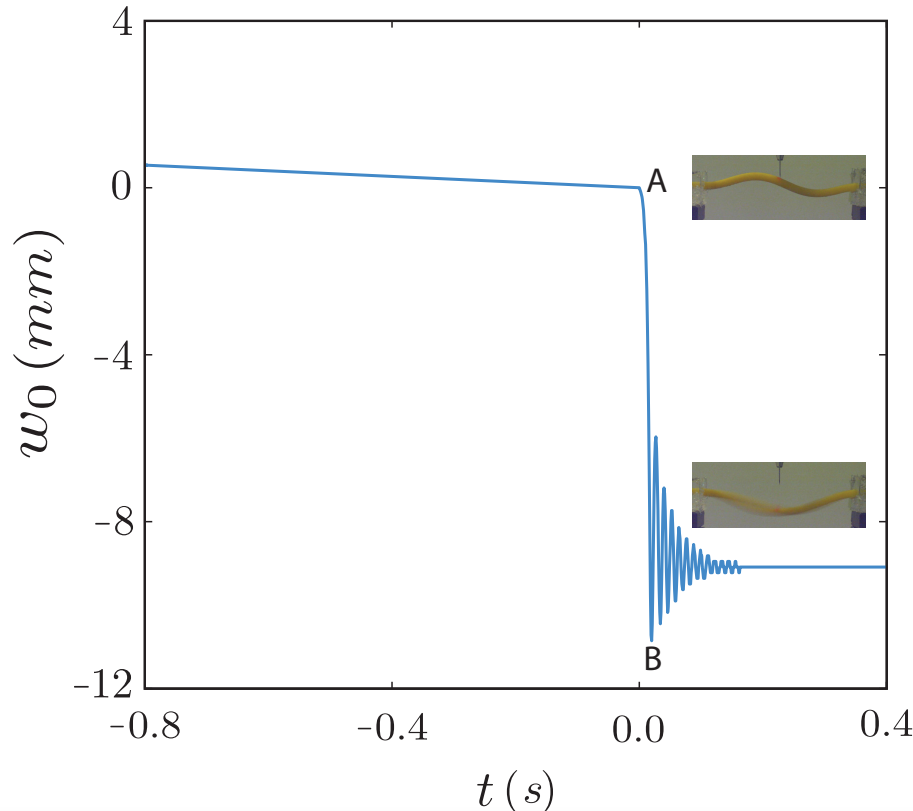






# Dynamics

$$\underbrace{\rho_s h \frac{\partial^2 w}{\partial t^2}}_{\text{dynamics}} + \underbrace{B \frac{\partial^4 w}{\partial x^4} + T \frac{\partial^2 w}{\partial x^2}}_{\text{potential energy}} = \underbrace{F \delta(x)}_{\text{indentation}}$$

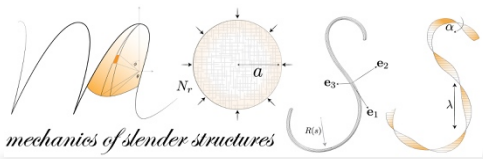


$$\underbrace{\frac{\partial^2 w}{\partial t^2} + \frac{\partial^4 w}{\partial x^4} + \tau^2 \frac{\partial^2 w}{\partial x^2}}_{\text{non-dimensional}} = 0$$

Solutions:

$$w(x, t) = \underbrace{w_\alpha(x)}_{\text{shape}} + \underbrace{\epsilon w_p(x) e^{\sigma t}}_{\text{perturbation}}$$

$$\tau = \underbrace{\tau_\alpha}_{\text{comp.}} + \underbrace{\epsilon \tau_p e^{\sigma t}}_{\text{perturbation}}$$



# Dynamics

At leading order in  $\epsilon$ , eigenvalue problem for  $\sigma$ , with eigenfunction  $w_p(x)$  satisfying

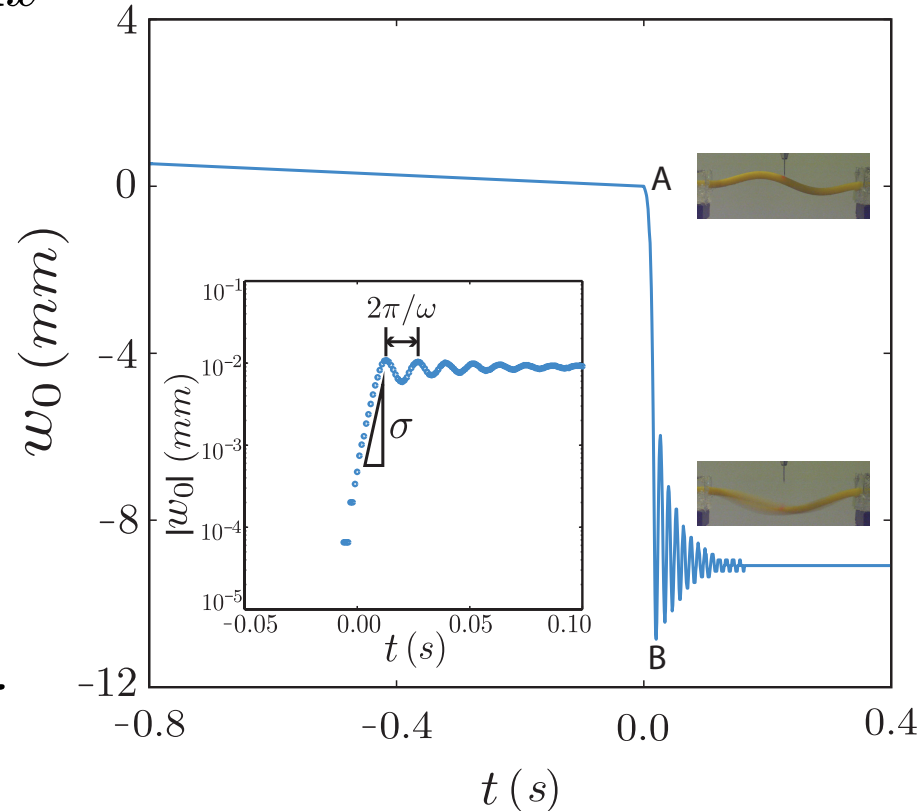
$$\frac{d^4 w_p}{dx^4} + \tau_p^2 \frac{d^2 w_p}{dx^2} + \sigma^2 w_p = -2\tau_\alpha \tau_p \frac{d^2 w_\alpha}{dx^2}$$

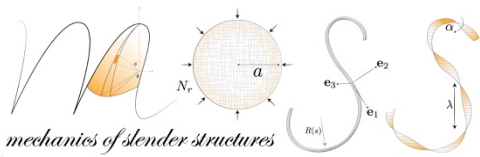
$$\int_{-1/2}^{1/2} \frac{dw_\alpha}{dx} \frac{dw_p}{dx} dx = -2\mathcal{S}\tau_\alpha \tau_p$$

Boundary Conditions

$$w_p(\pm 1/2) = w_p'(\pm 1/2) = 0$$

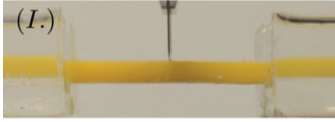
Transcendental equation for  $\sigma$



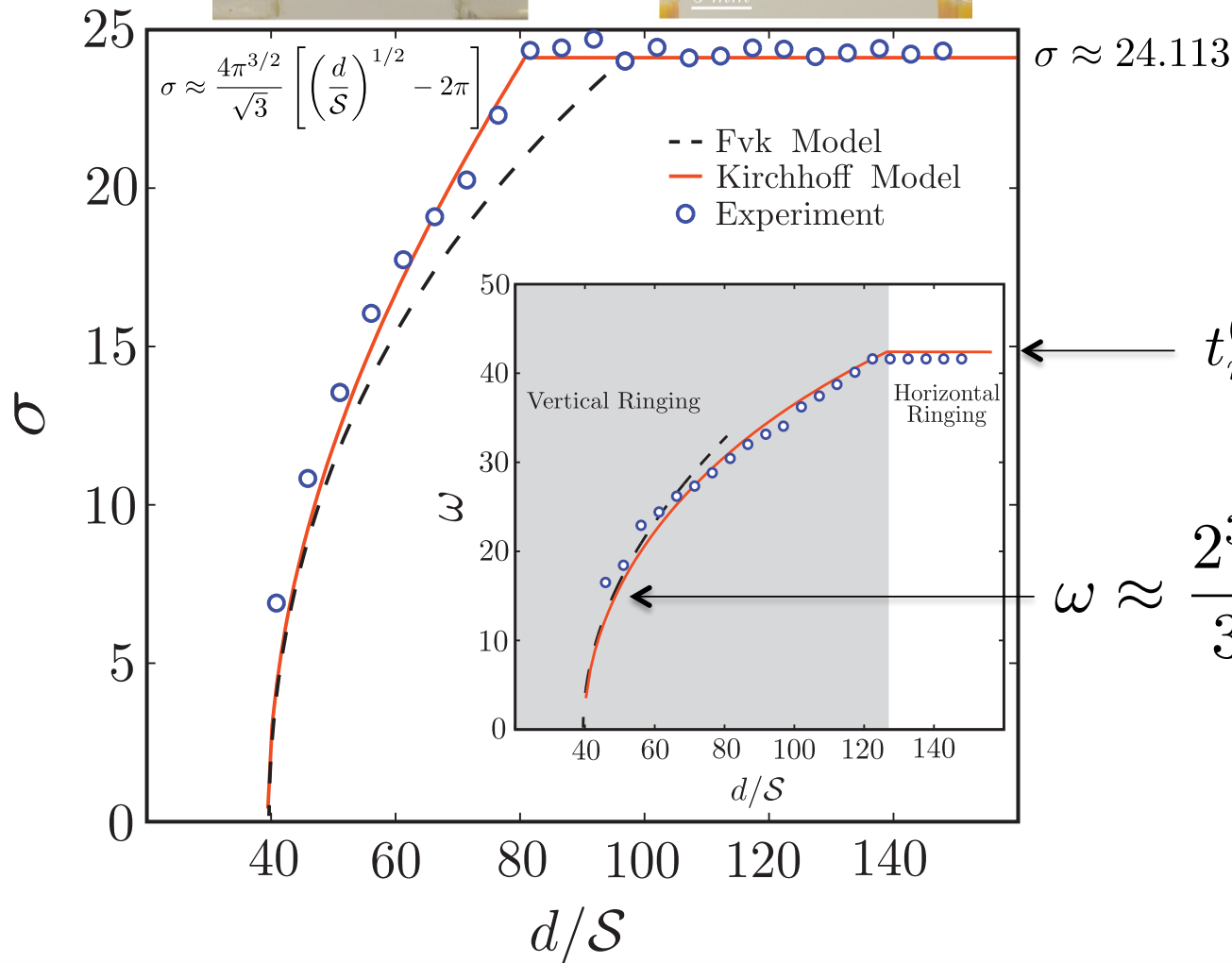


# Dynamics

Symmetric

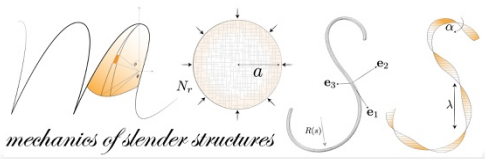


Asymmetric

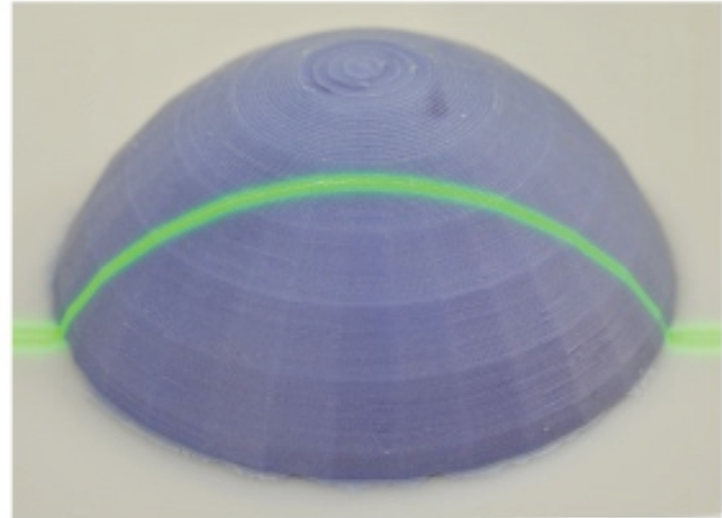
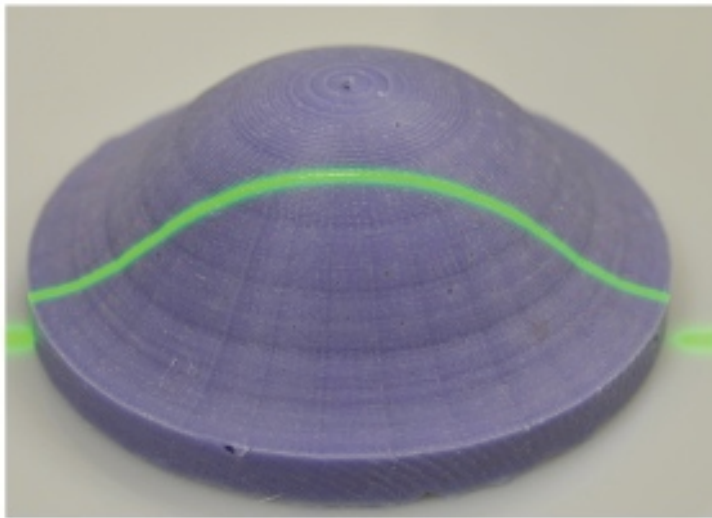
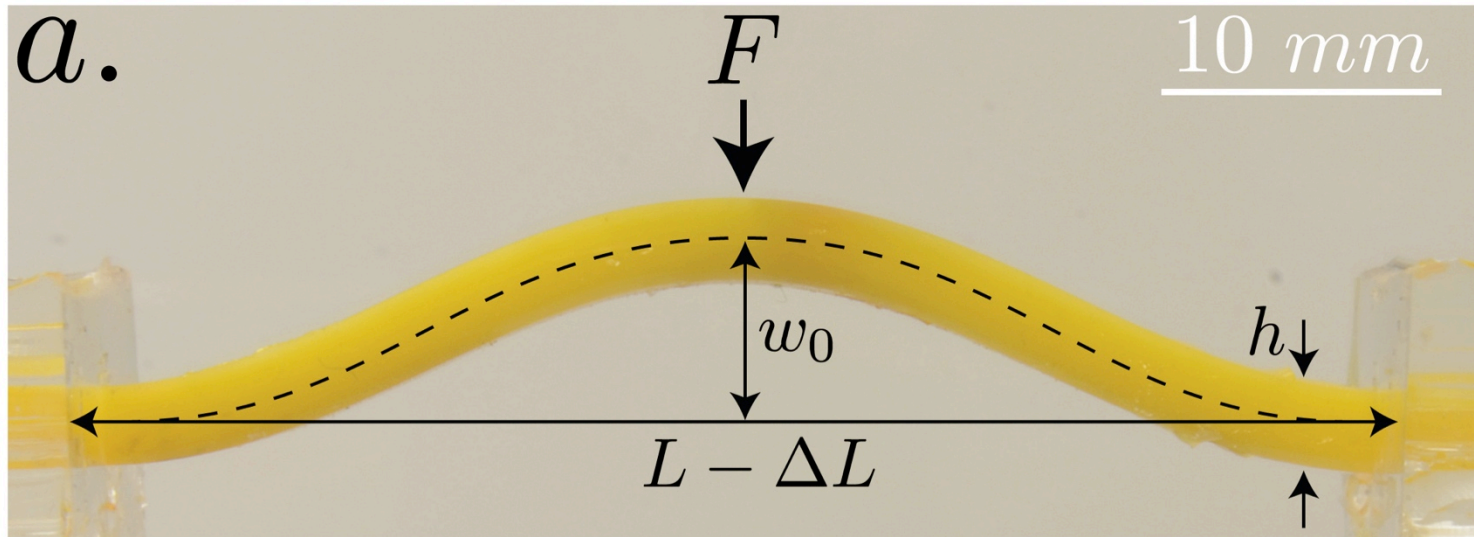


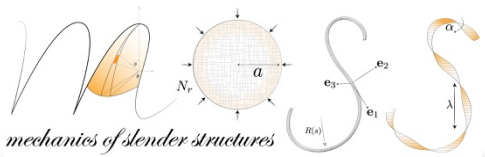
$$t_r^{(b)} = \frac{2\pi}{\omega} t_* \approx 0.513 \frac{L^2}{ch}$$

$$\omega \approx \frac{2^{3/2}\pi}{3^{1/2}} \left( \frac{d}{S} - 4\pi^2 \right)^{1/2}$$

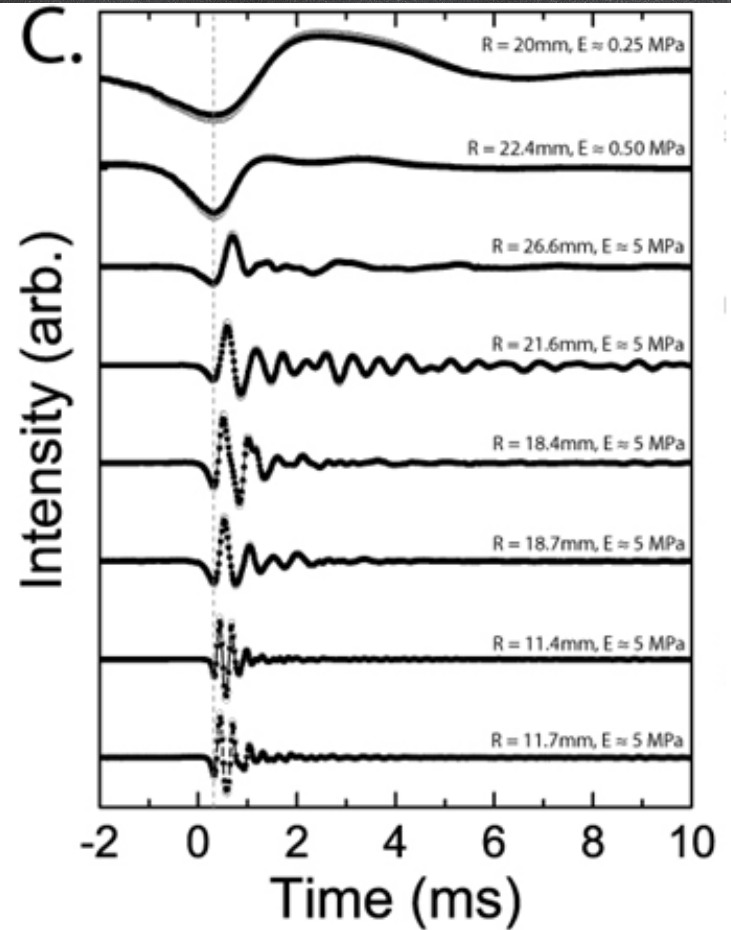
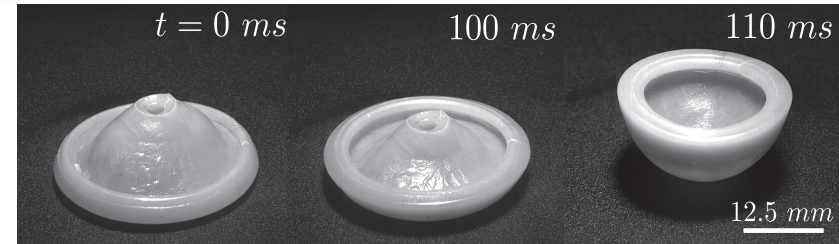
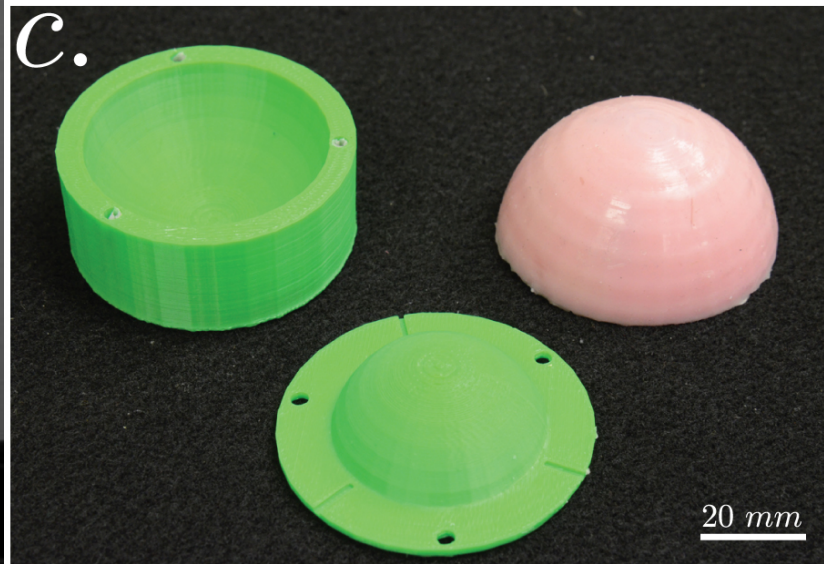
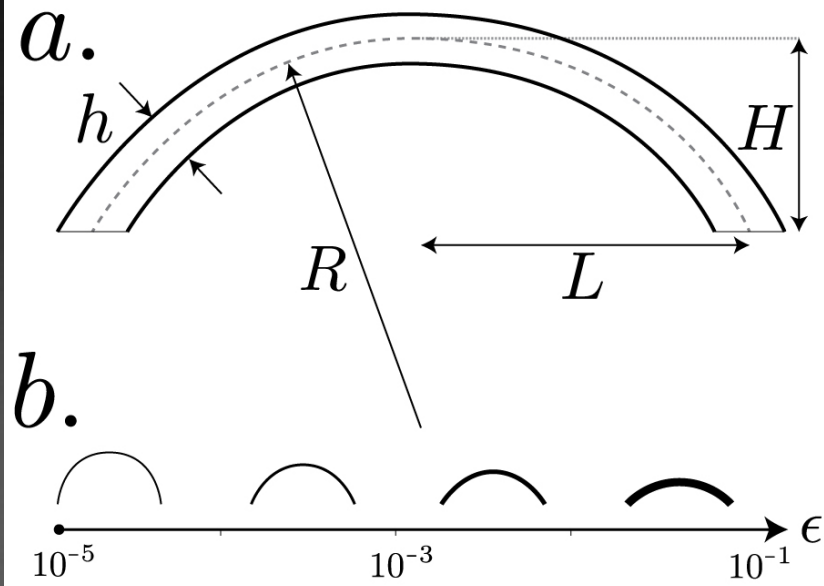


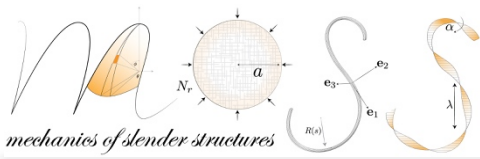
# From Beams to Shells



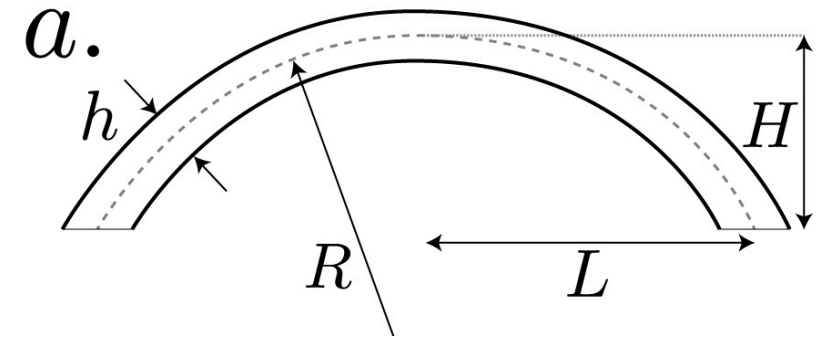


# Snapping Dynamics





# Snapping Dynamics



Donnell-Mushtari-Vlasov

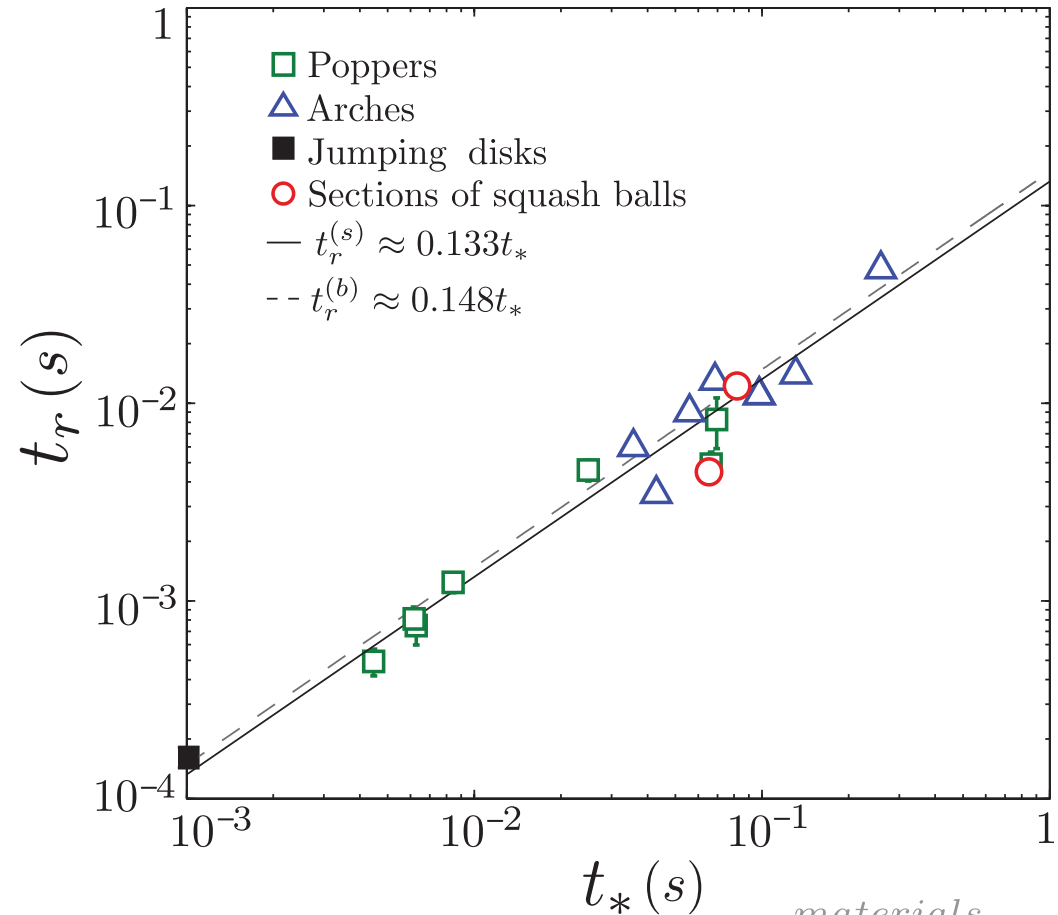
$$\rho_s h \frac{\partial^2 w}{\partial t^2} + B \nabla^4 w + \underbrace{\frac{Eh}{R^2} w}_{\text{curvature}} = 0$$

Boundary Conditions

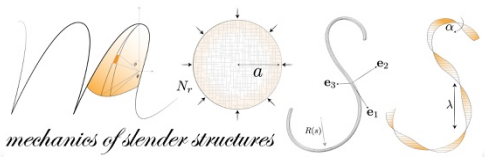
$$\underbrace{\nabla^2 w|_{r=L/2}}_{\text{moment-free}} = \underbrace{\frac{d}{dr} (\nabla^2 w)|_{r=L/2}}_{\text{shear-free}} = 0$$

Ringling Timescale ( $\lambda \approx 3.196$ )

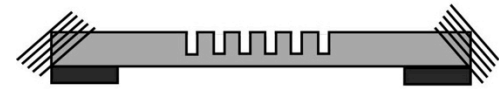
$$t_r^{(s)} = \frac{\pi}{2} \frac{\sqrt{1-\nu^2}}{\lambda^2} t_* \left( 1 + \frac{3}{4} \frac{1-\nu^2}{\lambda^4} \frac{L^4}{h^2 R^2} \right)^{-1/2}$$



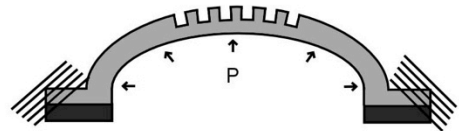
Timescale:  $t_r \sim \underbrace{\frac{L^2}{h}}_{\text{geometry}} \underbrace{\sqrt{\frac{\rho_s}{E}}}_{\text{materials}}$



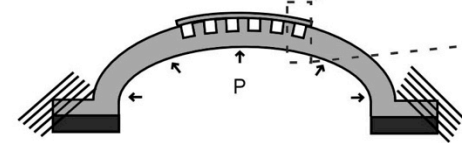
# Snapping Surfaces



a. Clamp the PDMS elastomer array over a hole.



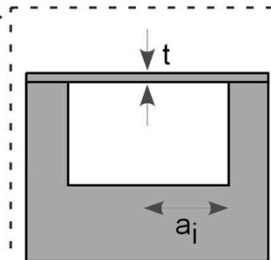
b. Inflate sample with air, creating biaxial tension.



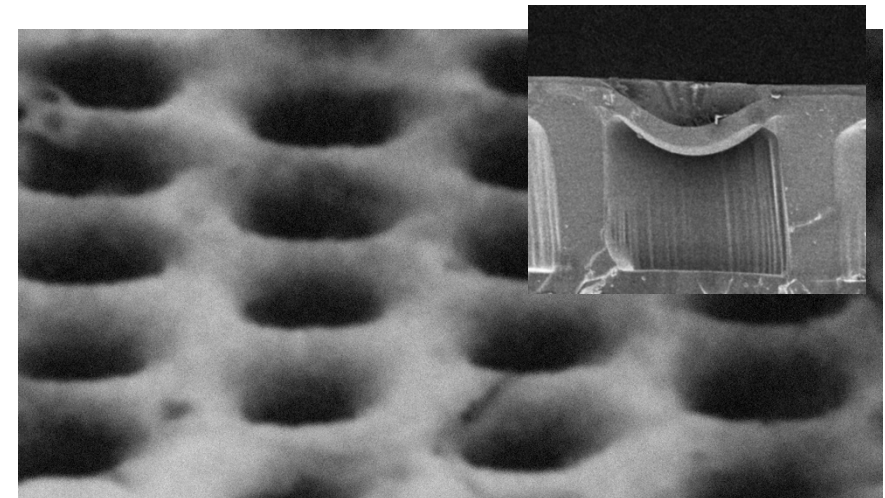
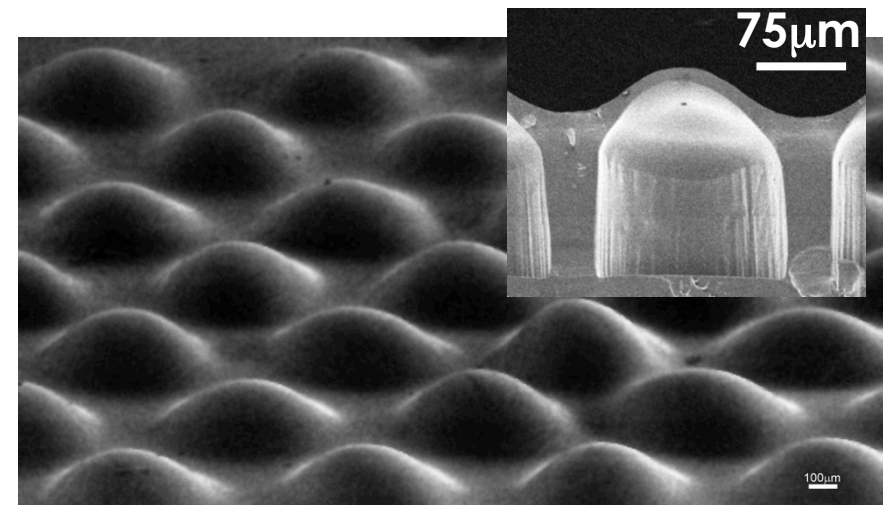
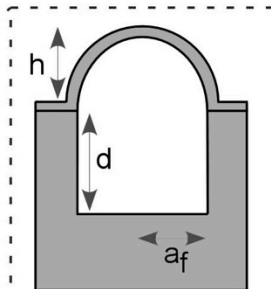
c. Spin-coat uncured PDMS onto a film of PDMS elastomer, lay it over the surface of holes and crosslink the uncured PDMS at 110 °C

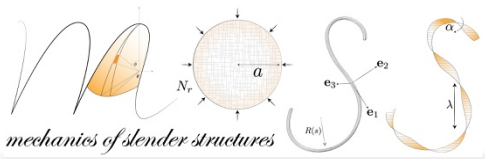


d. Release the pressure to form hollow, hemispherical shells.

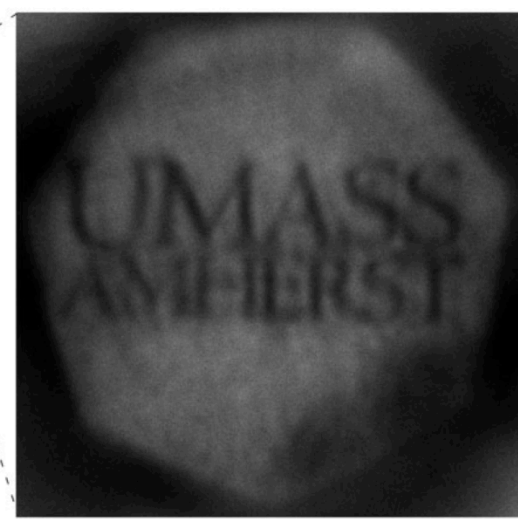
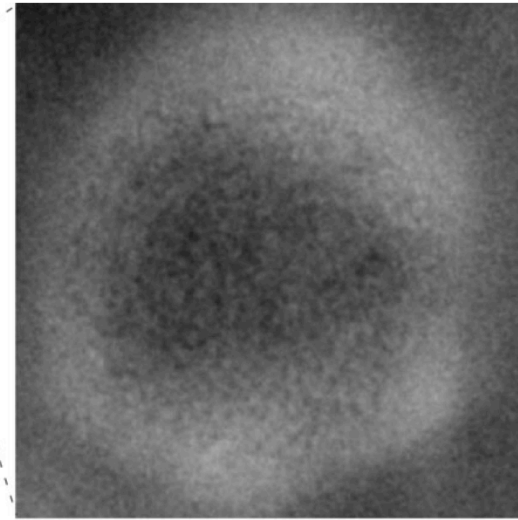
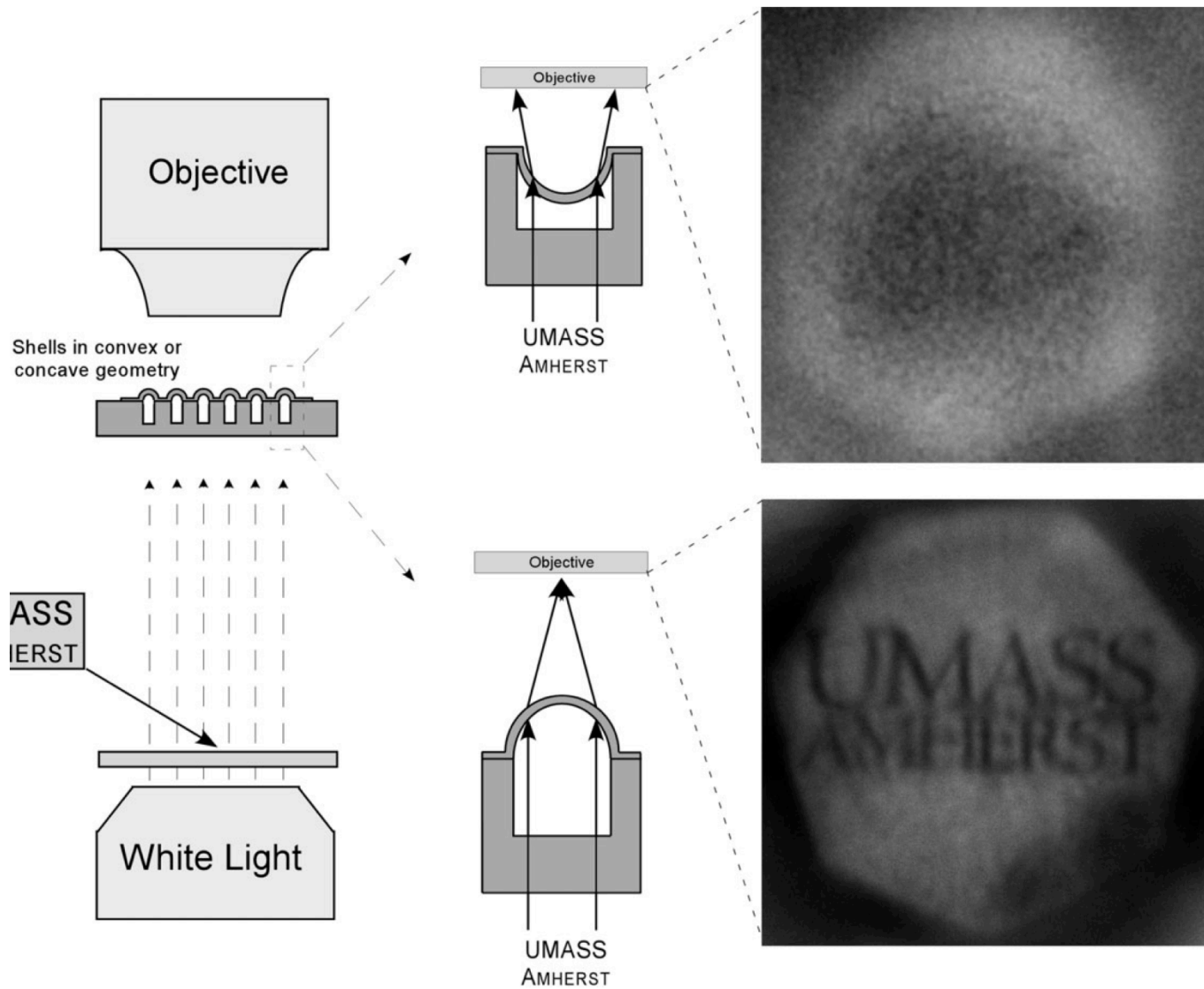


release biaxial strain  
 $\epsilon = \frac{(a_f - a_i)}{a_f}$

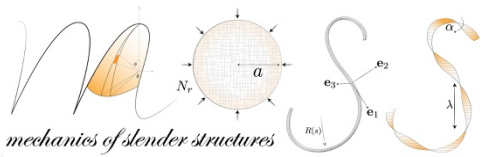




# Rapid Optical Switch

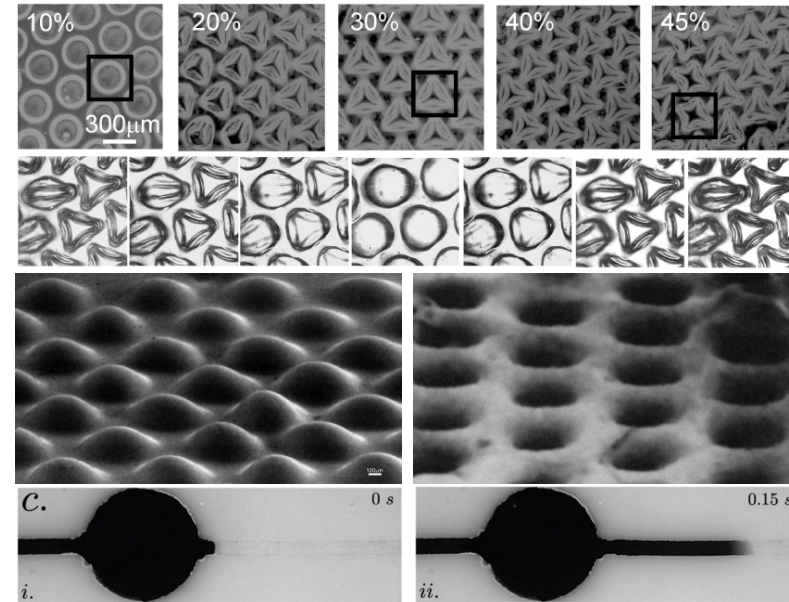






# Summary

- **Understanding** and **controlling** the shapes of **slender** structures.
- Harness **instabilities** for advanced **functionality**.
- **Large, fast, reversible** deformations for **adaptable** materials.



## Relevant Publications by PI

- **D.P. Holmes** and A.J. Crosby, "Snapping Surfaces," *Advanced Materials*, **19**, 21, 3589-3593, 2007.
- **D.P. Holmes**, and A.J. Crosby, "Crumpled Surface Structures", *Soft Matter*, **4**, 82, (2008).
- A. Pandey, D. Moulton, D. Vella, and **D.P. Holmes**, "Dynamics of Snapping Beams and Jumping Poppers", *EPL (Europhysics Letters)*, **105**(2), 24001, (2014).
- **D.P. Holmes**, B. Tavakol, G. Froehlicher, and H.A. Stone, "Control and Manipulation of Microfluidic Fluid Flow via Elastic Deformations", *Soft Matter*, **9**(29), 7049, (2013).
- B. Tavakol, M. Bozlar, G. Froehlicher, H.A. Stone, I. Aksay, and **D.P. Holmes**. "Buckling Instability of Dielectric Elastomeric Plates for Flexible Microfluidic Pumps." *Under Review*, 2014.
- B. Tavakol, **D.P. Holmes**, and H.A. Stone, "Extended Lubrication Theory", *Under Review: Physics of Fluids*, (2015).

## Acknowledgements

Behrouz Tavakol (Virginia Tech)  
 Anupam Pandey (Virginia Tech)

Howard A. Stone (Princeton)  
 Ilhan Aksay (Princeton)  
 Michael Bozlar (Princeton)  
 Guillaume Froehlicher (Princeton)

Dominic Vella (Oxford)  
 Derek Moulton (Oxford)

## Funding

- Army Research Office – MURI (#W911NF-09-1-0476)
- NSF CMMI – Mechanics of Materials (#1300860)